Vertical Bunsen Burner Test

The Evaluation of the 12-seconds Vertical Bunsen Burner Test Appendix F Part I(a)(4) to Part 25





INTERNATIONAL AIRCRAFT MATERIALS FIRE TEST WORKING GROUP ATLANTA, GEORGIA OCTOBER 19-20, 2005 MEETING WJH FAA Technical Center John W. Reinhardt Fire Safety Section, AAR-422 Atlantic City Int'l Airport, New Jersey 08405

Outline



- Project Objectives
- Brief Overview (London Meeting)
- Answer to Previous TG Questions
- Discuss Tests Result Data
- Discuss Intermediate-Scale Test vs Radiant Panel Test







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Objective



 FAA's goal is to raise the standard for the airplane such that fires in inaccessible areas do not spread and create catastrophic conditions.

 For ducts, the current test does not predict the behavior of the part in actual conditions and therefore suggests the need for a new standard.





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Overview

BRIEF OVERVIEW

During our last meeting in Gatwick, UK (June 28-29, 2005) the following was discussed:

• The fire test results of some (10) aircraft duct materials were presented.

 An approach to a new fire test protocol (based on the current FAA radiant panel test) was presented as a possible means of compliance to current regulations.

First task group meeting was conducted.

 All of the task group members agreed that the 12-seconds vertical Bunsen burner test is not a good discriminator to determine the fireworthiness of a material.

 A couple of members did not agree that the current FAR test protocol should be evaluated under the current venue (recommended use of ARAC Committee).



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PREVIOUS TASK GROUP MEMBERS QUESTIONS

Q1. "But, as an industry task group member, it is not clear to me that there has been enough industry discussion about the need for a new test requirement that takes into account in-service history relative to ducting."

A1: As mentioned in the Objective section, the current FAA's mission is to raise the standard for the airplane such that fires in inaccessible areas do not spread and create catastrophic conditions. During the evaluation of the current test protocol for aircraft ducting, it was determined that this existing test is not up to par with the FAA's mission (Internal Deficiency). The FAA safety strategy today is to be proactive rather than reactive as in the past.











Accident Date: 9/5/96, Location: Newburgh, New York Operator: Federal Express, Cause: Unknown Fire Source in Cargo











Accident Date: 9/4/93, Location: Santo Domingo, Dominican Republic Operator: Dominicana, Cause: Unknown Fire aft of lavatory







PREVIOUS TASK GROUP MEMBERS QUESTIONS (CONT.)

Q2: "Prioritization of this effort relative to other safety improvements should also be discussed, particularly with FAR 25.856(a) [Thermal/Acoustic Insulation Materials] now in effect."

A2: When compared to FAR 25.856(a), this is a low priority project.







PREVIOUS TASK GROUP MEMBERS QUESTIONS (CONT.)

Q3: "Having well defined goals along with the supporting rationale will provide a fundamental basis for the Ducting Task Group activities."

A3: Ducting components are major parts in the aircraft, running in the attic, cheek, cargo compartments, and other areas, that have the probability of becoming a fuel source during a fire. Since this ducting system interconnects along the aircraft, it may provide a propagation path to the fire. To evaluate the propagation characteristic of a ducting material, the FAA requires that it be tested according to Appendix F to Part 25, Part I (a) (ii). After re-evaluating the Appendix F test, commonly known as the 12-seconds Vertical Bunsen Burner, results show that this test does not predict the behavior of the part in actual conditions and therefore suggests the need for a new test standard.







12-sec Vertical Bunsen Burner Test





Intermediate-Scale Test





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12 & 60-Sec Vertical Bunsen Burner Test



OSU Heat Release Test



Intermediate-Scale Test



NBS Smoke Test



Radiant Panel Test



Micro-Scale Combustion Calorimeter Test





TESTS RESULT DATA (CONT.)

• About 33 different types of specimens were donated by the working group members and tested by the FAATC (included different materials, shapes, configuration, thickness)

- Rigid Ducts
- Flexible Ducts
- Thermoplastic & Thermosetting Composites
- Old and New Materials Used for Aircraft Ducting
- More than 550 tests were conducted





12 & 60 SECONDS VERTICAL BUNSEN BURNER TESTS:

Test Protocol: Chapter 1 of DOT/FAA/CT-89/15 Aircraft Material Fire Test Handbook

Sample Size: 75mm x 305mm

Heat Source: Methane Flame (41 kW/m², 925 °C)

Heat Source Exposure: 12 (60) seconds

Flame Extinguishing Time: <15 seconds

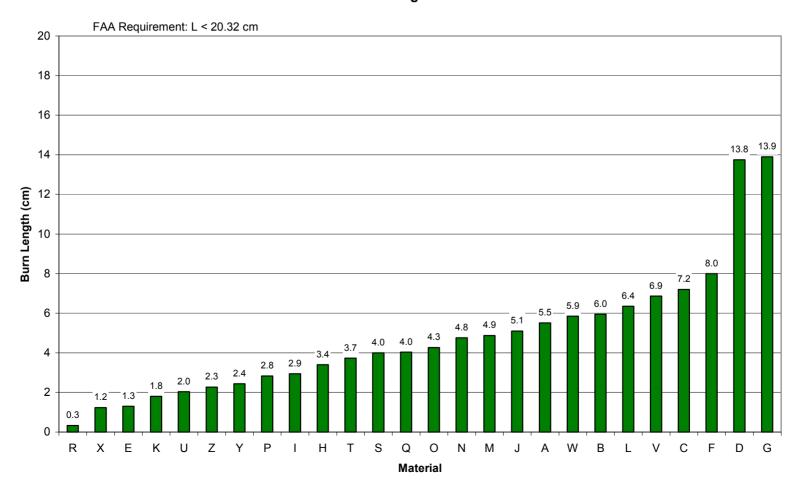
Burn Length: <20.32cm (<15 cm)

Drip Extinguishing Time: <5 seconds

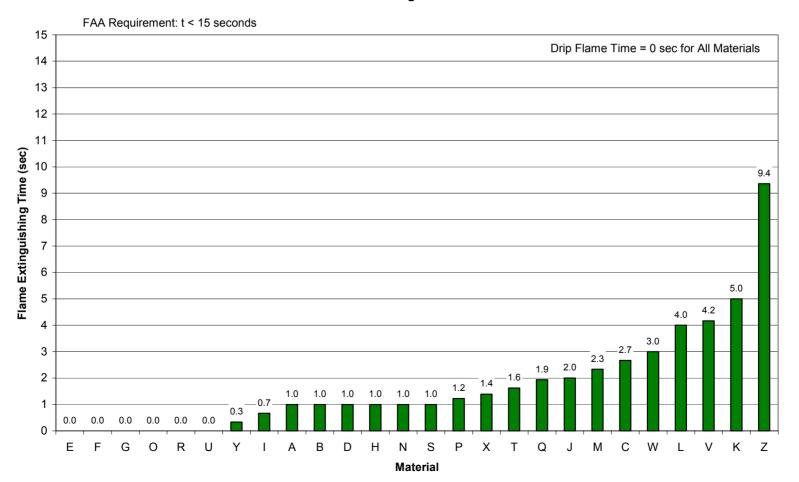






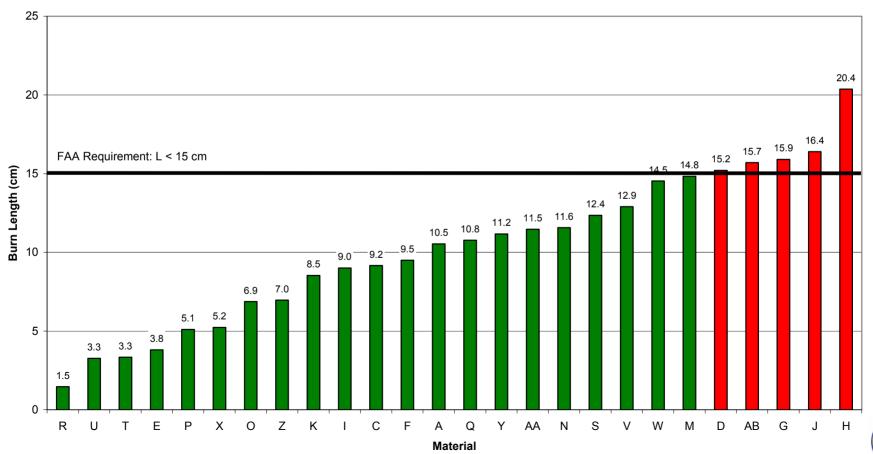






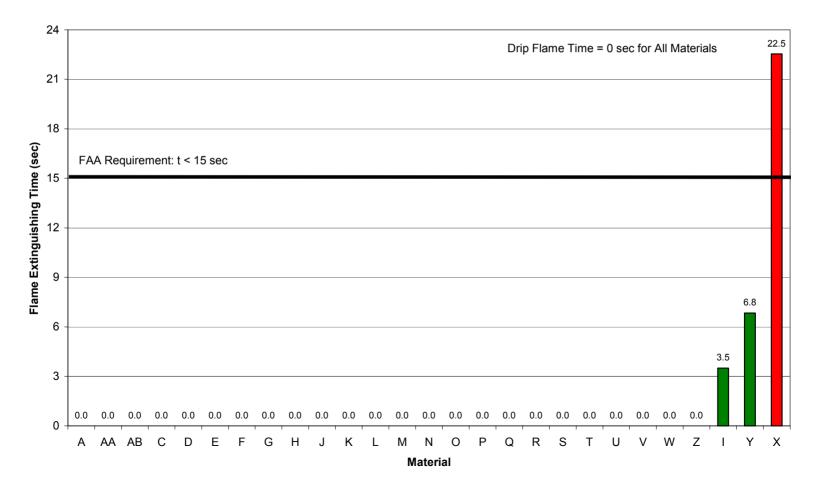
















INTERMEDIATE-SCALE FIRE TEST:

Test Protocol: FAA Report DOT/FAA/AR-99/44 - Dev. Of Improved Flammability Criteria for Aircraft Thermal Acoustic Insulation

Sample Size: 15.2cm (30.4cm) x 243.8cm

Heat Source: Polyurethane Foam Block (49 kW/m², 781 °C)

Heat Source Exposure: ~6 minutes

Not a compliance test



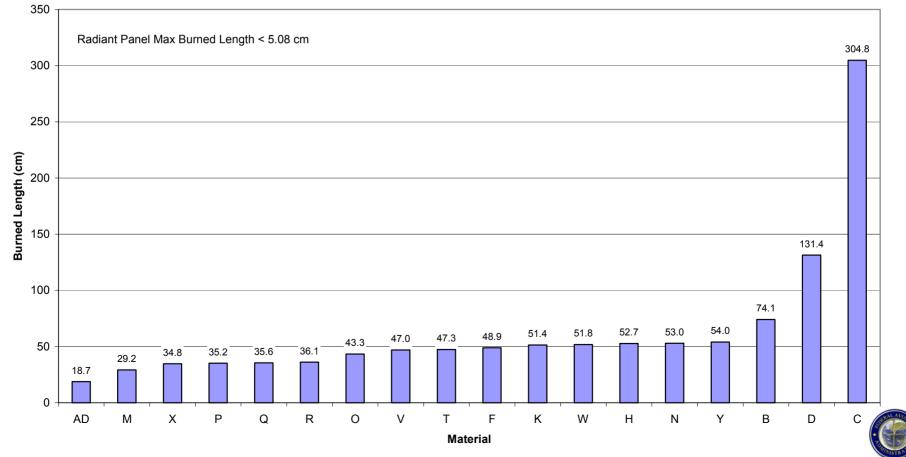






INTERMEDIATE-SCALE FIRE TEST Aircraft Ducting Materials / Narrow-Body Configuration

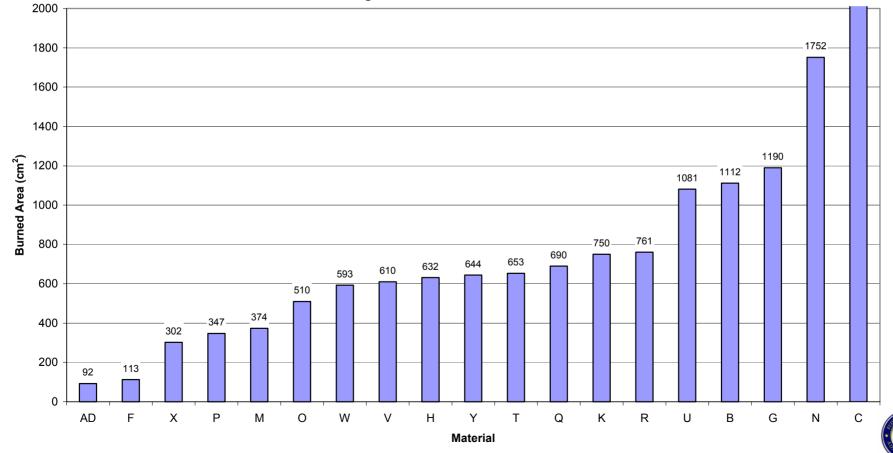
Ignition Source Outside





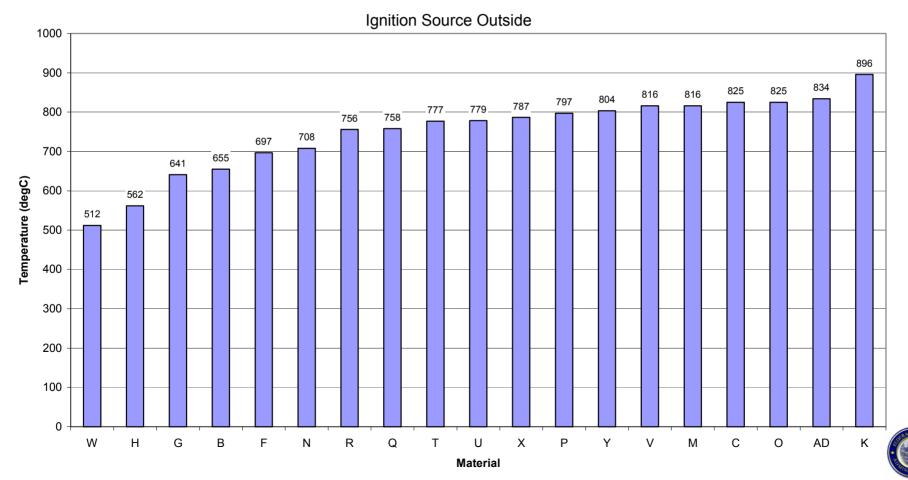
INTERMEDIATE-SCALE FIRE TEST Aircraft Ducting Materials / Narrow-Body Configuration

Ignition Source Outside



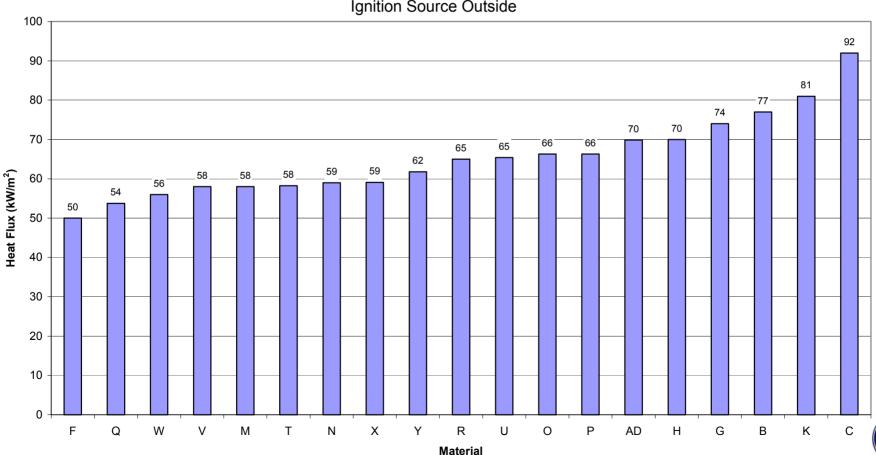


INTERMEDIATE-SCALE FIRE TEST Aircraft Ducting Materials / Narrow-Body Configuration





INTERMEDIATE-SCALE FIRE TEST Aircraft Ducting Materials / Narrow-Body Configuration



Ignition Source Outside

Narrow-Body Configuration: Ignition Source Outside



Large Duct IST Test







Narrow-Body Configuration: Ignition Source Inside



Narrow-Body Configuration: Ignition Inside



Wide-Body Configuration: Ignition Outside





Narrow-Body Configuration: Ignition Inside



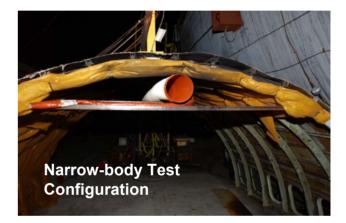
Wide-Body Configuration: Ignition In-Outside





INTERMEDIATE-SCALE TEST (CONT.)

- The narrow-body test configuration was more severe than the wide-body test configuration (heat confined)
- The narrow-body configuration with the ignition source inside the duct was more severe that with the ignition source next to the duct outside (more heat from conduction, convection and radiation sources).









RADIANT PANEL TEST:

Test Protocol: Appendix F to Part 25 (Part IV) – Test Method To Determine the Flammability and Flame Propagation Characteristics of Thermal/Acoustic Insulation Materials

Sample Size: 318mm x 584mm

Heat Source: Propane Flame & Radiant Heating Coils (17 kW/m² panel, 147 kW/m² pilot)

Heat Source Exposure: 15 seconds (pilot) and until flames are extinguished (radiant panel)

Max Flame Propagation: < 5.08 cm

Max Flame Time: < 3 seconds

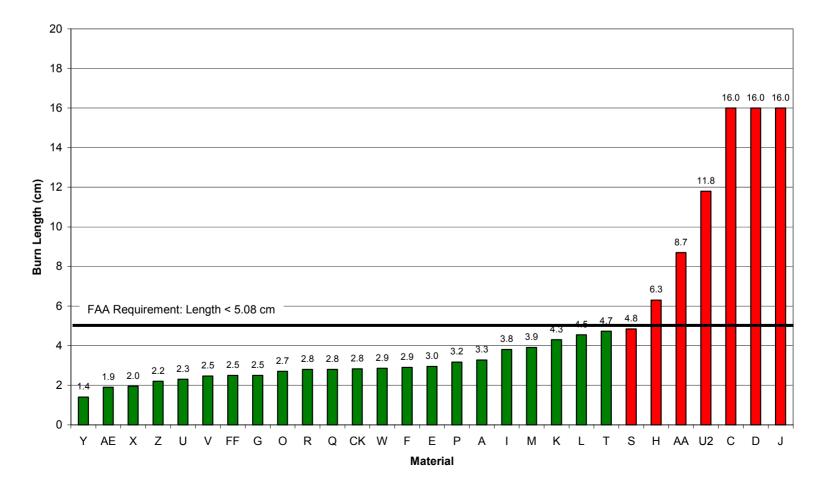








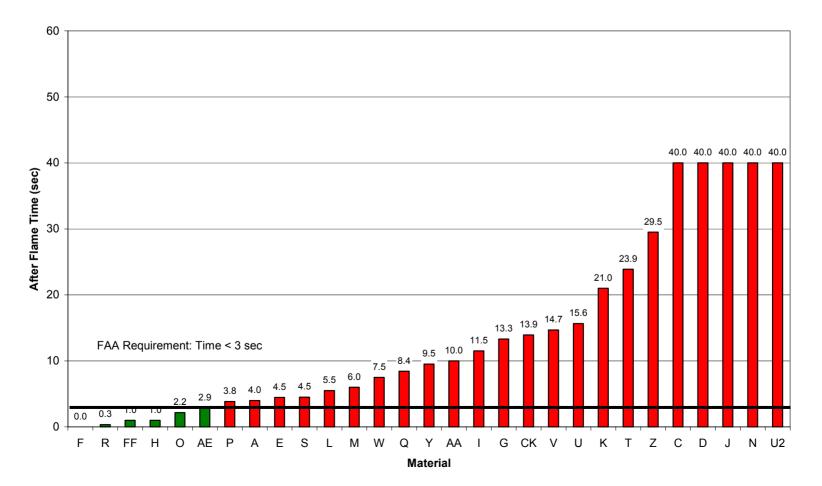
RADIANT PANEL TEST RESULTS Aircraft Ducting Materials







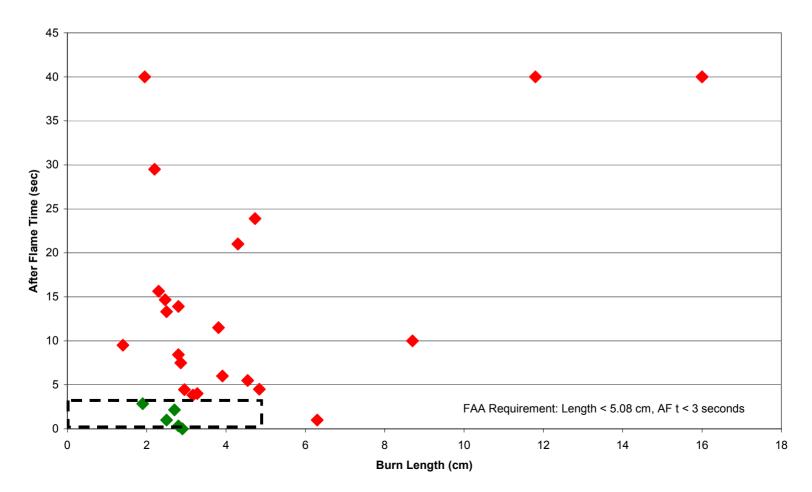
RADIANT PANEL TEST RESULTS Aircraft Ducting Materials





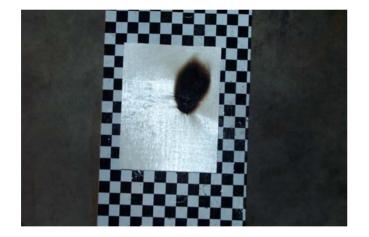


RADIANT PANEL TEST RESULTS Aircraft Ducting Materials









Passed Test - Sample O: Burn Length = 2.7 cm After Flame Time = 2.2 sec



Failed Test - Sample J: Burn Length = +16 cm After Flame Time = +40 sec





FIRE PROTECTION COATING ON SAMPLE K



Rample Ck

Failed Test, Sample K: Burn Length = 4.3 cm After Flame Time = 21 sec

Failed test, but improved Sample K: Burn Length = 2.8 cm After Flame Time = 13.9 sec





RADIANT PANEL EXPERIMENTAL TEST:

Test Protocol: Same

Sample Size: Same

Heat Source: Same

Heat Source Exposure: 1 minute exposed to radiant heat, 10 seconds pilot flame impingement

Max Flame Propagation: < 5.08 cm

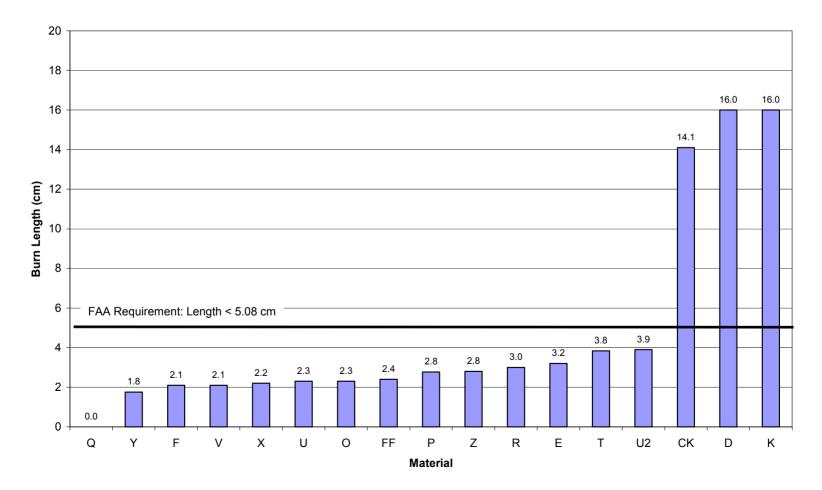
Max Flame Time: < TBD







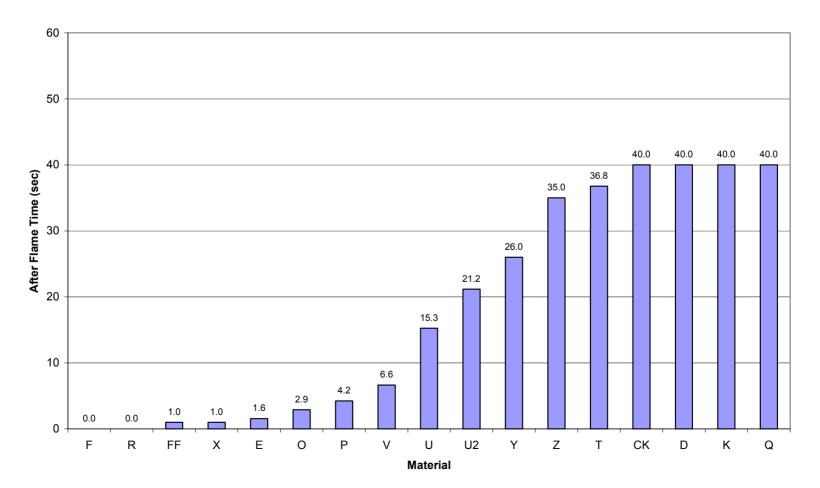
EXPERIMENTAL RADIANT PANEL TEST RESULTS Aircraft Ducting Materials





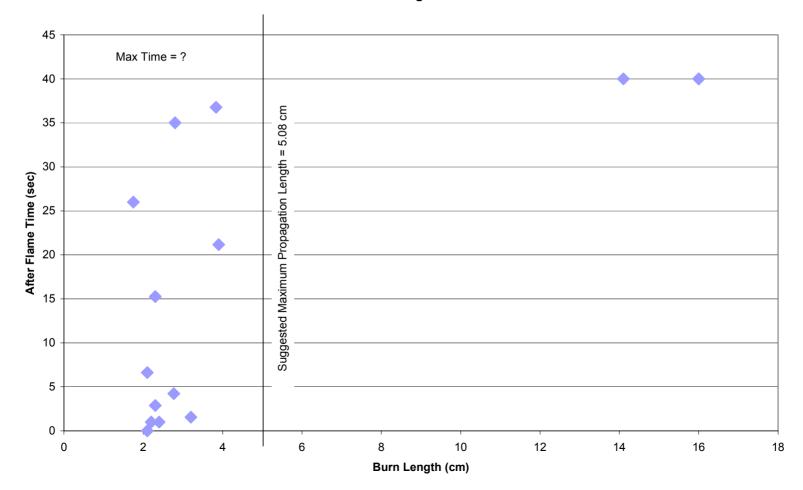


RADIANT PANEL TEST RESULTS Aircraft Ducting Materials





EXPERIMENTAL RADIANT PANEL TEST RESULTS Aircraft Ducting Materials

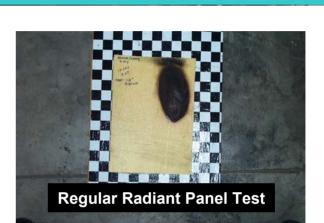


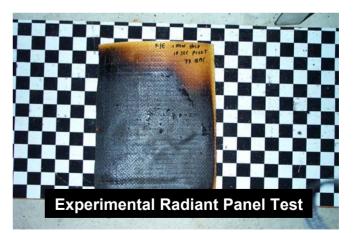






Sample O: Burn Length = 2.3 cm After Flame Time = 2.9 sec





Sample K: Burn Length = +16 cm After Flame Time = +40 sec





HEAT RELEASE RATE TEST FOR CABIN MATERIALS:

Test Protocol: Chapter 5 of DOT/FAA/CT-89/15 Aircraft Material Fire Test Handbook

Sample Size: 150mm x 150mm

Heat Source: Methane Flame & Radiant Heating Coils (35 kW/m²)

Heat Source Exposure: 5 minutes

Max Avg. Heat Release Rate: <65 kW/m²

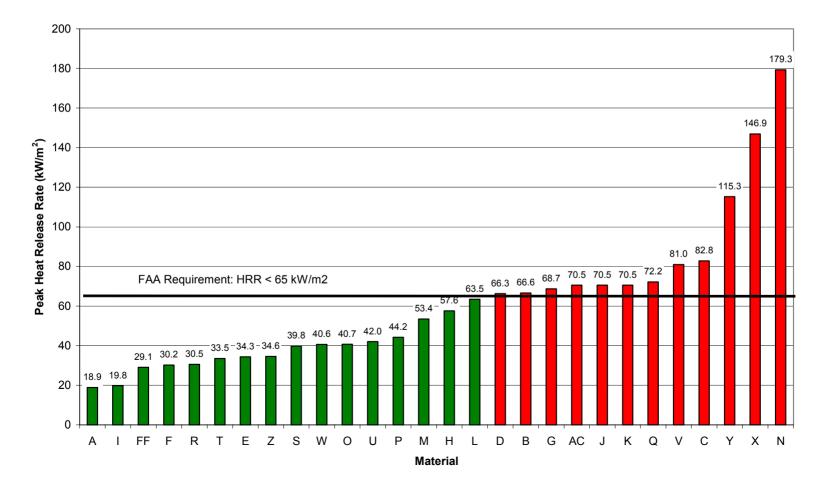
Max Avg. Total Heat Released (2 min): <65 kW*min/ m²







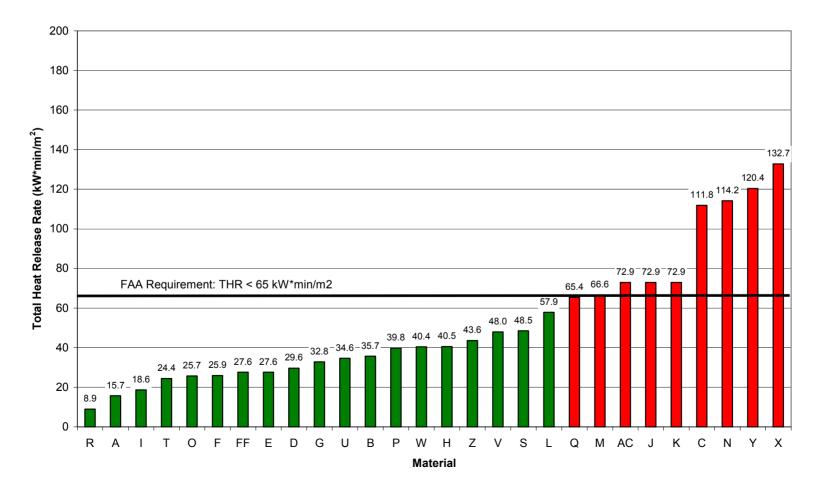
FAA/OSU HEAT RELEASE TEST RESULTS Aircraft Ducting Materials







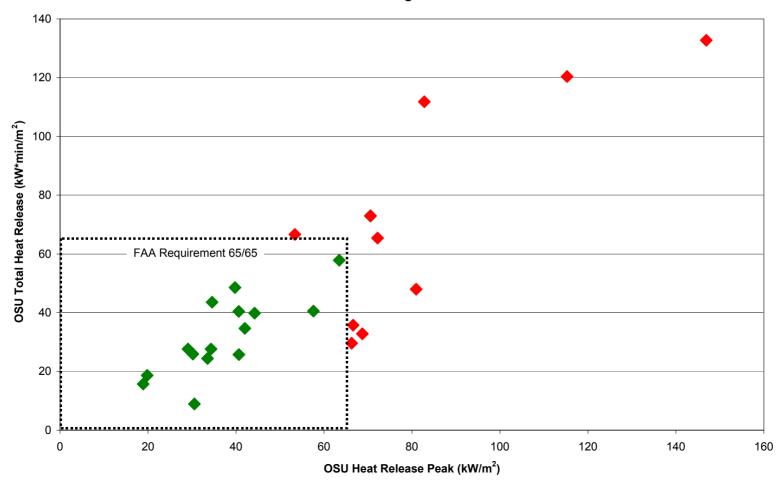
FAA/OSU HEAT RELEASE TEST RESULTS Aircraft Ducting Materials







OSU HEAT RELEASE TEST Aircraft Ducting Materials





SMOKE TEST FOR CABIN MATERIALS:

Test Protocol: Chapter 6 of DOT/FAA/CT-89/15 Aircraft Material Fire Test Handbook

Sample Size: 73mm x 73mm

Heat Source: Methane Flame & Radiant Heating Coils (25 kW/m²)

Heat Source Exposure: 4 minutes

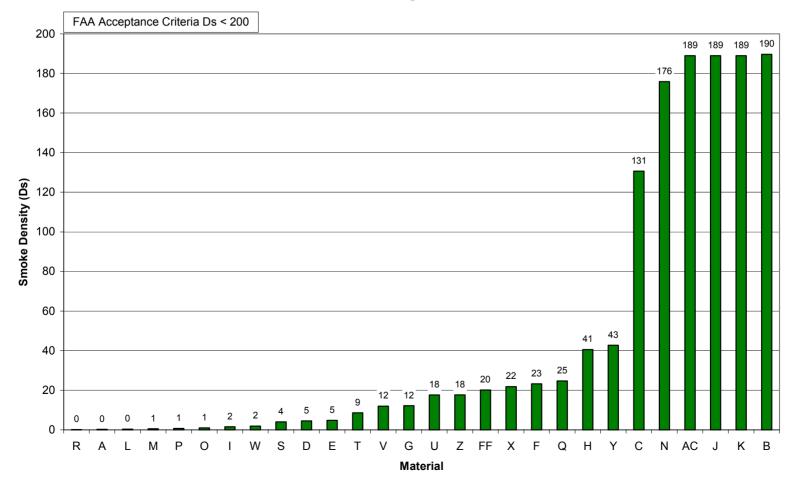
Max Avg. Specific Optical Density, Dm: <200







FAA/NBS SMOKE TEST RESULTS Aircraft Ducting Materials







MICRO-SCALE COMBUSTION CALORIMETER TEST:

Test Protocol: FAA Report DOT/FAA/AR-01/117 A Micro-scale Combustion Calorimeter

Sample Size: milligram range

Heat Source: Heating Coils (900 °C)

Heat Source Exposure: 10 to 120 seconds to effect pyrolysis

Not a compliance test

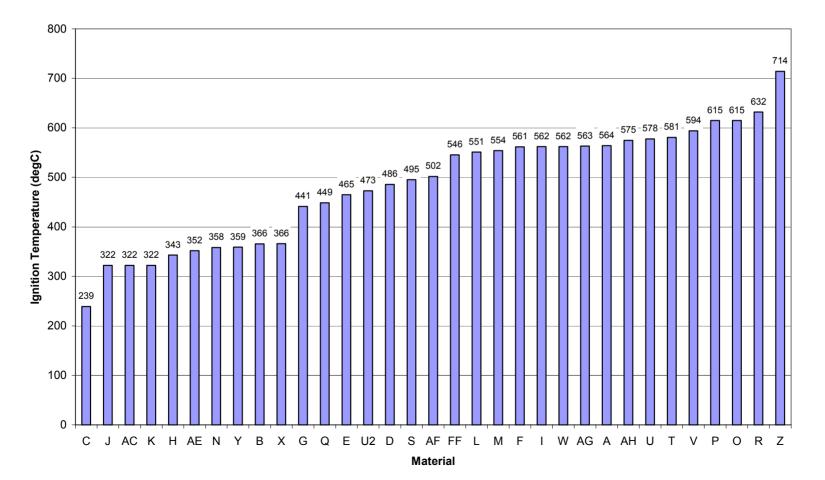








MICRO-SCALE COMBUSTION CALORIMETER TEST RESULTS Aircraft Ducting Materials

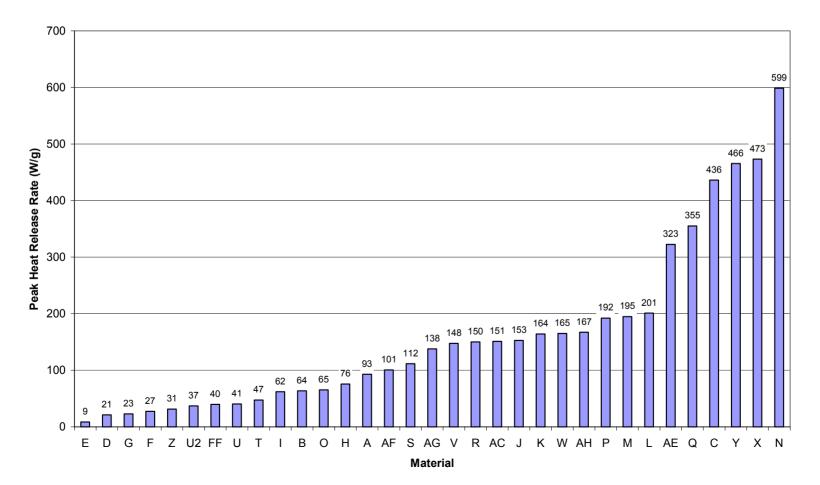








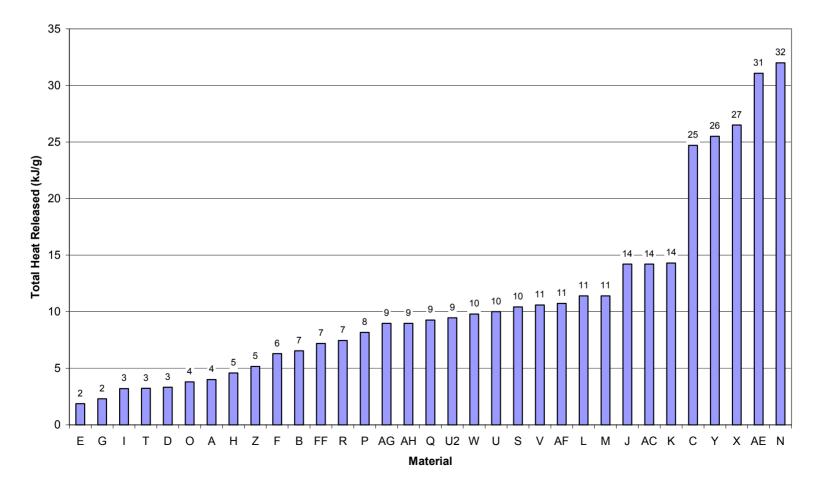
MICRO-SCALE COMBUSTION CALORIMETER TEST RESULTS Aircraft Ducting Materials







MICRO-SCALE COMBUSTION CALORIMETER TEST RESULTS Aircraft Ducting Materials







Therefore, a good fire worthy material is one that has a:

- High Ignition Temperature (at least higher than the expected fire threat)
- Low Total Heat Release
- Low Heat Release Rate
- Low Smoke Emission
- Low Propagation Capacity







Material Fireworthiness Ranking:

Best

Worst

12 VBB Material	Final Rank
R	1
U	2
Х	3
Y	3
Н	4
Р	4
0	5
N	6
Т	6
К	7
Q	8
B/AB	9
F	10
G	11
М	12
W	13
С	14
V	15

MATERIAL	FINAL RANK
R	1
0	2 3
Р	3
W	4
F	5
Т	6 7
М	
U	8
U G V	9
	10
Н	11
H Q Y	12
	13
Х	14
B C	15
С	16
K	17
N	18

* Ranking based on FAA's OSU heat release, NBS smoke, and Radiant Panel propagation tests.



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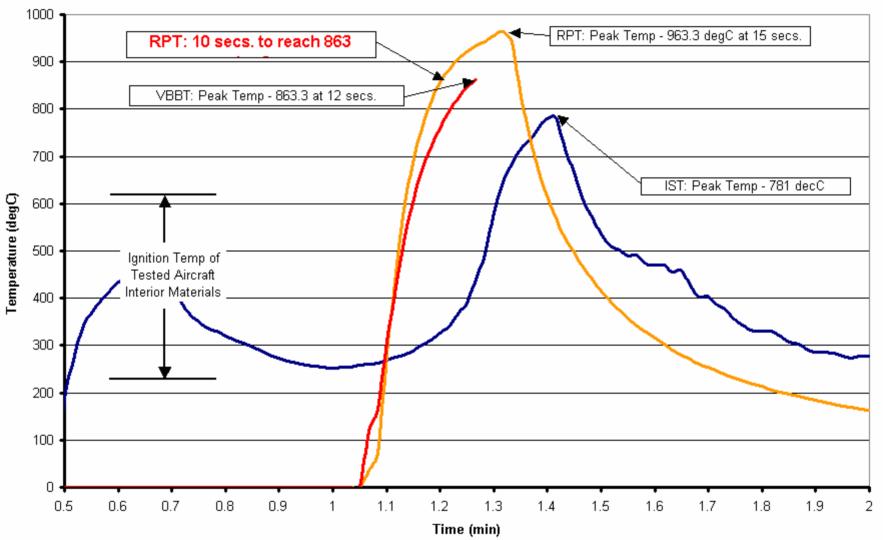
INTERMEDIATE-SCALE TEST VS RADIANT PANEL TEST





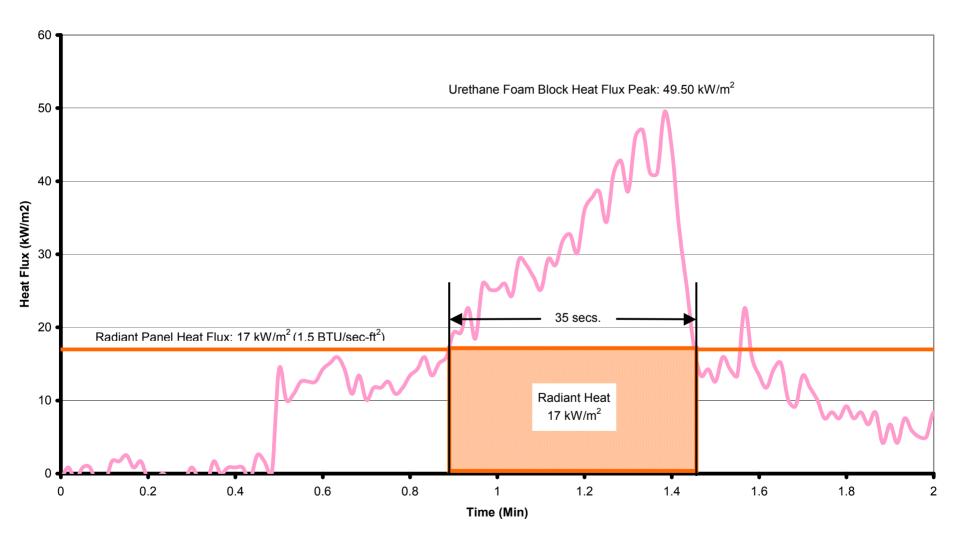


Ignition Sources Temperature Profile Comparison

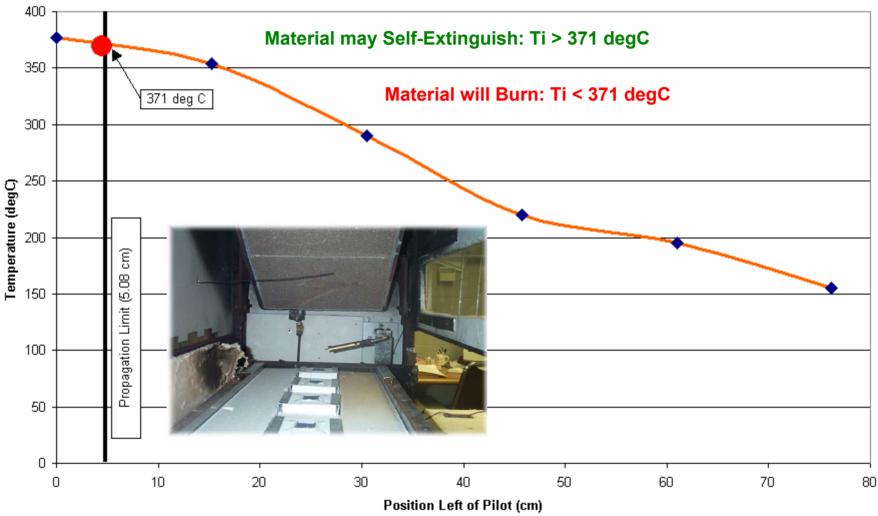




Urethane Foam Block (Igniter) Heat Flux Profile During Intermediate-Scale Test



Radiant Panel Temperature Profile on Tray (Gas System with Pilot Off)







RANKING COMPARISON

IST Material	Final Rank
F	1
W	2
Q	3
Х	3
М	4
Н	5
Т	5
V	5
Р	6
N	7
R	7
Y	8
AD*	9
0	10
G	11
U	11
B/AB	12
K	13
С	14

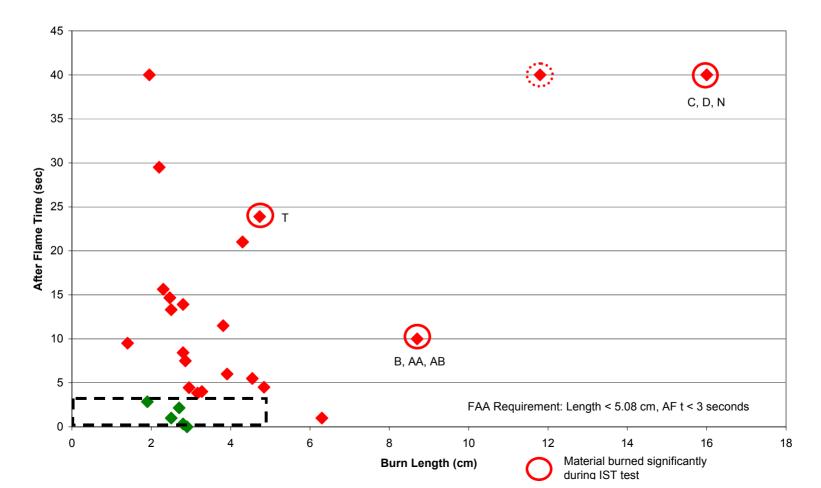
RP	FINAL RANK
F	1
R	1
0	2
Р	2
Y	2 2 2 3 3
Q W	3
W	3
G	4
Н	4
М	4
U	4
V	4
Х	5
B/AA/AB	6
K	7
Т	7
С	8
N	8

EX RP	FINAL RANK
F	1
0	2
R	2
R V	2 2 2 2 3 4 4 5
X Y	2
Y	3
Р	4
Q	4
U	5
Т	6
С	7
K	7
N	7

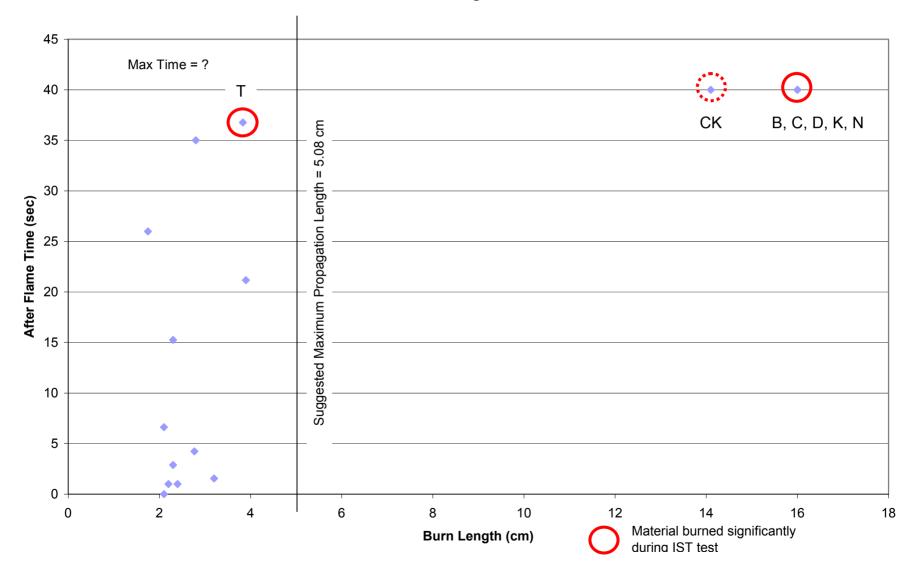




RADIANT PANEL TEST RESULTS Aircraft Ducting Materials



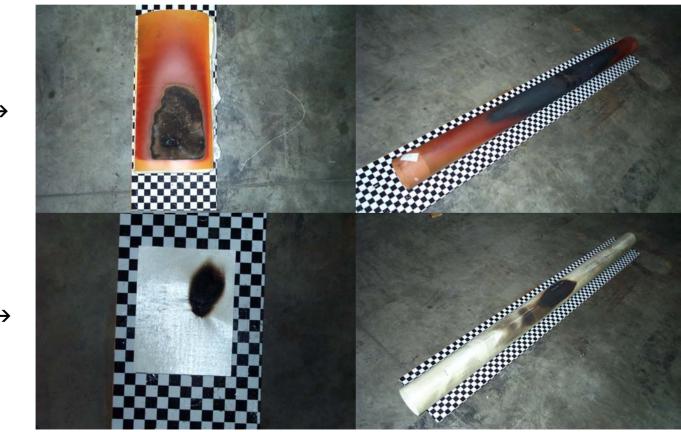
EXPERIMENTAL RADIANT PANEL TEST RESULTS Aircraft Ducting Materials





Radiant Panel Test

Intermediate-Scale Test (Fire Outside)



Sample D \rightarrow

Sample O \rightarrow





Exp Radiant Panel Test

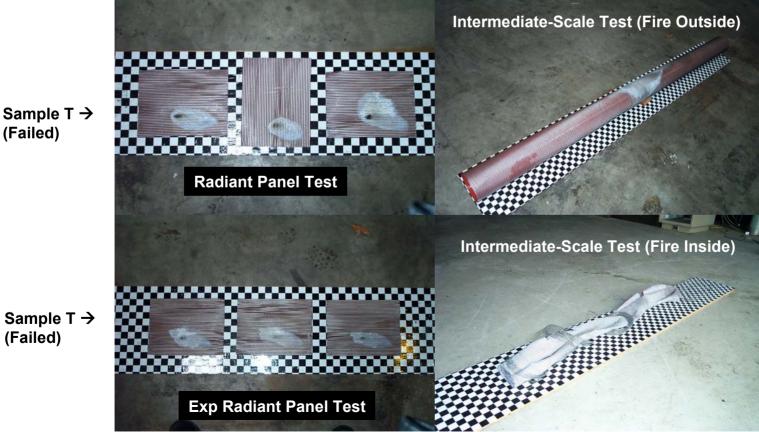
Intermediate-Scale Test (Fire Inside)



Sample D \rightarrow (Failed)

Sample O \rightarrow (Passed)





Sample T \rightarrow (Failed)

(Failed)





What's Next?:

- Test ducting union and other duct component materials (tapes, silicones, bellows)
- Test fire retardant coatings
- Test new Kaowool board
- Complete testing by 2Q 2006
- Initiate AMOC Draft (By 4Q 2006)



Acknowledgement





degussa.

creating essentials





