

Full-scale Demonstration Testing with a Solid Aerosol Fire Extinguishing Agent, Discussion Continued



Federal Aviation
Administration

Presented to: International Aircraft Systems Fire
Protection Working Group

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Presentation Overview

Major Discussion Points

- **Purpose**
- **Test Conditions**
- **Project Status**
- **Various Views of the Test Article**

References within this presentation to businesses, their services and/or products does not constitute endorsement.



Full-scale Demonstration Testing, Solid Aerosol

Purpose

- FAA invoked “Part B” of MPSe rev04
 - generally,
 - “part B” is pertinent testing to demonstrate the proposed certification criteria from MPSe “part A” are plausible on an airframe
 - invoked due to differences between (a) the halon-replacement candidate, its use, and/or its measurement, and (b) the state-of-the-art¹
 - the subject candidate differs from the state-of-the-art
 - is a solid aerosol fire extinguishing agent
 - utilizes a different quantifying concept & analyzer

1. State-of-the-art is defined by the use of halon 1301, stored in a bottle having a pressurized/gaseous ullage, is delivered/injected to the nacelle fire zone by tube, and its distribution within the volume is measured with a Statham-derivative gas analyzer.

Full-scale Demonstration Testing, Solid Aerosol

Test Conditions

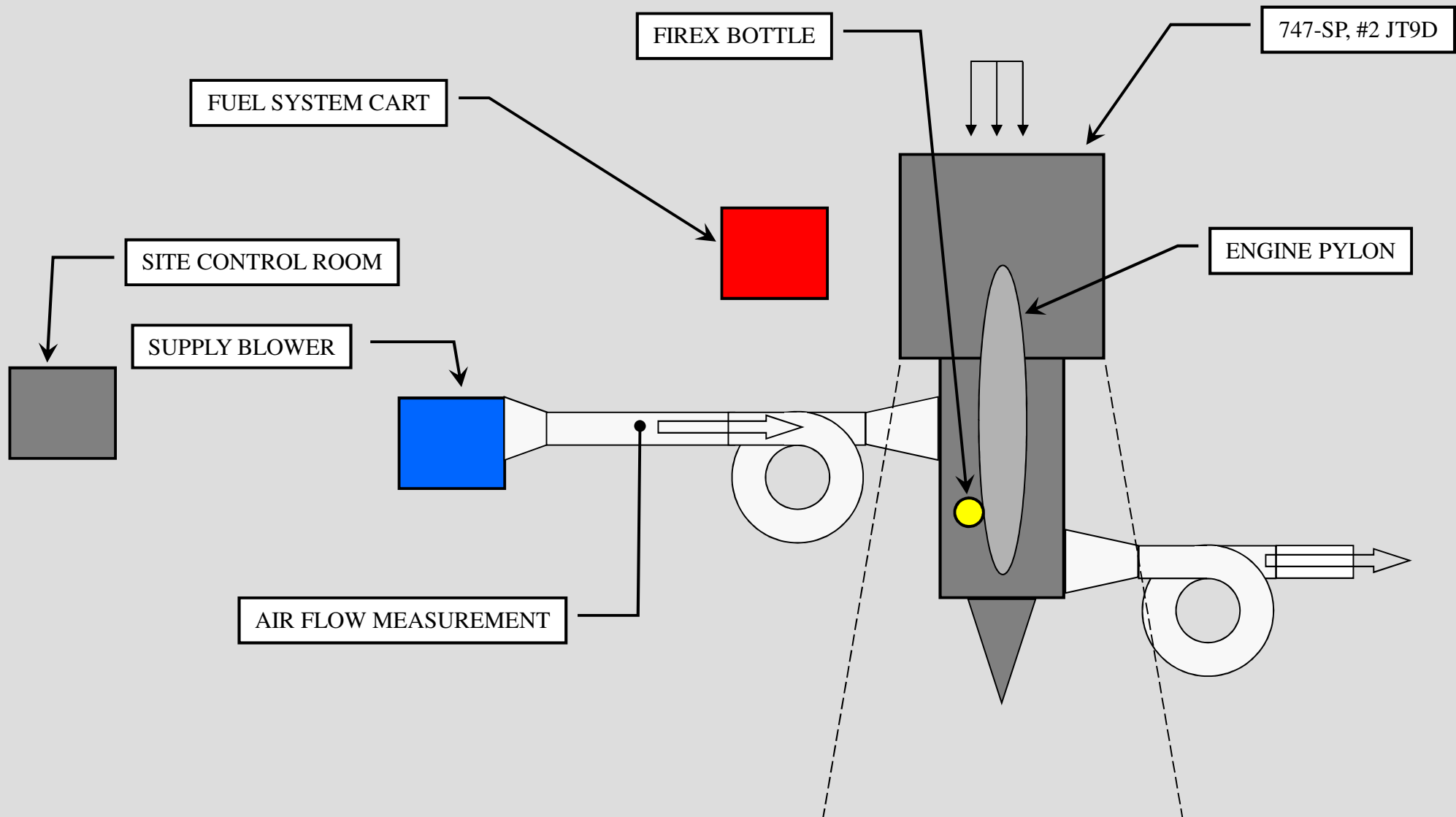
- Using a JT-9D engine on the FAA-owned 747-SP
- JT-9D nacelle forcibly ventilated externally from atmosphere
 - air mass flow ≈ 0.5 kg/s (1 lbm/s)
 - engine run will precede each test to “heat” the nacelle
- Fire threats are spray- & pool-based, fueled with JP-8
- Industry is providing firex system & concentration analyzer
 - sodium bicarbonate-based solid aerosol
 - optical system used to measure agent distribution

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Project Status

- Currently nearing the end of the “build-up” phase
 - industry decided to use a JT-9D located at the FAATC, May 2011
 - complete external engine nacelle ventilation, Oct 2011
 - finalize aircraft engine use, Nov 2011
 - complete spray & pool fire threats, Feb 2012
 - firex agent distribution, completion imminent; begun Feb 2012
- Anticipating :
 - completion, Jun 2012
 - draft report into publication process, Sep 2012

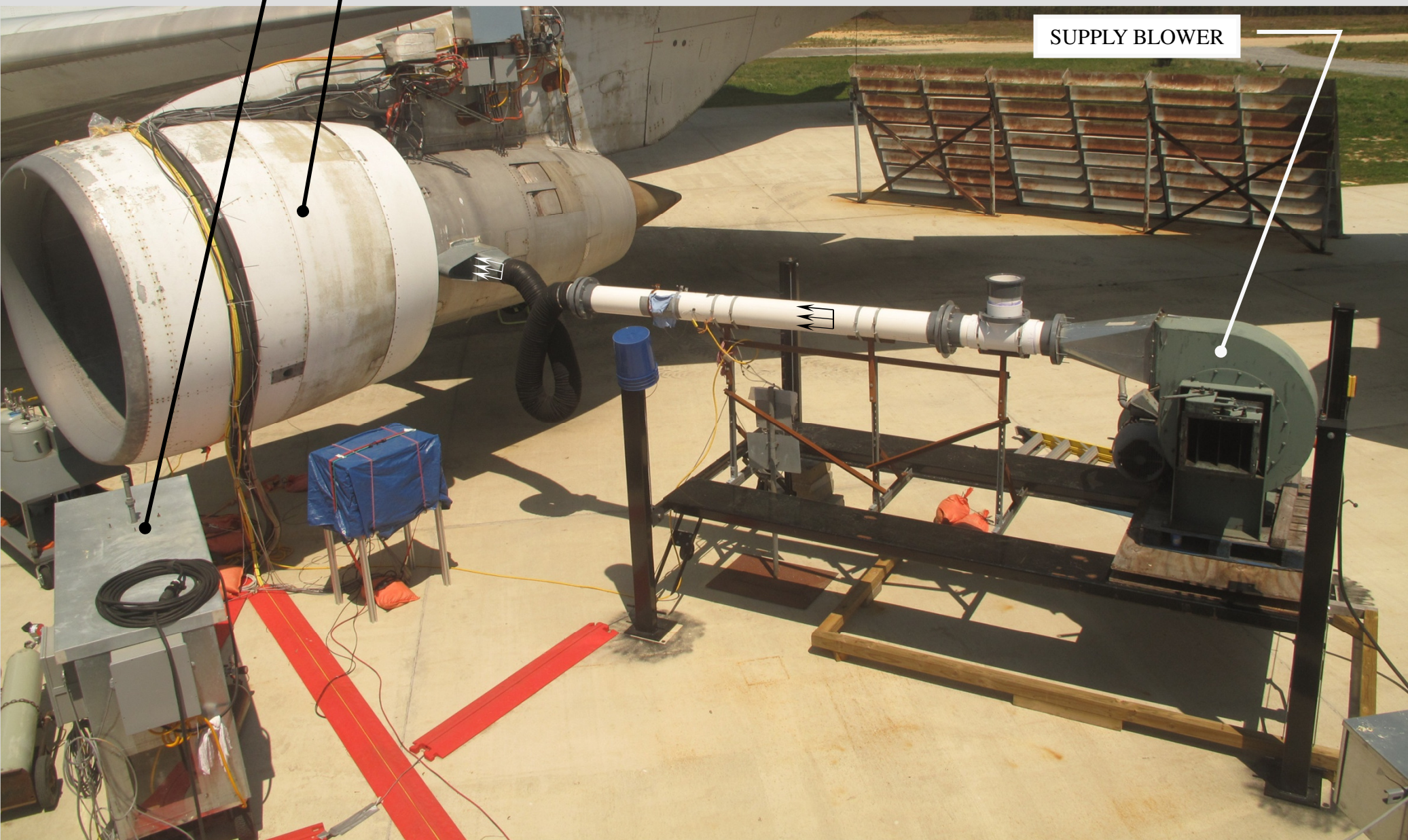
Various Views, Schematic Site Plan

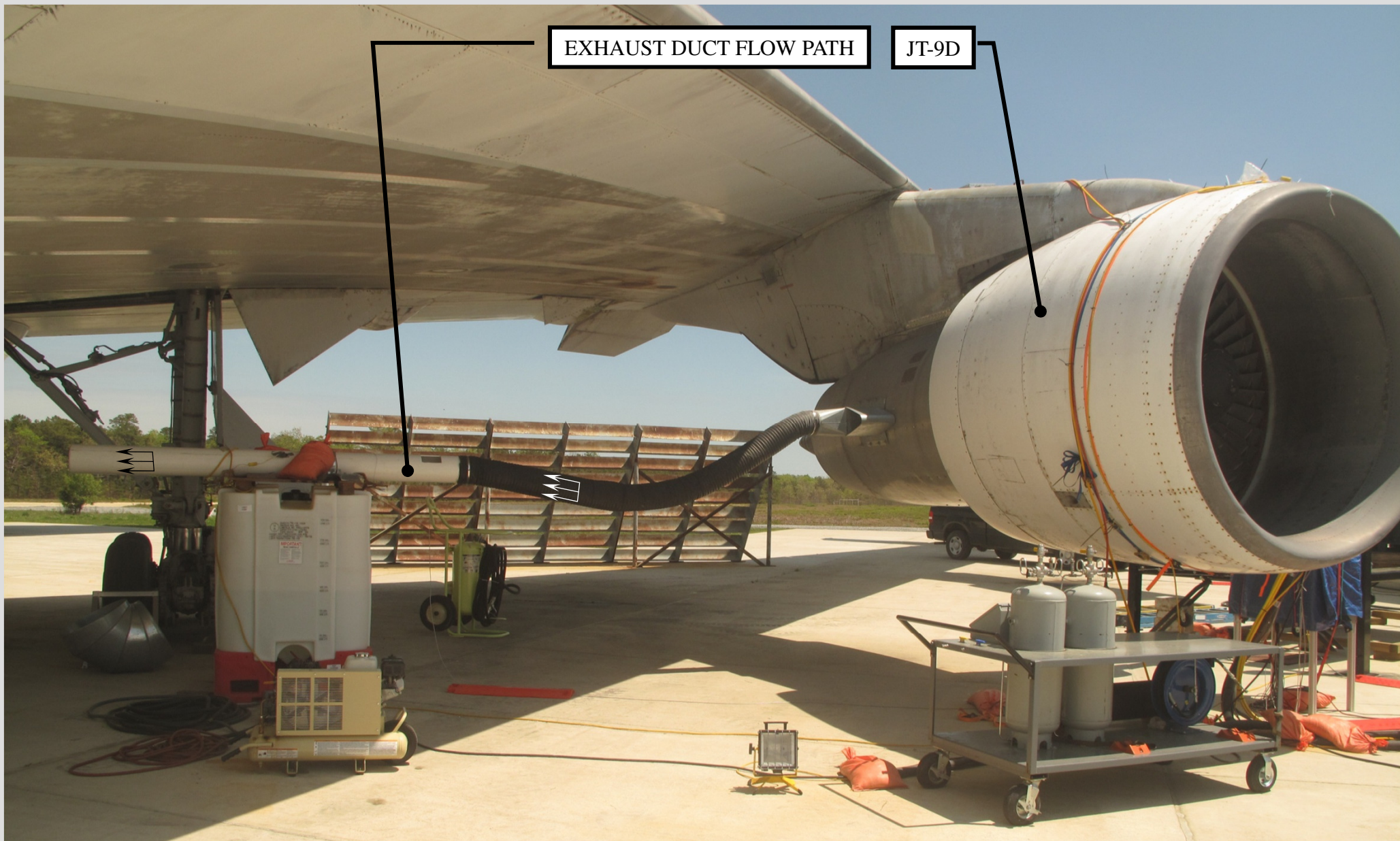


FUEL SYSTEM CART

JT-9D

SUPPLY BLOWER

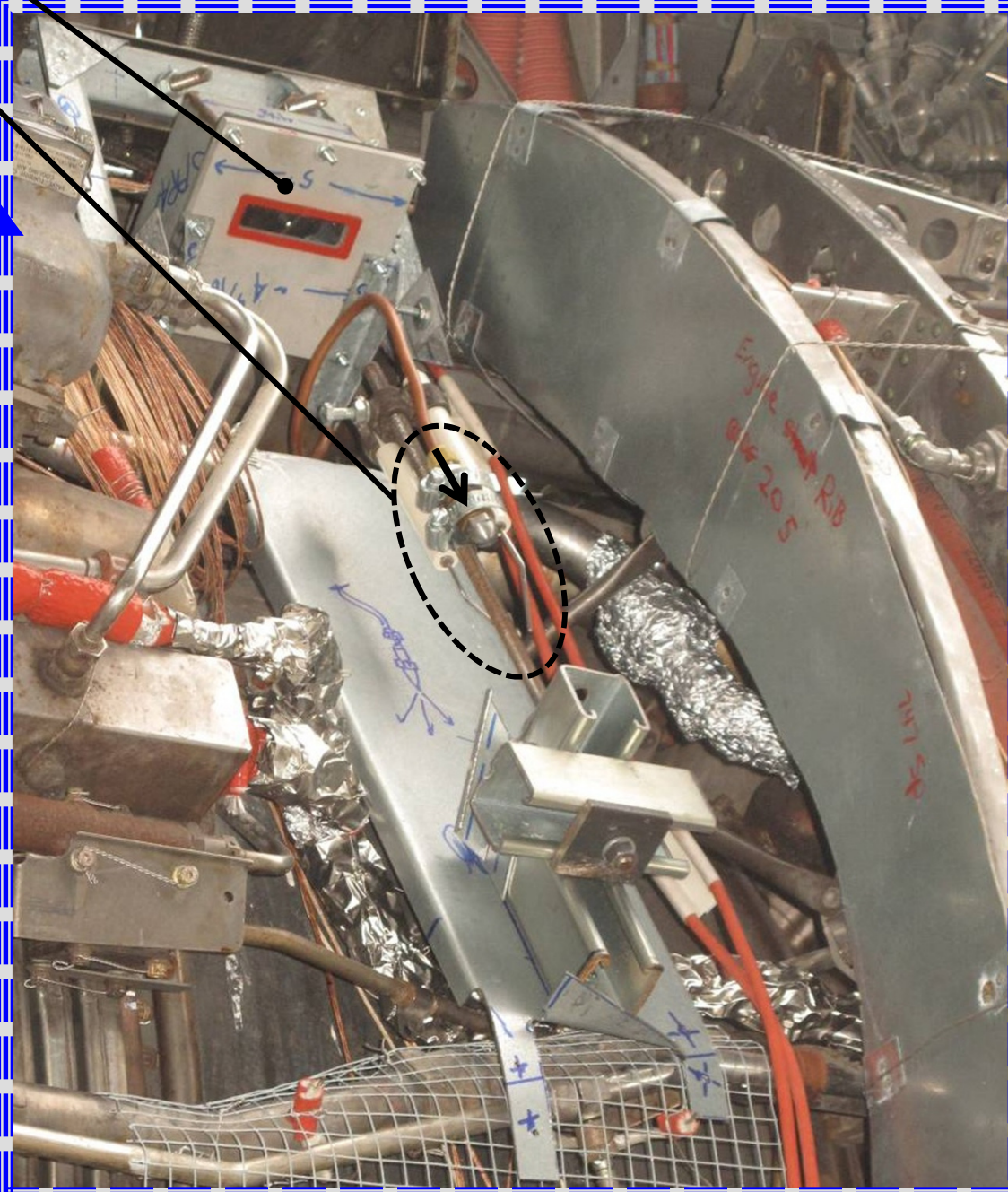




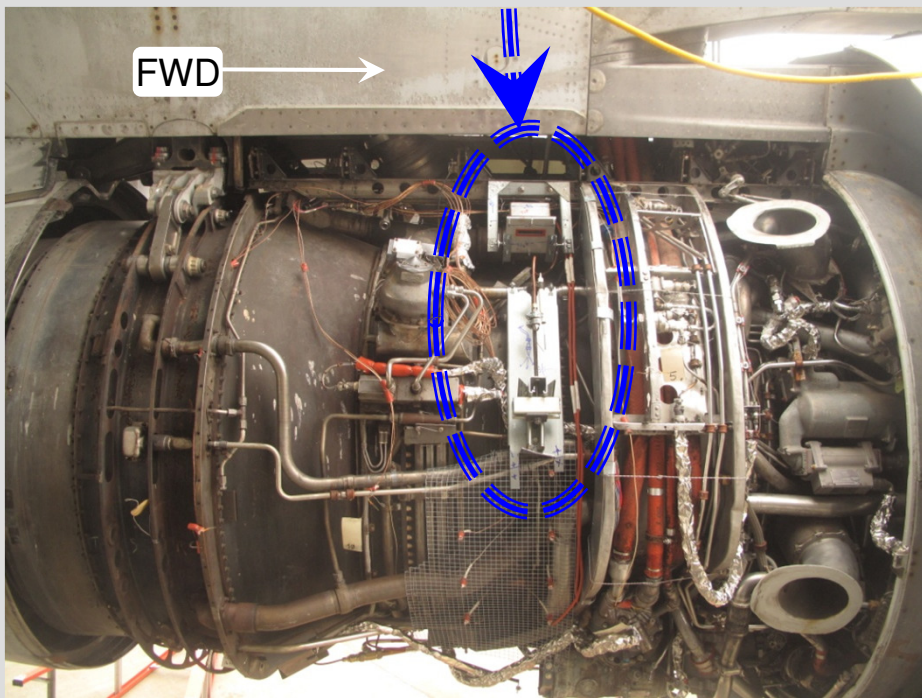
CAMERA, SPRAY FIRE

Various Views, Nacelle Fire Threat, Spray ("new")

FUEL SPRAY NOZZLE &
ELECTRICAL IGNITION ARC GAP



FWD

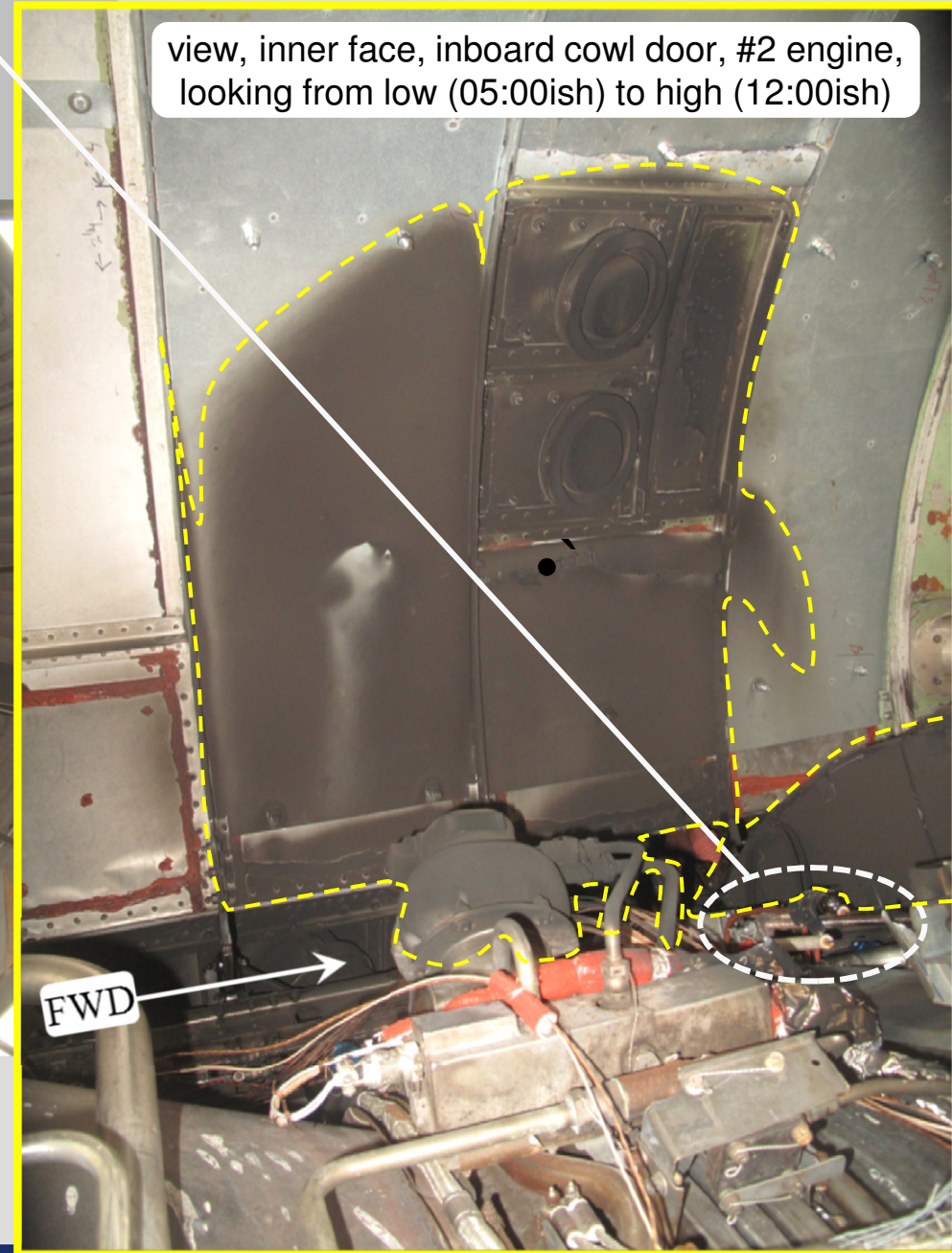
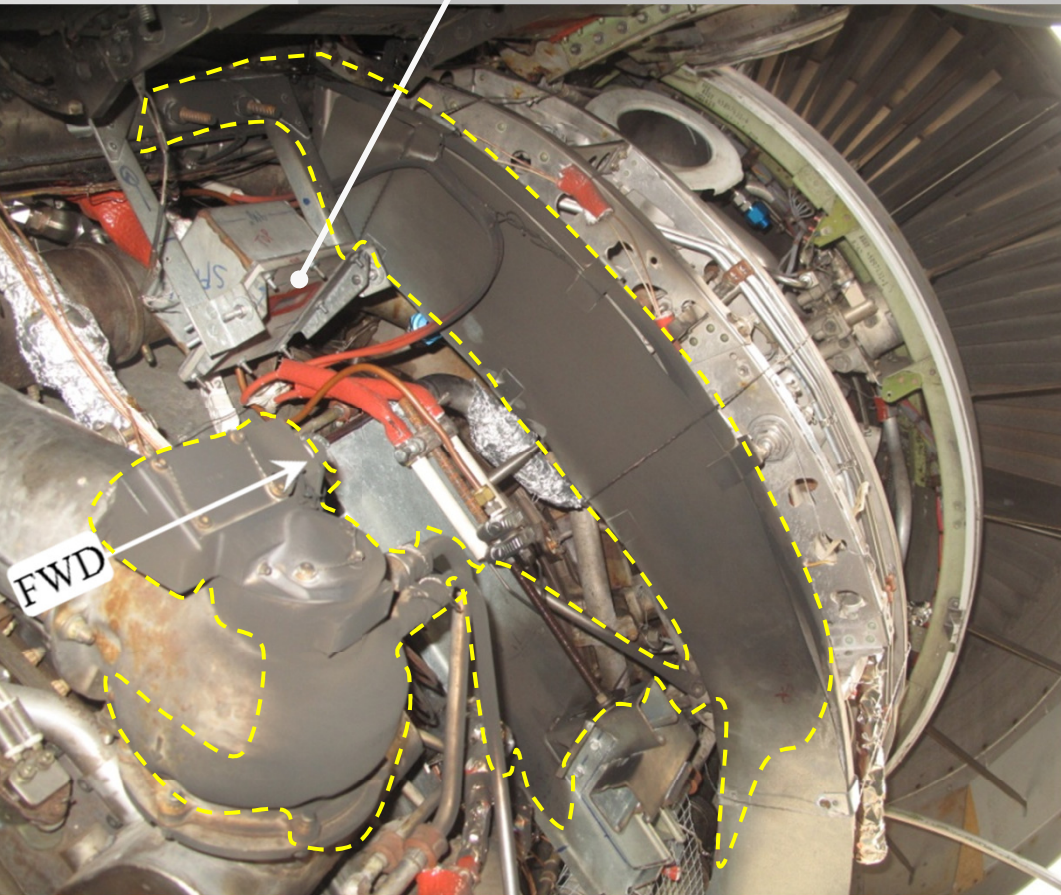


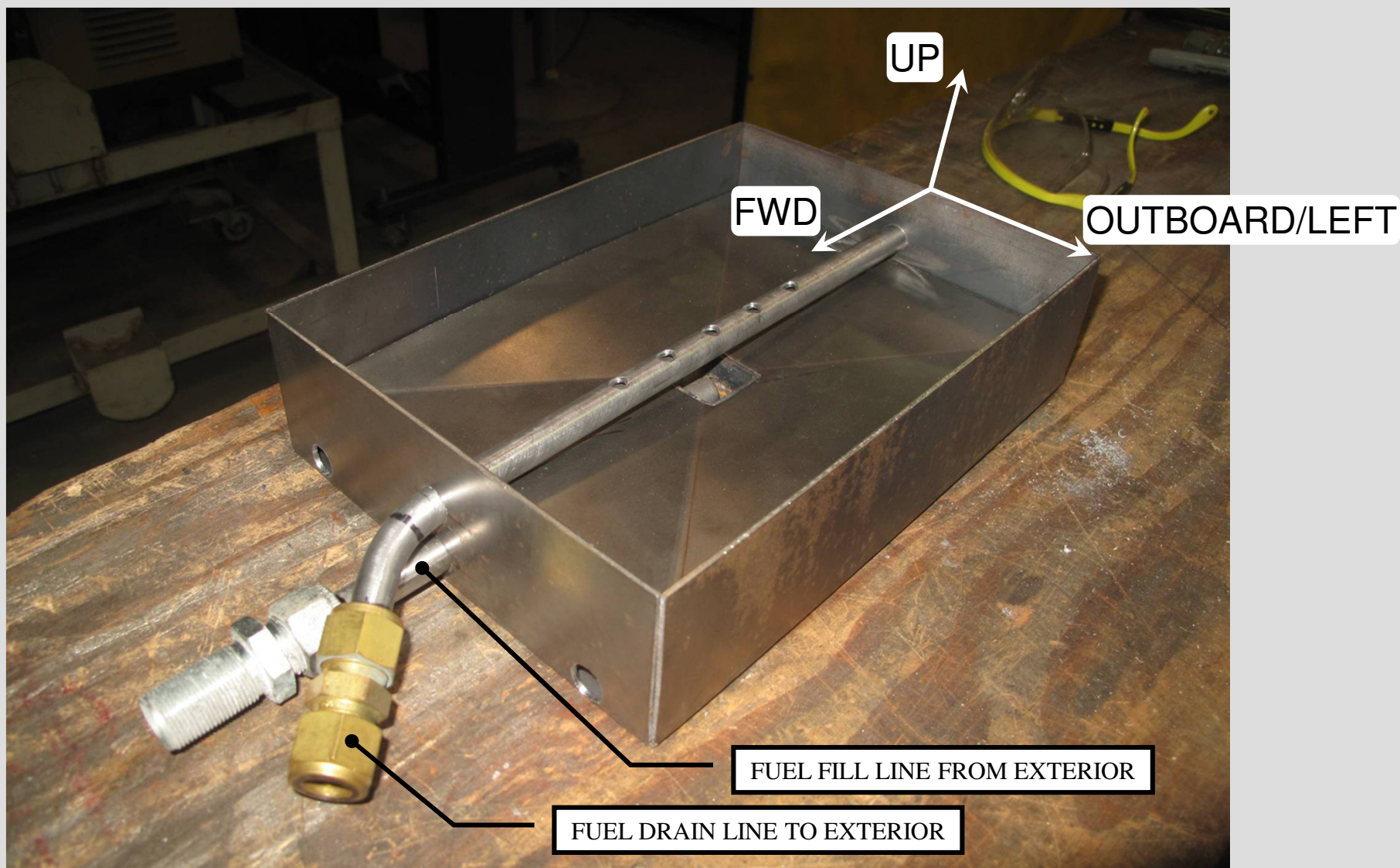
Various Views, Nacelle Fire Threat, Spray (after 1ST fire)

FUEL SPRAY NOZZLE &
ELECTRICAL IGNITION ARC GAP

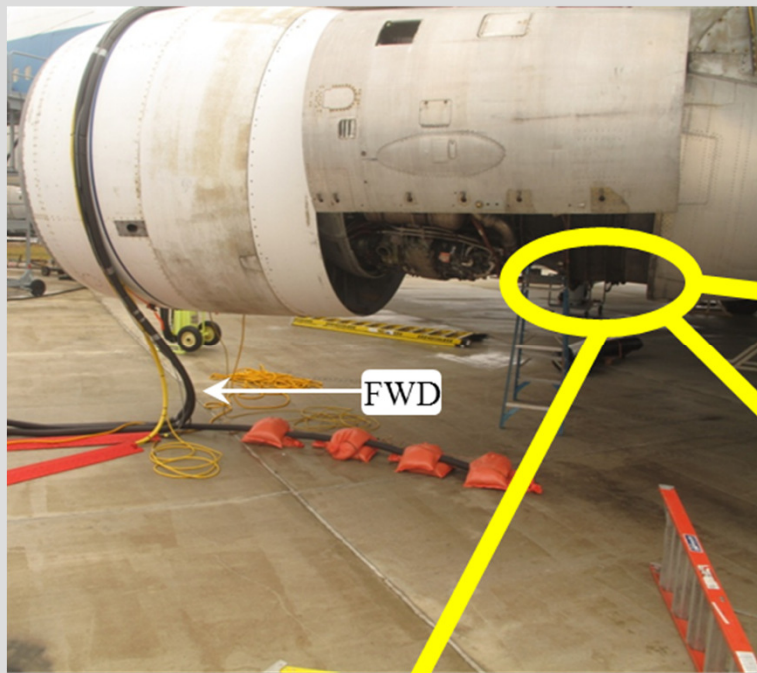
CAMERA, SPRAY FIRE

view, inner face, inboard cowl door, #2 engine,
looking from low (05:00ish) to high (12:00ish)

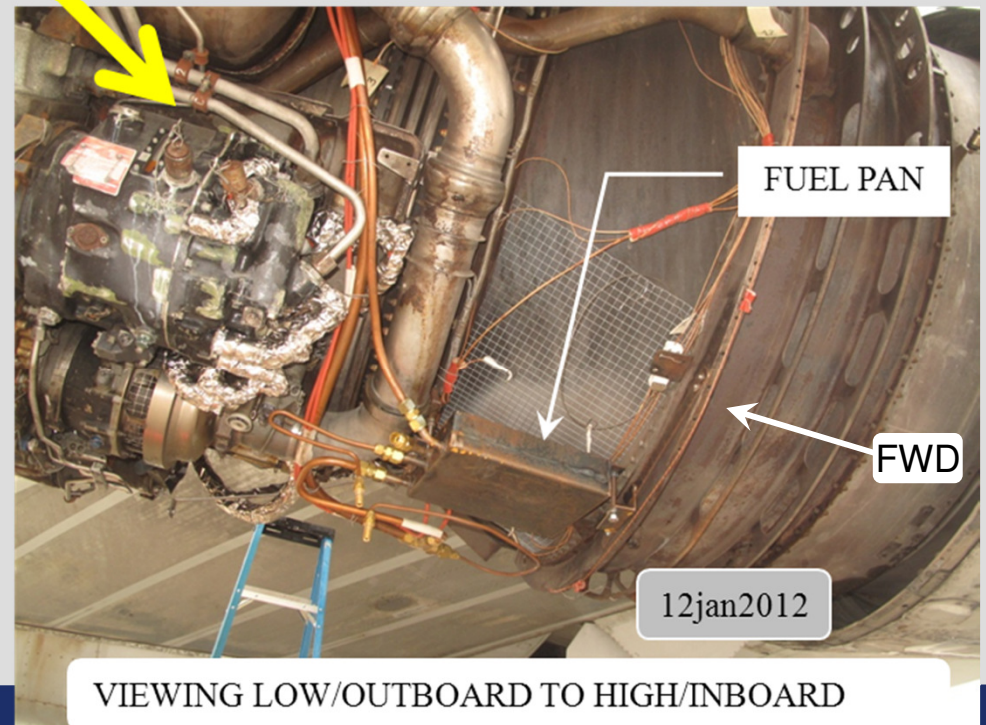
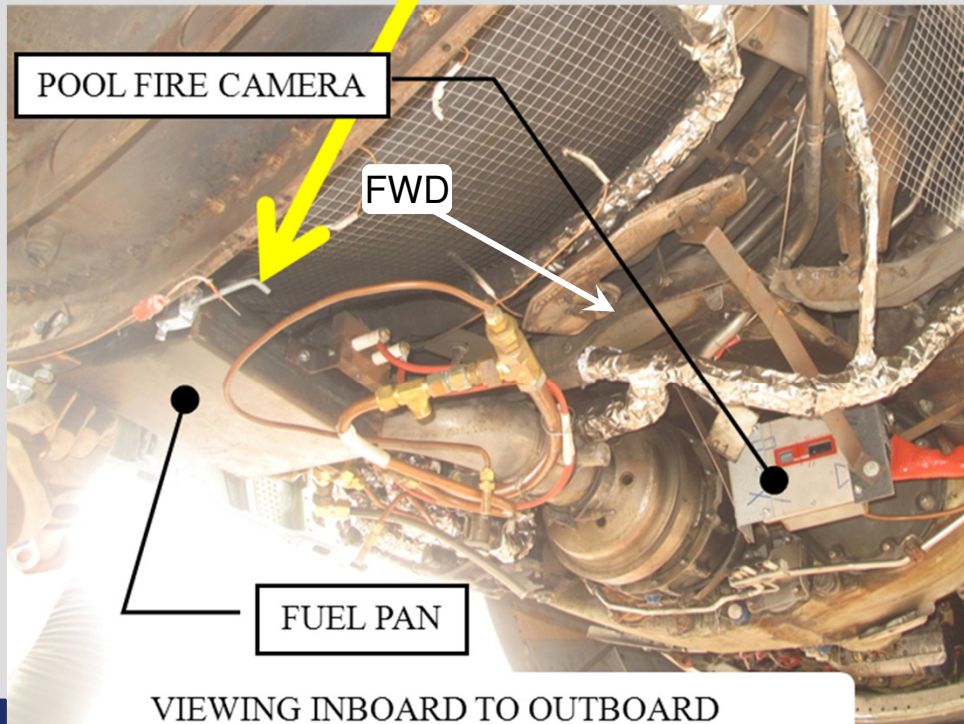
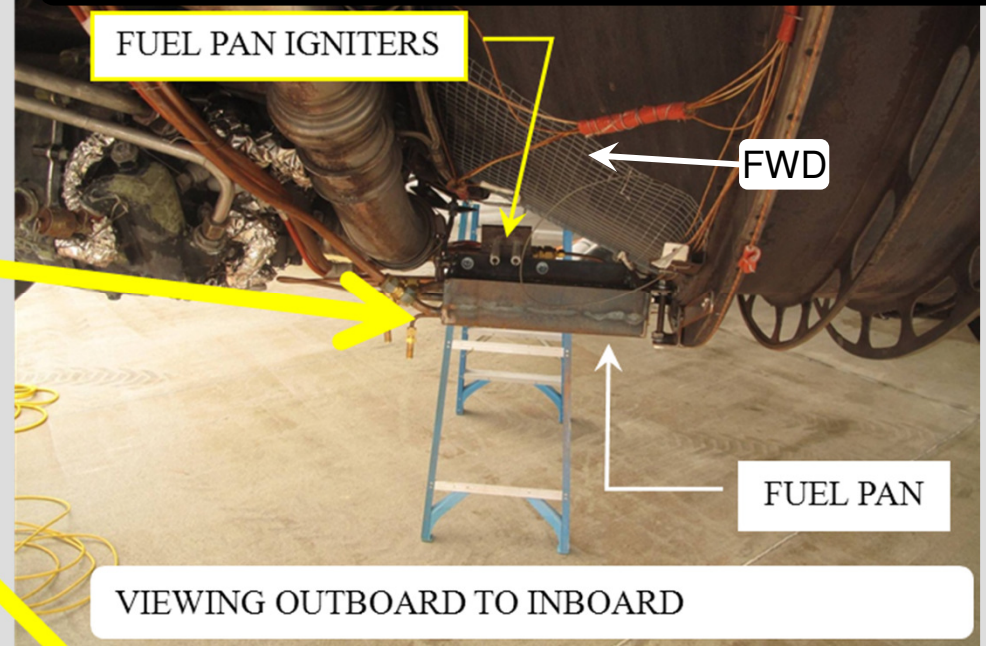




- Fuel pan is made from 14 ga steel & approximately 191 x 267 x 64 x mm tall (7.5 x 10.5 x 2.5 in tall)
- Fuel puddle depth approximately 13 mm (0.5 in) & freeboard (dry lip) height approximately 25 mm (1 in)
- Base of fuel pan is hollow permitting water flow from a water jacket

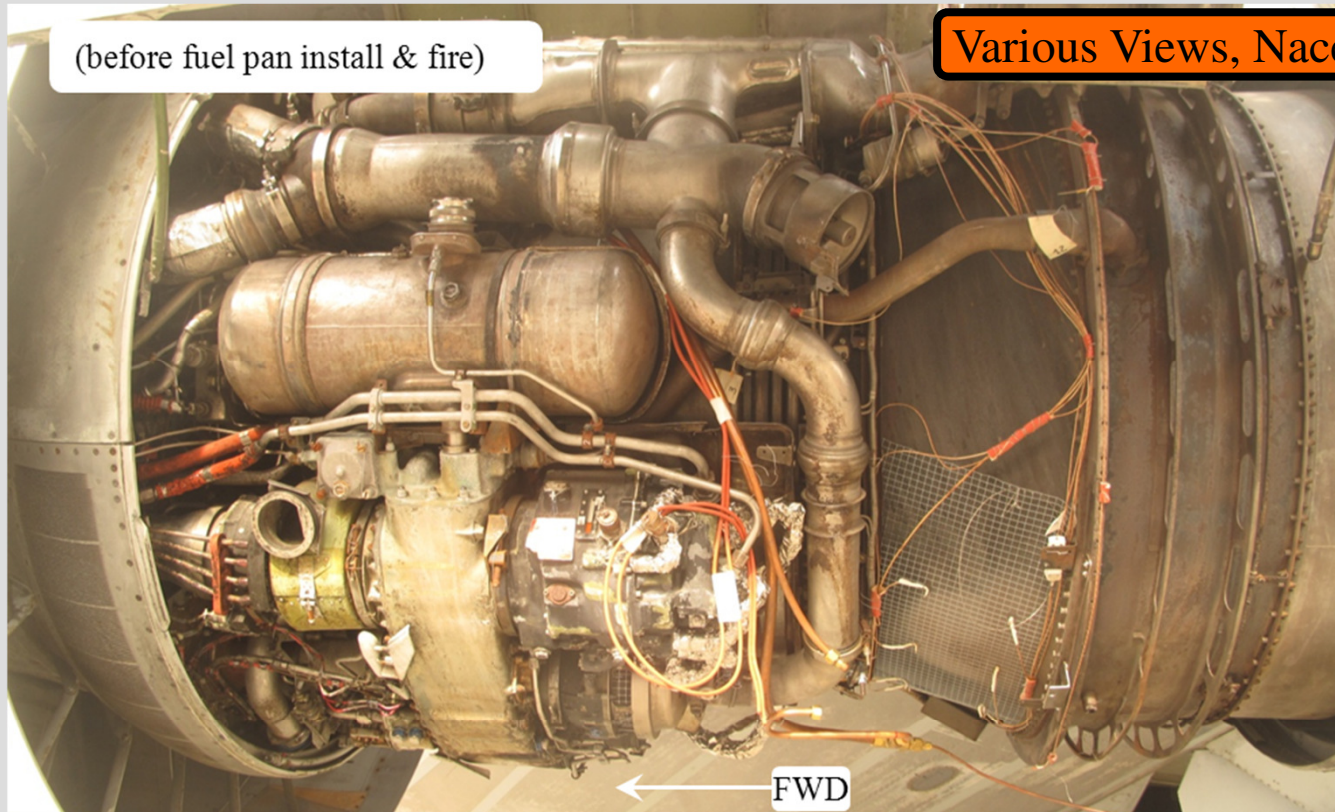


Various Views, Nacelle Fire Threat, Pool ("new")

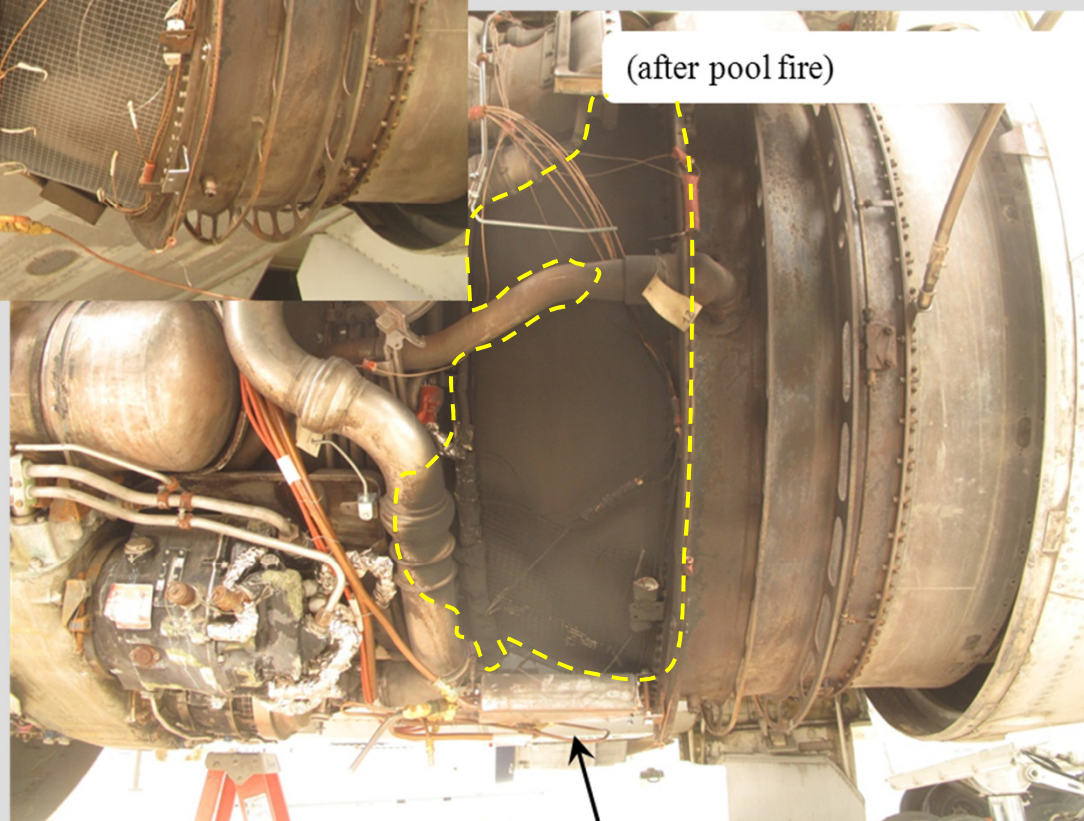


(before fuel pan install & fire)

Various Views, Nacelle Fire Threat, Pool (after 1ST fire)



VIEWING OUTBOARD TO INBOARD



(after pool fire)

FUEL PAN

End

- **Acronyms, definitions, short-hand notations**

FAA = United States Federal Aviation Administration

FAATC = FAA W.J. Hughes Technical Center

firex = fire extinguishing or fire extinguishing system

FWD = forward

MPSe = Minimum Performance Standard for Halon Replacement in Civil Aircraft Engine Nacelle & APU
Compartments

rev = revision