International Aircraft Materials Fire Test Working Group

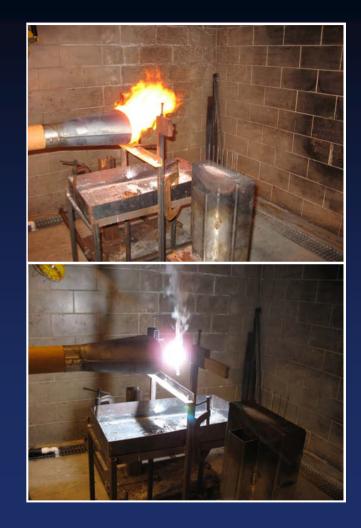
Update on Flammability Testing of Magnesium Alloy Components

Presented to: IAMFT WG, Niagara Falls By: Tim Marker, FAA Technical Center Date: June 17, 2008



Federal Aviation Administration

- Background
- Fire Threats
- Test Method Development
- Lab Testing Thus Far
- Planned Full Scale Testing
- Closing





Background

- Renewed interest in using mag-alloys in commercial aviation
- Current FAA TSO C127 "Rotorcraft and Transport Airplane Seating Systems" makes reference to SAE specification, which bans use of magnesium in seats
- SAE specification references tests conducted 30 years ago at FAATC





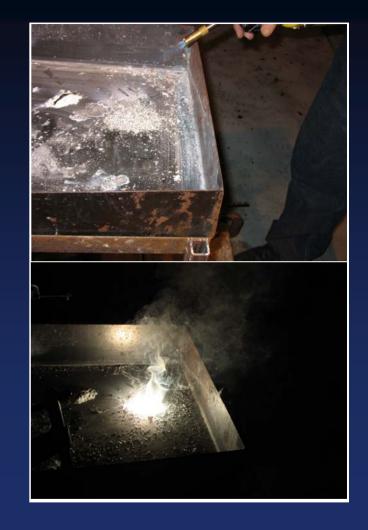


What are advantages and drawbacks?

Benefits

- Lightweight*
- Machinability
- Recyclable
- Abundant

*Boeing estimates \$1 million lifetime saving to the operator for every 100 lb of weight saving.

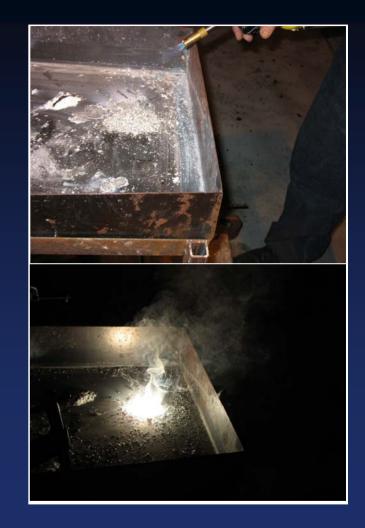




What are advantages and drawbacks?

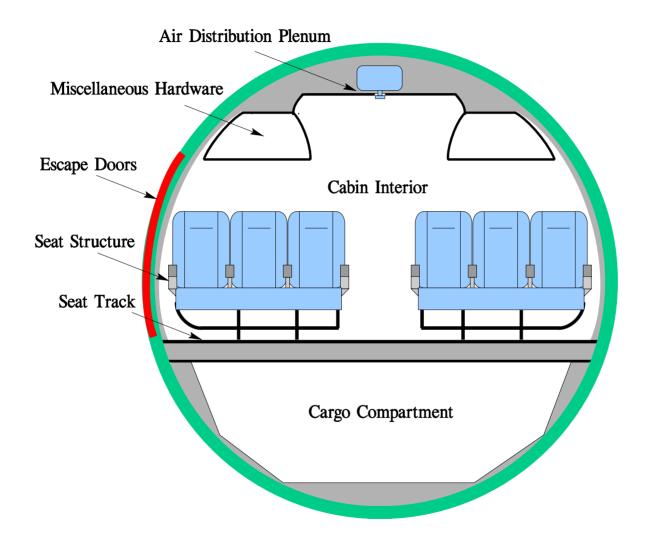
Disadvantages

- Corrosion
- Flammability
- Manufacturing/housekeeping challenges during machining





Magnesium Alloy Flammability: Potential Use Locations







Electrical arc, hidden fire adjacent to mag-alloy component

Direct threat of fire entering cabin, flashover, passenger and firefighter protection



How do we develop an appropriate test method?

Clearly define the threat(s)

Replicate as many aspects of threat conditions as possible

Correlate with results of full-scale testing



What Has Been Done?

Initial Laboratory Scale "Fact-Finding" Experimentation

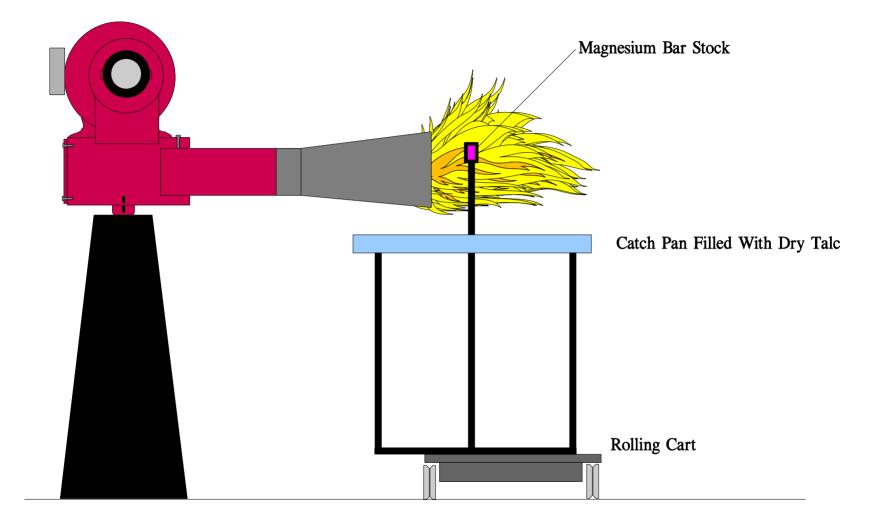
Oil Burner Testing

Handheld Extinguisher Testing

Miscellaneous Lab-Scale Flammability Testing



Initial Oil Burner Testing of Mag Alloy





Initial Oil Burner Testing of Mag Alloy





Magnesium Burning After Burner Shut Off



Update on Flammability Testing of Magnesium Alloy Components June 17, 2008



Federal Aviation Administration

Mag Alloy Test Results Using Oil Burner



Update on Flammability Testing of Magnesium Alloy Components June 17, 2008



Federal Aviation Administration

Findings of Oil Burner Testing

None of the magnesium bar samples melted prior to 2 minutes

Extending exposure time beyond 2 min caused melting and ignition

78% of samples (18 of 23) continued to burn after burner flame removed

22% of samples (5 of 23) self extinguished within 5 seconds

Sample performance (i.e., flammability) largely dependent on alloy type and section thickness

WE43, Elektron 21, and Elektron 675 more ignition resistant than other alloys

Vertical orientation of sample promoted continued burning

Use of intumescent coating increased time-to-melt substantially



Critical Elements of Postcrash Lab Test for Magnesium

Flame duration/exposure time

Size, shape, thickness of sample

Orientation of sample

Time to reach melting

Ignition following melting (y/n)?

Duration of after-flame following ignition

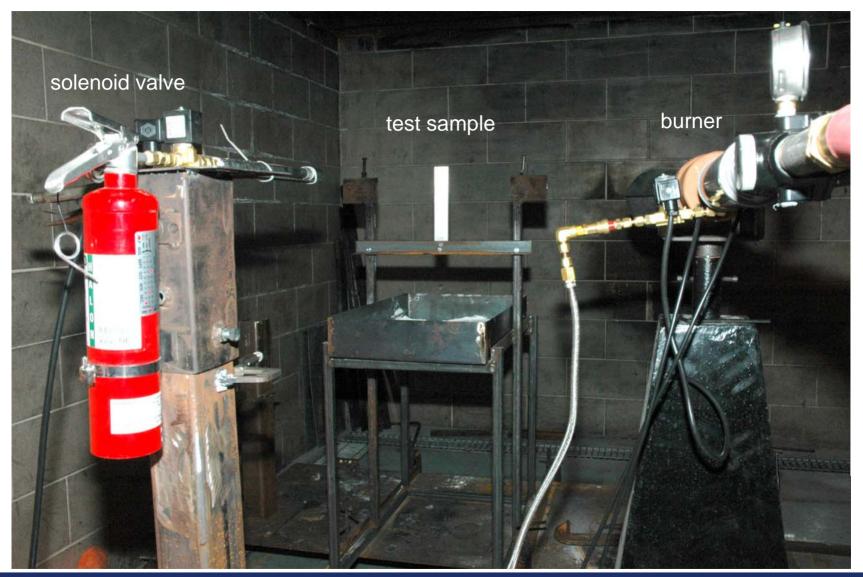




Update on Flammability Testing of Magnesium Alloy Components June 17, 2008



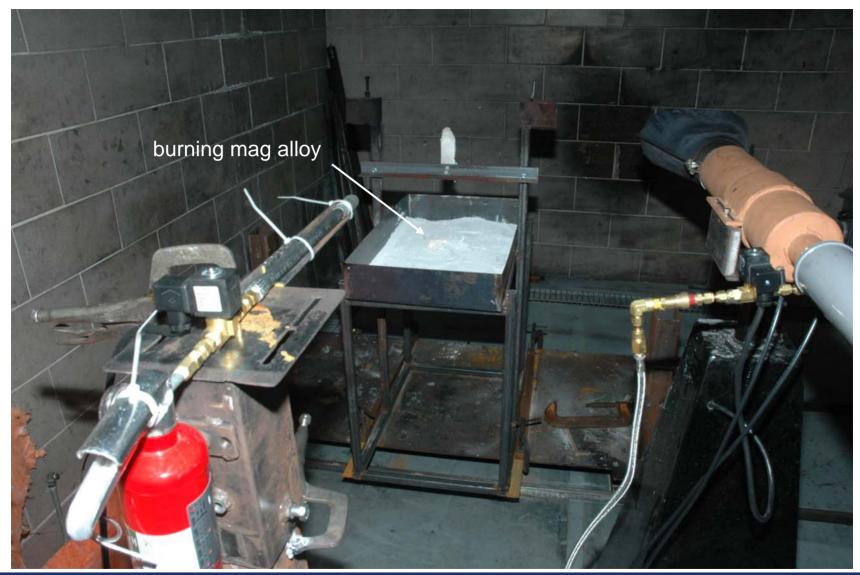
Federal Aviation Administration





























7 tests conducted on ignited mag-alloy samples (3 Halon-1211, 2 water, and 2 FE-36)

All extinguishing agent applications caused minor flare-up, sparking, and excitation of the burning samples, but no explosions or detonation

Halon-1211 least effective at extinguishing fire; water most effective

FE-36 caused rapid oxidation of burning samples



Additional Lab-Scale Flammability Testing of Mag Alloy Samples

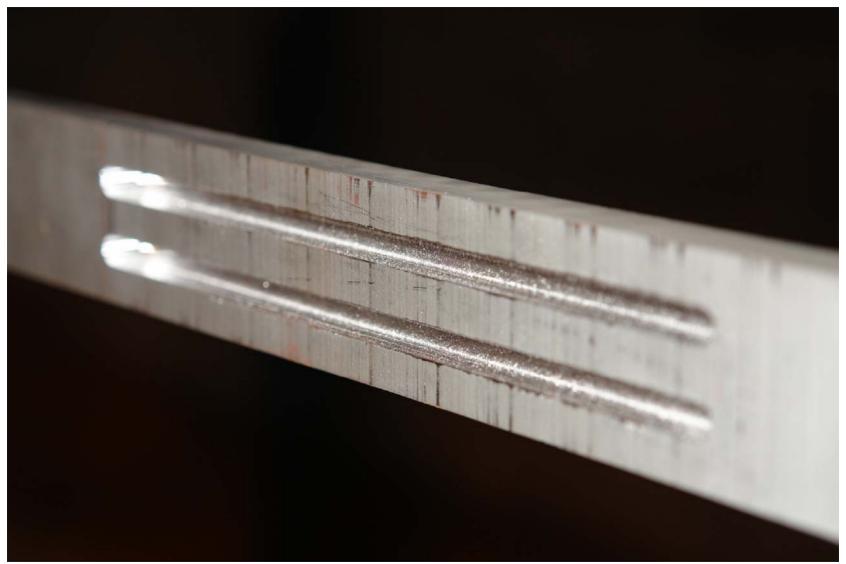


Burner test of sample with modified cross section





Hemispherical grooves machined into test sample





Machining did not increase flammability of test sample





Vertically oriented machined test sample -no change





Flammability of turnings from lathe





Mag-alloy turnings ignited by torch





Burning mag-alloy turnings





Flammability of thin slice of mag-alloy





Flammability of thin slice of mag-alloy





Ignition of thin slice of mag-alloy





Flammability of thin slice using burner





Flammability of thin slice using burner





Thin slice not ignited using burner





Burner test of sample with modified cross section





Burner test of sample with modified cross section





Ignition of sample with modified cross section





Magnesium Alloy Flammability

Preliminary lab scale fact-finding testing

Handheld extinguisher testing

Define critical elements of preliminary testing

Conduct full scale test using mag-alloy seat frames

Develop lab scale test based on full-scale results



How Should a Full Scale Seat Test Be Conducted?

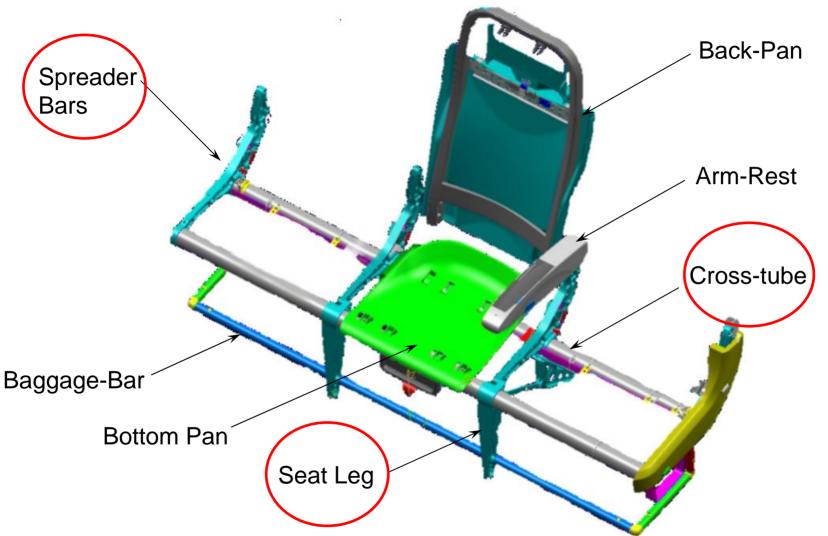


Typical Seat Assembly



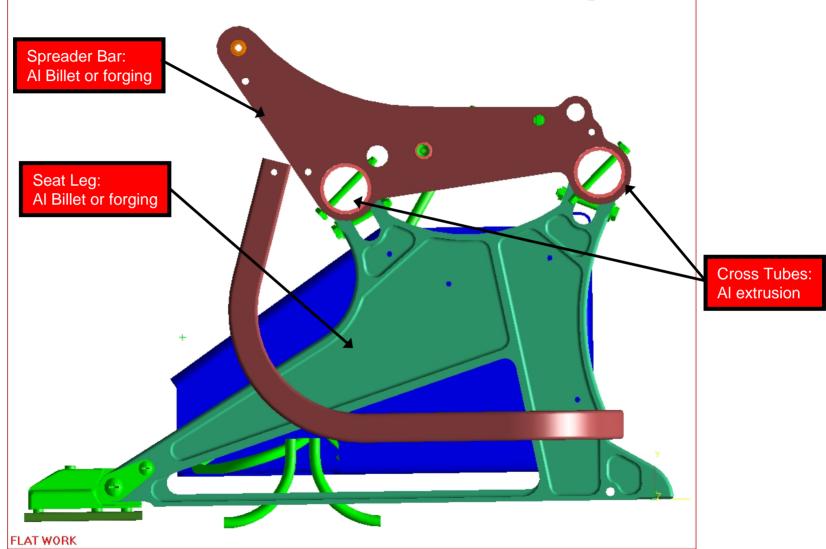


Typical Seat Assembly





Billet and Extruded Seat Components





Initial Planned Testing at FAA Tech Center

Full-Scale Postcrash FireTesting i.e., 3 tests

Baseline using OEM aluminum frames, FB seat cushions Substitute poor-performing mag alloy for aluminum frames Substitute good-performing mag alloy for aluminum frames

Outcomes

Determine if any difference exists between 3 scenarios Determine if difference exists between mag alloys



Interim Task Group Meeting @ FAATC 2/21/08

Attended by representatives from seat manufacturers, airframers, and mag-alloy supplier

Consensus was that full-scale tests are obvious next step

Discussed proposed mock-up seat, advantages, drawbacks, how realistic is it?

Consensus to use actual aircraft seats in full-scale testing, not mock-up

Conduct 4 full-scale tests:

•aluminum baseline test

•poor performing mag-alloy used in primary components

•good performing mag-alloy used in primary components

•good performing mag-alloy used in <u>all</u> components



Interim Task Group Meeting @ FAATC 2/21/08

Additional Points of Discussion

Considering "good" performing mag-alloy, which alloy should be used?

Considering "bad" performing mag-alloy, which alloy should be used?

Interaction of other materials (feedback)?

Effect of fire blocked seats vs. fire hardened foam (unblocked)?

Impact of using water on burning mag-alloy seats? Impact on CFR crews?

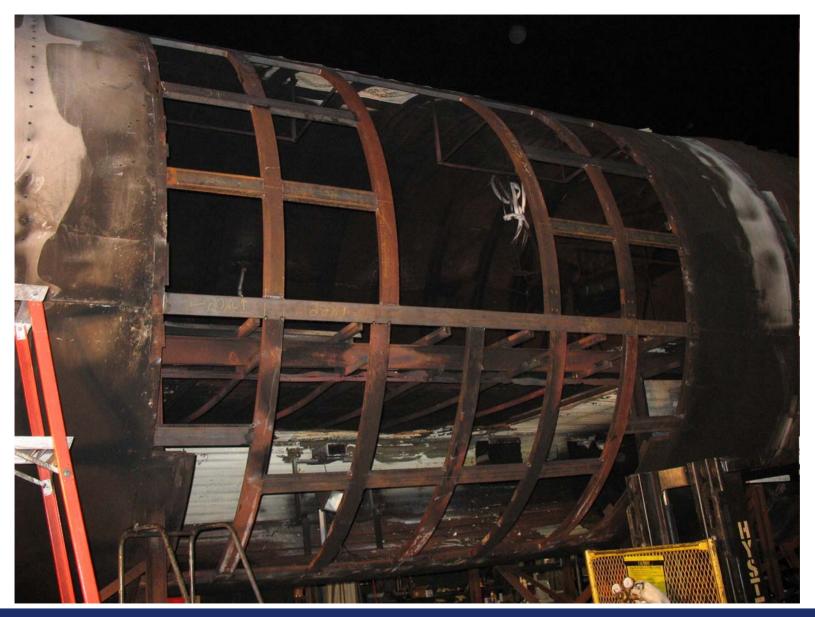
Propose AZ31 for poor performer, WE43 for good performer

Baseline test could be performed by June 2008

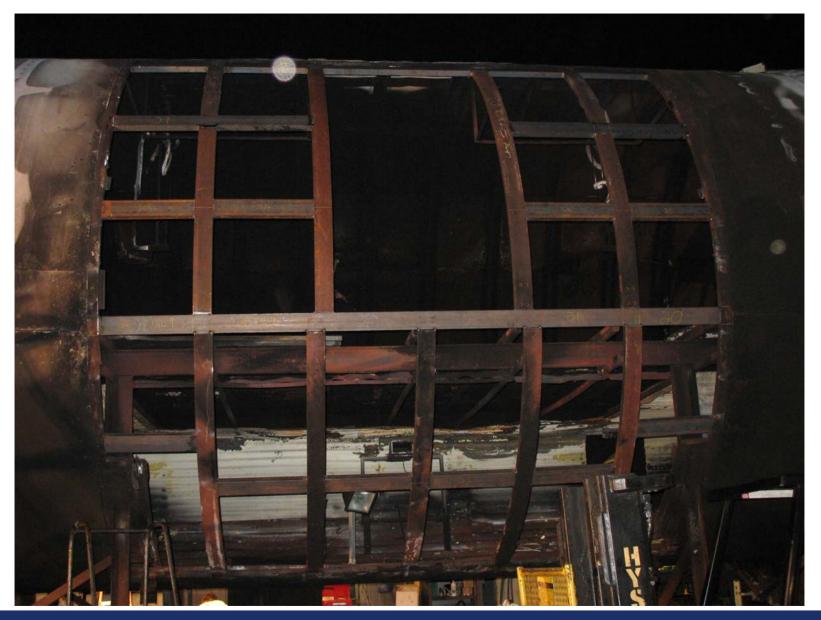


Full Scale Testing Update

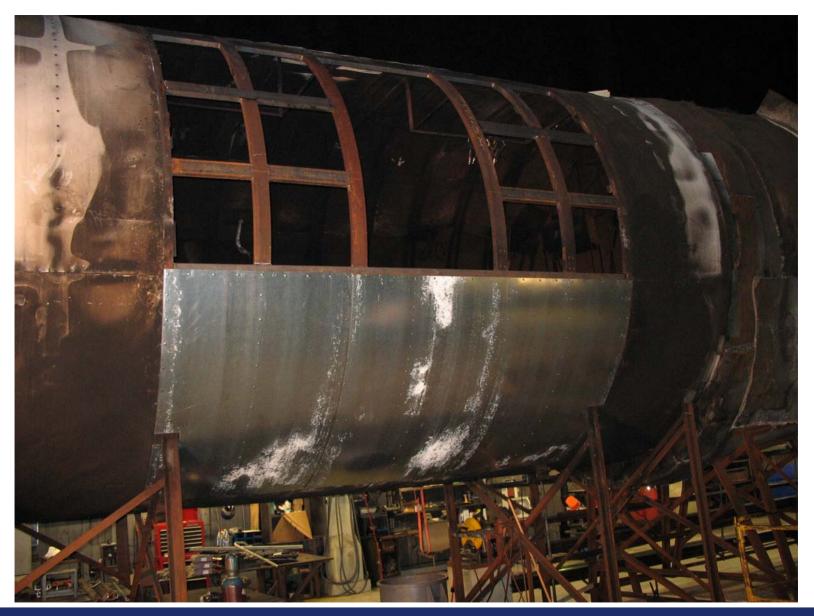


















Previous Test Configuration in 707





Previous Test Configuration in 707





Previous Test Configuration in 707





View Into Fire Door in 707





Full-Scale Test Parameters

Test Article

B707 fuselage, fully fire hardened interior

Instrumentation

Continuous gas analysis at 2 locations, 2 heights + FTIR at locations

Temperature measurement: 3 thermocouple trees + additional seat thermocouples

Smoke measurement: smoke-meters at 2 locations, 3 heights each

4 interior video cameras, 2 external

Interior Materials

0.25-Inch thick crushed-core Nomex honeycomb panels, meets 65/65 Aircraft-grade carpet, meets VBB

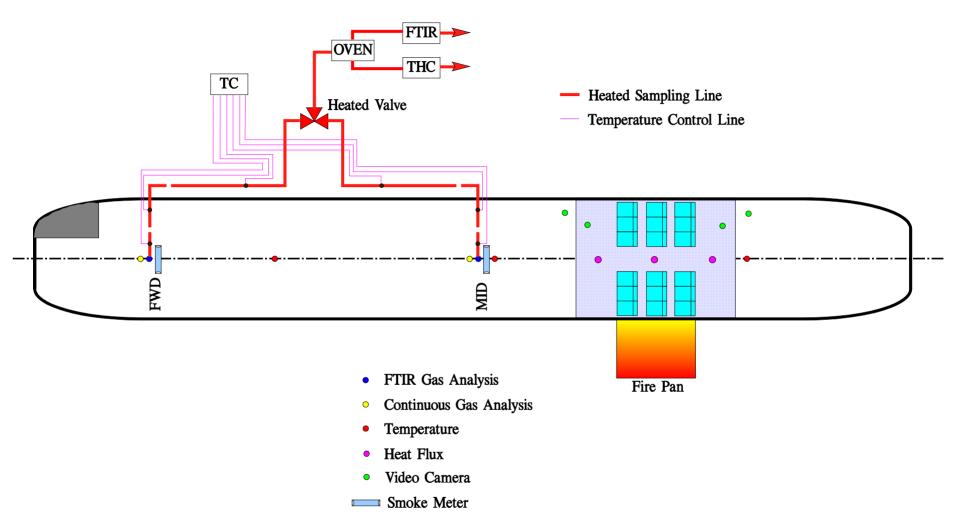
Test Execution

55 gallon JP8 fuel fire in 8' by 10' pan adjacent to fuselage

External fuel fire extinguished following noticeable flashover

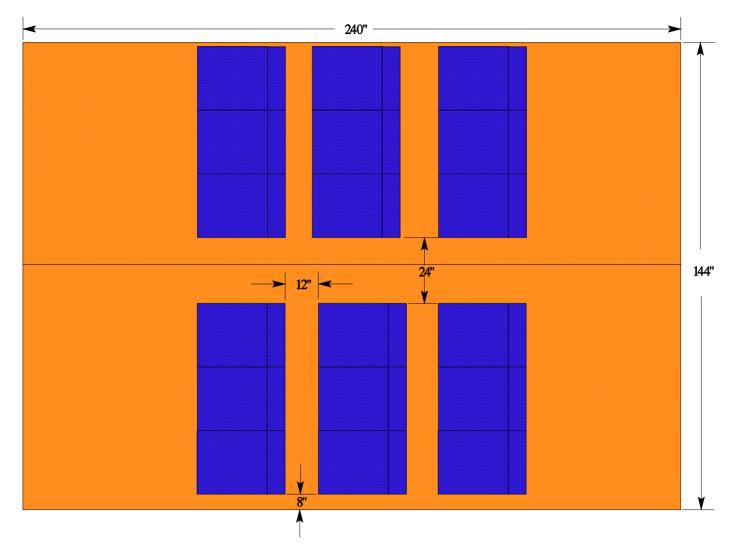


Full-Scale Test Apparatus





Seat Configuration & Location





Procurement of Seats for Full-Scale Testing



























Seat Disassembly



Update on Flammability Testing of Magnesium Alloy Components June 17, 2008



Federal Aviation Administration

Cross Tube Assembly





Spreader Assembly





Leg Assembly





Future Considerations

All full-scale test results would help define an appropriate lab-scale test method or methods, which is the primary goal of the research.

Manufacturer's perspective necessary to determine value of developing new test methodology.

