

Cargo Compartment Smoke Detector AS8036 Standard revision

International Aircraft Systems
Fire Protection Working Group
Cologne, May 22-23 2013

Dr. André Freiling, Head of Fire Protection, Airbus

FIRE

DUST

LIGHT

FOG

INSECTICIDE



CARGO COMPARTMENT FIRE DETECTION INSTRUMENTS

1. **PURPOSE:** This standard establishes minimum requirements for cargo compartment fire detection instruments primarily for use in reciprocating and turbine engine powered aircraft.
2. **SCOPE:**
 - 2.1 This standard covers the following types of fire detection instruments intended for use in protecting aircraft cargo compartments, galleys, electronic equipment bays and other similar installations.
 - 2.2 **Types:**

Type I: Carbon monoxide, an instrument which will actuate an alarm signal when the concentration of carbon monoxide in air exceeds a specified value.

Type II: Smoke detector, electronic, an instrument operating on the principle of smoke particles modifying the relationship between a light beam and electronic light sensor which will actuate an alarm signal when the concentration of smoke in air exceeds a specified value.

Type III: Smoke detector, visual, an instrument which, by visual means, will show in a positive manner the presence of smoke when the concentration of smoke in air exceeds a specified value.

Type IV: Smoke detector, electronic, an instrument operating on the principle of smoke particles modifying the current in an ionization chamber which will actuate an alarm signal when the concentrations of smoke in air exceeds a specified value.

Type V: Same as Type IV except maximum operating altitude is 18,000 ft. (5,486 M) when installed in a non-pressurized area.

Things To Do

COMPLETE

FOLLOW-UP

PRIORITY

1 Update AS8036 to achieve state-of-the-art smoke detector qualification in order to improve safety.

2 Define testing to reduce false alarms from smoke detectors

3 Include latest qualification standards and clean up test requirements - D0160B -> G

- FAA Approached SAE to update the AS8036 Standard to include nuisance tests
- Working group was put together in April/May 2011 to look at updating the document
- Face-to-face meetings and bi-weekly / weekly teleconferences were organized

- Committee agreed that a minimum performance standard for testing of smoke detectors has been developed.
- Internal ballot finalized
- Document ready for public ballot process

The Committee Team

David Alexander, SAE

Keely Andrews, SAE

Ken Bell, Kidde

Dave Blake, FAA Tech Center

Ian Campbell, Meggitt

Laura Feix, SAE

Andre Freiling, Airbus

Loic Frère, Siemens

Stephen Happenny, FAA

Joan Hughson, FAA

Larry Lamberth, Honeywell

Bruce Mahone, SAE

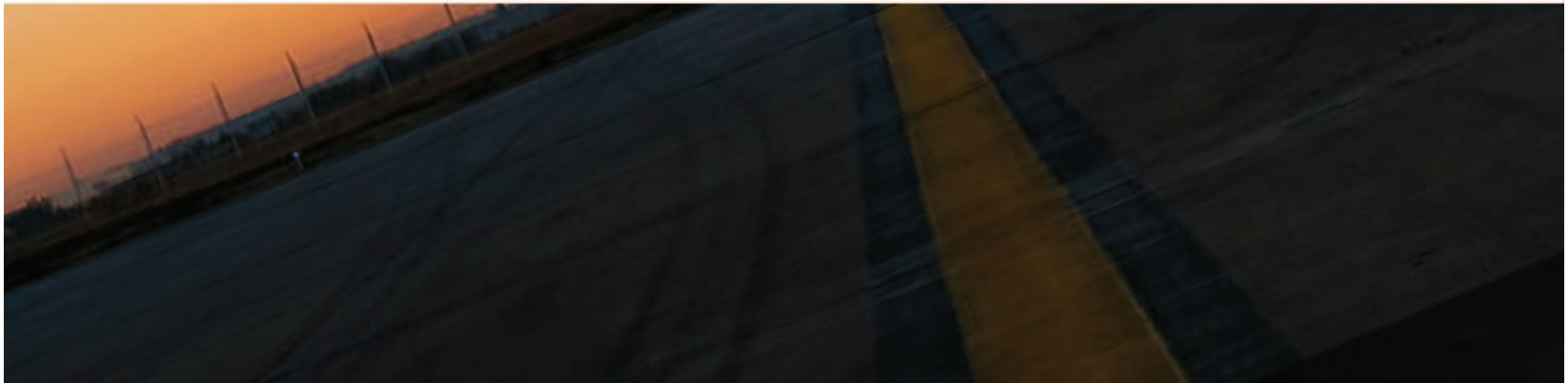
Bruce Miller, Boeing

Gerd Wedler, apparatebau Gauting





SAE Aerospace <i>An SAE International Group</i>	AEROSPACE STANDARD	AS8036	REV. A
		Issued 1985-04 Revised 2013-3-5	
		Superseding AS8036	
Cargo Compartment Fire Detection Instruments			



剛張弓



EN/UL smoke detector fire test standards

Test fires shall be selected to demonstrate that the detector can successfully detect fire.

UL268:

Test B (Wood Fire)

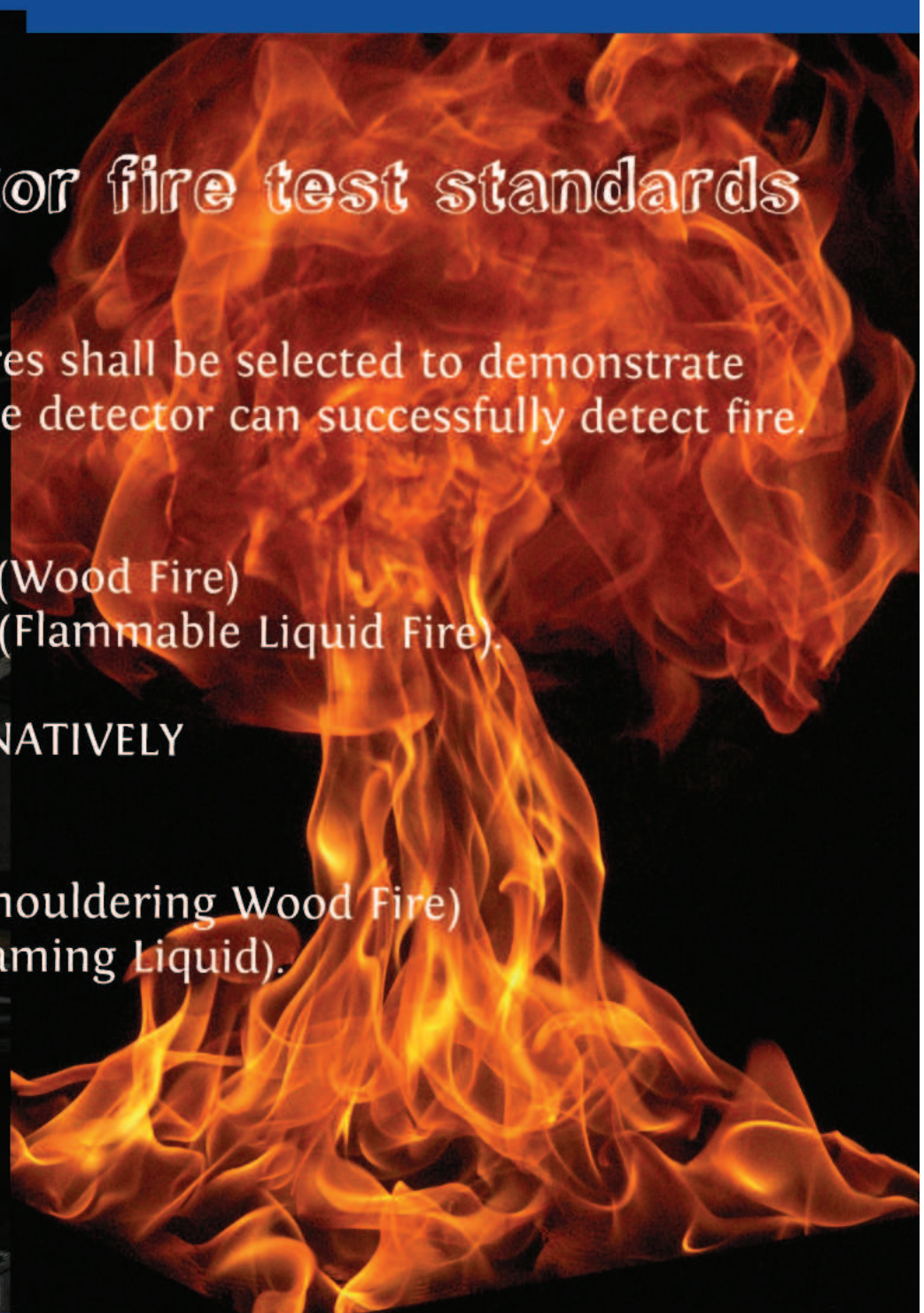
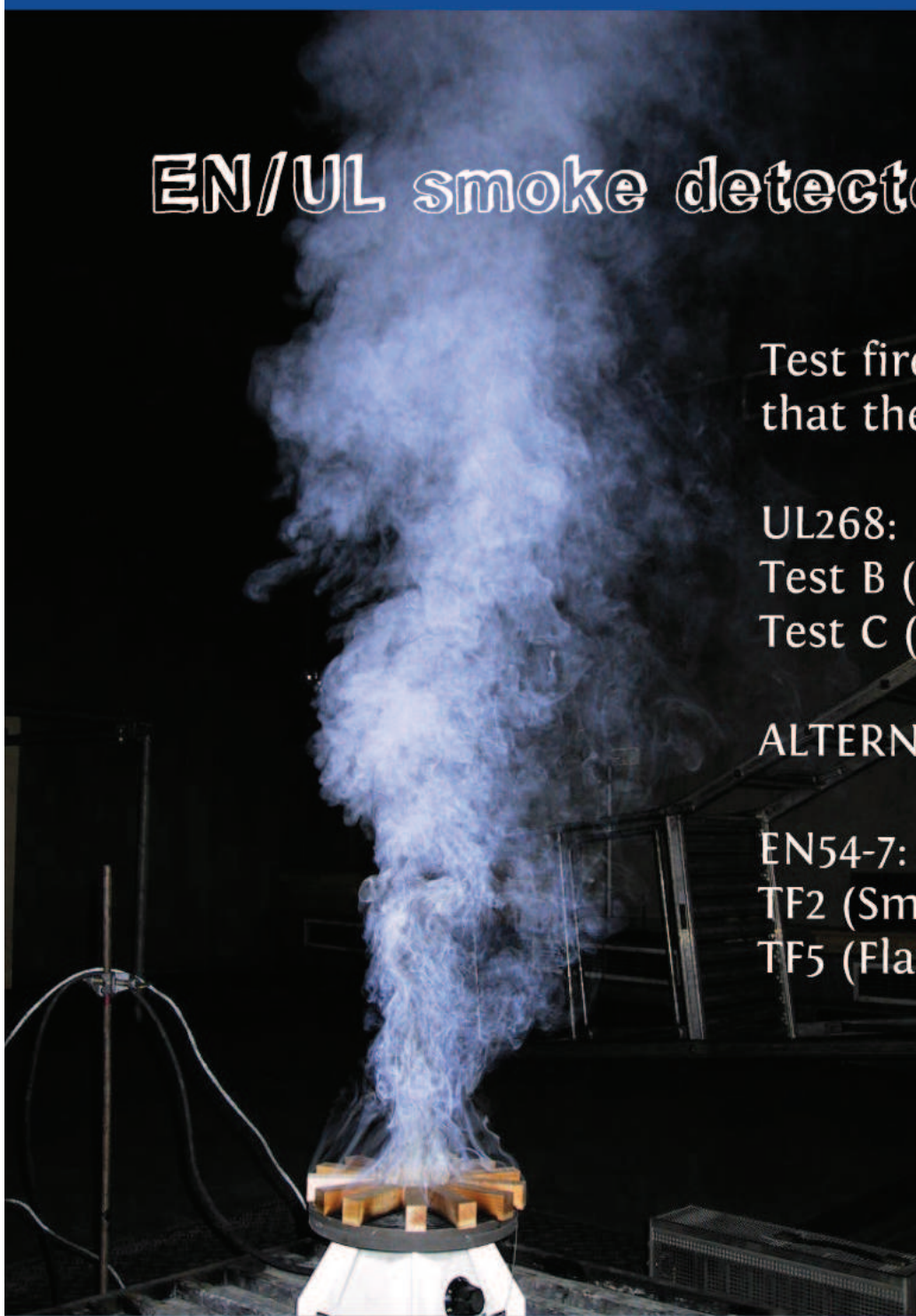
Test C (Flammable Liquid Fire).

ALTERNATIVELY

EN54-7:

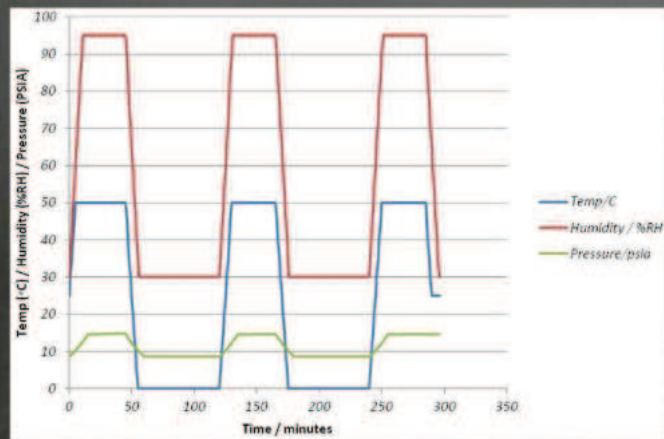
TF2 (Smouldering Wood Fire)

TF5 (Flaming Liquid).



FOOG

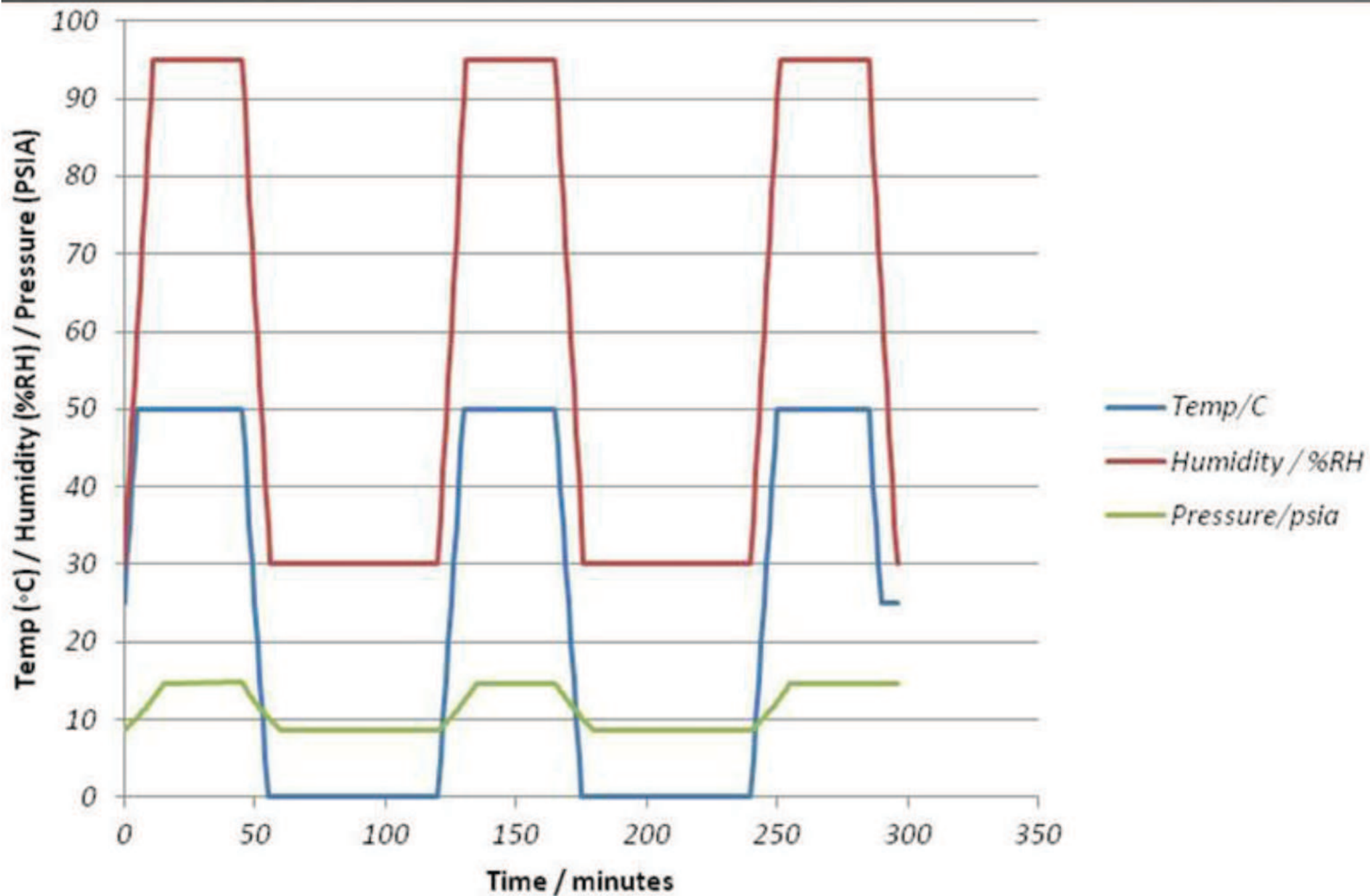
Combined temperature, pressure and humidity cycling



Test Procedure

- 1) Bring the environment to Warm/Moist. With the test item operating, the chamber from the test site altitude pressure to 15000 ft pressure altitude at a ramp rate of 1000 ft/min ramp. Ramp the humidity to 95 percent relative humidity (RH) at an average 6%/min. Ramp the temperature to 50°C at a rate of 5K per minute.
- 2) Warm/Moist Dwell. Maintain 50°C, site pressure, and 95 percent relative humidity for 30 minutes.
- 3) Ramp to Cold Altitude. With the test item operating, ramp the temperature to 0°C at a rate of 5K per minute. Ramp the humidity to ambient conditions at an average ramp rate of 6% RH/min. Ramp the chamber from the test site pressure to 15000 ft pressure altitude. Perform the pressure ramp at 1000 ft/min.
- 4) Cold Soak. Allow the test item to soak at 0°C, 15000ft and uncontrolled humidity for 1 hour.
- 5) Repeat the cycle (steps 1 to 4) two (2) times
- 6) Bring to ambient conditions

Note: Depending on equipment capability, temperature, humidity and pressure ramping should be performed as coincidentally as possible.



Test Procedure

1) Bring the environment to Warm/Moist. With the test item operating, the chamber from the test site altitude pressure to 15000 ft pressure altitude at a ramp rate of 1000 ft/min ramp. Ramp the humidity to 95 percent relative humidity (RH) at an average 6%/min. Ramp the temperature to 50°C at a rate of 5K per minute.

2) Warm/Moist Dwell. Maintain 50°C, site pressure, and 95 percent relative humidity for 30 minutes.

3) Ramp to Cold Altitude. With the test item operating, ramp the temperature to 0°C at a rate of 5K per minute. Ramp the humidity to ambient conditions at an average ramp rate of 6% RH/min. Ramp the chamber from the test site pressure to 15000 ft pressure altitude. Perform the pressure ramp at 1000 ft/min.

4) Cold Soak. Allow the test item to soak at 0°C, 15000ft and uncontrolled humidity for 1 hour.

5) Repeat the cycle (steps 1 to 4) two (2) times

6) Bring to ambient conditions

Note. Depending on equipment capability, temperature, humidity and pressure ramping should be performed as coincidentally as possible.

°C
Humidity / %RH
Pressure / psia

A photograph of an airplane wing, likely from a Boeing 737, showing the fuselage and the wing structure. The wing is white with black markings, including a registration number 'N11111' and a cross symbol. The background is a clear blue sky. The word "DUST" is overlaid in a large, stylized, orange, hand-drawn font across the center of the image. The letters are thick and have a slightly irregular, artistic appearance. The 'D' is particularly large and has a small white rectangular object attached to its top left corner. The 'U' and 'S' are also quite large and have a similar hand-drawn style. The 'T' is also large and has a simple, blocky design. The overall composition is centered, with the wing and the text being the main focus.

DUST

Test campaign at
FAA Tech Center



Test Procedure

When subjected to dust [...] the instrument must not suffer a false alarm and operate electrically and/or mechanically, where appropriate.

The test dust shall be ISO12103-1 Ultra Fine or equivalent.

Minimum test concentration: 7%/ft (23%/m). The average test concentration shall be maintained for at least one minute.

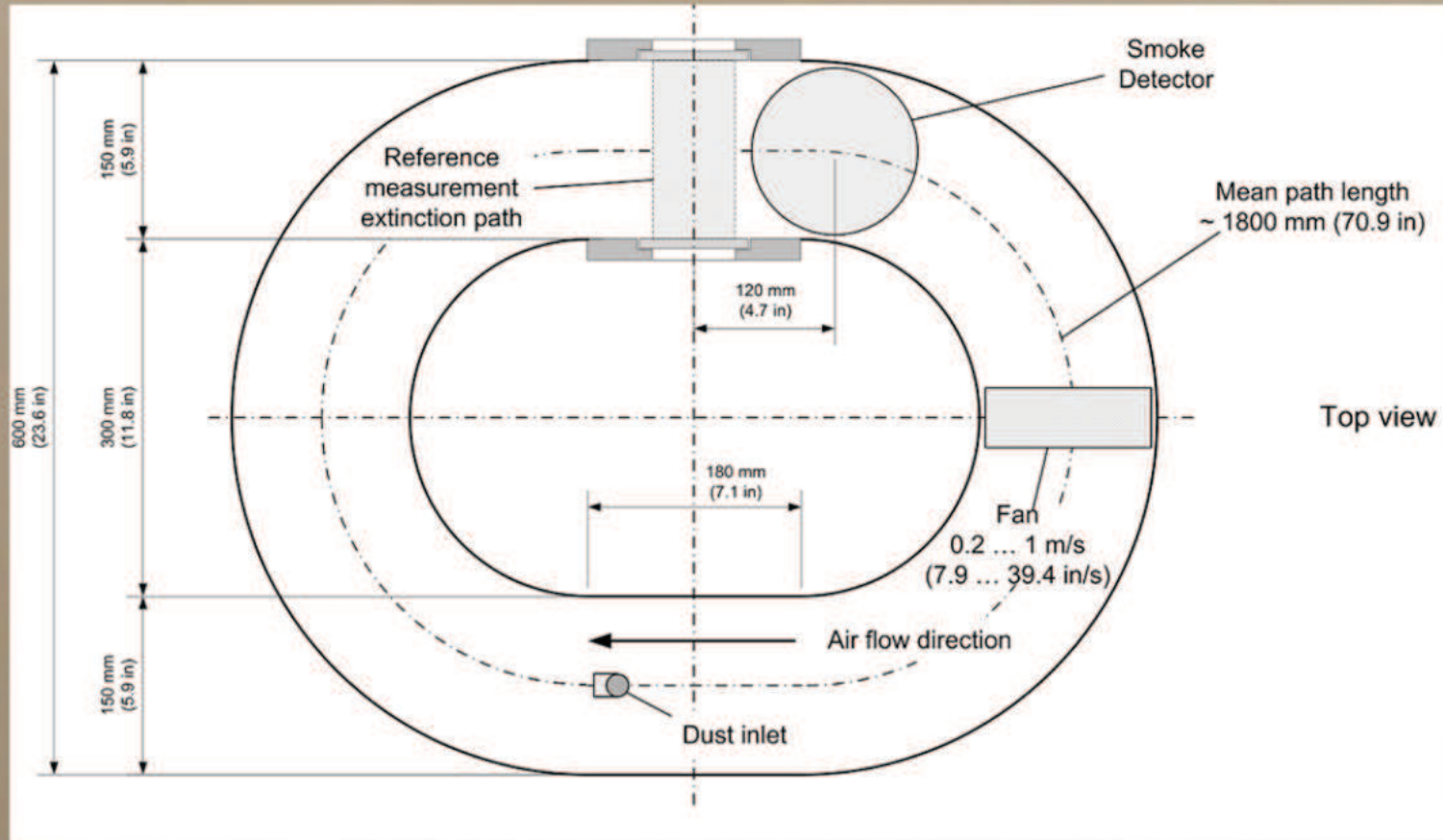


For detectors calibrated to alarm at a smoke level X (where $X \geq 5\%/ft$ (16.4%/m)) obscuration, the detector shall not alarm when subjected to a minimum test dust concentration of $7\%/ft$ (23%/m).

For detectors calibrated to alarm at a smoke level Y (where $Y > 5\%/ft$ (16.4%/m)) obscuration, the detector shall not alarm when subjected to a minimum test dust concentration of $Y + 2 \%/ft$ (6.6%/m).

The test shall be repeated 3 times. All consecutive tests shall be passed.

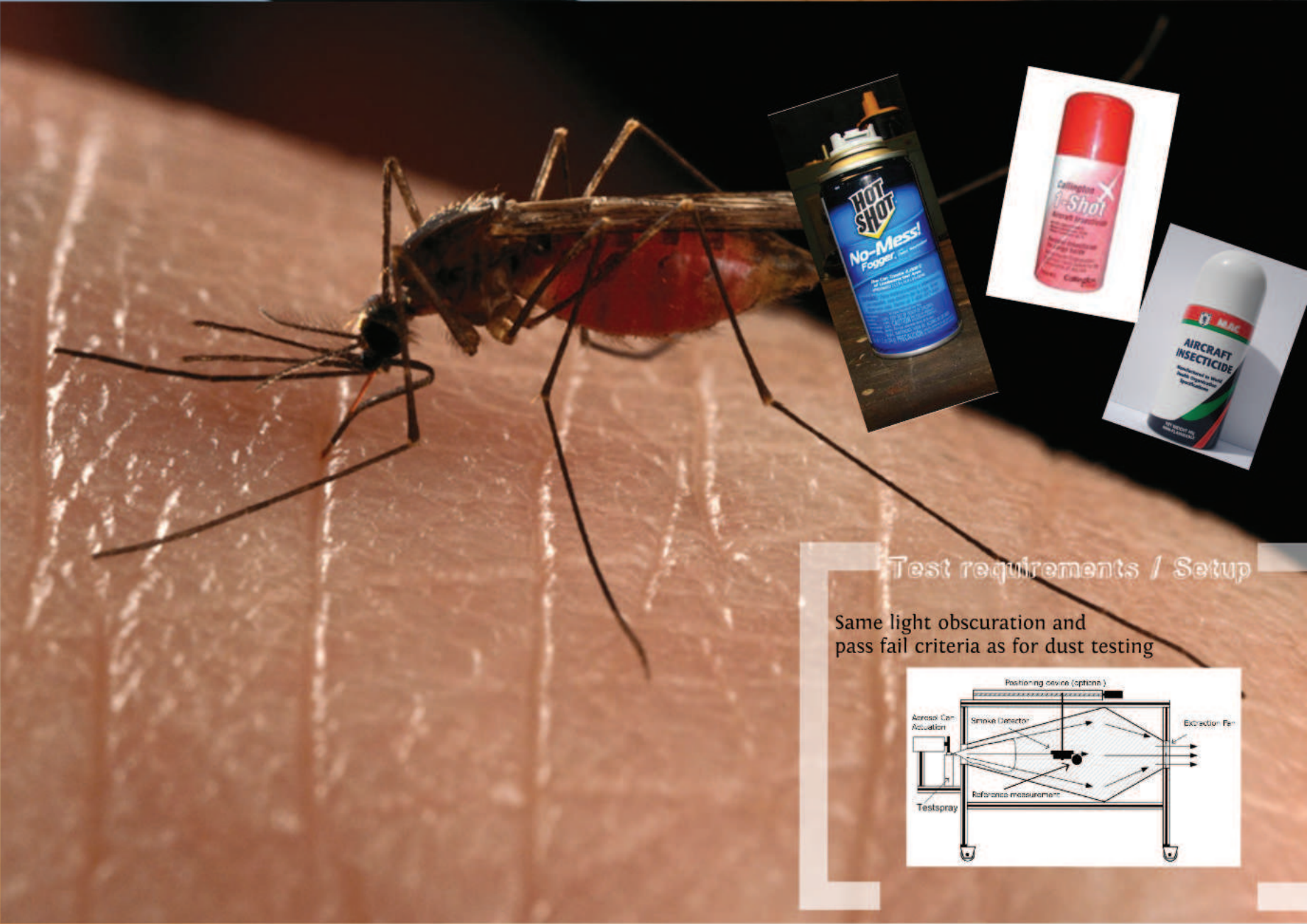
Test setup



- reproducible slope
- controlled dust injection
- rotational air speed
- laminar flow

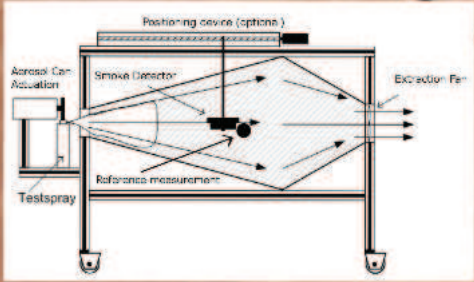
A photograph taken from a low angle looking up at the underside of a white airplane. The sky is a gradient of light blue to orange, suggesting a sunset or sunrise. The word "INSECTICIDE" is overlaid in large, bold, red, outlined letters across the center of the image. The airplane's underbelly features several small windows and markings, including two crosses on the fuselage. The background shows a dark horizon with some distant structures and a runway or tarmac area at the bottom.

INSECTICIDE



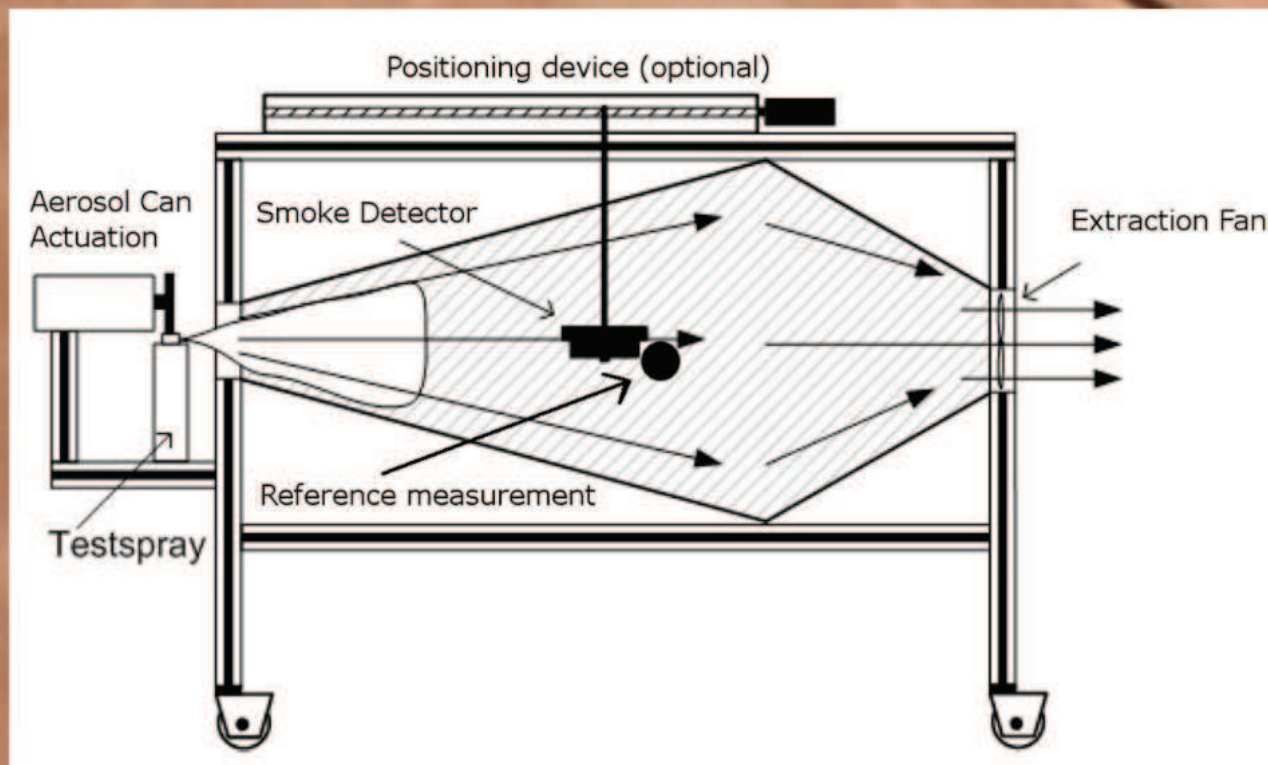
Test requirements / Setup

Same light obscuration and pass fail criteria as for dust testing



Test requirements / Setup

Same light obscuration and pass fail criteria as for dust testing





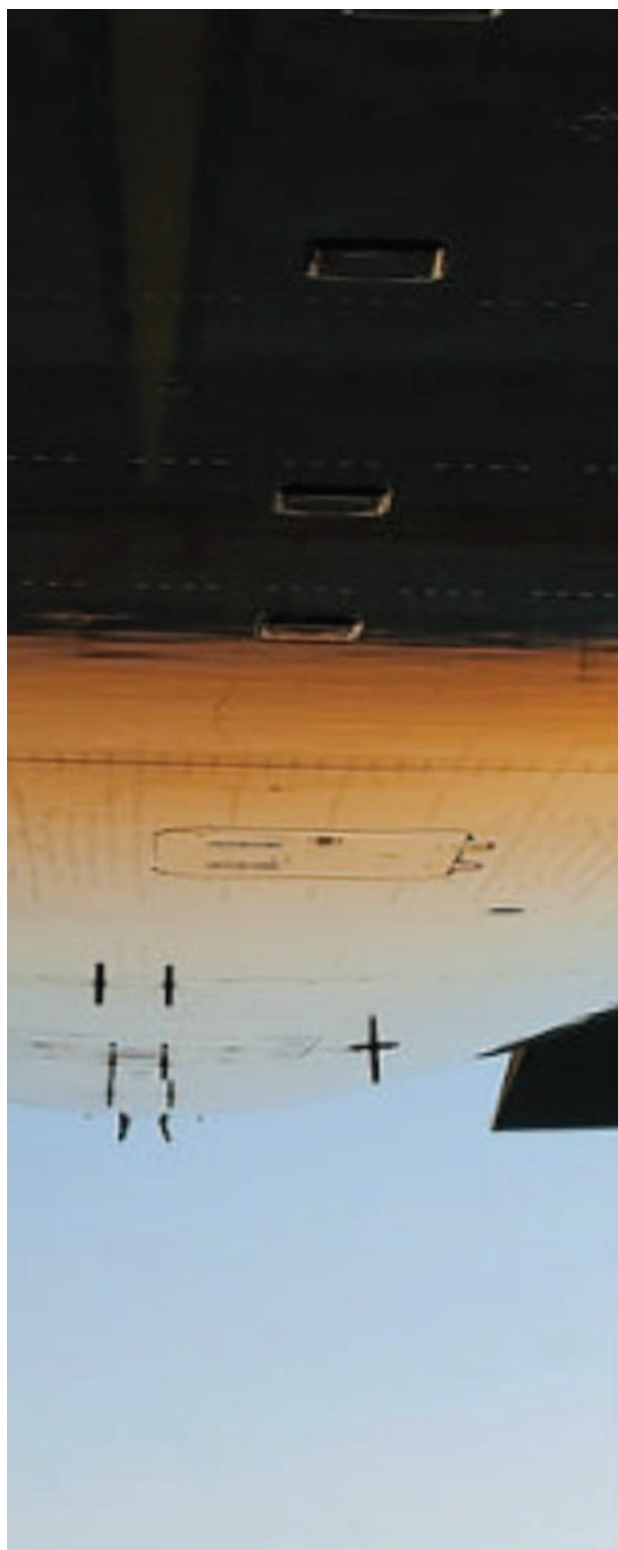
LIGHT

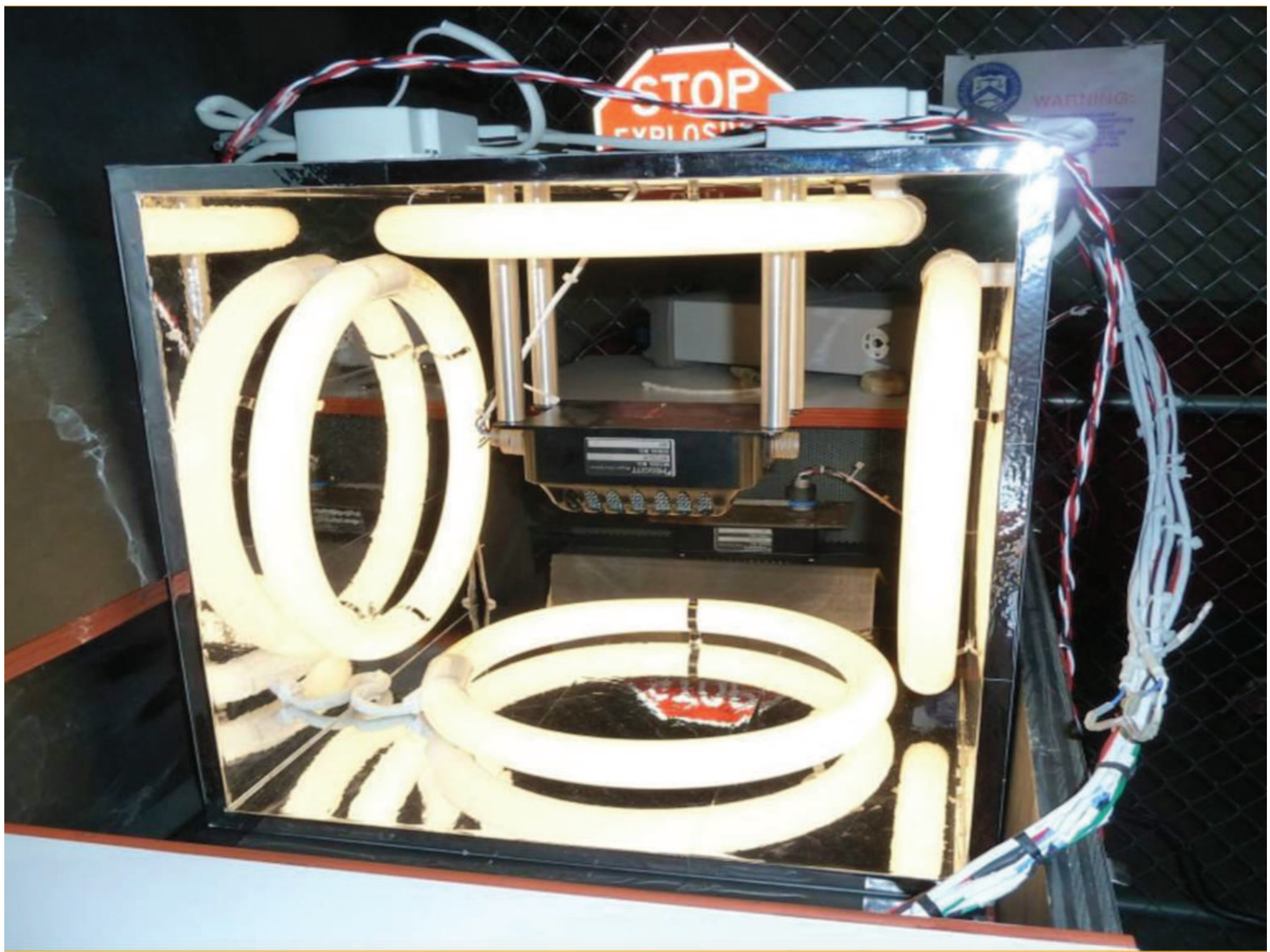
Ambient Light

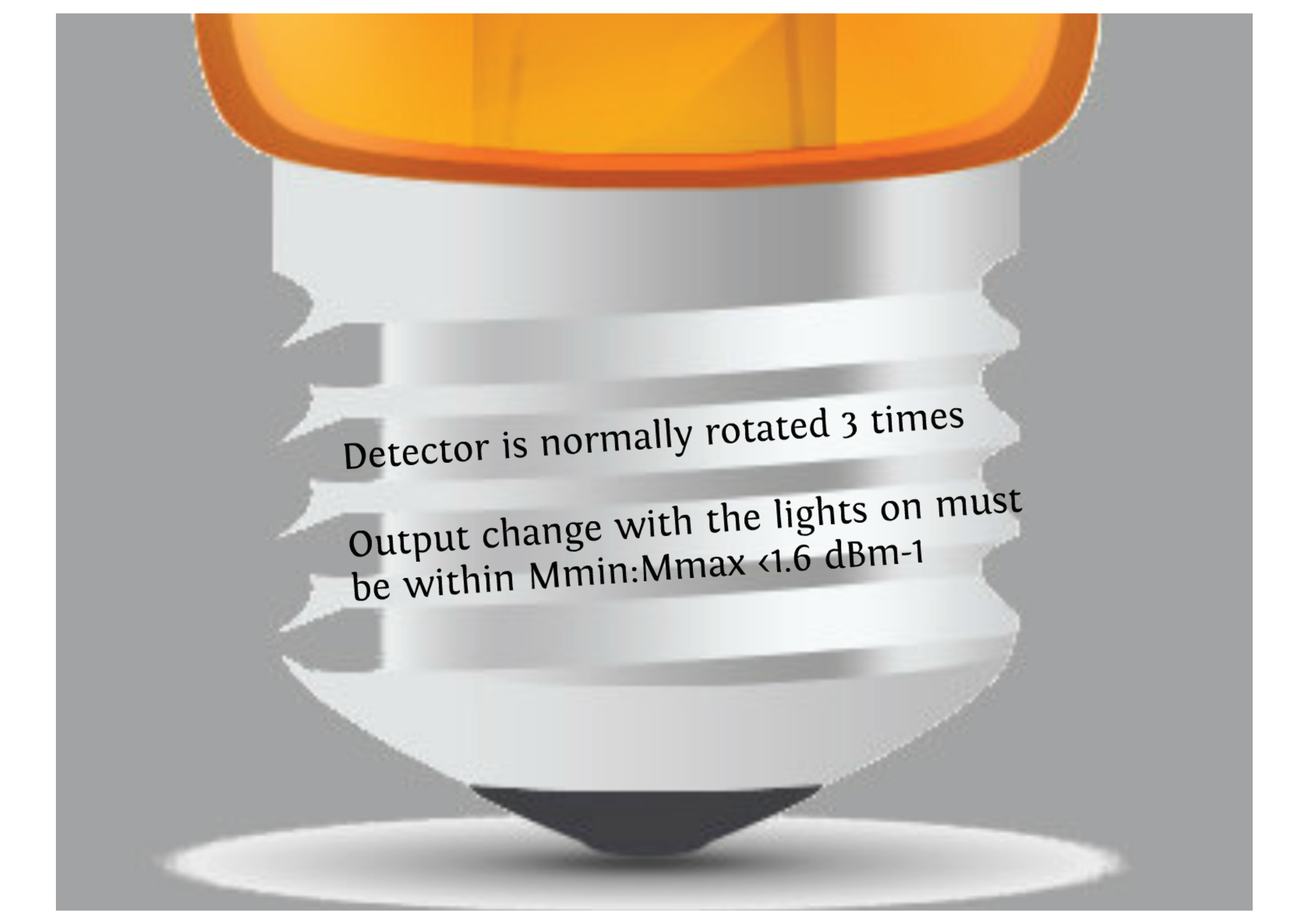
Test procedure
according to EN54

Detector is normally rotated 3 times

Output change with the lights on must
be within $M_{min}:M_{max} \leq 1.6 \text{ dBm}^{-1}$







Detector is normally rotated 3 times

Output change with the lights on must
be within $M_{min}:M_{max} < 1.6 \text{ dBm}^{-1}$

Summary

- AS8036 working committee has finalized the work on the SAE standard
- Public ballot to start
- TSO C1e to be issued after release of AS8036