## International Aircraft Materials Fire Test Forum Meeting

# Development of New Flammability Test for Magnesium-Alloy Cabin Components

Presented to: International Aircraft Materials Fire Test

Forum, Virtual Meeting

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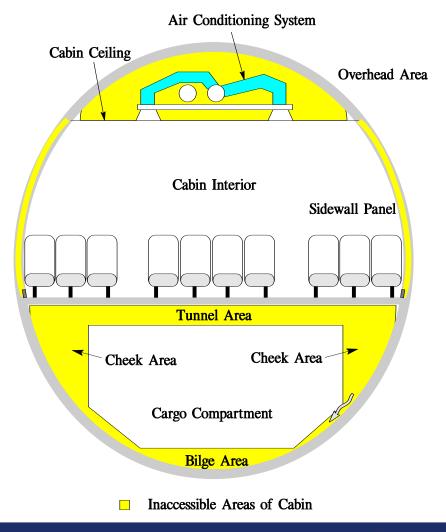
Date: June 8, 2020



Objective: Develop a flammability test for magnesium alloy components located in inaccessible areas of the cabin

- Representative
- Repeatable
- Reproducible

## Development of Flammability Test for Magnesium Components Used in Inaccessible Areas



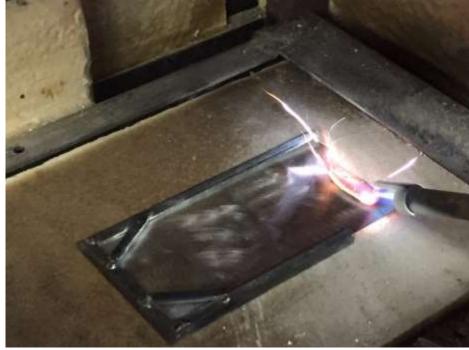
#### **Current Test Parameters**

- Test developed using Radiant Panel Apparatus
- 3- by 6-inch sample size, 0.025-inch thickness
- 2-minute pilot ignition
- 4-minute exposure to radiant heat
- Maximum weight loss of 30% (proposed)

...Test Method inserted as Chapter 26 in current Fire Test Handbook!

### Truncated Perimeter Sample Holder





#### Interlab Study

Prepare identical samples for participating laboratories, to determine lab-to-lab reproducibility:

- <u>Test materials received from Luxfer (Magnesium Elektron) 2019</u>
- Materials were manufactured to 0.125-inch thickness, which were then milled down further to 0.025-inch thickness by FAA for testing
- 8 laboratories (Airbus, Boeing, Accufleet, DGA, Skandia, FAA, Govmark, Honda)
- For first round of testing, 2 types magnesium alloy (EL43, Boeing material "alloy1")
- 21 samples of EL43, 10 samples alloy1 per lab
- Test results compiled by FAA

<sup>\*</sup>Refine test parameters and pass/fail criteria based on results of interlab study

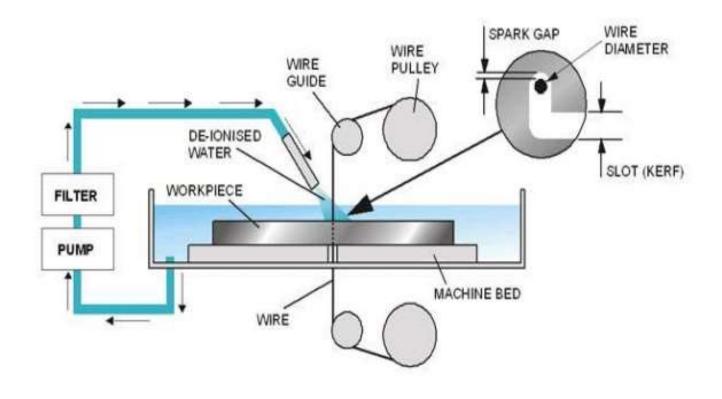
#### **EDM Electrical Discharge Machining**

FAA contracted US company to mil samples down from 0.125-inch supplied thickness to specified 0.025-inch thickness for testing.

Wire EDM is an electro thermal manufacturing process where components are made using electrical discharges. A thin strand of metal wire accompanied by de-ionized water allows the wire to cut through metal just from the heat of the sparks.

#### **EDM Electrical Discharge Machining**

## EDM Wire cutting

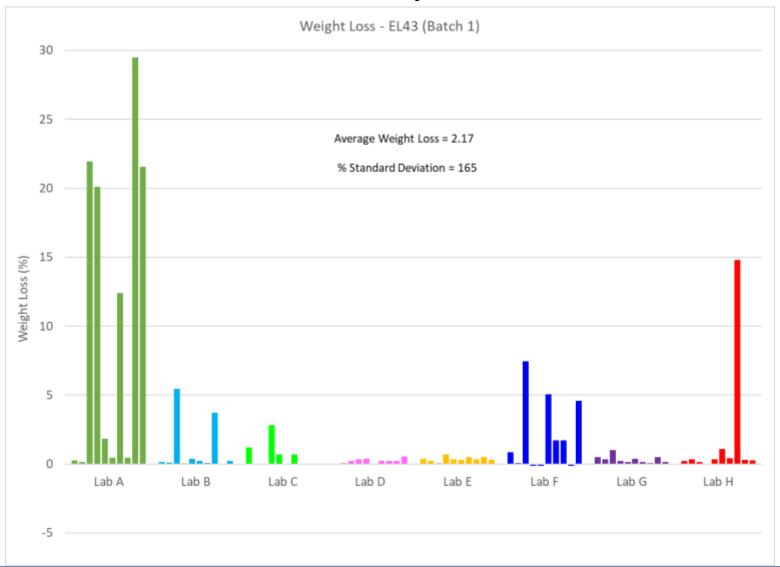




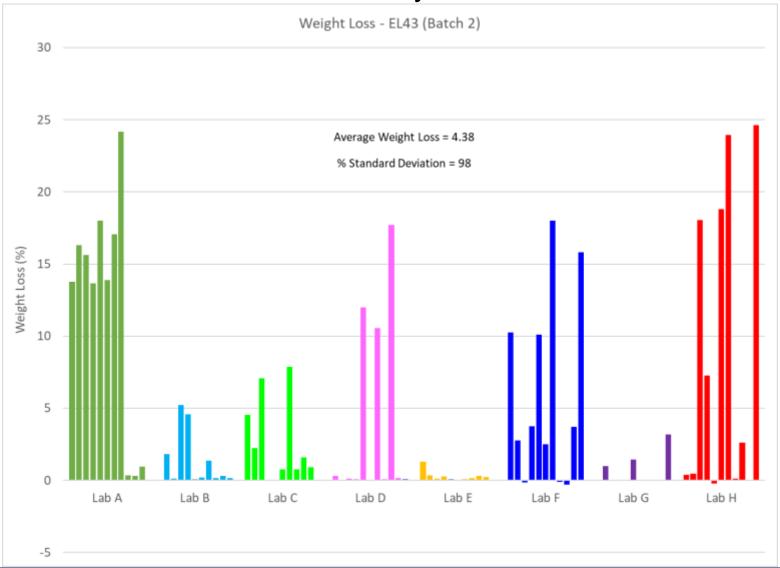




#### Interlab Study Results



### Interlab Study Results



## Interlab Study Results



## Lab A Test Configuration



#### Interlab Study Observations

EDM samples in Batch 1 and Batch 2 were very inconsistent in terms of thickness, which likely played a role in the weight loss results.

Pre-heating of samples may produce more consistent results. Update procedure accordingly, conduct additional study.

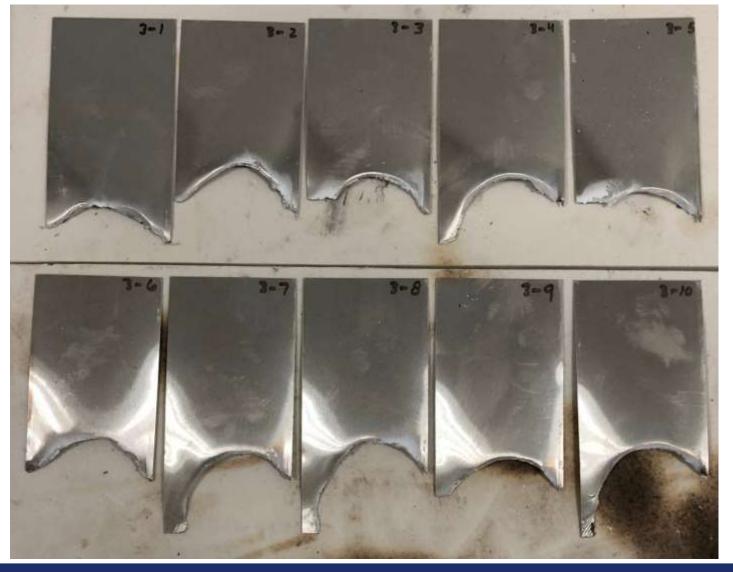
### Lab H Results on Batch I Samples



## Lab H Results on Batch II Samples



## Lab H Results on Batch III Samples



#### Summary of Task Group Discussion from 3/10/2020

#### Interlab Study Discussion:

- Need for consistently-manufactured test samples
- EDM-produced samples were not consistent. Difficulties in mounting the thin samples
- More favorable samples would result from EDM process using thicker initial stock
- Oxidized surface of EDM samples may have contributed to poor results

#### Test Method Discussion:

- Task Group participants unanimously agree the test is easy to conduct, not time consuming
- Determine Impact of backer board on results (i.e., void in material allows air under sample)
- Need additional language in Chapter 26 to address this (frequency of replacing backer board)

#### Summary of Task Group Discussion from 3/10/2020

#### Test Method Discussion (con't):

- Reports of negative weight loss (i.e., weight gain) in several instances
- Measured weights are very low; possibility of HVAC air currents impacting sensitive scales?
- Documentation of tests was helpful to testers; FAA only received limited photos of tests
- FAA recommends photo and/or video overage of tests

#### Pre-Heating of Samples Discussion:

- Conditioning temperature → Pre-heat temperature → Actual test temperature
- · Large tolerances on temperatures could impact results
- FAATC has experimented with pre-heating for 1 minutes, which has produced excellent results
- FAATC plans to monitor the surface temperature of the samples to determine time required to reach steady state conditions

#### Summary of Task Group Discussion from 3/10/2020

#### Test Method Robustness Discussion:

- What is impact of current heat flux tolerance on test results?
- FAATC to conduct tests at low and high ranges of heat flux to determine impact

#### Randomization of Samples Discussion:

- During collaborative effort between FAATC/Boeing on HR2, Boeing developed randomization process. The process would prevent certain parameters from influencing test results.
- Boeing has offered to assist in the randomizing effort for future interlab studies on magnesium

#### Discussion Items for Task Group

Discuss the key elements of the new flammability test for components located in inaccessible areas:

- Time until ignition (cannot be less than 30 seconds)
- Should there be a limit on self extinguishment? (currently not required)
- Discuss sample milling options for next interlab study

Discuss any other items related to the use of magnesium alloy in either seats or other cabin components

## Questions?

