

FAA EASA

Large Surface Areas on Seats Acceptable Method of Compliance

IAMFTWG June 22/23, 2011



- 25.853 does not require non-metallic materials installed on seats to comply with Heat Release/Smoke Emission requirements Background
- When Heat Release/Smoke Emission requirements were introduced in FAR 25 in 1988, full scale testing had shown that the contribution of non-metallic panels installed on seats to a post-crash fire could be considered negligible.
- Back in the 1980s seat cushions used to constitute the most significant share of non-metallic material installed on seats. Part II of Appendix F was deemed sufficient to address the contribution of seats to a post-crash fire



Since then seat design has significantly evolved:

- privacy features (back shells, dividers, sidewalls, etc.)
- increased stowage capability through the installation of larger seat-associated furniture
- the amount and the size of non-metallic panels have increased to the point that their contribution to a postcrash fire could not be ignored anymore.
- The FAA issued Policy Memos to highlight that seat associated furniture to be considered interior components and therefore are required to comply with part IV and V of App. F.
- Evolved to development of special conditions after reviewing the regulatory basis.



- ➤ Two Main Compliance Questions
 - ➤ When to implement the special conditions
 - ➤ How to determine whether something is covered by the special conditions
- ➤ Industry Formed a working group to recommend solutions
- ➤ Authorities participated in the forming of the recommendations



- The HR/SE SC define new requirements but do not specify how to show compliance
- The Special Conditions introduce new terminology, e.g. non- traditional, large, non-metallic panels or new seat certification program but do not provide clear definitions
- the criteria for applicability of the special conditions are not clear enough
- In April 2009 EASA realized to have a different position with respect to the FAA regarding testing coverings glued on panels required to comply with the Special Conditions



Applicability

- EASA/FAA/TCAA/ANAC developed fully harmonized criteria to determine the applicability of the Special Conditions
- Different criteria apply to new deliveries of aeroplanes
- The developed criteria better define the meaning of new seat certification program



Applicability – Newly manufactured aeroplanes

Same Seat P/N	New Customer	Same Customer, New A/C Model (new family or new derivative)	Same Customer, New Block of A/C	Same Customer, Same Block, New seat count
New Seat P/N, no change to large panels	✓			
New Seat P/N, changes to large panels	√	√	✓	✓
New Seat Model	√	√	√	✓

√ = seats must comply with SC

See Notes on next page



Applicability – Newly manufactured aeroplanes

- "New Customer" relates to a situation where Airline "A" has, for example, 737-700 airplanes certificated
 with seats with large non-metallic panels prior to the issuance of special conditions. Airline B, after the
 effective date of the special conditions, purchases 737-700 airplanes and may (or may not) have exactly the
 same seating configuration as Airline "A."
- "Same Customer, New Model" means if Airline "A" has, for example, 737-700 airplanes certificated with seats with large non-metallic panels prior to the issuance of special conditions. The Airline now purchases 767, or 737-800 airplanes and installs seats in the new model 737 after the issuance of the special conditions.
- "Same Customer, New Block of A/C" means if Airline "A" has, for example, A320-200 airplanes certificated
 with seats with large non-metallic panels prior to the issuance of special conditions. The Airline now
 purchases, after the effective date of the special conditions, additional A320-200 airplanes.
- 4. "Same Customer, Same Block, New seat count" means if Airline "A" has, for example purchased 50 737-900 airplanes and the first of this block of 50 airplanes has been type certificated with seats with large non-metallic panels prior to the issuance of special conditions. Then on the 15th airplane to be delivered (which is after the effective date of the special conditions), the Airline decides to increase the number of first class seats in the airplane and reduce the number of economy class seats.
- 5. "Same seat P/N" means a seat that is unchanged. In the case where the same P/N is retained, but changes are made to all seats with that p/n (i.e., the prior configuration is eliminated on the drawing and in the field), this is considered as "New" for the purposes of implementation of the special conditions.

"Customer" refers to the end user, and not a leasing company, that supplies many operators.



Applicability - Changes introduced after delivery

	Installation of existing arrangement for fleet commonality	Re-arrangement (or removal) of existing seats (no additional seat installations)	New Installation of seats
Same Seat P/N			
New Seat P/N, no change to large panels			*
New Seat P/N, changes to large panels	✓	~	~
New Seat Model	*	*	*



Applicability – Changes introduced after delivery

- "Same", or "new" with respect to seat P/N refers to authority approval status, and whether there is an installation approval, even if it was granted to someone else. The applicant is responsible for both identifying and supplying evidence of the prior approval
- "Fleet commonality" means the installer already has this arrangement in their fleet, and is configuring other airplanes to match. Modification is being performed by other than airframe mfr.
- "Re-arrangement" means the seats are moved around, or maybe some are removed, but no additional seats are installed.
- 4. "New Installation of seats" means that the modification includes installation of more (Row 1) or different (Rows 2-4) seats than the current arrangement. Includes both a new installation as well as adding to an existing arrangement.
- 5. With respect to Row 2, column 3, when the same operator/user obtains a post-TC approval (e.g., via STC), to essentially provide fleet commonality with the arrangement as delivered from the factory, R2:C1 would apply instead. Where a customer/user is obtaining a new installation approval for a new P/N, which happens to be similar to another customer's approval, the special conditions would apply.



Definitions

- Panel
- Nonmetallic
- Large
- Exposed
- Traditional

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Traditional vs. Non-Traditional

 Definitions for Traditional or Non-Traditional are specific to different features of a seat.

Head Rests

Privacy Dividers

Center Consoles

Arm Rests and Armrest Closeouts

End Bays

Rub Strips / Bumpers

Leg Rests

Foot Bars

Line Replaceable Units



Seat Pans

Shrouds

Seat Primary Structure

Life Vest Containers

Food Trays Shells Monitors and Bezels Literature Pockets

Kick Panels Foot Wells

Seat Backs

Independent Furniture Safety Features



Item	MOC slide number	Industry proposal including 10% buffer (in square inches)	Industry proposal without 10% buffer (in square inches)	Limits identified by the AAs (in square inches)
Centre consoles	21-22	1208	1098	1008
Seat end closeout	23-24	378	344	344
End bay	25-26	1186	1078	1008
Armrest closeout	25-27	576	524	432
footbars	30-31	152	138	108 with non-metallic support bar / 72 with metallic support bar
Food trays	35-36	231	200	200
Video bezel	38	n/a	115	115
Kick panels	41-42	213 (per seat place)	193.5 (per seat place)	144 (per seat place)



Coverings 1/3

- Non-traditional, large, non-metallic panels covered with fabrics or leathers in a more "traditional" way (e.g. only affixed around the edges) will be tested without their coverings. However, non-traditional, large, non-metallic panels covered with fabrics which are affixed all over (e.g. glued) will be tested with their coverings.
- In April 2009 the FAA clarified that large non-metallic panels covered with traditional fabrics or leathers will be tested without their coverings, regardless of the method of attachment.
- At the moment this is the only aspect that is not harmonized between the FAA and EASA Special Conditions. However, the FAA has clarified that showing compliance with the EASA SC, i.e. testing panels together with their coverings, is acceptable to show compliance also with the FAA SC.



Coverings 2/3

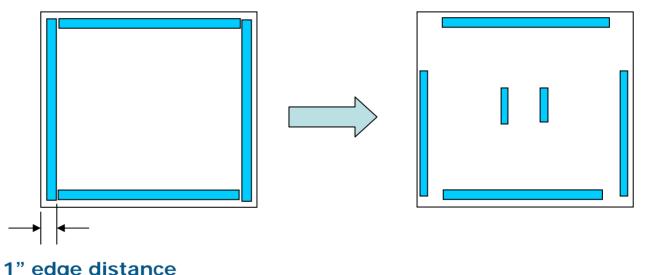
- ➤ EASA has developed a criteria to determine when the covering can be considered "affixed around the edges" of the supporting panel and therefore is not required to be tested together with the supporting panel.
 - If glue is applied within 1 inch from the edge of a covering, the panel shall be tested without the covering.
 - If glue is applied in locations other than within 1 inch from the edge of the covering, the panel may be tested without the covering if the total amount of glue used is equivalent to the amount that would be applied only around the edges.

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Coverings 3/3

Equivalent area example



1" edge distance

EASA considers that the same criteria must be applied also to coverings attached by means of hook and loop, tape and other similar attachment methods.