

Development of a Lab-Scale Flame Propagation Test for Composite Fuselages

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By: Robert Ian Ochs

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Federal Aviation
Administration



Introduction

- **With the increased use of non-traditional materials for modern aerospace applications, fire test methods must be continually updated and re-evaluated in order to maintain a high level of passenger safety**
 - Application of fire tests to modern materials
 - Re-evaluation of pass/fail criteria
 - Introduction of new safety threats with new materials
 - Develop new standards or test methods to address these issues
- **Composite materials (carbon fiber-epoxy) are being used in places where aluminum was traditionally used**
 - Fuselage skin
 - Structural members – stringers and formers
 - Seat frames
 - Fuel tanks
- **There is a need to evaluate the fire properties of these materials to ensure there is not a decreased level of safety**



Composite Fuselage

- **There is a need to evaluate the fire properties of a composite fuselage**
 - Burnthrough
 - Toxicity
 - In-flight burnthrough
 - Flame propagation
- **This objective of this study is to determine whether a composite fuselage will pose a flame propagation hazard**
 - Identify potential scenarios where a threat may be present
 - Evaluate threat with full or intermediate scale test
 - Analyze results to determine if there is an increased risk
 - Use full/intermediate scale test results to develop a lab-scale test for future certification purposes



Evaluation of Flame Propagation Risk

- **An intermediate scale test was performed using the foam block fire source**
- **Different configurations of the fire source, thermal acoustic insulation, and composite panel were attempted**
- **Test results indicated that the material being evaluated did not present a flame propagation hazard**
- **Other composites or composites of varying thicknesses may pose a threat**



Development of Lab-Scale Test

- **Use the results from previous intermediate scale test as a baseline for a “pass”**
 - The intermediate scale test results were used to certify that specific material for use in aircraft
 - The intermediate scale test will not suffice for certification, however, as it is a large test and takes time and money to perform
 - Certification tests must be performed when varying the material (different epoxies, thicknesses, etc.)
 - The lab scale test must provide the same discretion as the intermediate scale test, but be more efficient to perform
- **Radiant Panel Test Apparatus**
 - The radiant panel test is very useful for evaluation flame propagation tendencies for materials
 - The test is a “surface” test, as radiant heat and the burner impingement are applied to the material surface
 - Material thickness and thermal conductivity play a large role in this test
 - Test parameters must be adjusted to account for composite materials of varying thicknesses (warm-up time, flame exposure time, radiant heat energy, etc.)
 - Task here is to determine if the radiant panel test will be useful for evaluating the flame propagation threat of composite materials



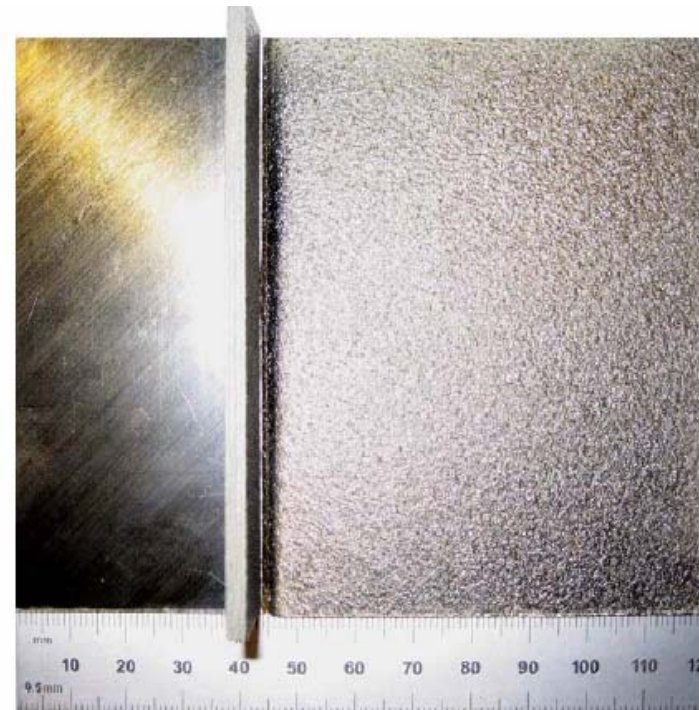
Previous Work on Composites

- **FAA Report DOT/FAA/AR-07/57 (Quintiere et al.) entitled “*Flammability Properties of Aircraft Carbon-Fiber Structural Composite*”**
- **Investigated a material manufactured by Toray Composites America to meet Boeing Material Specification 8-276**
- **Key objective was to investigate the heating and burning properties of the material with various methods**
 - Cone calorimeter
 - Microscale combustion calorimeter
 - Thermogravimetric analysis
 - Differential scanning calorimeter
 - Flame spread apparatus
 - Thermal conductivity apparatus
 - OSU and smoke chamber



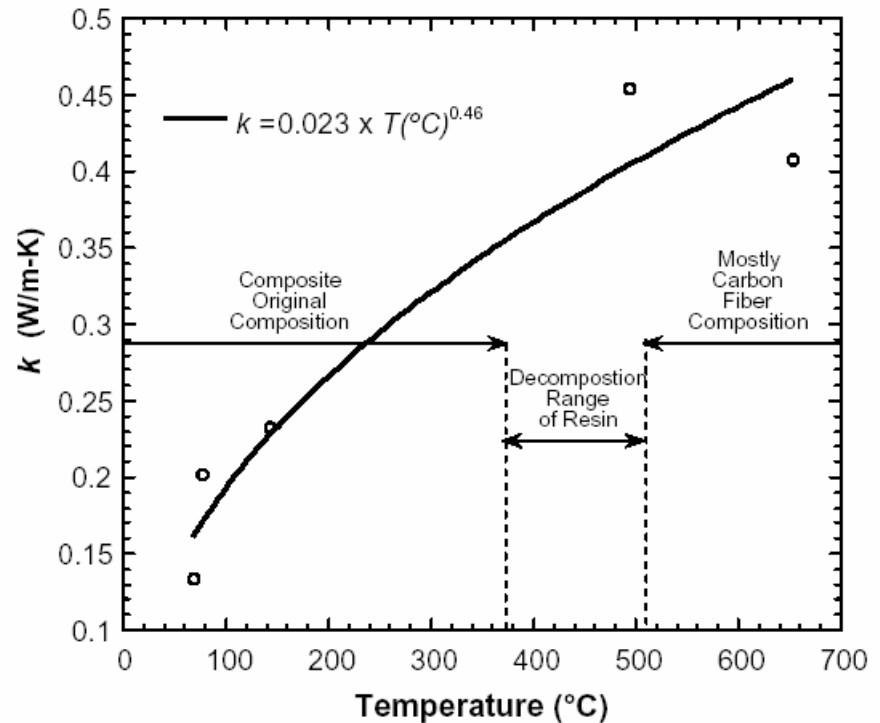
Material Details

- **Toray Composites, BMS 8-276**
 - [-45, 0, 45, 90]2s 16 plies
 - Material is composed of carbon fibers (7 μ m dia.) and resin
 - 3.2 mm thick (0.125 in.)
 - Density = 1530 kg/m³ (95.5 lb/ft³)
 - 60% by volume carbon fibers
 - Resin density = 1220 kg/m³ (76 lb/ft³)
 - Typical char fraction of resin ~ 25%



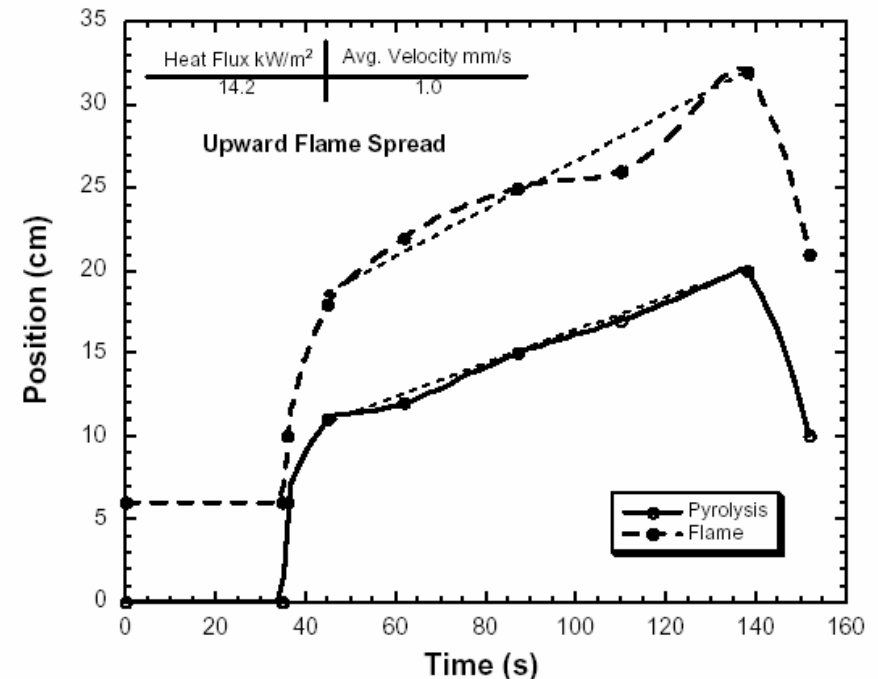
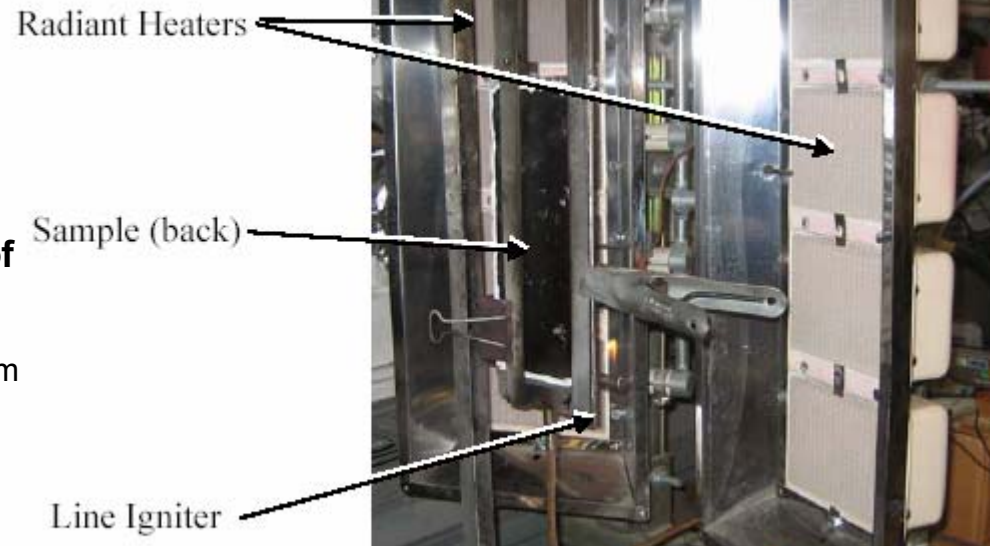
Properties

- **Resin is primary fuel for reactions**
 - Escapes the material through pores in the matrix
 - Heating causes material to swell as internal pressures rise
 - Flame jets can protrude from the material due to pressure release
 - Properties change as material changes in shape
 - Density decreases
 - Thermal conductivity of matrix decreases
 - Both flaming and non-flaming combustion can occur
 - Resin can produce flaming combustion
 - Char and carbon can smolder on the surface without flame
- **Thermal conductivity was measured with a home-built apparatus**
 - Measurements were made over a range of temperatures relevant to combustion
 - Heat loss errors were found with the apparatus, possibly up to 20% difference
 - More accurate methods are needed
 - Despite the errors, the dependency of thermal conductivity on temperature is clear



Flame Spread Experiments

- **Critical heat flux was found from cone calorimeter measurements**
 - Piloted ignition: 17.5 kW/m² (1.5 BTU/ft²s)
 - Non-piloted ignition: 31.5 kW/m² (2.8 BTU/ft²s)
- **Apparatus was developed in the work of Panagiotou and Quintiere**
 - Vertically mounted specimen
 - Surface exposed to radiant heat is 6 x 25 cm (2.4" x 9.8")
 - Pre-heat time of 4 min was used to create thermal equilibrium on sample surface (found from critical heat flux for piloted ignition)
 - Flame spread is only a function of heat flux for this pre-heating condition, as flame speed is dependent upon surface temperature
- **Key findings**
 - Exposed heat flux of 1.25 BTU/ft²s
 - Pilot flame in contact entire test
 - Flame spread of approx 7" over 2 ½ min
 - Pyrolysis spread of approx 12.6" over 2 ½ min



Status

- **Work is in the initial phase right now**
- **Initial work will involve tooling with the radiant panel and different composite material plaques to observe how the material behaves in this test**
 - Vary sample size, thicknesses
 - Vary radiant heat and flame exposure times
- **Gather samples of different composite materials for intermediate and lab scale tests**
- **Perform intermediate and lab scale tests, change test parameters such that the intermediate and lab scale results correlate**

Questions or Comments?

Contact:

Robert Ochs
DOT/FAA Tech Center
Bldg. 287
Atlantic City Int'l Airport, NJ 08405
(609)-485-4651
robert.ochs@faa.gov

