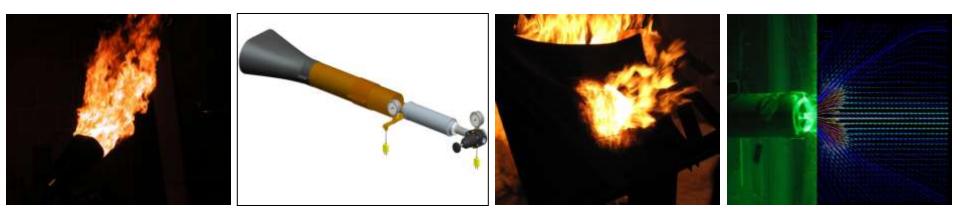
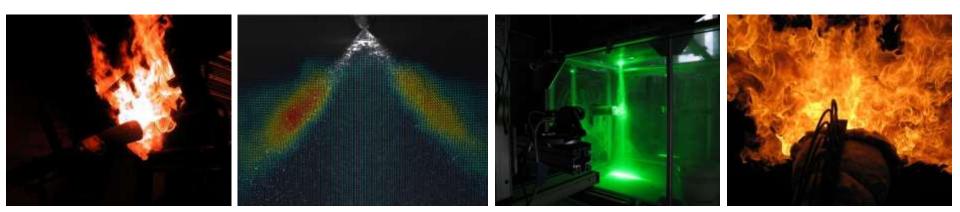
#### NexGen Burner Comparative Testing



Federal Aviation Administration



Presented to: IAMFTWG, Toulouse, France By: Robert I. Ochs Date: June 20-21



# Objectives

- Perform comparative burnthrough testing to determine the effect of various parameters on test results
  - Use picture frame sample holder and PAN material to determine burnthrough performance
- Test results will help to determine which parameters are most critical when specifying the burner in the new workbook



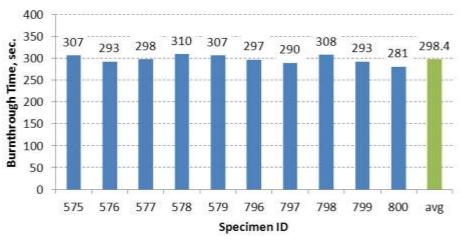


### Establish Baseline Dataset

400 350 Burnthrough Time, sec. 300 252 248 254 244 249 249 244.3 230 241 246 230 250 200 150 100 50 0 531 988 529 532 533 534 985 987 997 1002 avg Specimen ID

8579 Baseline Tests

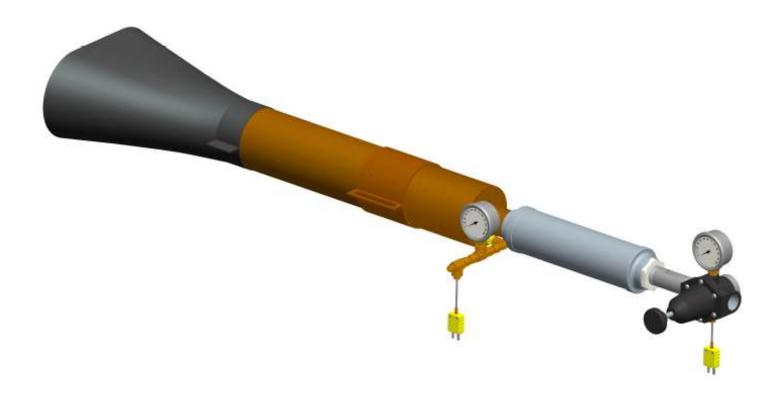
3.43% Standard Deviation 8611 Baseline Tests



3.17% Standard Deviation

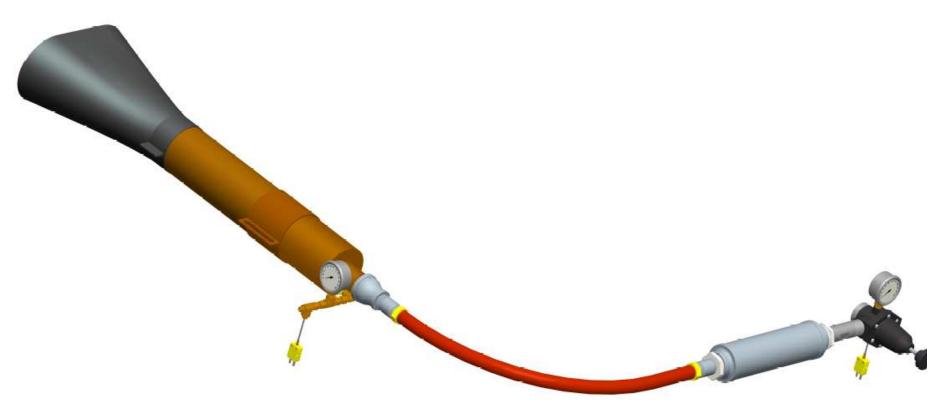


# Test 1: Location of sonic choke Standard Configuration





#### **Relocated Choke**



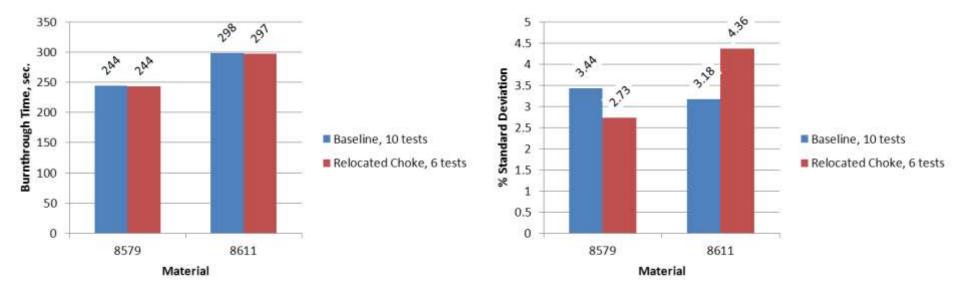






#### Burnthrough Time

#### Repeatability





# Test Series 1 - Summary

- The choke was relocated approximately 6' from the burner and a curved, flexible hose was added between the muffler and the burner
- Test results indicate no noticeable deviation from the baseline configuration



# Test Series 2 – Burner Cones

- Objective is to determine which cone parameters have an effect on burnthrough time
  - Thickness
  - Flange
  - Material
  - Age
- Besides the baseline cone, three additional cones were tested (all new)
  - Baseline Cone: 0.06" thickness with recessed flange
  - Cone #1: 0.048" thickness 18 gauge 310 Stainless Steel
  - Cone #2: 0.061" thickness 321h Stainless Steel
  - Cone #3: Same as Cone #1 with 1" flange welded on exit plane



#### Cones

#### Cone #1

#### Cone #2

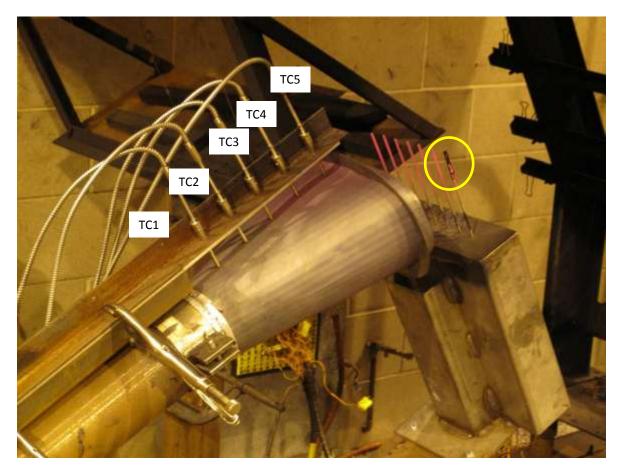
#### Cone #3





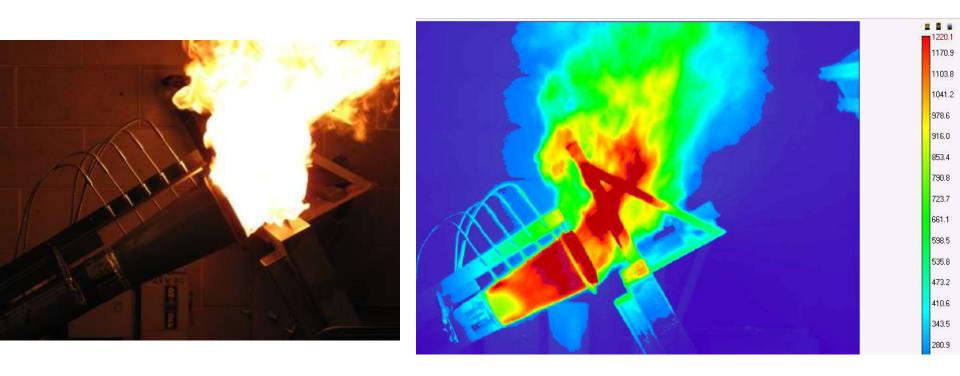
10

### Cone Surface Temperature Measurement



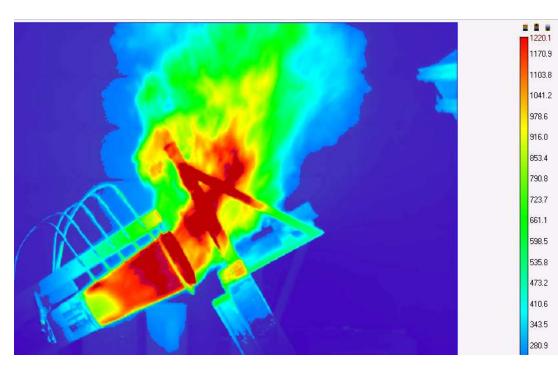


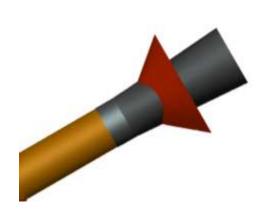
# **FLIR Imaging**





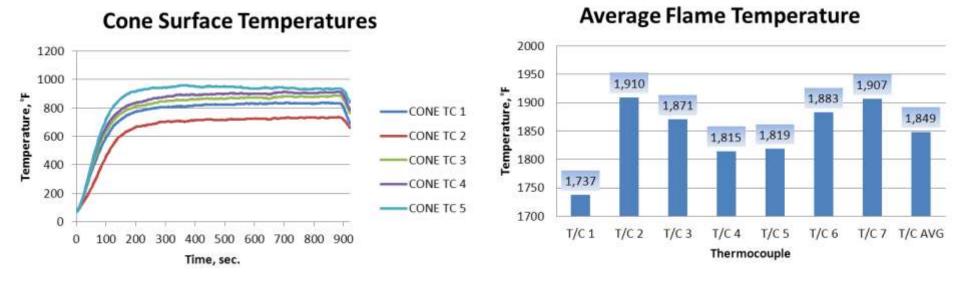
### Fuel Nozzle Spray Cone Angle







### **Cone and Flame Temperature**



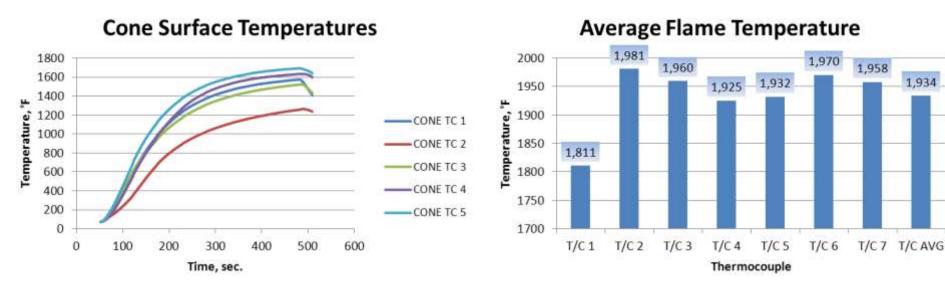


### **Insulated Cone**





#### **Insulated Cone**

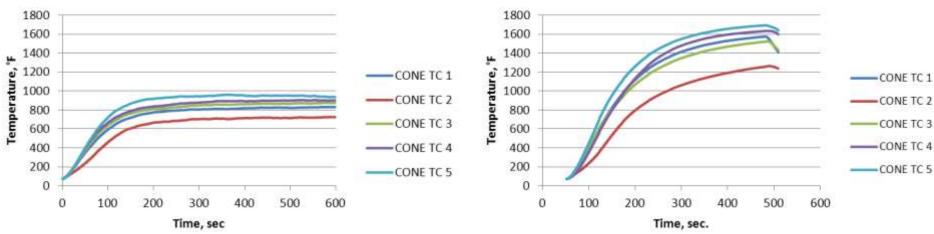




#### **Cone Surface Temperature Comparison**

**Un-Insulated Cone** 

Insulated Cone



Cone Surface Temperatures

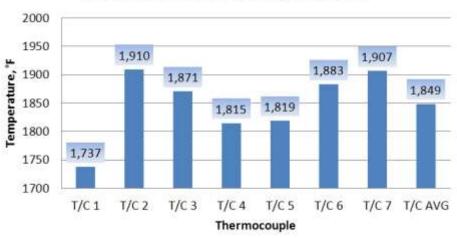
**Cone Surface Temperatures** 

NexGen Burner Comparative Testing IAMFTWG, June 20-21, 2012, Toulouse, France



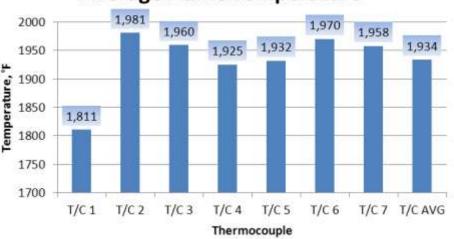
### Flame Temperature Comparison

#### **Un-Insulated Cone**



#### **Average Flame Temperature**

#### **Insulated Cone**



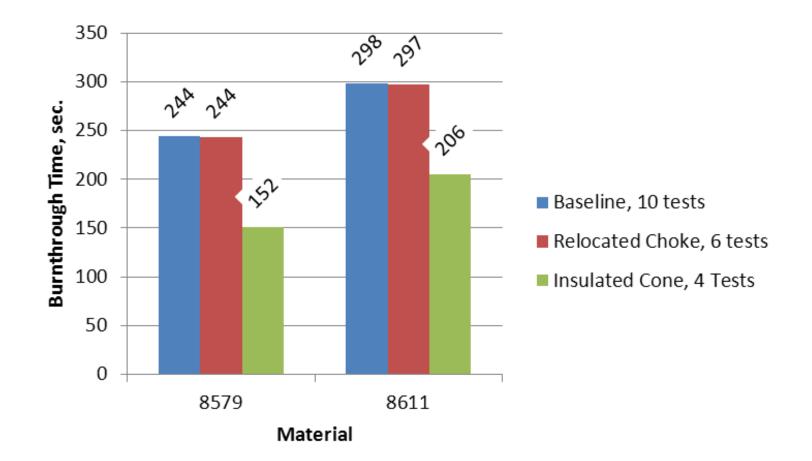
#### Average Flame Temperature

NexGen Burner Comparative Testing IAMFTWG, June 20-21, 2012, Toulouse, France



18

## **Burnthrough Times**



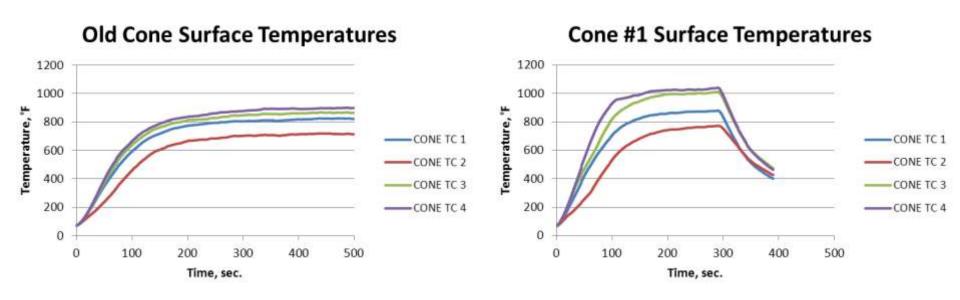


# Summary - Insulated Cone

- Insulating the cone increases the average flame temperature by about 85°F
- Insulation increased the cone surface temperatures by 500-600°F
- The insulated cone burned through the PAN material significantly quicker than the un-insulated cone
  - 8579: 92 sec. quicker
  - 8611: 92 sec. quicker
- Cone insulation used as an extreme example to determine how heat loss from cone can affect burnthrough results



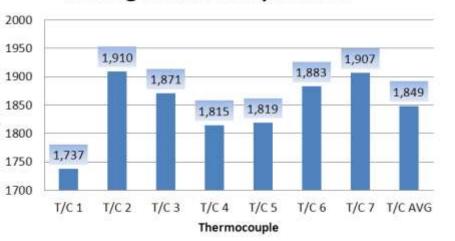
0.048" thickness 18 gauge 310 Stainless Steel





0.048" thickness 18 gauge 310 Stainless Steel

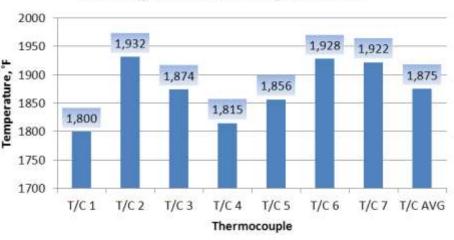
#### **Baseline Cone**



#### Average Flame Temperature

Femperature, 'F





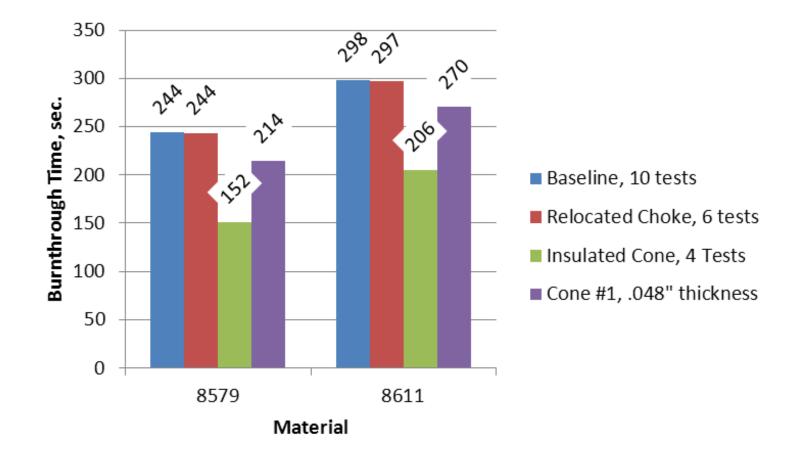
#### **Average Flame Temperature**

NexGen Burner Comparative Testing IAMFTWG, June 20-21, 2012, Toulouse, France



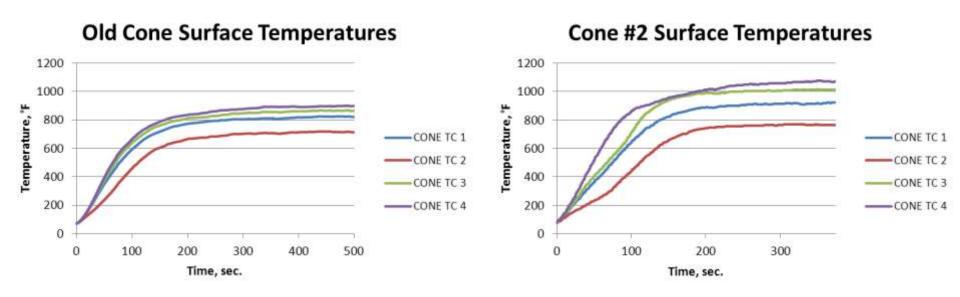
22

#### 0.048" thickness 18 gauge 310 Stainless Steel





#### 0.061" thickness 321h Stainless Steel





#### 0.061" thickness 321h Stainless Steel

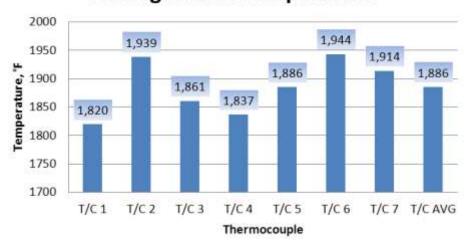
#### **Baseline Cone**

Average Flame Temperature

#### 2000 1950 1,910 1,907 1,883 Femperature, 'F 1900 1,871 1,849 1850 1,815 1,819 1800 1,737 1750 1700 T/C 1 T/C 2 T/C 3 T/C 5 T/C 6 T/C7 T/CAVG T/C 4 Thermocouple

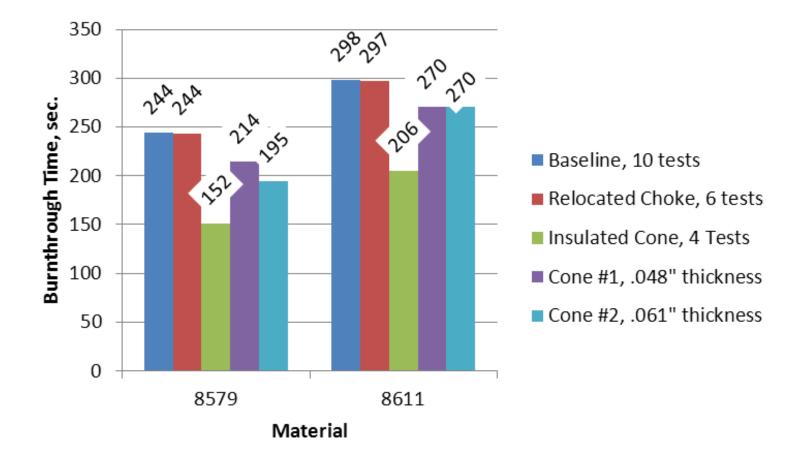
#### Average Flame Temperature

**Cone #2** 



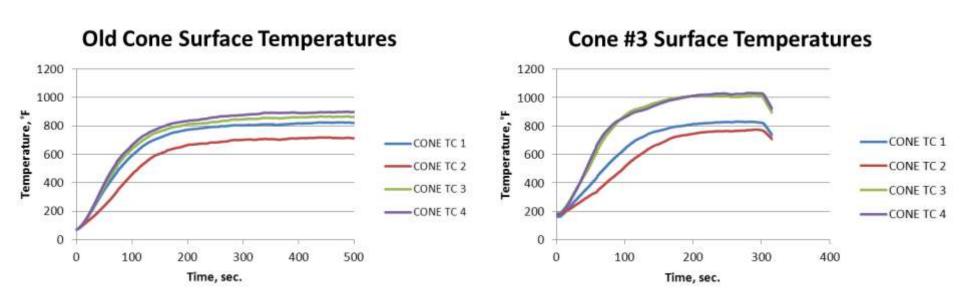


#### 0.061" thickness 321h Stainless Steel





Same as Cone #1 with 1" flange welded on exit plane





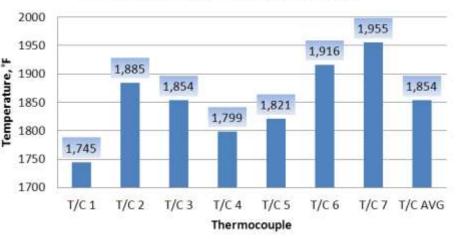
Same as Cone #1 with 1" flange welded on exit plane

#### **Baseline Cone**

2000 1950 1,910 1,907 1,883 Femperature, 'F 1900 1,871 1,849 1850 1,815 1,819 1800 1,737 1750 1700 T/C 1 T/C 2 T/C 3 T/C 5 T/C 6 T/C7 T/CAVG T/C4 Thermocouple

#### Average Flame Temperature

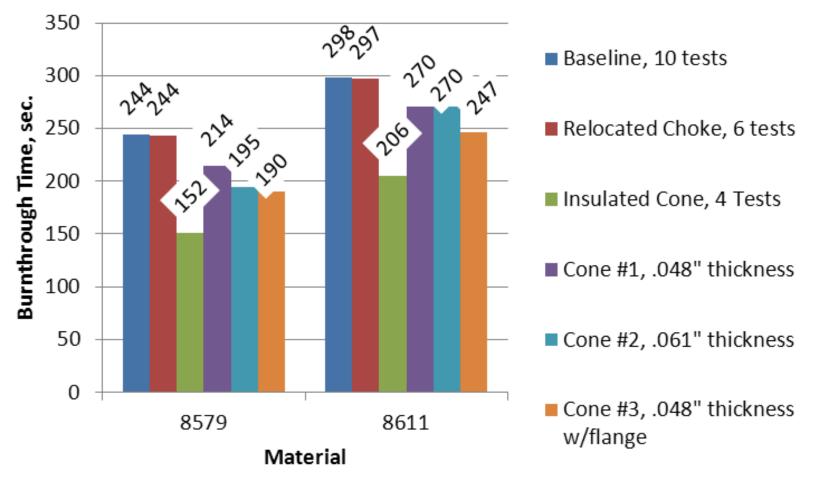
#### Cone #3



#### **Average Flame Temperature**

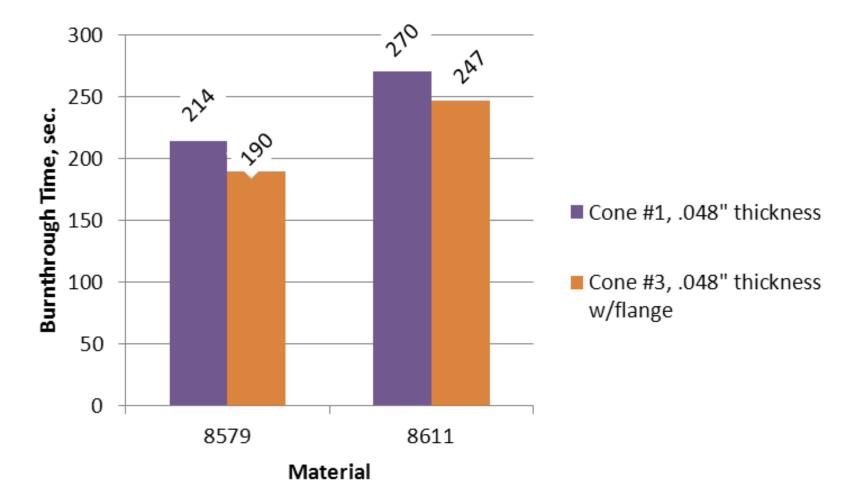


Same as Cone #1 with 1" flange welded on exit plane



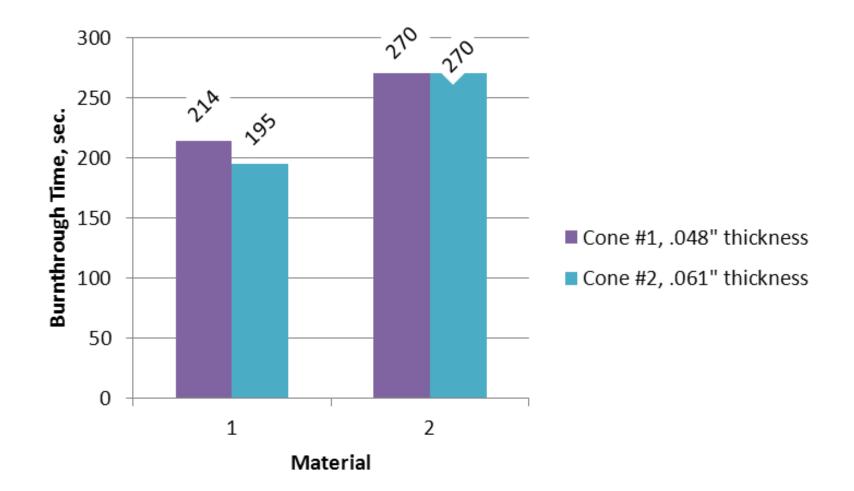


### Cones #1 vs. #3 – Effect of Flange



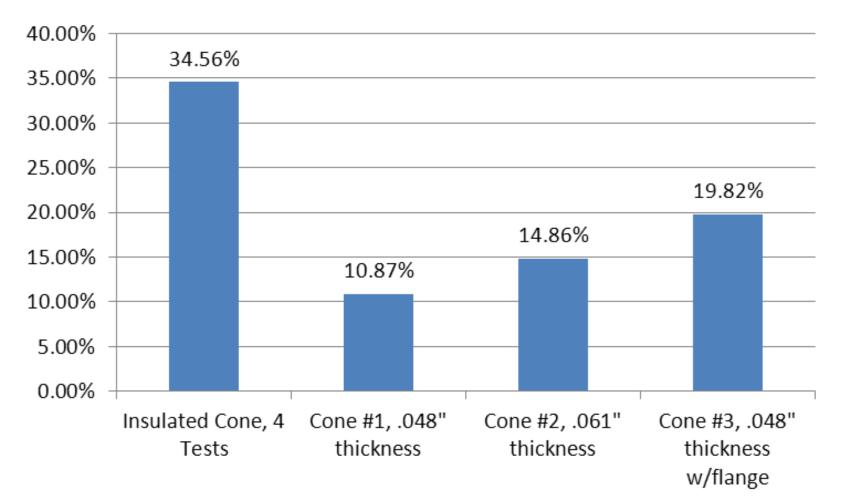


#### Cones #1 vs. #2 – Effect of Thickness





# Average % Change from Baseline





### **Cone Comparison - Summary**

• All cases tested reduced the burnthrough time of both 8579 and 8611 from the baseline case

- In order of impact on burnthrough severity
  - Insulating outer cone surface to prevent heat loss
  - 1" flange on end of cone
  - Slightly thicker cone material
  - New cone



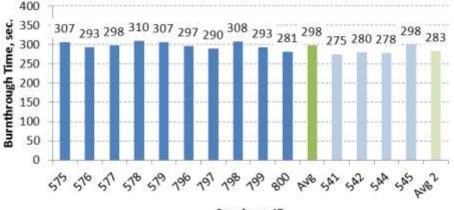
### **Re-run Baseline Tests**

400 350 Burnthrough Time, sec. 300 244 254 252 248 230 241 246 230 249 249 244 232 247 218 232 250 200 150 100 50 0 53° 53° 53° 53° 58° 58° 58° 59° 50° 54° 53° 54° 525 182 52 Specimen ID

8579 Baseline Tests

#### 244 ± 5% 231.8<x<256.2





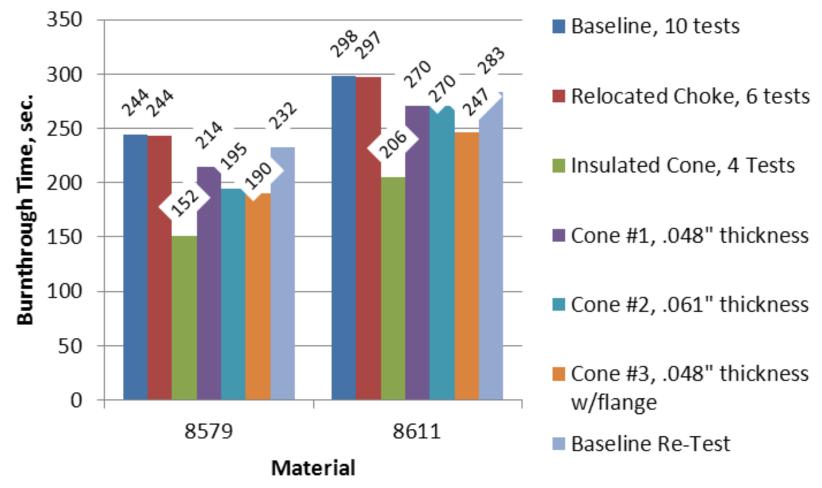
Specimen ID

298 ± 5% 283<x<313





### **Re-run Baseline Tests**

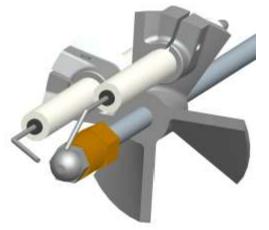




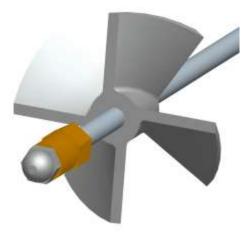
#### Test Series #3 - Stator

#### Definitions

"Old Stator" Marlin Engineering CNCmachined reproduction of original Monarch H-215



"New Stator" Marlin Engineering, symmetric, CNC-machined





#### New Stator – External Ignition



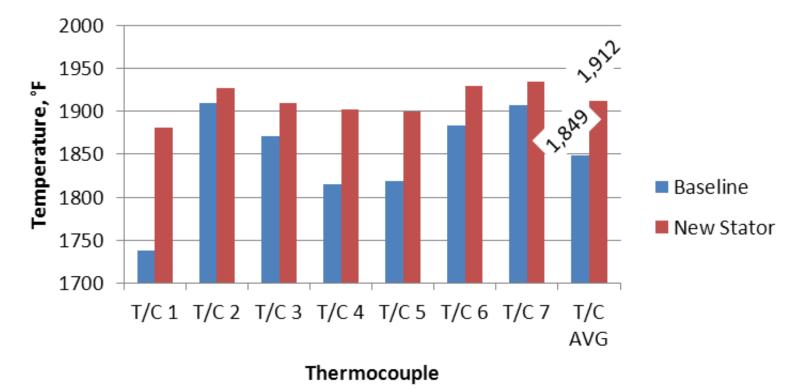


### **New Stator**

- Initially new stator was put in exact position as old stator
  - 4" back from nozzle tip
  - Centerline between vanes aligned 35° from vertical
- Ignition wires were removed from burner completely

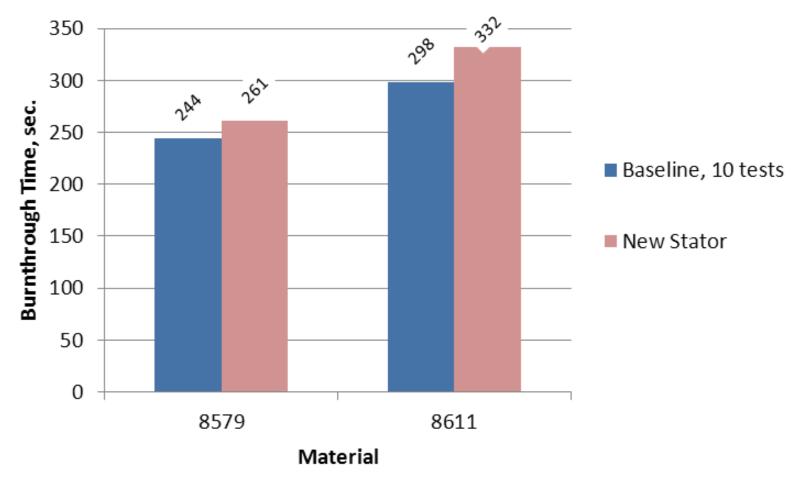


#### New Stator – Temperature Comparison Average Flame Temperatures





#### New Stator - Burnthrough



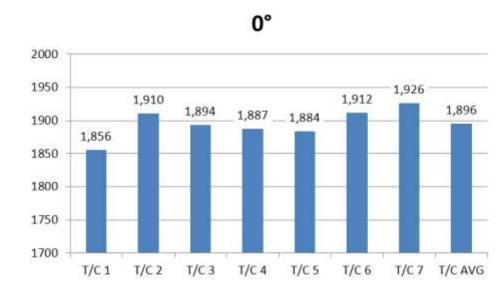


#### Summary of Initial New Stator Results

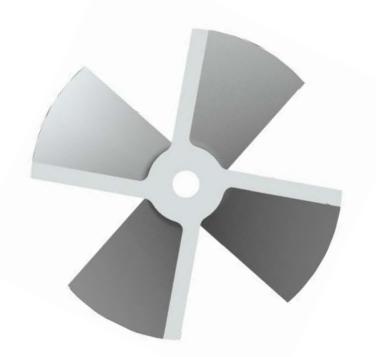
- New Stator Flame Temperature Measurement
  - More uniform flame temperature profile
  - Significant improvement on #1 T/C
  - Over 50°F increase in average flame temperature
- New Stator Burnthrough Tests
  - Longer overall burnthrough times for both 8579 and 8611 material
    - 8579 16.45 sec. longer
    - 8611 33.60 sec. longer

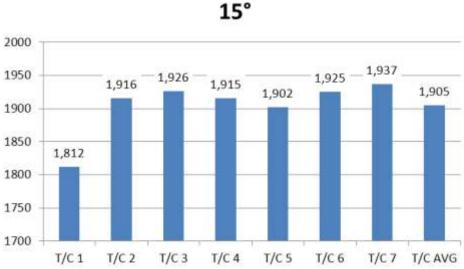




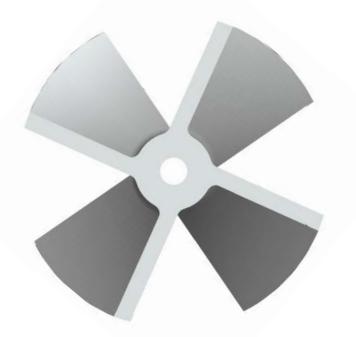


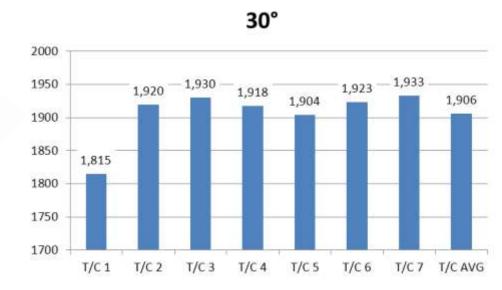




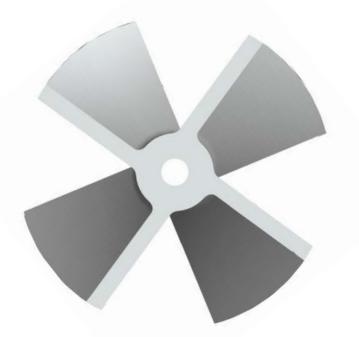


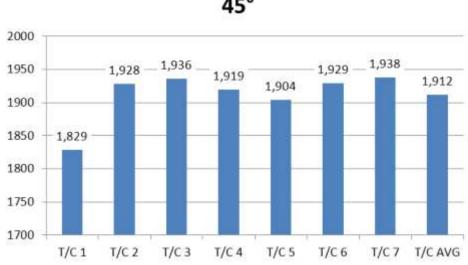






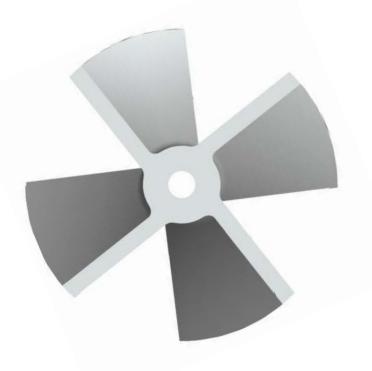


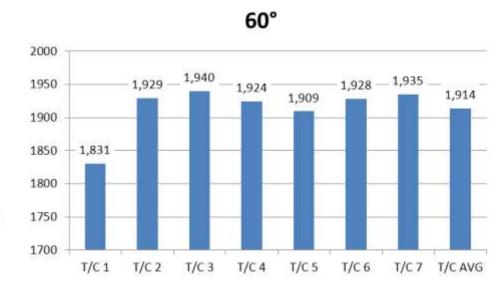




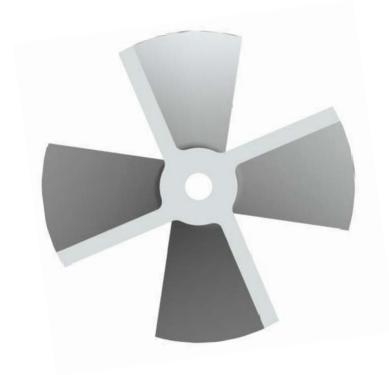
45°

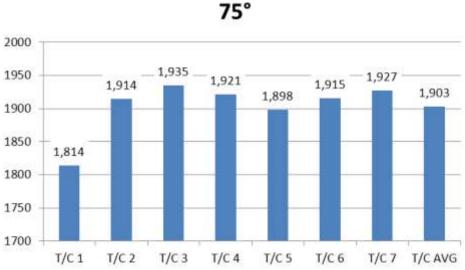




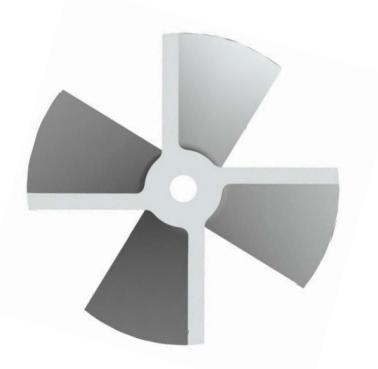


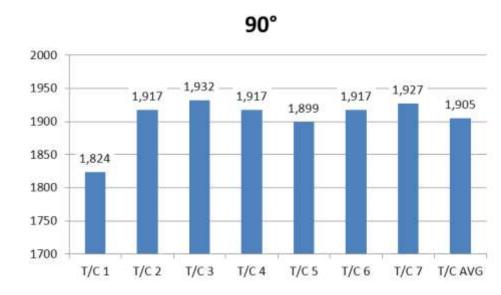








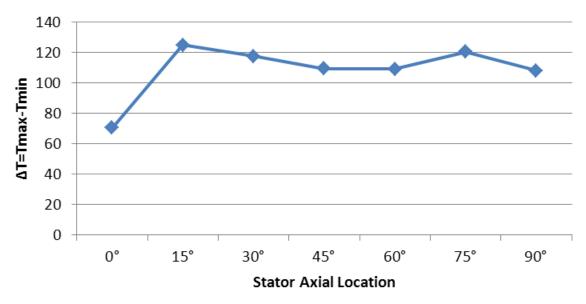






## Stator Rotation: $\Delta T = T_{max} - T_{min}$

ΔT=Tmax-Tmin



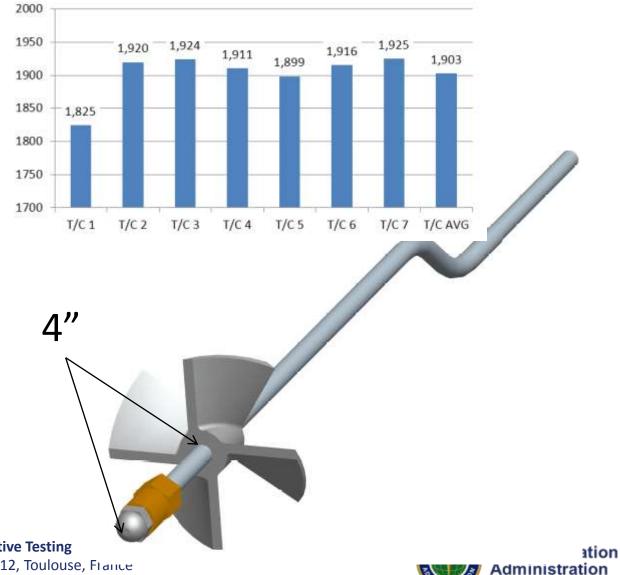
 ΔT is used to determine the uniformity of the flame temperature measurement, smaller ΔT, more uniform profile



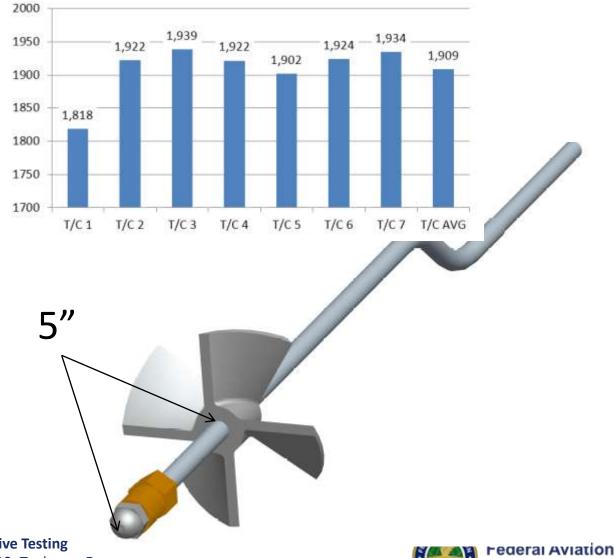
## **Stator Rotation - Summary**

- Rotating the stator over 90° in 15° increments resulted in slightly different flame temperature profiles
- The uniformity of the flame was assessed by subtracting the minimum temperature from the maximum temperature
- The best uniformity was found at 0°, with a spread of 70.36°F





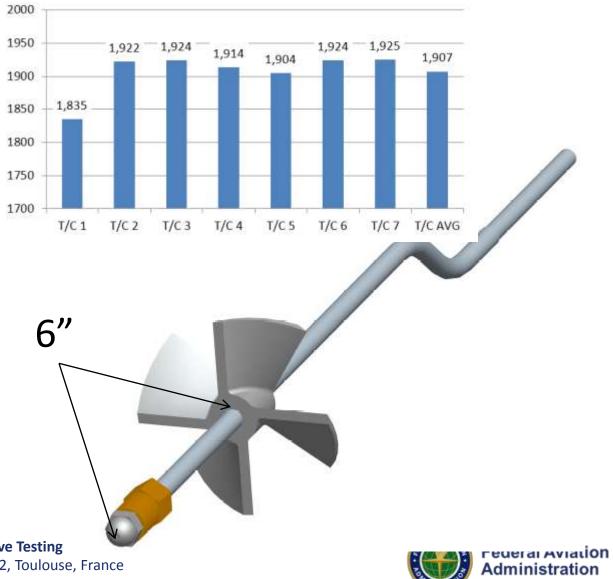
0° 5 in

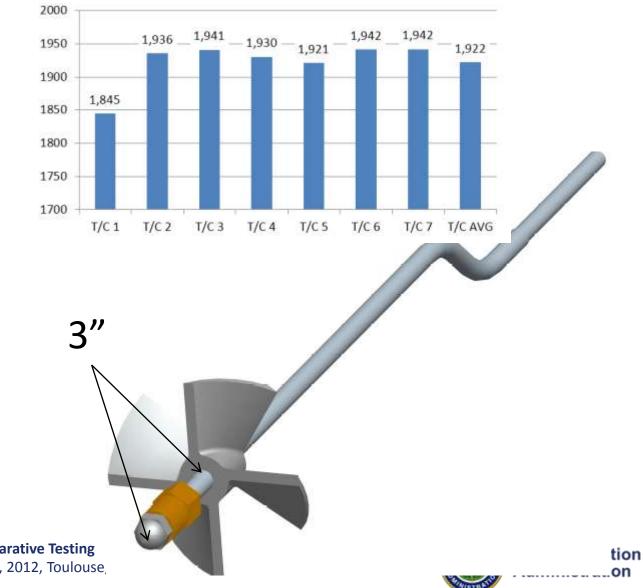


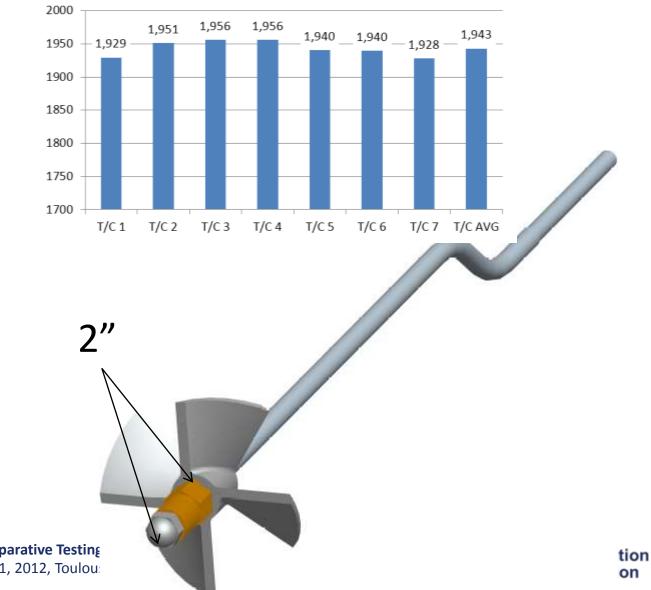
NexGen Burner Comparative Testing IAMFTWG, June 20-21, 2012, Toulouse, France



Administration

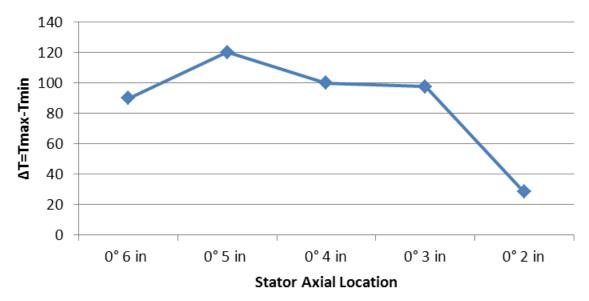






## Stator Axial Location: $\Delta T = T_{max} - T_{min}$

ΔT=Tmax-Tmin



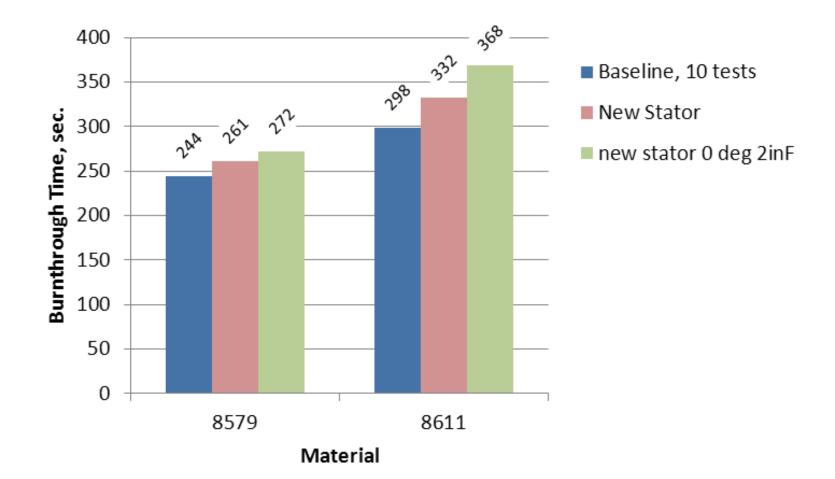


## Axial Translation – Summary

- Translating the stator on the axis of the burner over a range of 4 inches in 1 inch increments resulted in slightly different flame temperature profiles
- The overall temperatures increased as the stator was translated closer to the fuel nozzle
- The highest overall flame temperature and best uniformity was found at 2 inches back from the nozzle tip



#### New Stator, 0°, 2" from nozzle tip





#### **Rotation and Translation Burnthrough**

- The most uniform flame temperature profile and the highest overall measured flame temperature resulted in the longest burnthrough for both 8579 and 8611
  - 8579: 271.50
  - 8611: 368.25
- These burnthrough times are longer than the baseline test
  - 8579: 27.20 sec. longer
  - 8611: 69.85 sec. longer



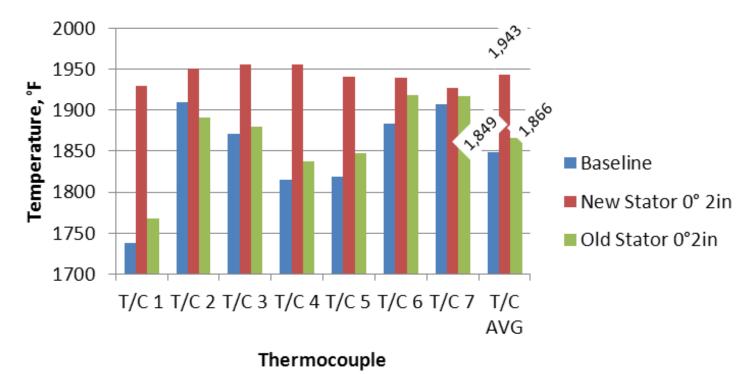
# Old Stator, same position as new stator, no wires





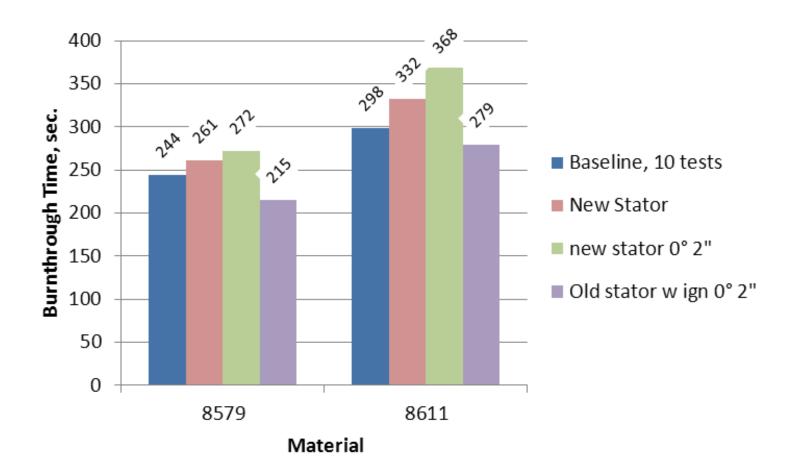
# Old Stator, same position as new stator, no wires

#### **Average Flame Temperatures**





### **Burnthrough Times**





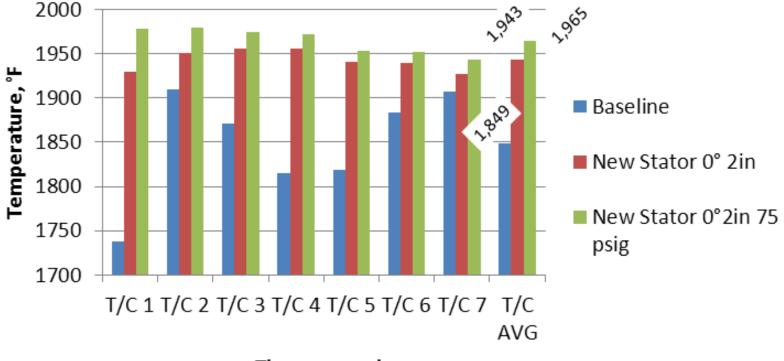
#### Old stator @ 2" from nozzle tip - Summary

- The old stator and igniters placed in the same position as the new stator resulted in lower flame temperatures with less uniformity
  - New Stator:  $T_{avg} = 1942^{\circ}F$ ,  $\Delta T = 28^{\circ}F$
  - Old Stator:  $T_{avg} = 1865^{\circ}F, \Delta T = 151^{\circ}F$
- The old stator and igniters resulted in significantly faster burnthrough times than the new stator and the original baseline
  - 8579: 56.75 sec. quicker than new stator
  - 8611: 89 sec. quicker than new stator
- These tests are proof that the magnitude of the measured flame temperature is not indicative of burner severity



#### New Stator @ 75psig sonic choke inlet pressure

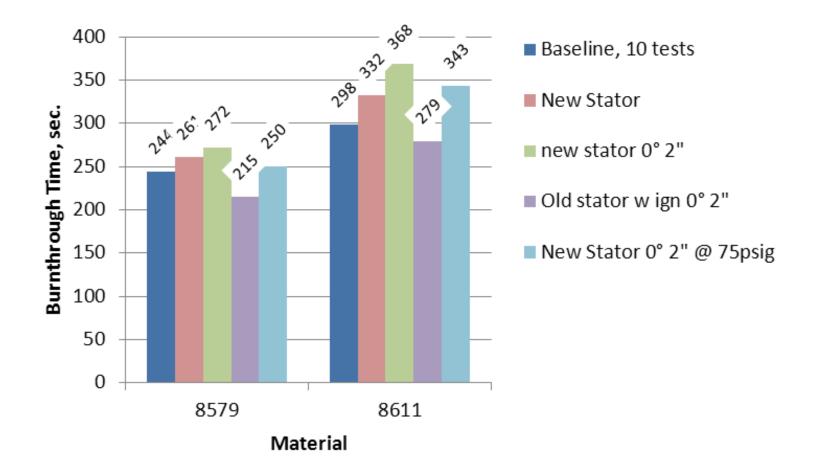
#### **Average Flame Temperatures**



Thermocouple



#### New Stator @ 75psig sonic choke inlet pressure



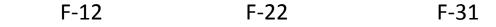


#### New Stator @ 75psig sonic choke inlet pressure Summary

- Increasing the air pressure from 60 psig to 75 psig resulted in the 8579 being closer to the baseline, but the 8611 was still significantly longer
  - 8579: 5.95 sec longer
  - 8611: 44.6 sec longer



#### Flame Retention Heads (FRH)

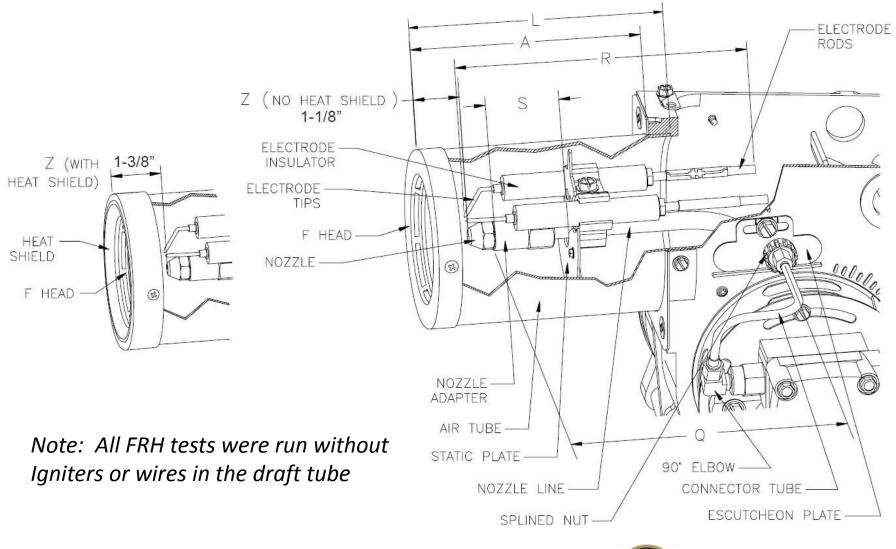




- Beckett flame retention heads were purchased from local supply store
- These heads are used on modern oil burners for more efficient burning
- The heads can be used to create inefficient fuel rich burning that we are looking for by mismatching the air flow and the fuel firing rate
- Benefits of Flame Retention Heads
  - One component replaces both the stator and turbulator
  - Reduces the amount of specification required for burner
  - Rotationally symmetric



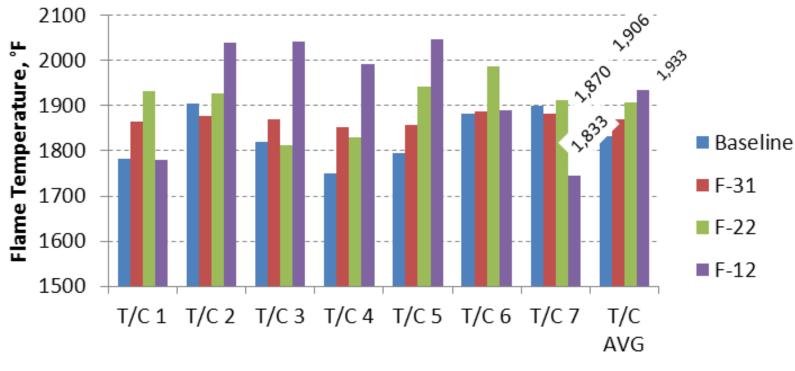
## Nozzle Depth





#### **Flame Retention Heads**

#### **Average Flame Temperature**

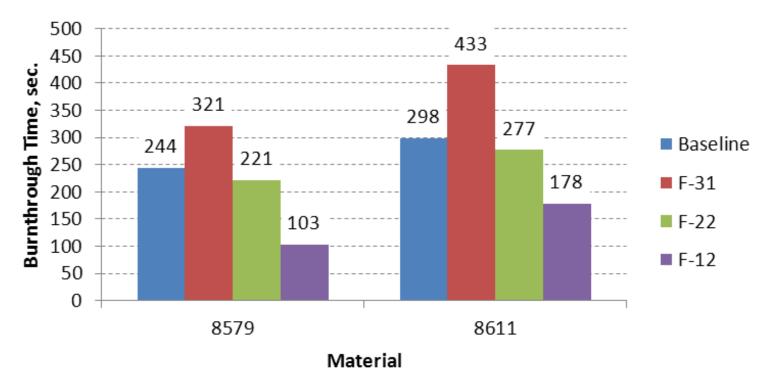


Thermocouple



#### Flame Retention Heads

#### **Burnthrough Times**





## Flame Retention Heads - Summary

- The flame retention heads give different temperature profiles depending on the size of the coflow air passages
- Overall the flame temperatures were higher with flame retention heads than with stator and turbulator
- A wide range of burnthrough times were obtained for the different heads, generally the larger the coflow air passages, the longer the burnthrough time
- The F-22 head seems to give the closest burnthrough time to the baseline nexgen burner configuration



#### FRH for Cargo Liner and Seat Burners



#### Flame Retention Head: Description

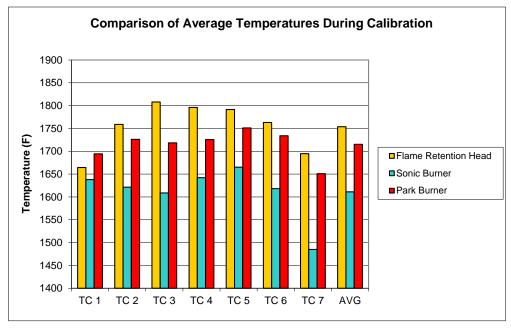
- Eliminates the need for a stator or turbulator
- Fits on end of burner draft tube
- Initial testing shows good potential





73

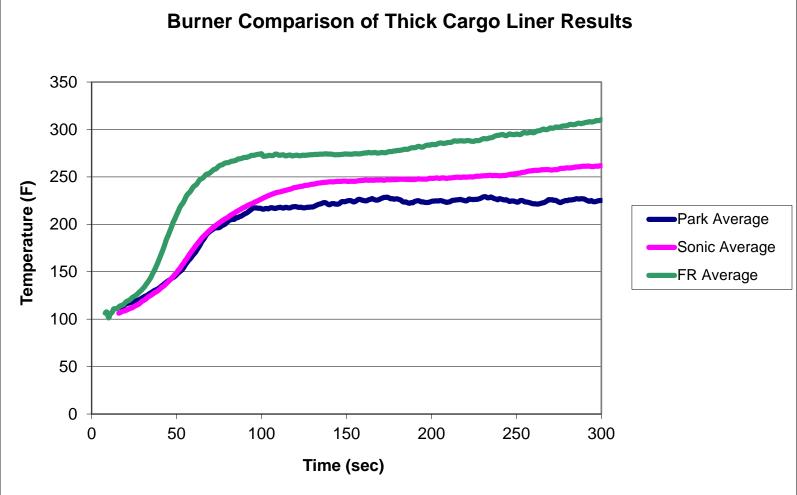
#### Flame Retention Head: Calibration



- Calibration readings are significantly higher using the flame retention head compared to sonic readings using the standard stator
- Readings also seemed more consistant from one calibration to the next



#### Flame Retention Head: Test Results





#### Flame Retention Head: Test Results

Burner Test Result Comparison: TexTech PAN 8579 450.00 400.00 Average Burn-through Times (sec) 326.4 345.9 344.5 350.00 300.00 Park Burner 250.00 Sonic Burner 200.00 □ Flame Retention Head 150.00 100.00 50.00 0.00 1

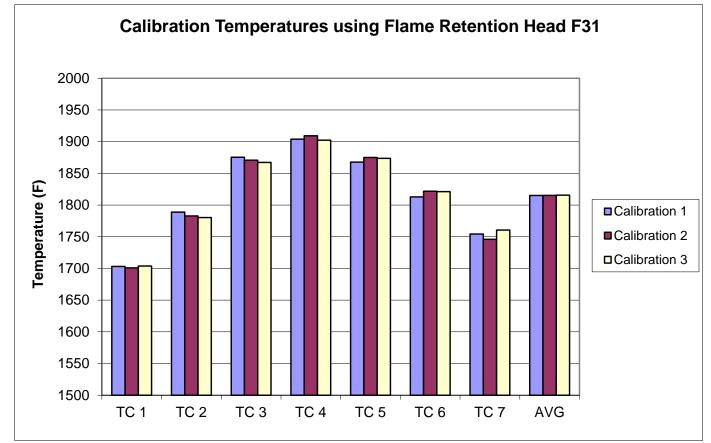


#### Cargo Liner Flame Retention Head: Conclusion

- Simplify setup and adjustments by eliminating stator and turbulator
- Capable of producing higher temperatures
- Flame temperature can be tailored by changing size of holes in flame retention head
- Head tested produced temperatures higher than the Park or sonic burner, as well as decreased burn-through time



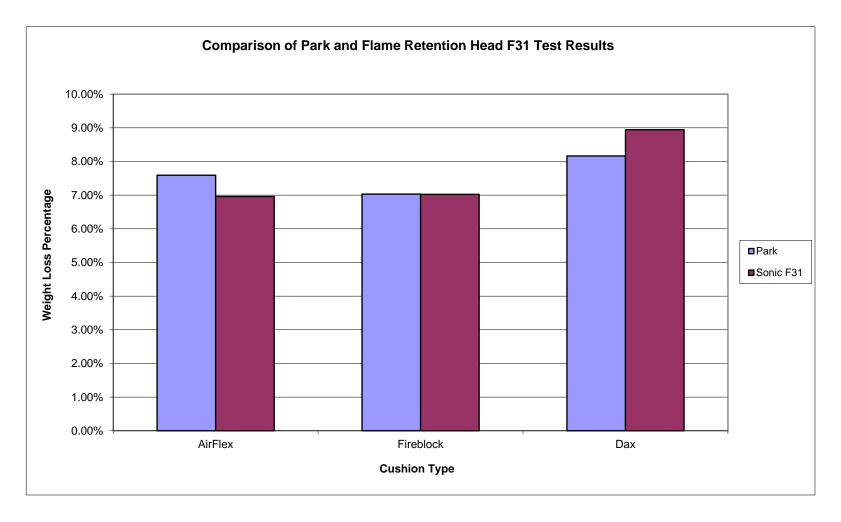
## FRH for Seat Cushion Burner



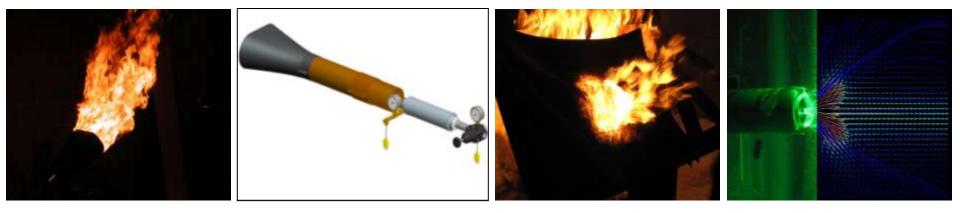
- Extremely low variation of temperature
- Less than 1°F variation of averaged temperatures



### FRH for Seat Cushion Burner







Contact: Robert I. Ochs Fire Safety Branch William J. Hughes Technical Center ANG-E212; Bldg 287 Atlantic City, NJ 08405 T 609 485 4651 E robert.ochs@faa.gov



#### Federal Aviation Administration

