Development of a Lab-Scale Test For Evaluating Decomposition Products of Burnthrough Compliant Insulation Systems







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Fuselage Burnthrough Chronology

Full-sale test article built at FAATC in mid 1990's for evaluating performance of burnthrough-resistant thermal acoustic insulation materials.

Testing indicated burnthrough-resistant insulation provided a much more survivable cabin atmosphere when compared to current insulation materials.

FAA issued NPRM, 2003 Final Rule issued, 2009 compliance.

Although burnthrough resistant materials provide a benefit, the ingress of toxic gases resulting from <u>decomposition</u> of the insulation needs to be quantified.

2005 FAATC began development of a lab-scale test for evaluating toxic gas decomposition products that could be generated inside fuselage during a postcrash fire.

Development of Lab-Scale Test For Measuring Decomposition Products During a Postcrash Fire

It is anticipated that this test method could be used to evaluate the potential toxicity of insulation constructions and innovations meeting the new burnthrough test requirements, in order to ensure that an *adverse* condition will not result inside an intact fuselage when exposed to an external fuel fire, despite the high burnthrough performance associated with a particular system.

This test method could also be used to evaluate the toxic contribution of the basic fuselage structure, whenever a nonmetallic material is used as the primary component.

Methodology

Conduct lab-scale burnthrough test on 2 types of burnthrough resistant insulation, and 1 type of structural composite material (without insulation). Measured the build-up of toxic and flammable gases within an enclosure simulating a fuselage

complete

Conduct subsequent full-scale tests with identical insulation materials to establish realistic baseline data using FTIR.

complete

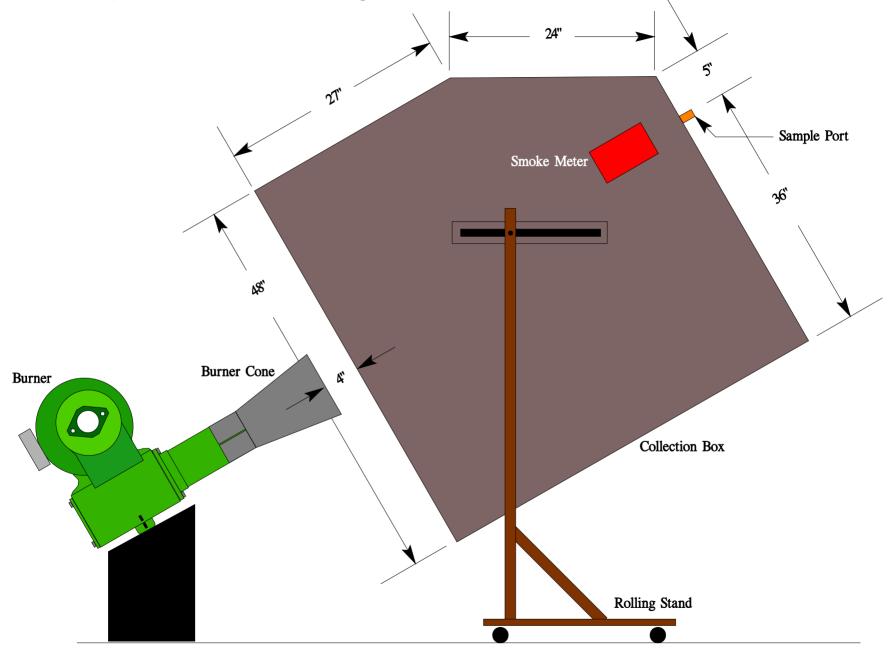
Determine concentration scaling factor between lab and full-scale tests in order to develop appropriate pass/fail criteria for lab-scale test.

complete

Produce Final Report and guidance on the acceptable level of decomposition products generated during lab-scale box test.

pending

Apparatus for Evaluating Toxic Gas Decomposition Products



Lab-Scale Apparatus for Evaluating Toxic Gas Decomposition Products

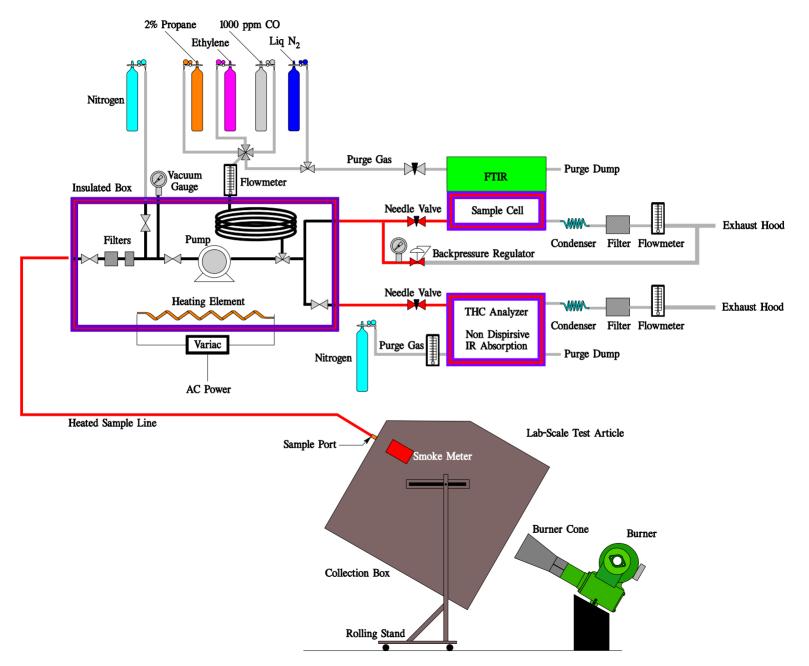
Burner configuration according to 25.856(b) Appendix F, Part VII.

Steel cube box simulates intact fuselage and serves as enclosure to collect emitted gases.

Fourier Transform Infrared (FTIR)/Total Hydrocarbon Gas analysis system used to collect and measure toxic and flammable gases yielded during tests.

Additional analyzers measured the concentration of carbon monoxide, carbon dioxide, oxygen, and total hydrocarbons (THC) as propane.

FTIR and THC Sampling System Used in Lab-Scale Testing

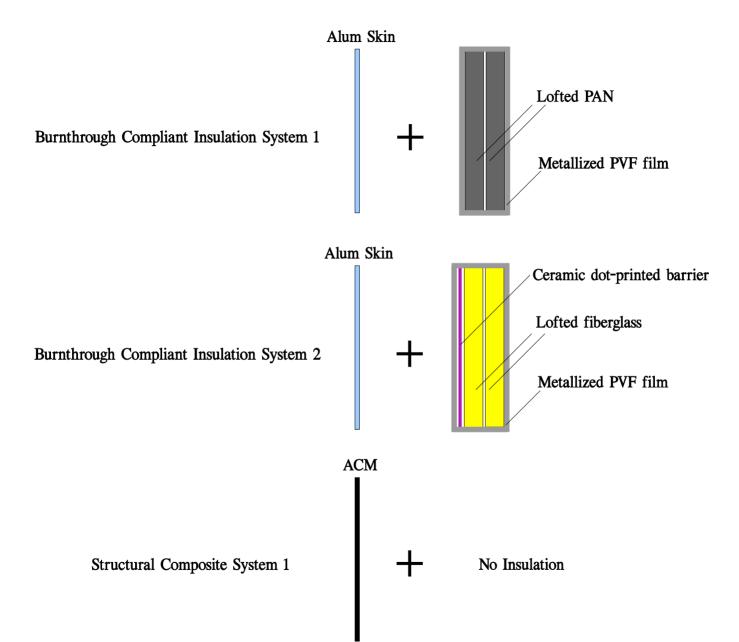


Gases Measured By FTIR

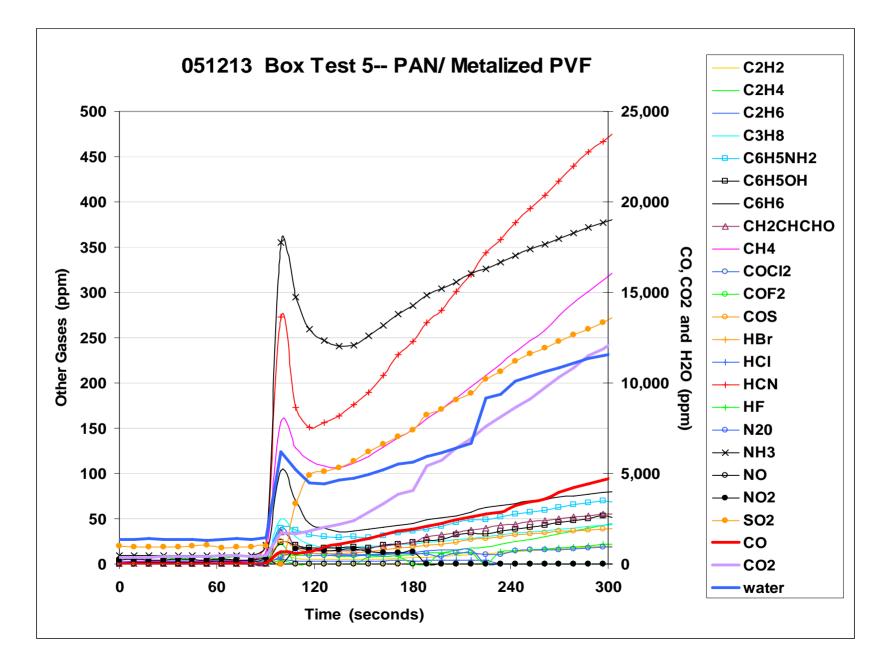
Toxic Gases	
C ₆ H ₅ NH ₂	Aniline
C ₆ H ₅ OH	Phenol
C ₆ H ₆	Benzene
CH ₂ CHCHO	Acrolein
CH ₄	Methane
СО	Carbon Monoxide
CO ₂	Carbon Dioxide
	Phosgene
COF ₂	Carbonyl Fluoride
COS	Carbonyl Sulfide
HBr	Hydrogen Bromide
HCL	Hydrogen Chloride
HCN	Hydrogen Cyanide
HF	Hydrofluoric Acid
NH ₃	Ammonia
NO	Nitrogen Oxide
NO ₂	Nitrogen Dioxide
SO ₂	Sulfur Dioxide

Flammable Gases	
C ₂ H ₂	Acetylene
C ₂ H ₄	Ethylene
C ₂ H ₆	Ethane
C ₃ H ₈	Propane
C ₆ H ₅ NH ₂	Aniline
C ₆ H ₅ OH	Phenol
C ₆ H ₆	Benzene
CH ₂ CHCHO	Acrolein
CH ₄	Methane
Other Gases	
CO ₂	Carbon Dioxide
H ₂ O	Water
N ₂ O	Nitrous Oxide

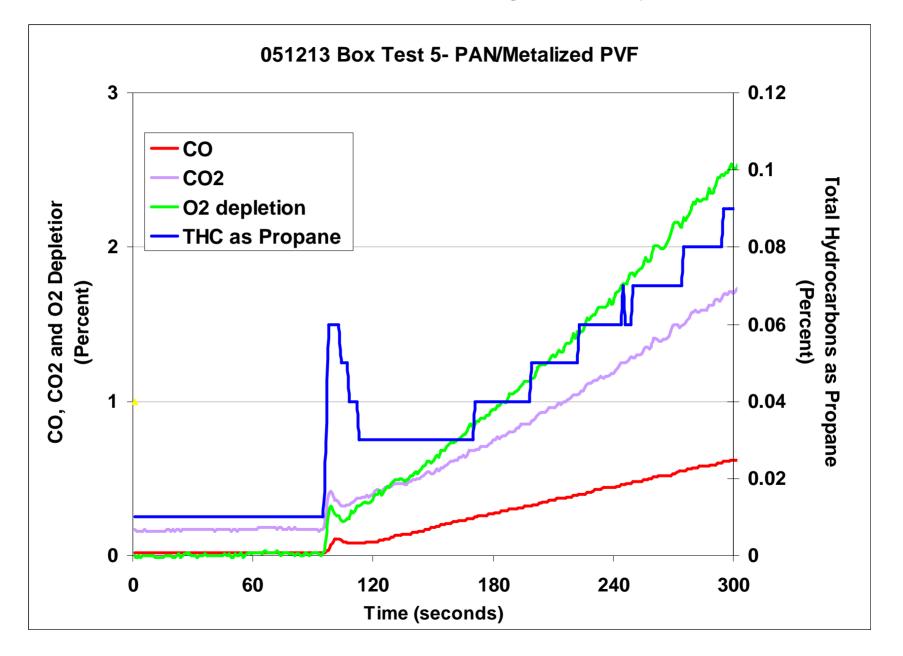
Material Systems Tested in Lab-Scale Apparatus



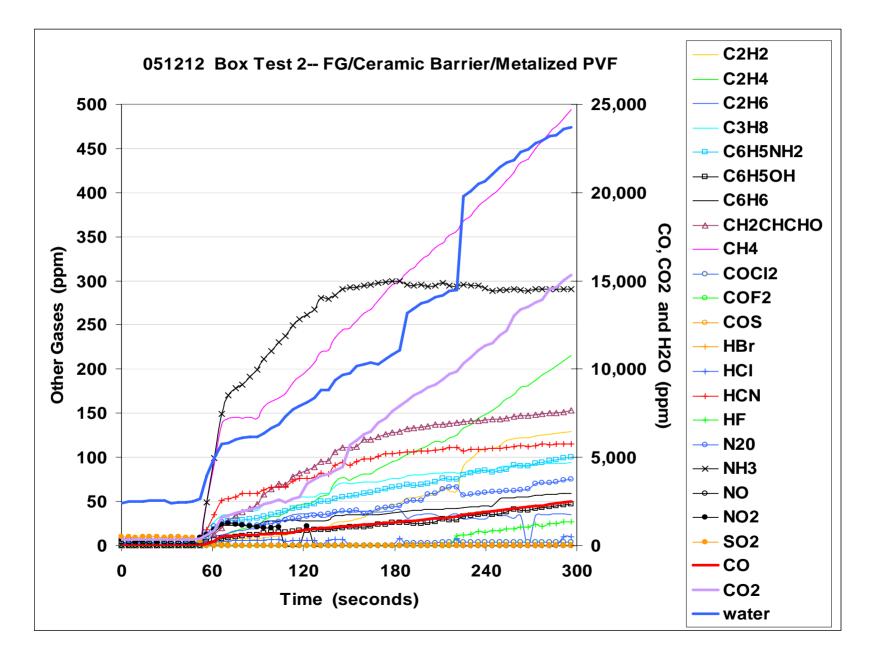
PAN Insulation Test Using FTIR Analysis



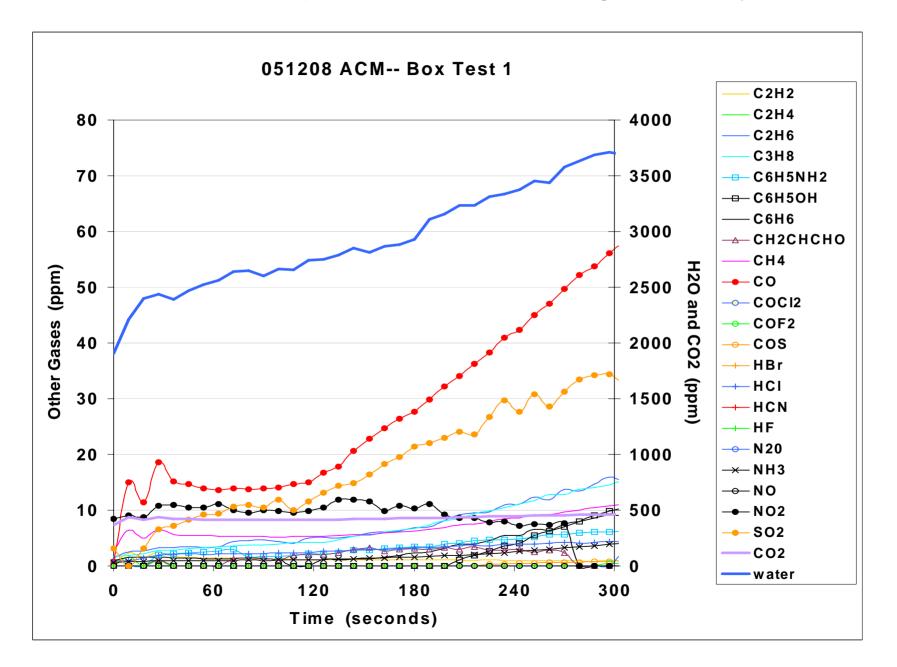
PAN Insulation Test Using Gas Analyzers



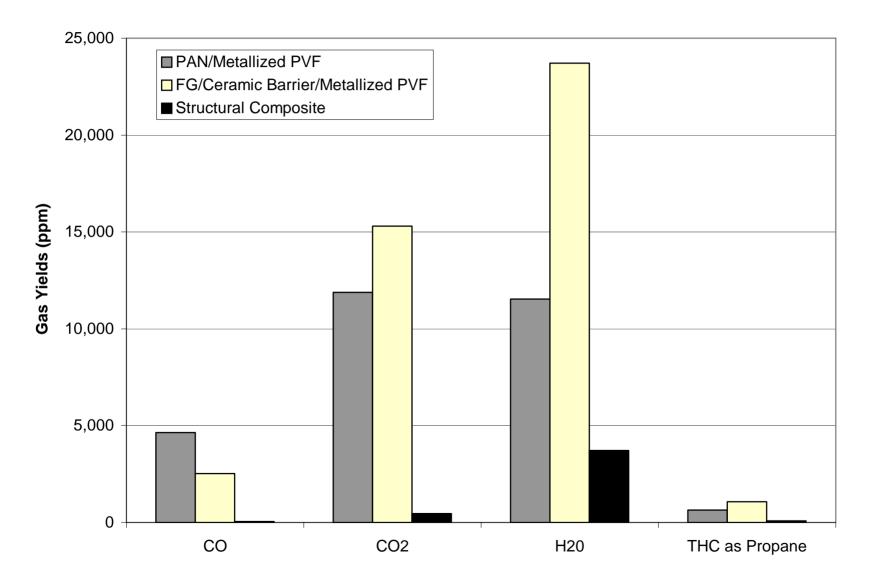
FG/Ceramic Barrier Insulation Test Using FTIR Analysis



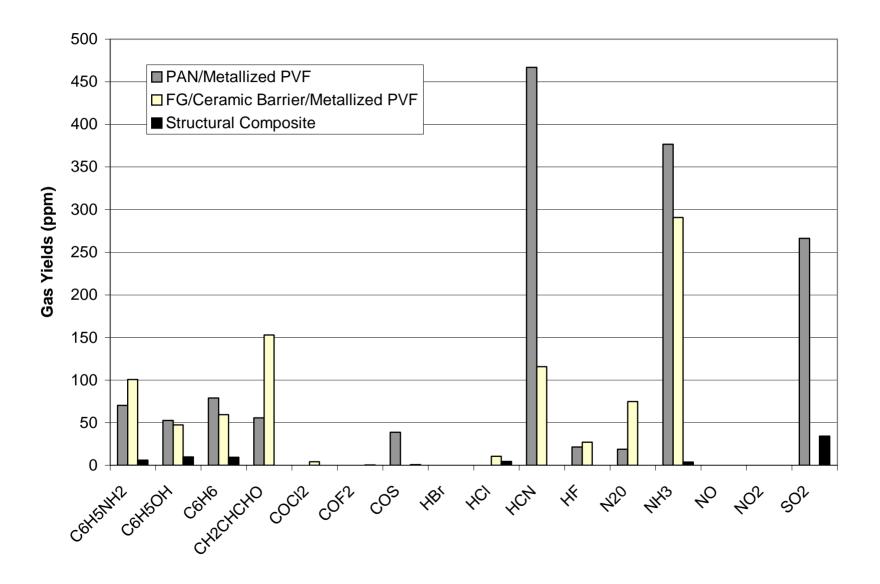
Structural Composite Material Test Using FTIR Analysis



Comparison of Box Test Results at 5 Minutes



Comparison of Box Test Results at 5 Minutes



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Development of a Laboratory-Scale Test for Evaluating the Decomposition Products Generated Inside an Intact Fuselage During a Simulated Postcrash Fuel Fire

Timothy R. Marker Louise C. Speitel

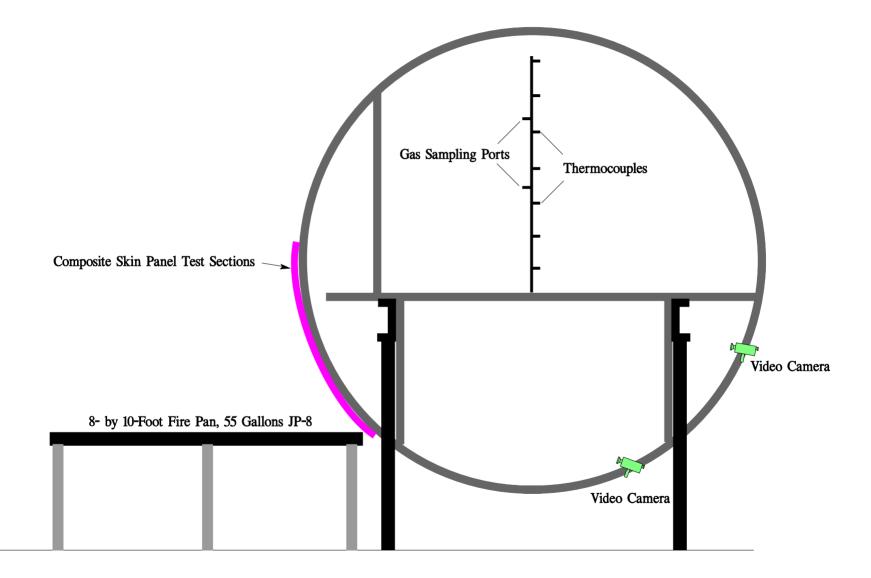
April 2007

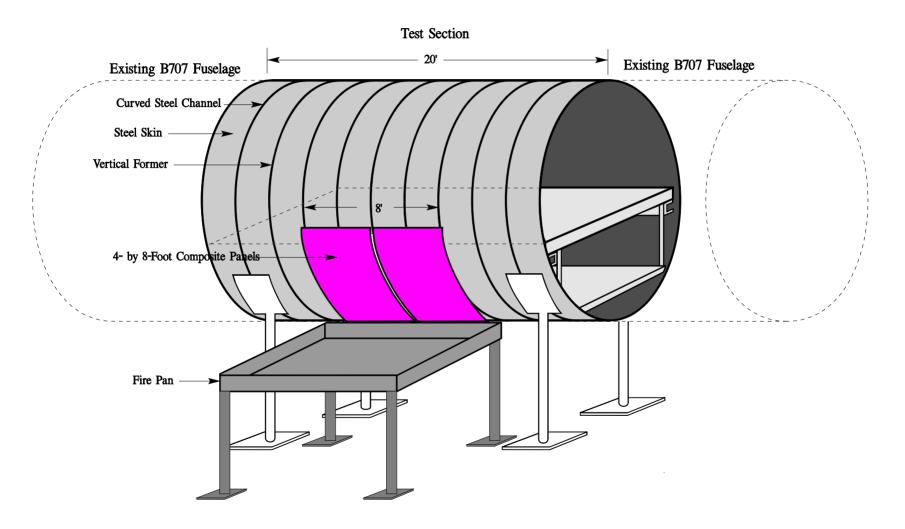
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DOT/FAA/AR-TN07/15

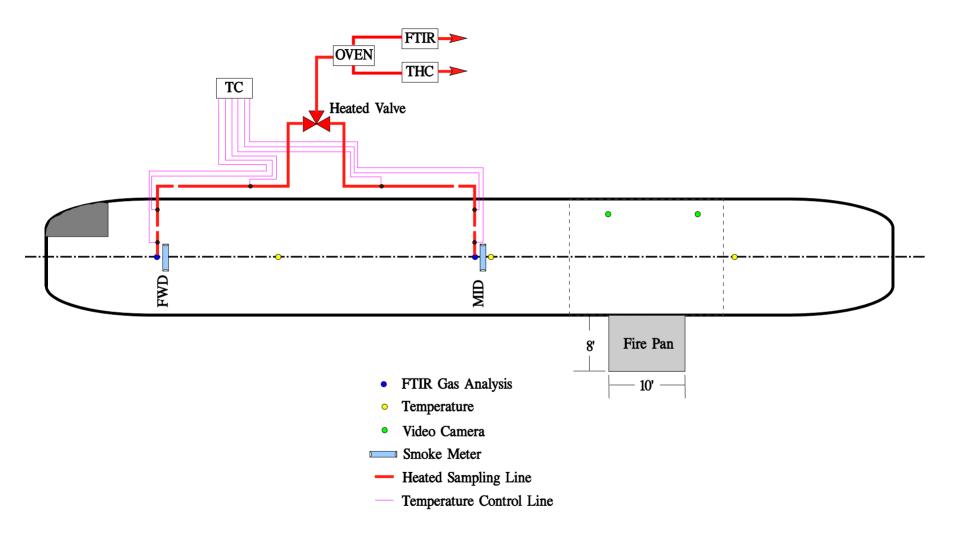
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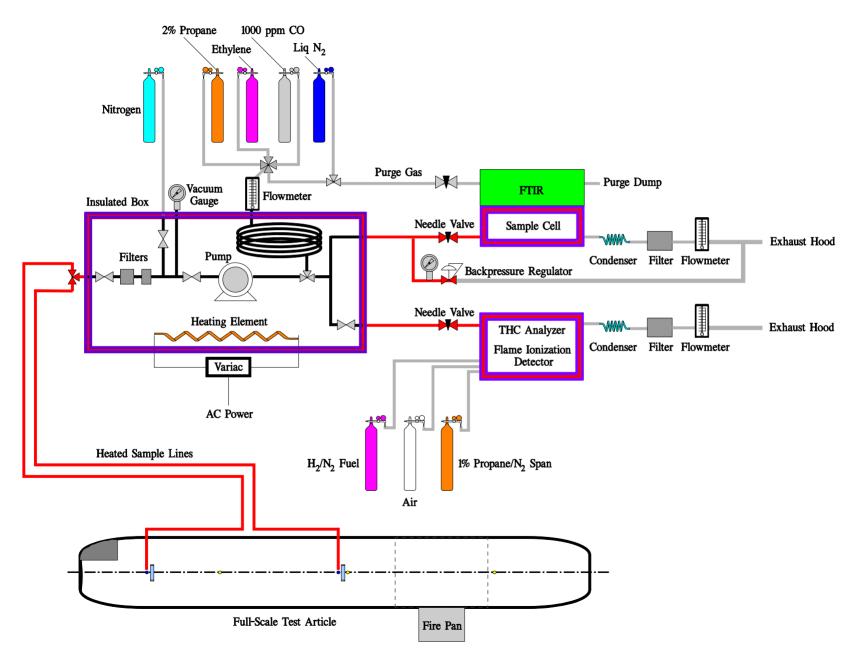








FTIR and THC Sampling System Used in Full-Scale Testing





Pre-test



Post-test



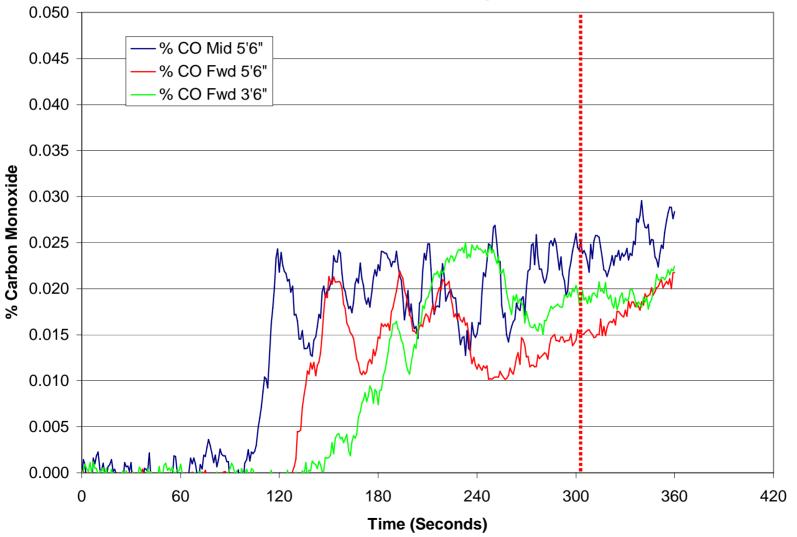
Post-test



Full-Scale Results, PAN Insulation, Gas Analyzers

Carbon Monoxide Levels

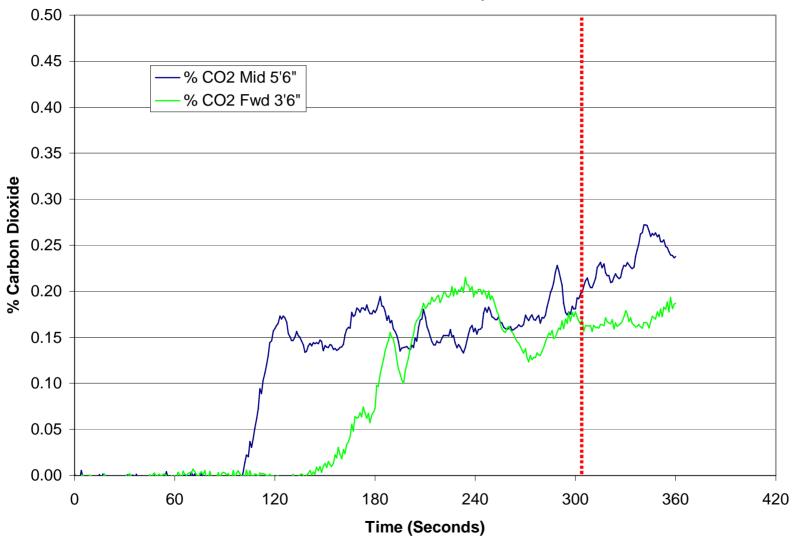
PAN Insulation, already corrected



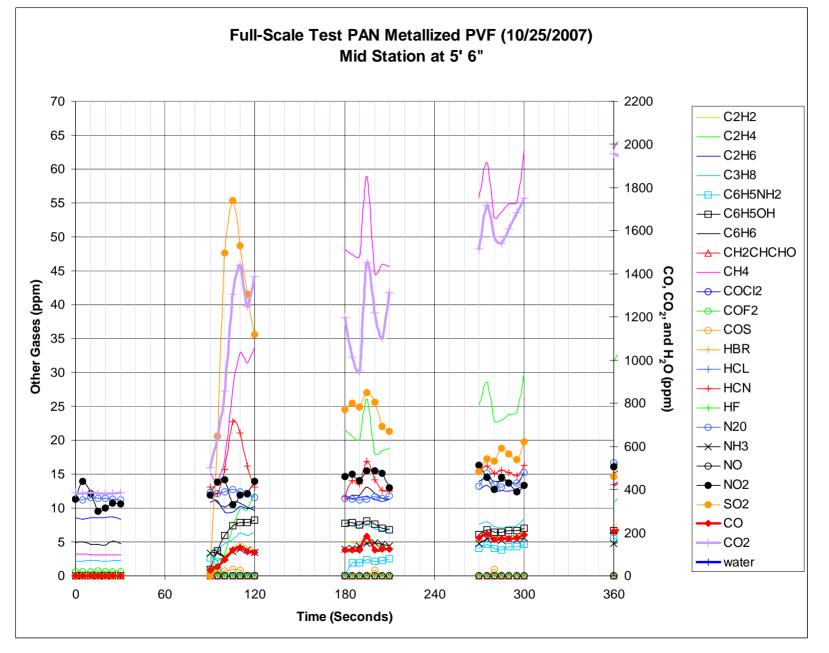
Full-Scale Results, PAN Insulation, Gas Analyzers

Carbon Dioxide Levels

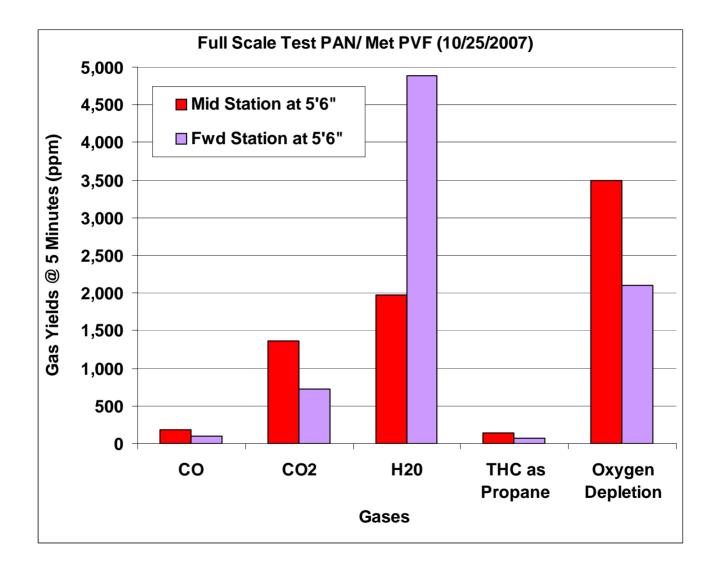
PAN Insulation, already corrected



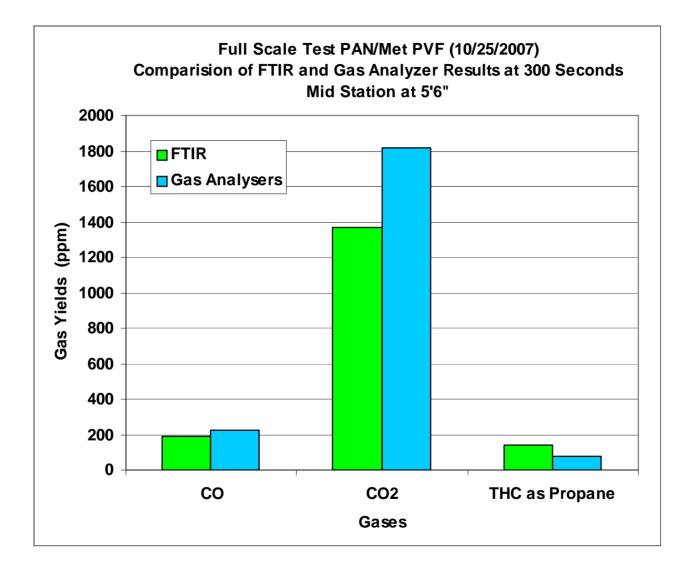
Full-Scale Results, PAN Insulation, FTIR



Full-Scale Results, PAN Insulation, FTIR



Full-Scale Results, PAN Insulation, Comparison

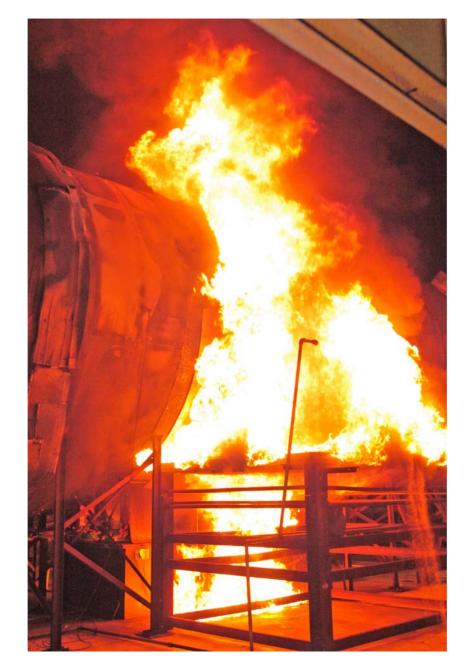


Full-Scale Test Results Ceramic Barrier Insulation System (Modified Configuration)

Full-Scale Test Results, Ceramic Barrier Insulation System II Pre-test



Full-Scale Test Results, Ceramic Barrier Insulation System II



Full-Scale Test Results, Ceramic Barrier Insulation System II Post-test



Full-Scale Test Results, Ceramic Barrier Insulation System II Post-test



Full-Scale Test Results, Ceramic Barrier Insulation System II Post-test



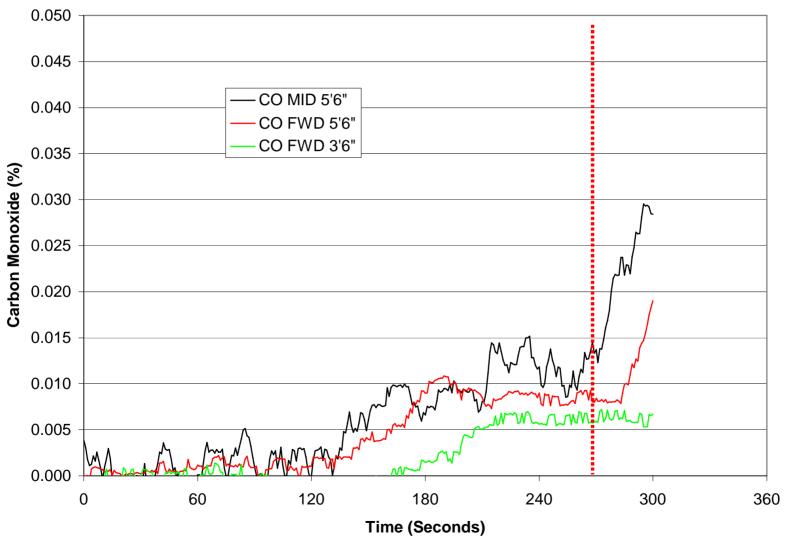
Full-Scale Test Results, Ceramic Barrier Insulation System II Post-test



Full-Scale Results, Ceramic Barrier Insulation II, Gas Analyzer

Carbon Monoxide Levels

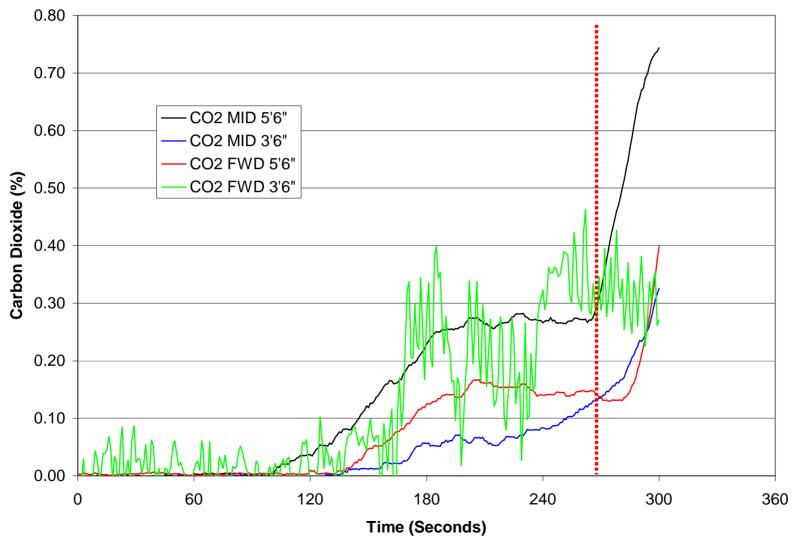
FG/Nextel 2nd test, already corrected for lag



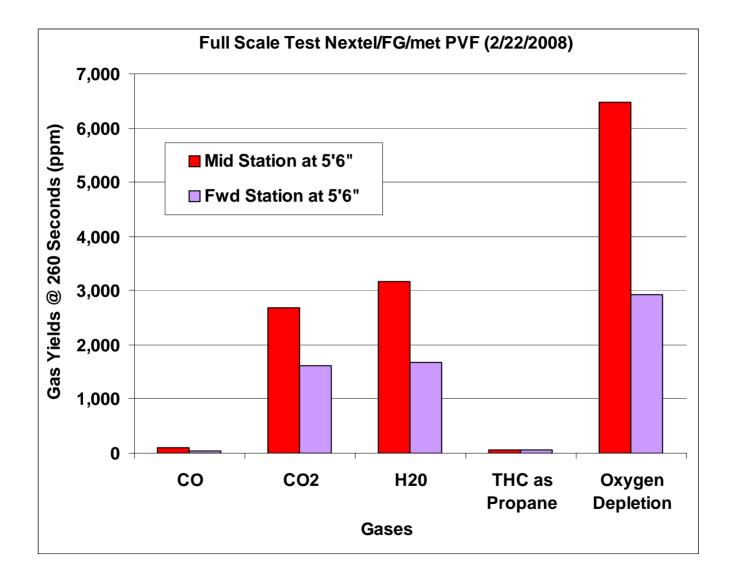
Full-Scale Results, Ceramic Barrier Insulation II, Gas Analyzer

Carbon Dioxide Levels

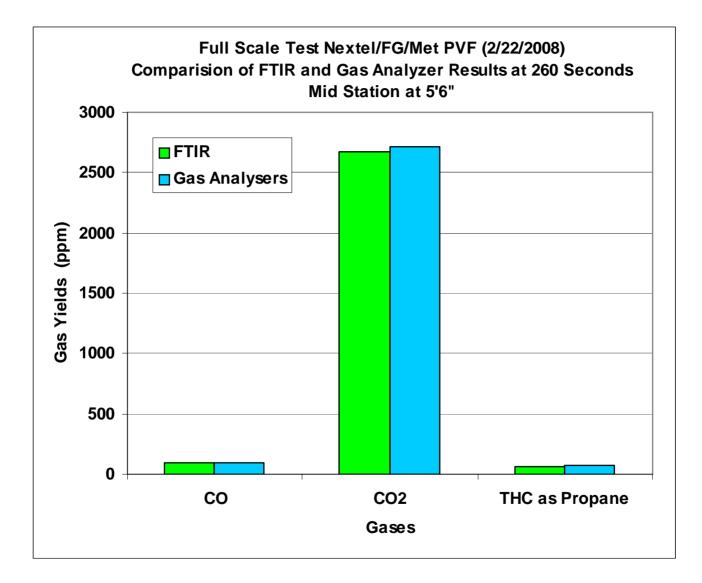
FG/Nextel 2nd test, already corrected for lag



Full-Scale Results, Ceramic Barrier Insulation II, FTIR

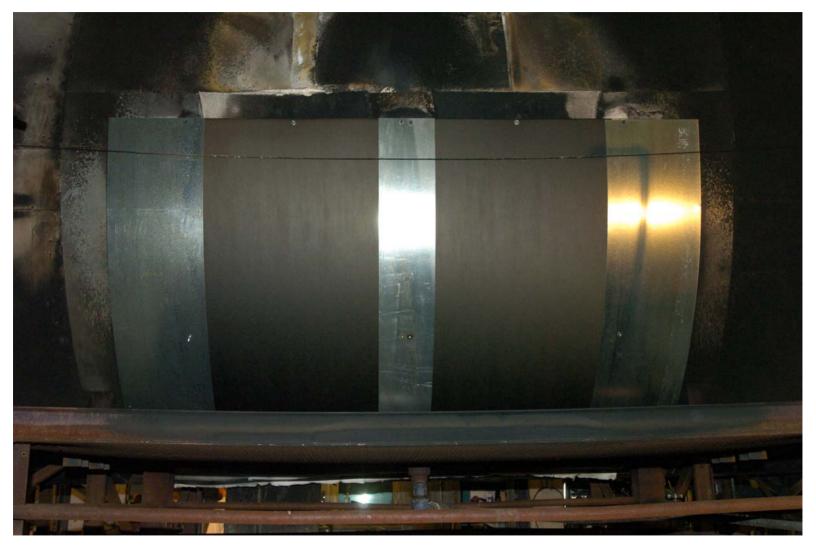


Full-Scale Results, Ceramic Barrier Insulation II, Comparison



Full-Scale Test Results Structural Composite System

Full-Scale Test Results, Structural Composite System *Pre-test*



Full-Scale Test Results, Structural Composite System



Full-Scale Test Results, Structural Composite System

Post-test



Full-Scale Test Results, Structural Composite System *Post-test*



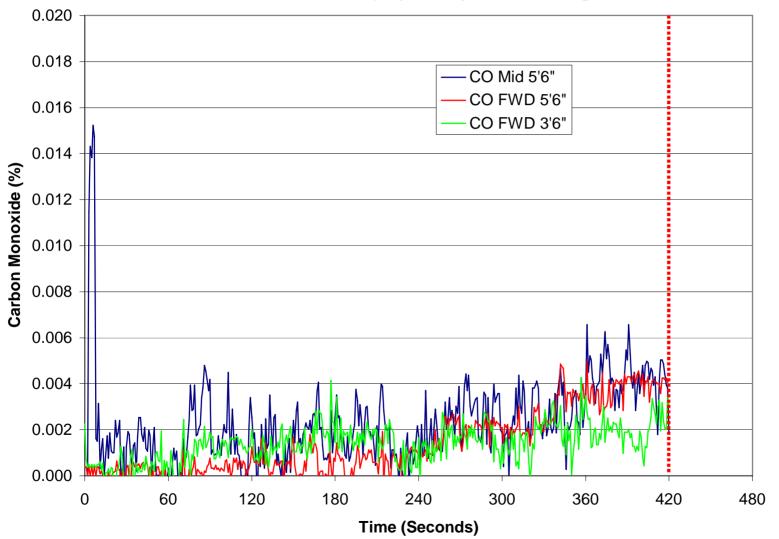
Full-Scale Test Results, Structural Composite System *Post-test*



Full-Scale Results, Structural Composite, Gas Analyzer

Carbon Monoxide

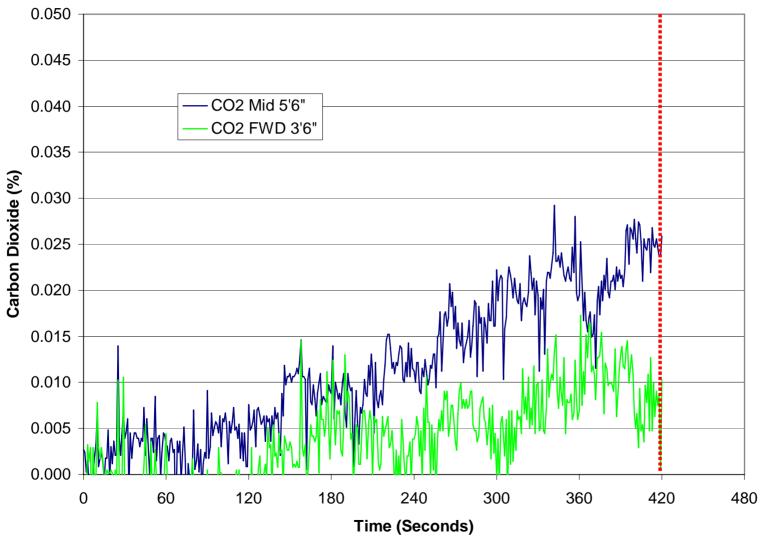
Carbon/Epoxy, already corrected for lag



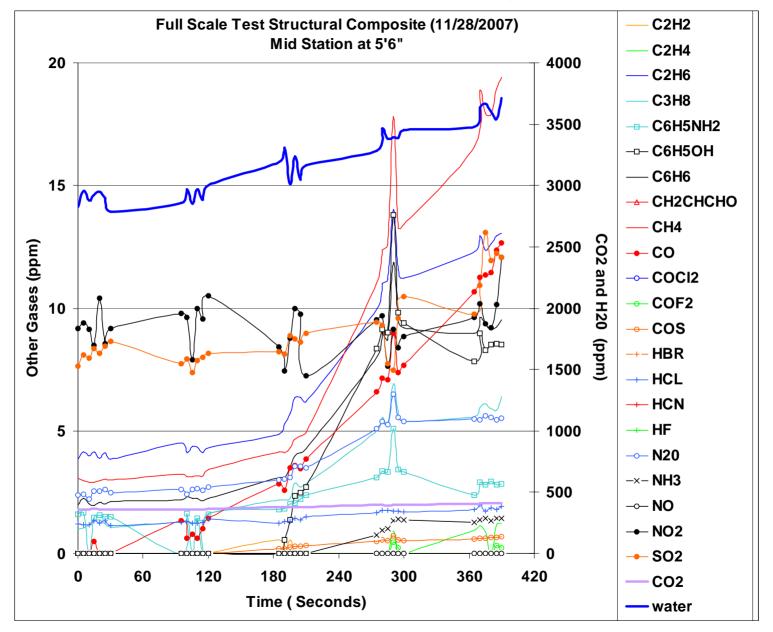
Full-Scale Results, Structural Composite, Gas Analyzer

Carbon Dioxide Levels

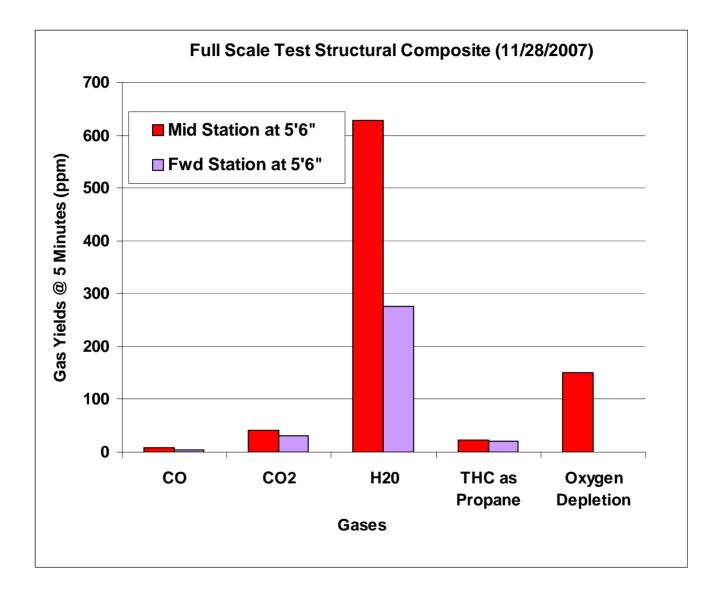
Carbon/Epoxy, already corrected for lag



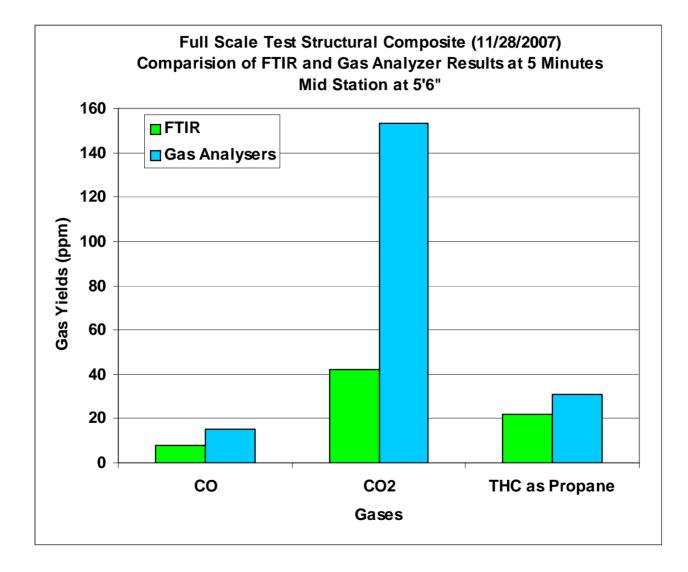
Full-Scale Results, Structural Composite, FTIR



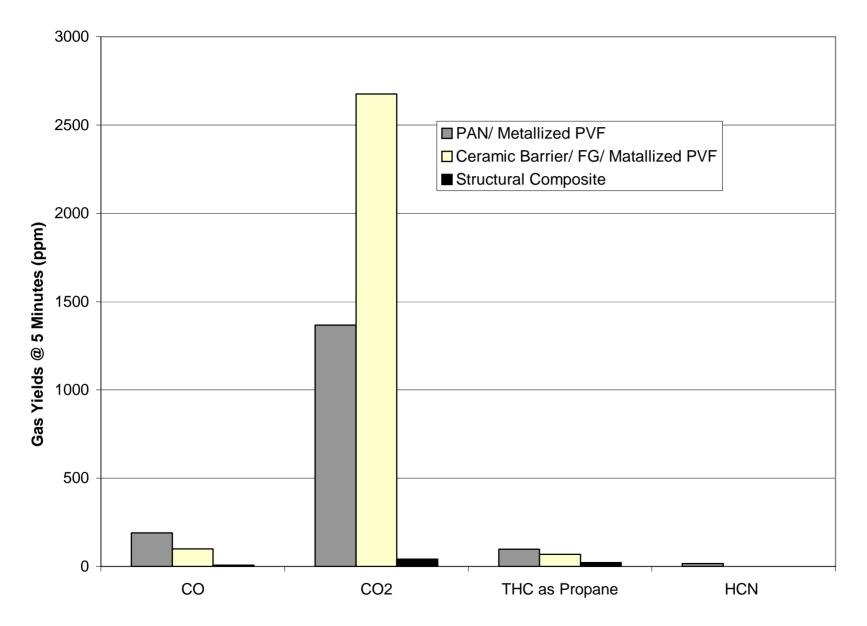
Full-Scale Results, Structural Composite, FTIR



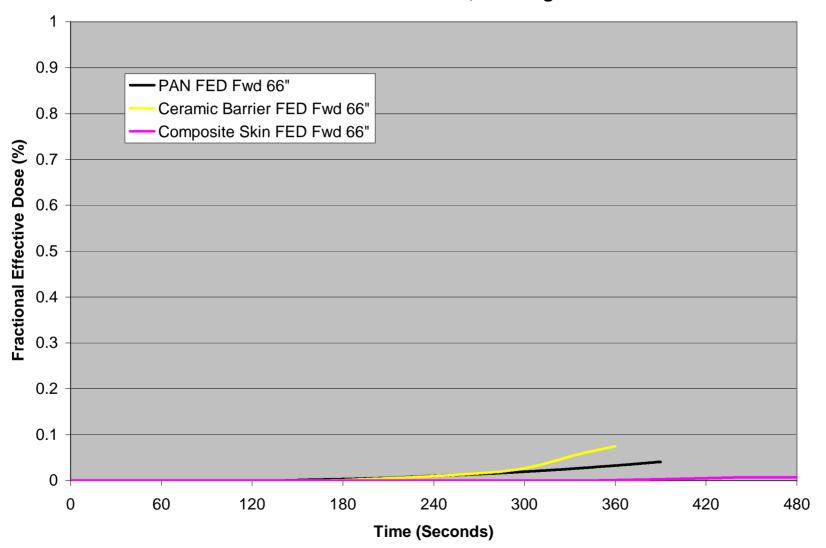
Full-Scale Results, Structural Composite, Comparison



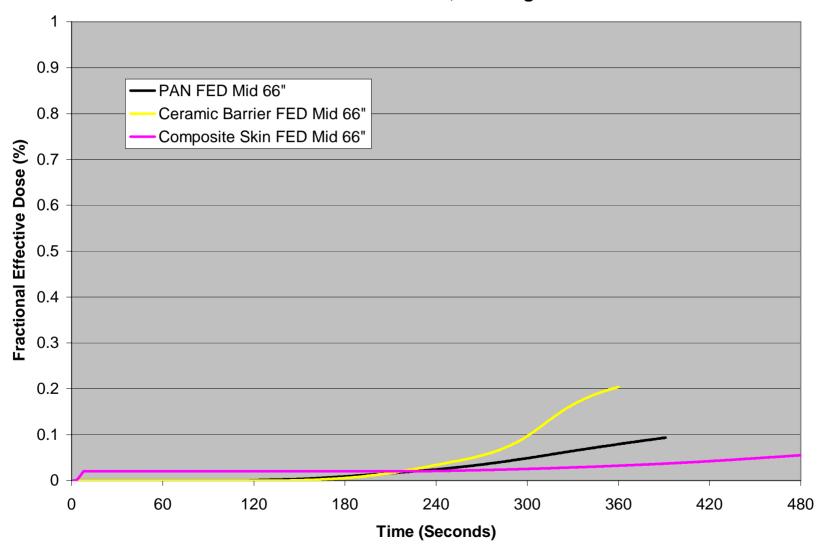
Full-Scale Results, Comparison of 3 Insulation Systems



Fractional Effective Dose Comparison Forward Station, 66" Height



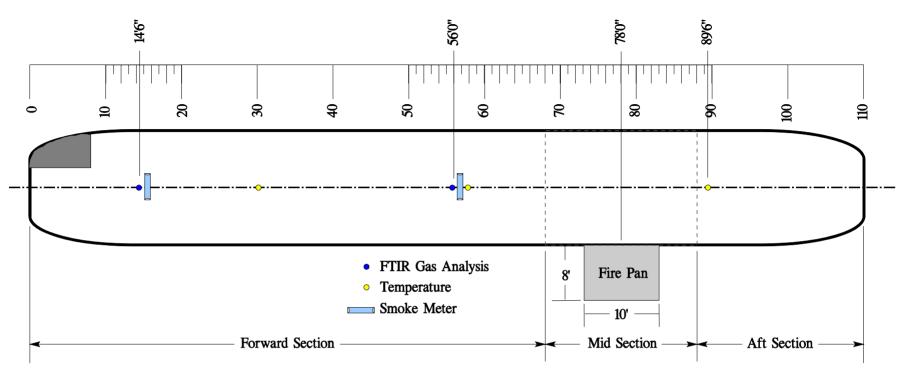
Fractional Effective Dose Comparison Mid Station, 66" Height



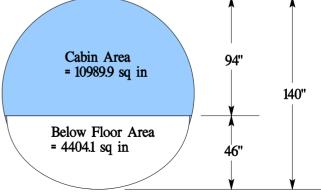
What do we do with all this data?

How does data compare to small scale results?

Determination of Full Scale Test Article Volume



Forward Volume = Cabin Area x Fwd Length = (10989.9/144) x 68 = 5189.7 cu ft Mid Volume = Total Cabin Area x Mid Length = (17203/144) x 20 = 2389.4 cu ft Aft Volume = Cabin Area x Aft Length = (10989.9/144) x 22 = 1679 cu ft Total Volume = Forward Volume + Mid Volume + Aft Volume Total Volume = 5189.7 + 2389.4 + 1679 Total Volume = 92.58.1 cu ft



Determination of Gas Concentration Scaling Factor

Ratio of Volume_{Box} to Burn Area_{Box} = 60.33 ft³ / 9.25 ft² = 6.52

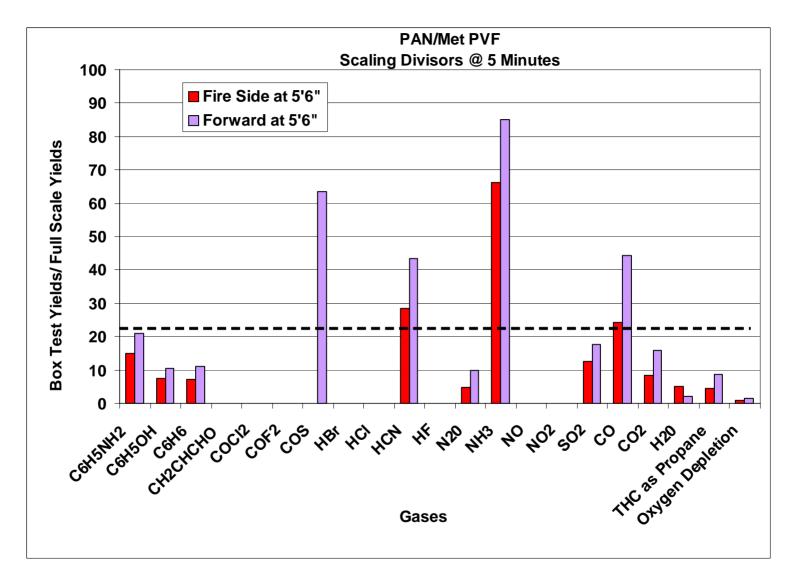
Ratio of Volume_{FSTest} to Burn Area_{FSTest} = 9258.1 ft³ / 64 ft² = 144.7

Ratio of Full Scale to Lab Scale = 144.7 / 6.52 = 22.2

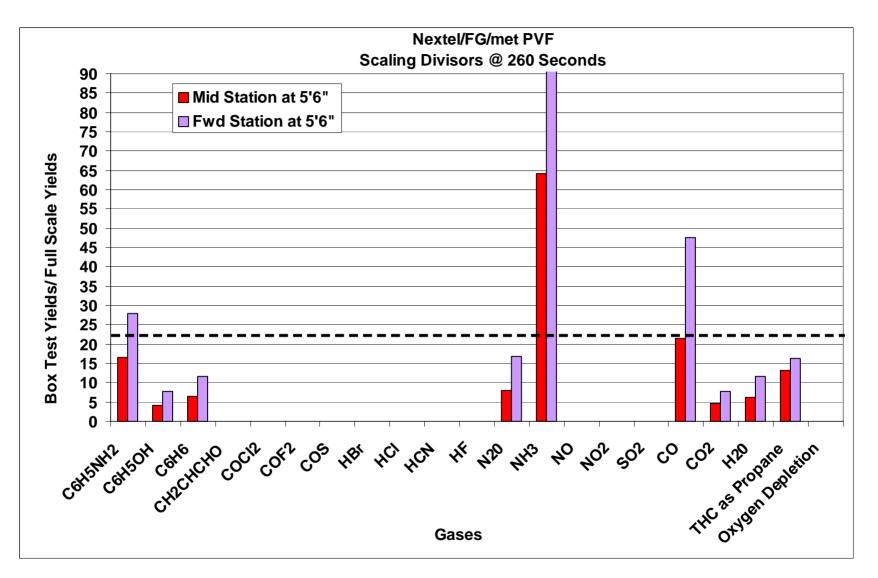
Full-Scale Test Article has 22.2 Times More Volume per Burn Area than Lab Scale Box

Theoretical Lab Scale Box Concentration is 22.2 Times Greater than Full Scale Concentration

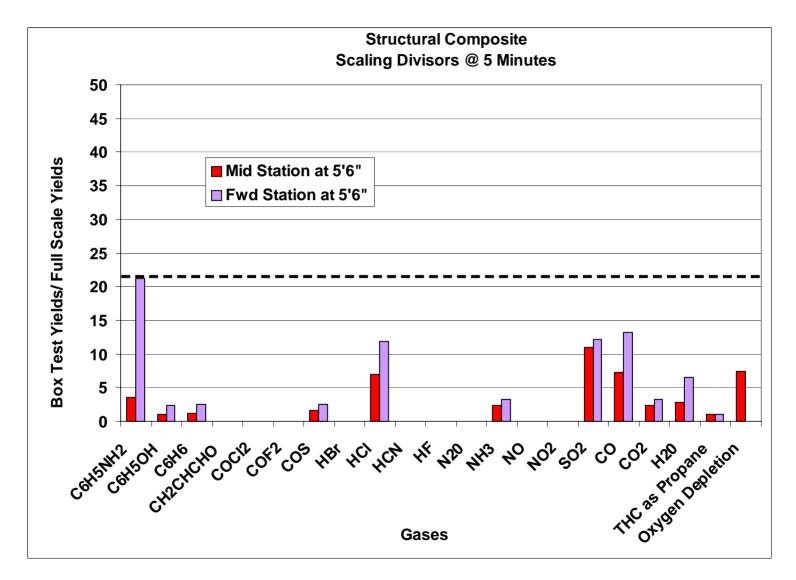
Gas Concentration Scaling, PAN Insulation System



Gas Concentration Scaling, Ceramic Barrier Insulation System



Gas Concentration Scaling, Structural Composite System



Gas Concentration Scaling, Findings

Analysis only considers volumetric aspects

Analysis assumes perfect mixing

Analysis does not consider surface area effects

Not all of gases scale similarly (example: COS)

Primary intoxicants (CO, HCN) scaled similarly

Development of Decomposition Products Limits for Burnthrough Compliant Insulation Systems

Difficult to use volumetric scaling as basis for setting limits in lab-scale test.

Since full-scale tests did not result in adverse conditions inside fuselage,

Take maximum (peak) values obtained in box test for each gas,

Add reasonable safety factor,

Establish acceptable decomposition limits in box test

Development of Decomposition Products Limits for Burnthrough Compliant Insulation Systems

Example: HCN

During full-scale tests, HCN did not reach toxic levels for any of three materials

During lab-scale tests, HCN reached the following levels:

PAN material 470 ppm FG/ceramic barrier 120 ppm Composite skin 0 ppm

HCN acceptable limit would be set at 500 ppm in box test.

If a burnthrough compliant material produced greater than 500 ppm HCN during a box test, then a full-scale test would be necessary.

																								1	-
	Full-Scale Test Data						Allowable Exposure from Various References							Lab-Scale Data							Scaling 5'6"	Factor Mid	d Station at	-	Max Allowe
	PAN/M	et PVF	FG/Ceramic Barrier/Met PVF (260 Sec)		Structural Composite		5 Minute Exposure				60 Minute 30 Minute Exposure Exposure		FED Effect				Acceptable Lab-					FG/			Conc. (ppm) Max B
ields 5 Ites	Station	Fwd Station at 5'6"	Station	Fwd Station at 5'6"	Station	Fwd Station at 5'6"	Incap Conc		Derived from 60 min ERPG3		ERPG 3 (2007)	IDLH (1995)		PAN/Met PVF	FG/Cera mic Barrier/M et PVF	Structural Composit	Scale Tox Limit= 5 Min Exp Limit x Scaling Factor	Acceptable Lab-Scale Tox Limit (ppm)	FED Effect		PAN/Me t PVF		Structural Composit	Gases to Measure FED > .02	Test Conc >
test IH2	4.63	3.27	5.5	3.27	1.73	0.29	Not Avail		00	600	?	100	0.009	68.73	91.14	e 6.14	(ppm) 600 x 15.7 (IDLH)	(ppin) 9,420	0.010	C6H5NH2	14.8	16.6	3.5	FED 2.02	1.4
nr∠)H	7.02	4.59	9.57	5.01	9.4	0.29	Not Avail		2400	1500	200	250	0.003	52.22	38.95	-	2400 x 4.2 (ERPG)	10,080	0.005	C6H5OH	7.4	4.1	1.0		
	10.46	10.41	8.05	4.56	7.21	3.21	?	?	12,000	3000	1000	500	0.001	76.60	52.5	8.33	12,000 x 5.0	60,000	0.003	С6Н6	7.3	6.5	1.2		
юно	0	0	0	0	0	0	10928	7783	18	12	1.5	2	0.000	55.50	146.04	0	7783 x 4 (LC50)	31,132		сн2снснс		#DIV/0!	ND		
	0	0	0	0	0	0	?	102 d	12	12	1	2	0.000	0.00	3.9	0	102 x 4 (LC50)	408		COCI2	ND	#DIV/0!	ND		
	0	0.21	0	0	0	0	?	?	300		25 (est)	?	0.001	0.00	0	0.43	300 x 4 (ERPG3)	1,200	0.000	COF2	ND	ND	#DIV/0!		
	0	0.61	0	0	0.53	0.34	500* (500 for 15min- brain damage)	(1000 for 15 minutes)	1200 (H2S)		100 (H2S)	?	0.001	38.66	0	0.84	500 x 4	2.000	0.019	cos	~10	ND	1.6	Yes	
	0	0.01	0	0	0.00	0.04	16830	15900	1800	180	150 (est)	30	0.000	0.00	0	0.04	15900 x 4 (LC50)	63,600	0.000	HBr	ND	ND	ND		
	0	0	0	0	0.49	0.29	16830	15900	1800	300	150	50	0.000	0.00	0	3.43	. ,	63,600	0.000	HCI	ND	ND	7.0		
	16.4	10.75	0	0	0	0	176	560	300	300	25	50	0.093	467.00	111.74	0	176 x 20 (Incap)	3,520	0.133	HCN	28.5	#DIV/0!	ND	Yes	
beak)	22.7	26.9					176	560	300	300	25	50	0.153	467.00	111.74	0	176 x 20 (Incap)	3,520	0.133	HCN	20.6			Yes	
	0	0	0	0	0	0	7663	7227	600	180	50	30	0.000	14.46	19.3	0	7227 x 4 (LC50)	28,908	0.001	HF	#DIV/0!	#DIV/0!	ND		
	3.95	9.94	7.81	3.72	2.99	1	?	?			?	?		18.75	62.56	0		No Limit		N20	4.7	8.0	ND		
	5.55	4.32	4.5	1.82	1.36	1	?	?	9000	1800	750	300	0.003	367.20	289.19	3.3	9000 x 65.3 (ERPG)	587,700	0.001	NH3	66.2	64.3	2.4		
	0	0	0	0	0	0	12850	4260	1800	600	150	100	0.000	0.00	0	0	4260 x 4 (LC50)	17,040	0.000	NO	ND	ND	ND		
	2.02	1.19	13.13	6.19	0	0	2570	852	360	120	30	20	0.007	0.00	0	0	852 x 4	3,408	0.000	NO2	ND	ND	ND		
	19.81	2.06	2.04	1.33	2.82	2.56	?	2115	180	600	15	100	0.009	246.57	0	31.17	2115 x 11.8	24,957	0.010	SO2	12.4	ND	11.1		
oeak)	55.4	65.5					?	2115	180	600	15	100	0.031	246.57	0	31.17	2115 x 4.5 16600 x 17.6	9,518	0.026	SO2	4.5			Yes	
	190.9	104.8	99.18	44.49	7.7	4.2	6850	16600	6000	7200	500	1200	0.028	4645.76	2116.23	55.32	(ERPG3)	292,160	0.016	со	24.3	21.3	7.2	Yes	6.
	1367.6	730.3	2674.66	1608	42	30	88000					40000	0.030	11506.60	12657	96.7	88,000 x 6.6	580,800	0.022	CO2	8.4	4.7	2.3	Yes c	17,
	1973.9	4885	3160.63	1684	627	276								10164.77	19430	1808.29	No Limit		N/A	H2O	5.1	6.1	2.9		
s ne	97.9	72.2	68.17	55.21	22	20.8							0.005	629.71	903.5	22.0	21,000 x 6.9	144,900	0.006	THC as Propane	6.4	13.3	1.0		
n ion	3500	2100	6470	2920	150	0	136000							3000.00		1120	Remove			Oxygen Depletion	0.9		7.5		