

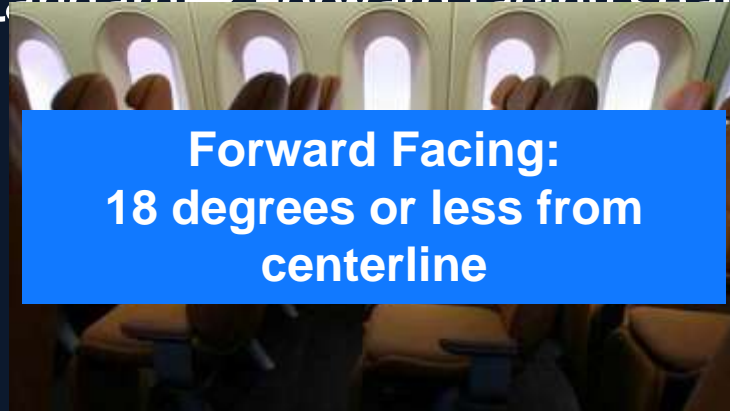
Occupant Response in Oblique Aircraft Seat Environment



John R. Humm, Narayan Yoganandan, Frank A. Pintar, Brian Peterson
David Moorcroft, Amanda M. Taylor, Richard DeWeese

AVIATION OCCUPANT PROTECTION

- Regulated by Federal Aviation Administration (FAA)
- Static Test → Seat strength
- Dynamic Test → Assess occupant injury
- Original standard → Forward facing seats



**Forward Facing:
18 degrees or less from
centerline**

OBLIQUE SEATS

- Motivation: Increase seating density/maximize comfort
- What type of injuries can we expect?
- Injury Criteria/Suitable ATD



**Equivalent Safety
Standard as Forward
Facing Seats**



AIRCRAFT ENVIRONMENT



© PA Archiv

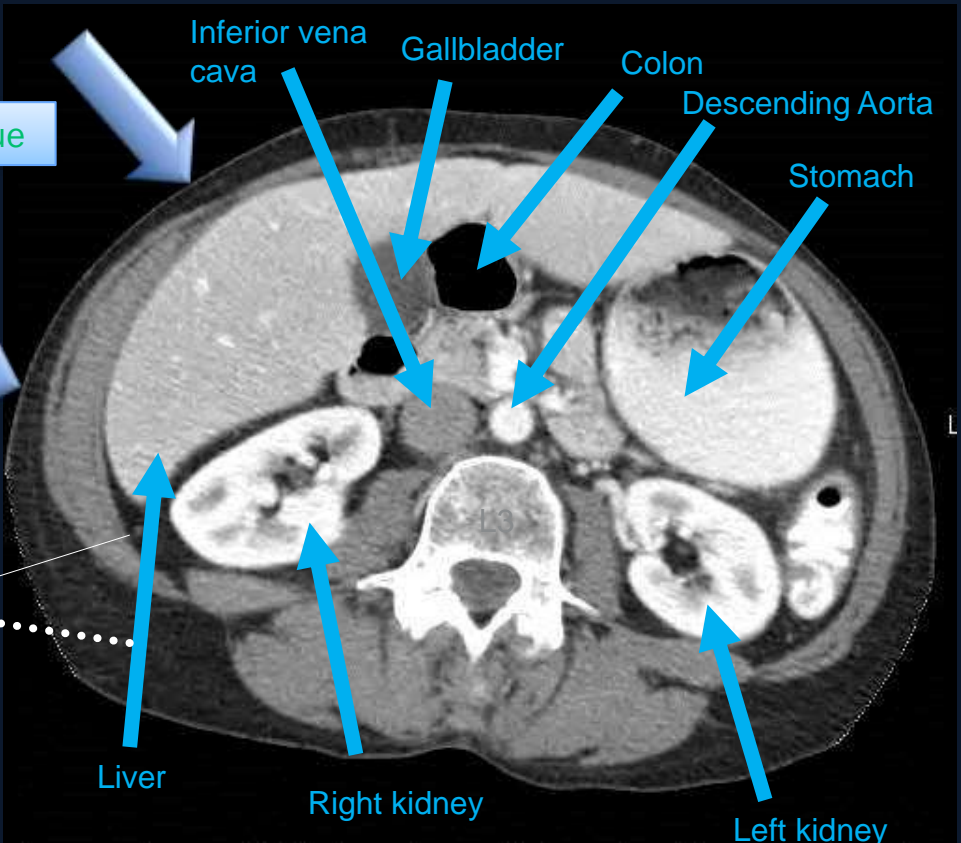


LATERAL V. OBLIQUE



Oblique

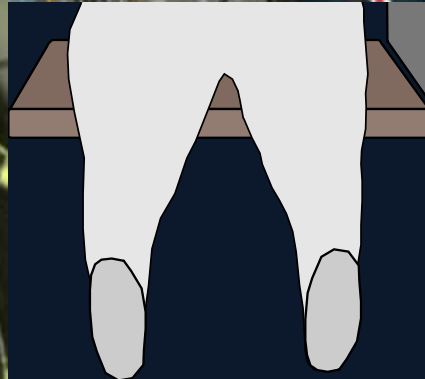
Lateral



OBLIQUE IMPACT BIOMECHANICS



Intruding vehicle structure



CAMI SLED TESTS



**Pelvis and legs
restrained but not**

restrained plus

CAMI SLED TESTS



INJURY CRITERIA

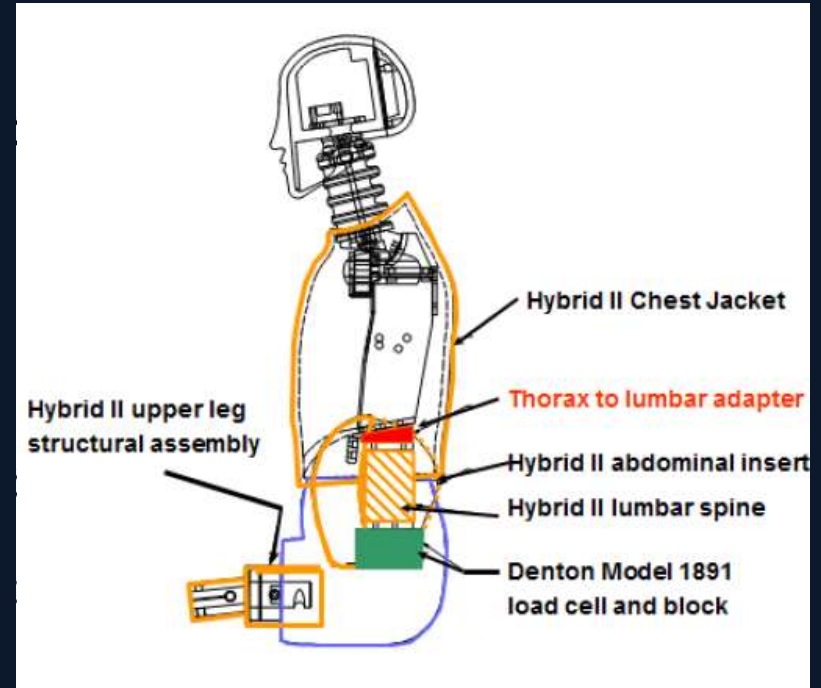
- Post Mortem Human Surrogate (PMHS) to define human response
- Same environment/loading conditions
- “Matched-paired” test w/ATD
- Injury Assessment Reference Values (IARV)



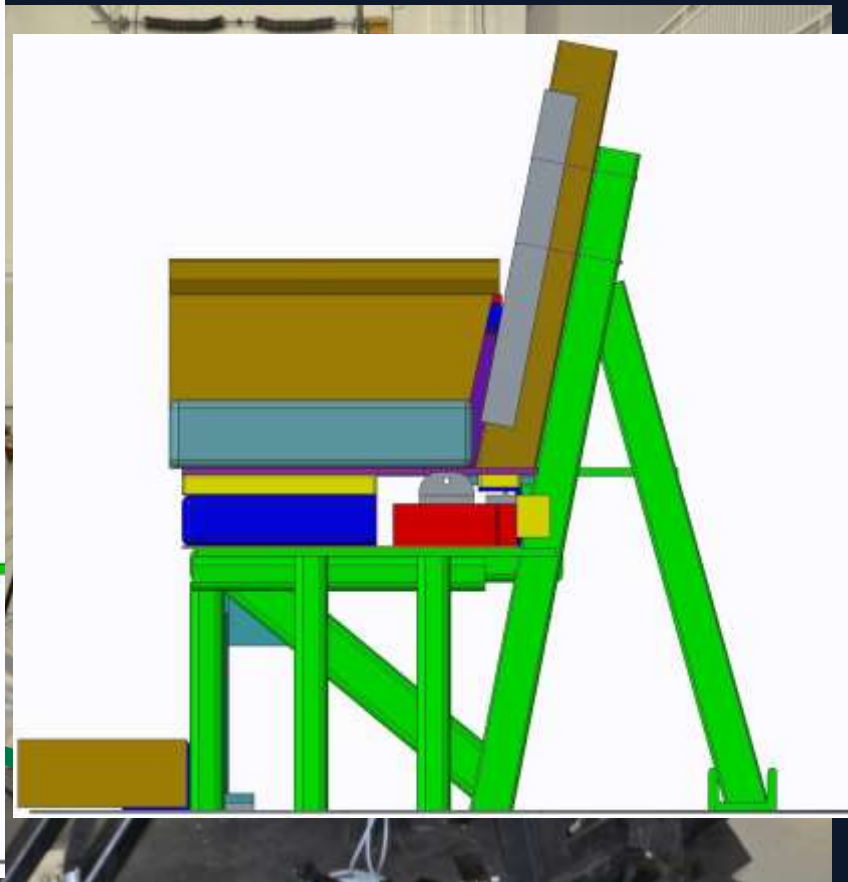
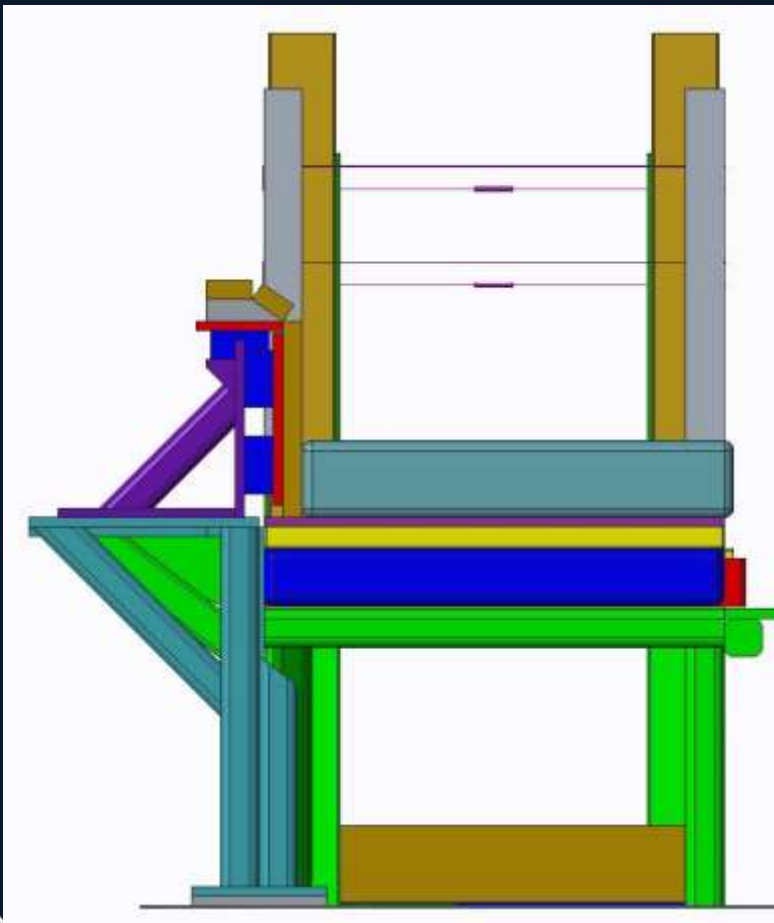
FAA-HYBRID III

FAA-Hybrid III is predominately made up of Hybrid III parts except:

- Hybrid II lumbar spine
- Hybrid II abdominal insert
- Hybrid II chest jacket
- Hybrid II upper leg bone
- Hybrid II lumbar load cell and
- Pelvic adaptor block
- Custom thorax/lumbar adaptor →
 - Thoracic load cell



SLED BUCK – AIRCRAFT INTERIOR



BOUNDARY CONDITION #1



1



Acceleration

BOUNDARY CONDITION #2



2



Acceleration

BOUNDARY CONDITION #3












3

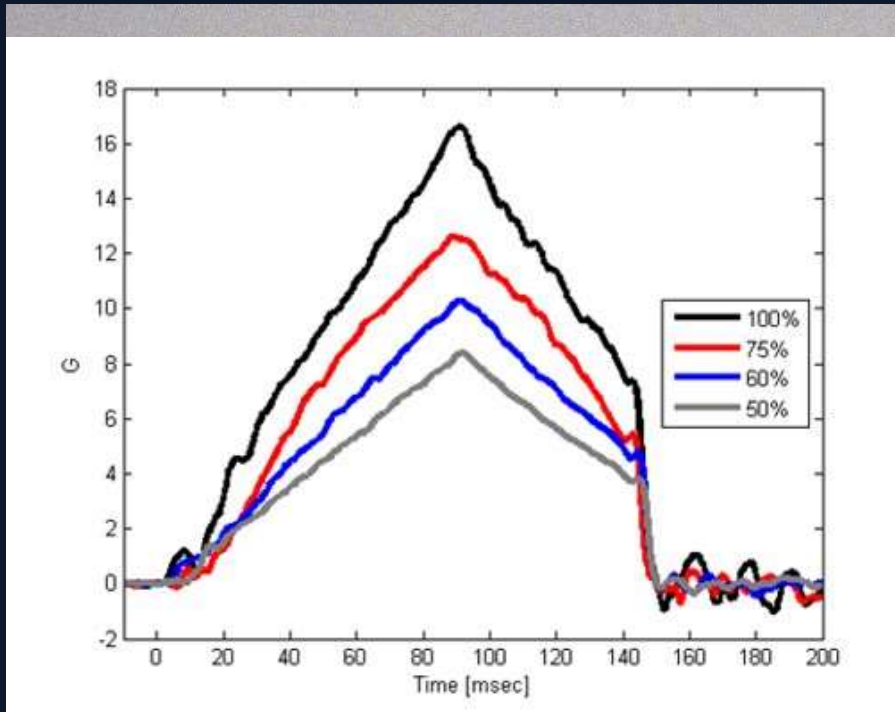


Acceleration

SUMMARY: TEST CONDITIONS

Condition				3
Angle	1	2	3	30
Belt				2 lap belts
Leg				Free
Loadwall	Acceleration	Acceleration	Acceleration	None

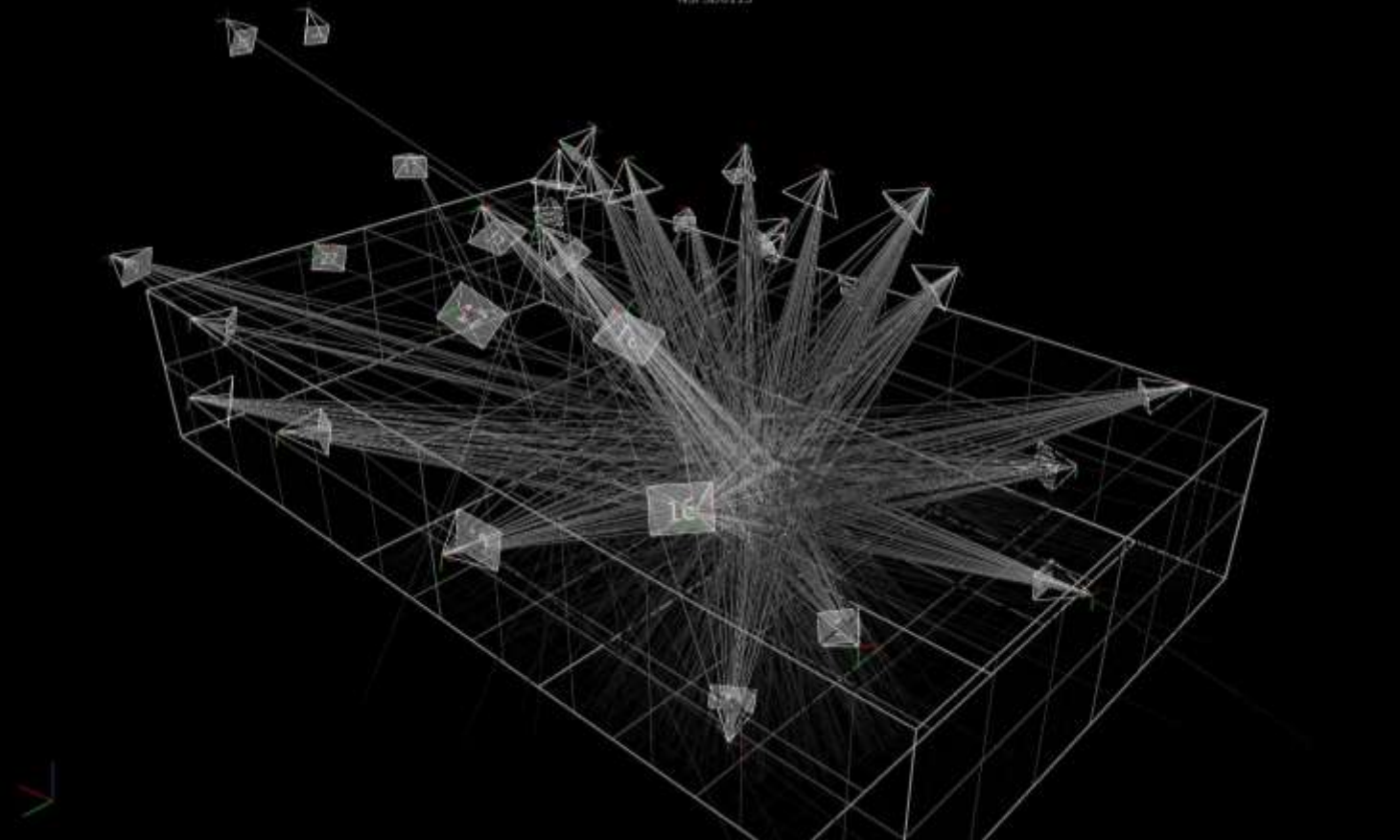
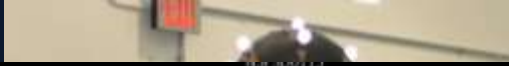
ACCELERATION PULSE



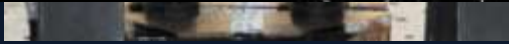
- Scaled pulse – magnitude only
- 100, 75, 60, 50 %
- 4 pulses @ 3 Conditions



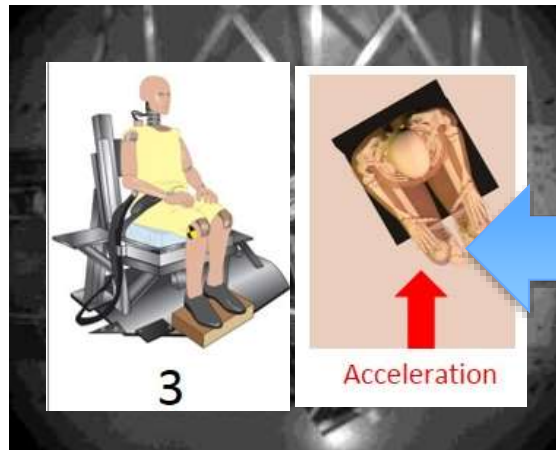
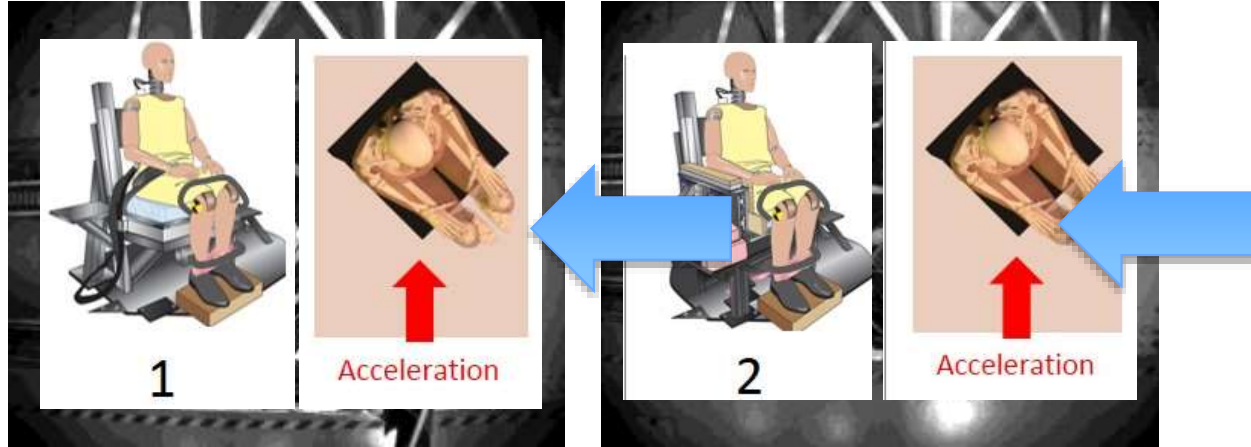
INSTRUMENTATION



Th
Lu



OVERHEAD: TIME 0 MSEC

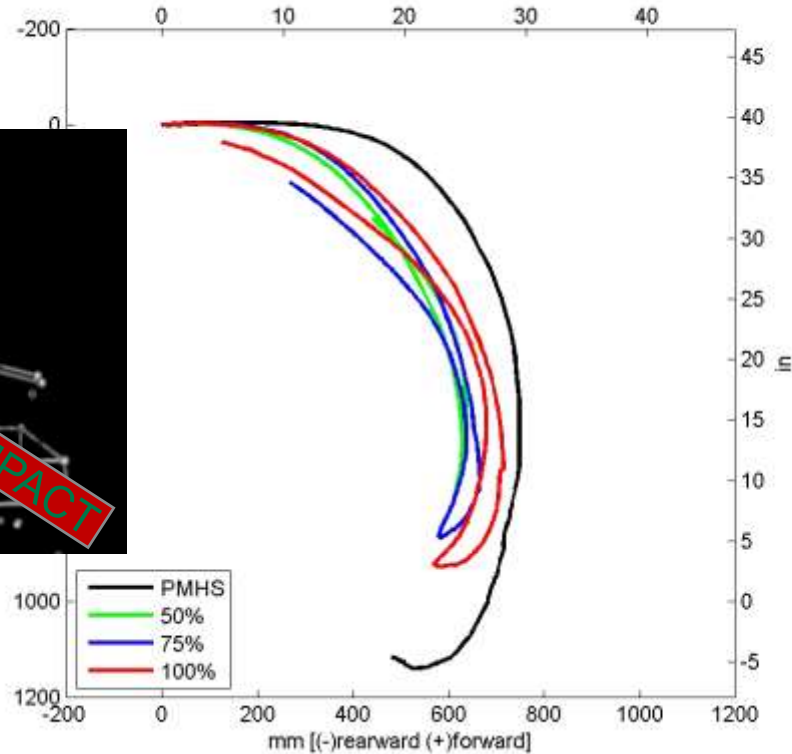
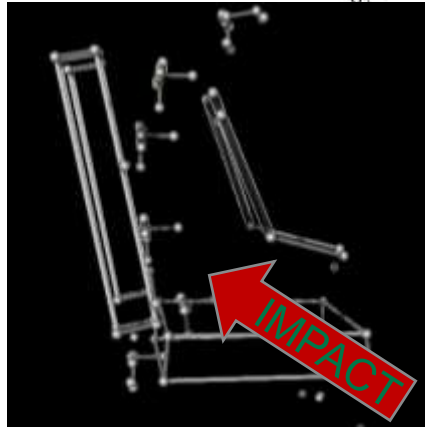


OVERHEAD: TIME 125 MSEC

OVERHEAD: TIME 225 MSEC

LATERAL: TIME 175 MSEC

CONDITION 1-HEAD WRT SEAT - SAGITTAL PLANE

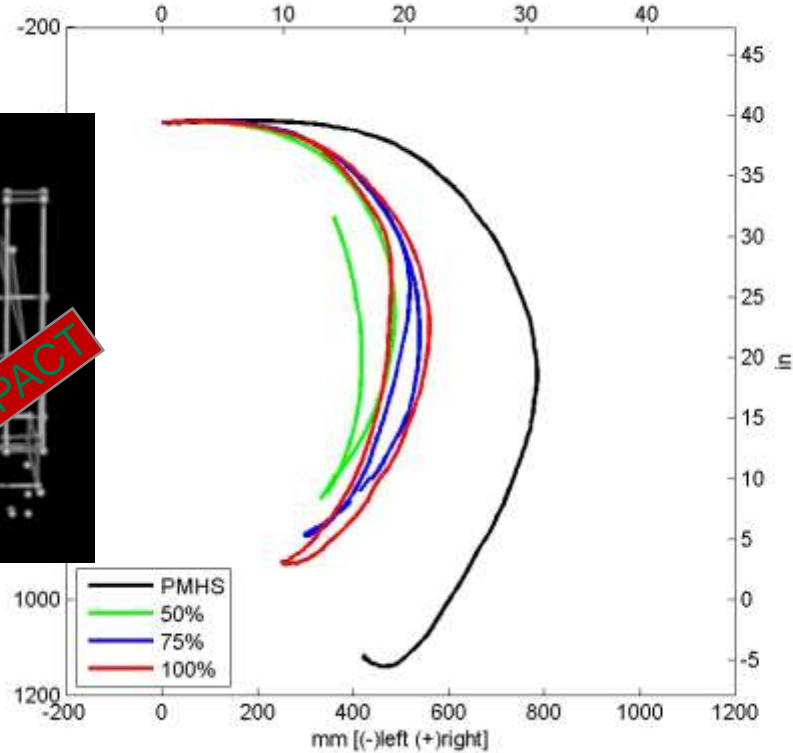
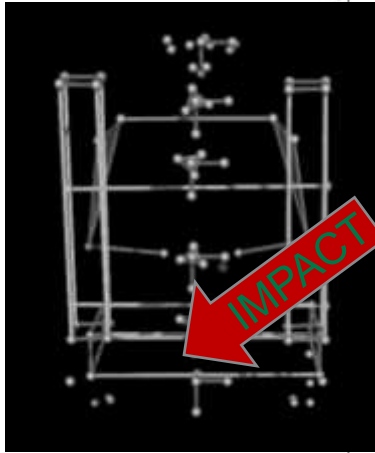


1



Acceleration

CONDITION 1-HEAD WRT SEAT - CORONAL PLANE

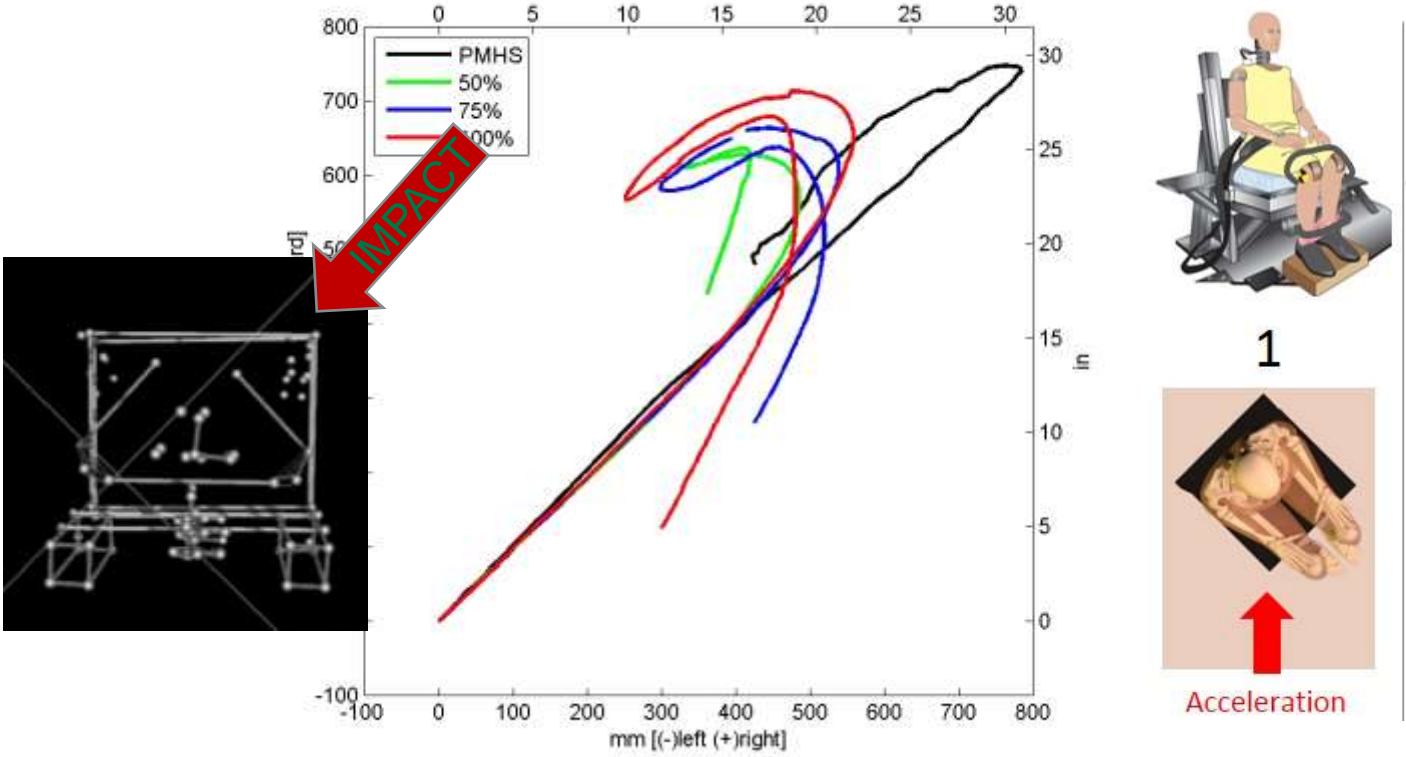


1

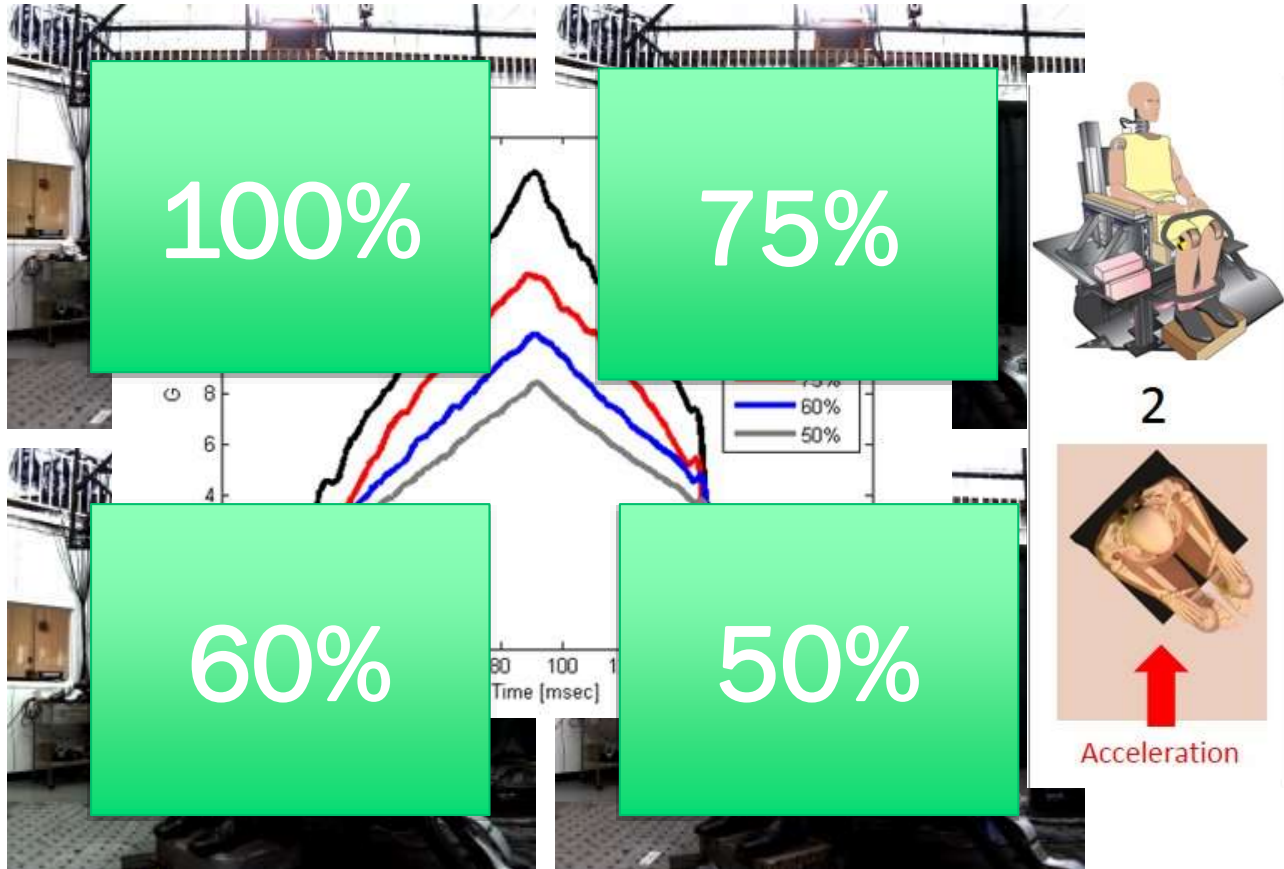


Acceleration

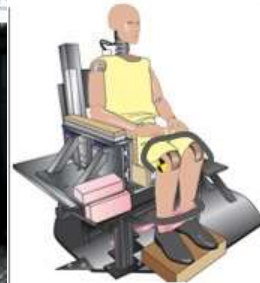
CONDITION 1-HEAD WRT SEAT - TRANSVERSE PLANE



Onboard Oblique: Time 0 msec



Onboard Oblique: Time 100 msec

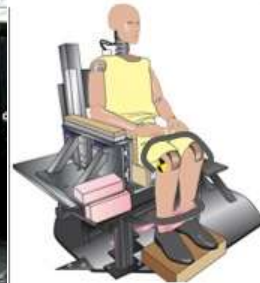


2



Acceleration

Onboard Oblique: Time 160 msec

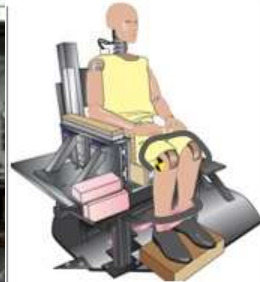


2



Acceleration

Onboard Oblique: Time 220 msec



2



Acceleration



100% Pulse

INJURY SUMMARY CONDITION 1



1



Acceleration



Coronal (spine) CT

Sagittal (spine) CT



Transection of vertebral column C5-C6

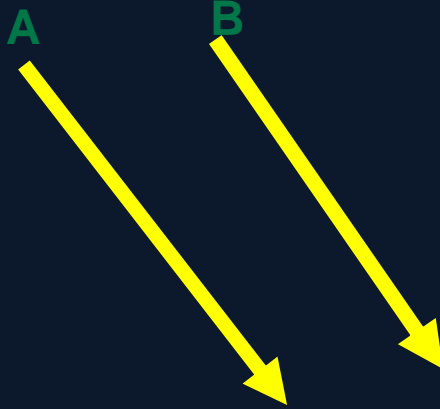




sagittal CT



Transection of vertebral column at L5-S1

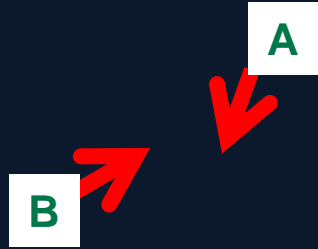


A Transection of vertebral column at L5-S1; **B** Left pelvis ala fx



Sagittal CT

Coronal CT



A Linear fracture, left pelvic ala; **B** chip fx left sacral body



INJURY SUMMARY CONDITION 2



MEDICAL
COLLEGE
OF WISCONSIN



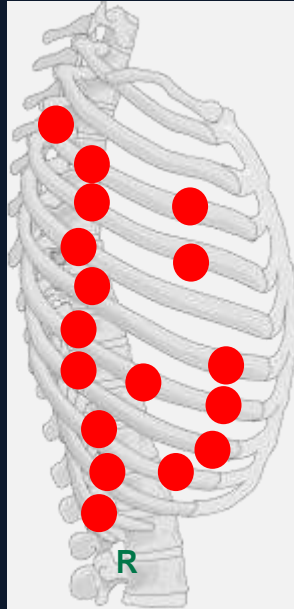
Sagittal (spine) CT

A



B







INJURY SUMMARY CONDITION 3



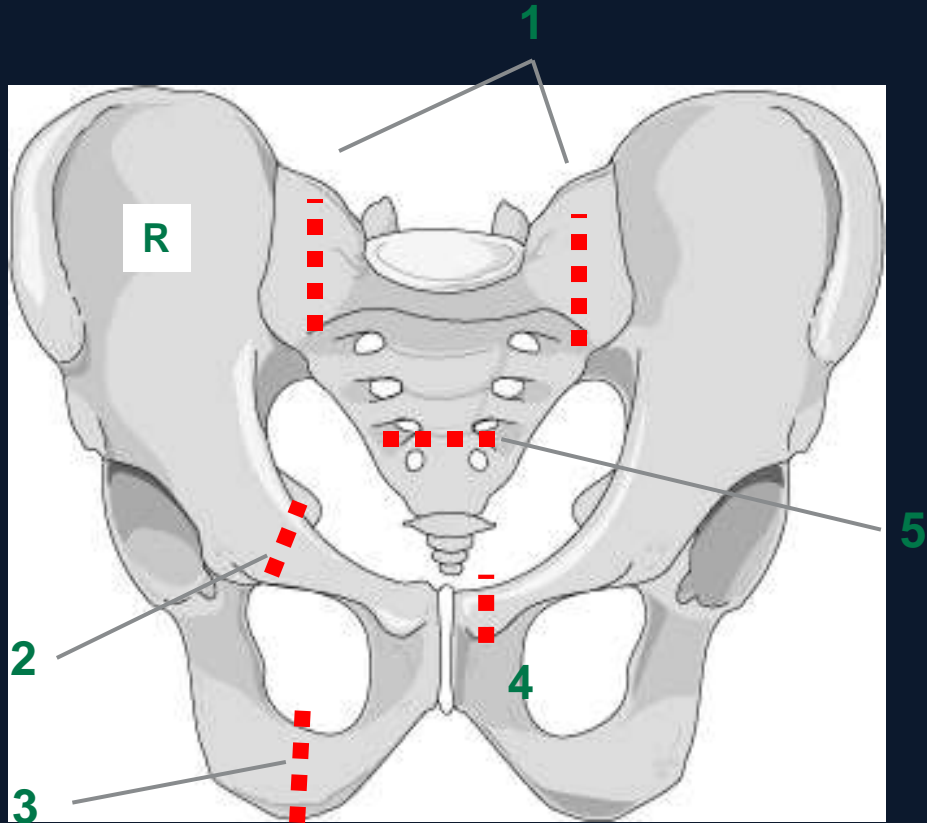
MEDICAL
COLLEGE
OF WISCONSIN

Sagittal CT

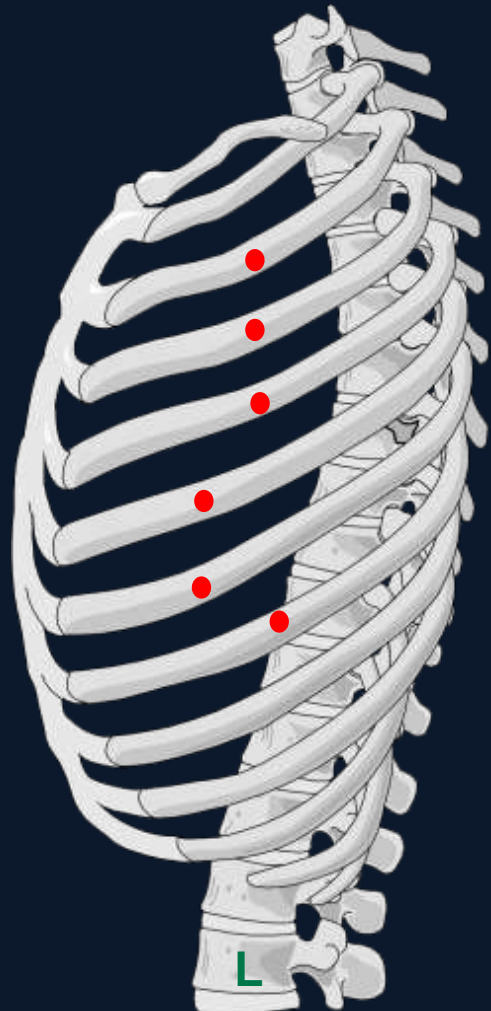
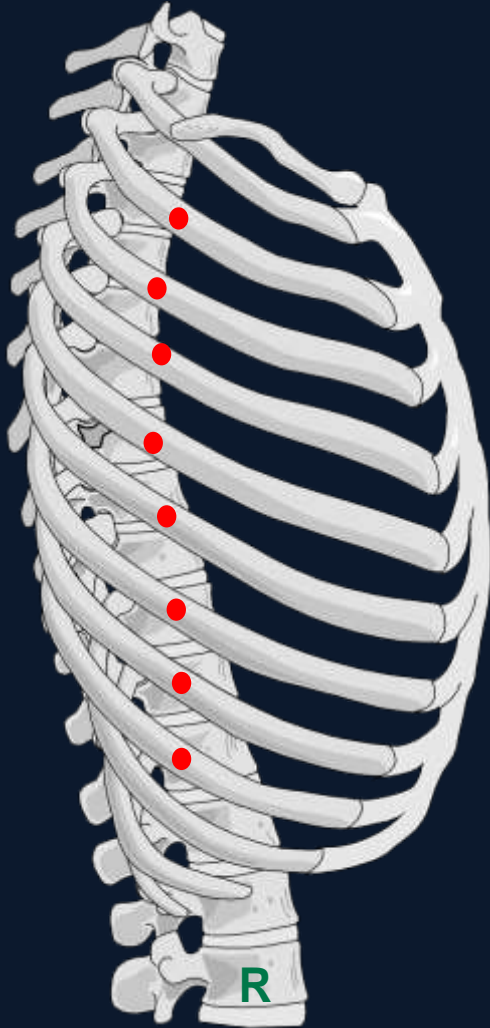
Coronal CT



A Spinal fracture-dislocation, T5



Pelvis Injuries





**MODELLING OF DIRECT HEAD IMPACT INJURY MECHANISMS
APPLIED TO TRANSPORT AIRCRAFT: ARE LONG PITCH SEATS SAFE?**

**T.H. Barth, AmSafe Aviation
S.M.R. Hashemi, CIC
A.C. Walton, CIC**



**D Right tibial shaft
fracture**

Condition 1



Condition 2



Condition 3



MATCHED-PAIRED TESTS

Conducted 100% Pulse PMHS for Conditions 1, 2, and 3

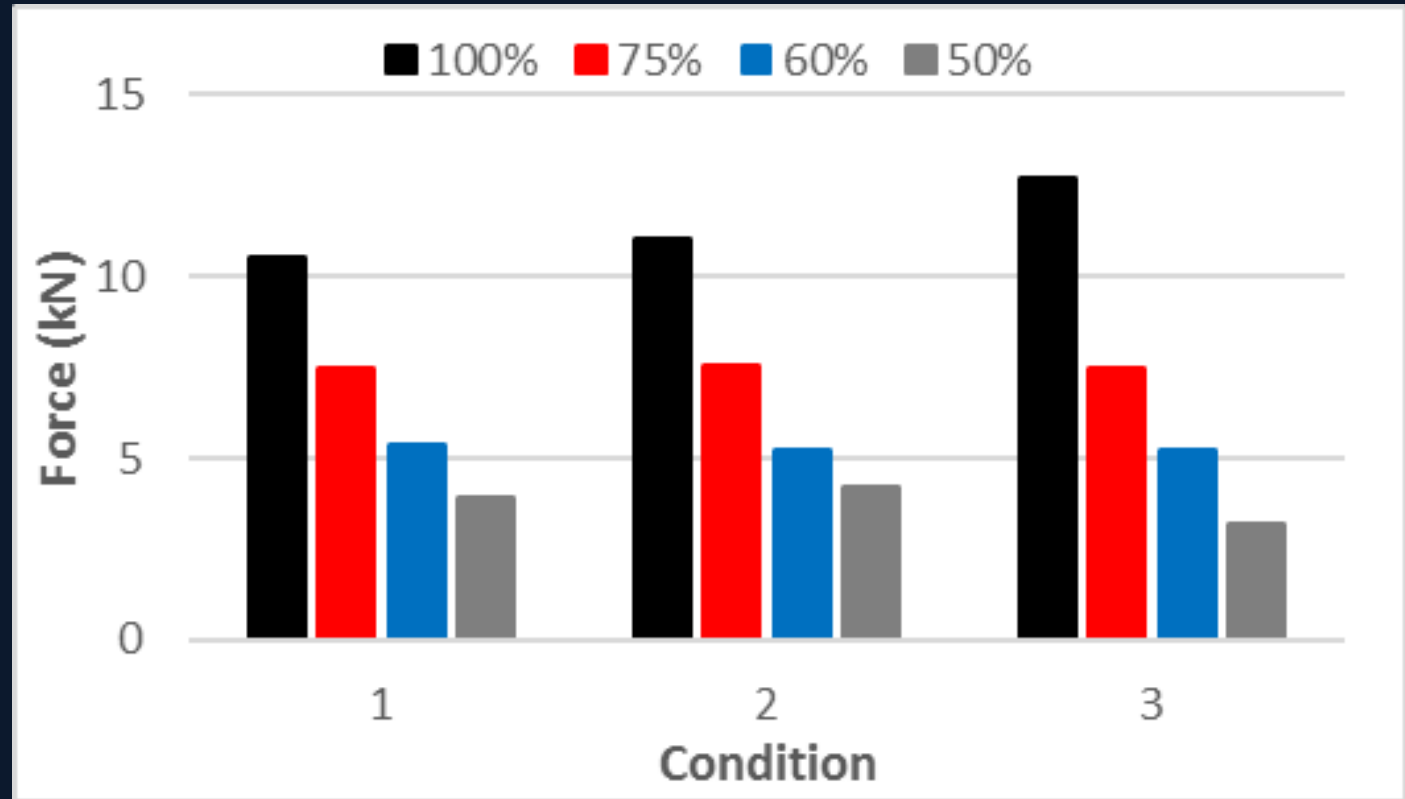
Conducted 100, 75, 60, and 50% Pulse ATD for
Conditions 1, 2, and 3.

From ATD data targeted 60% Pulse for no injury PMHS

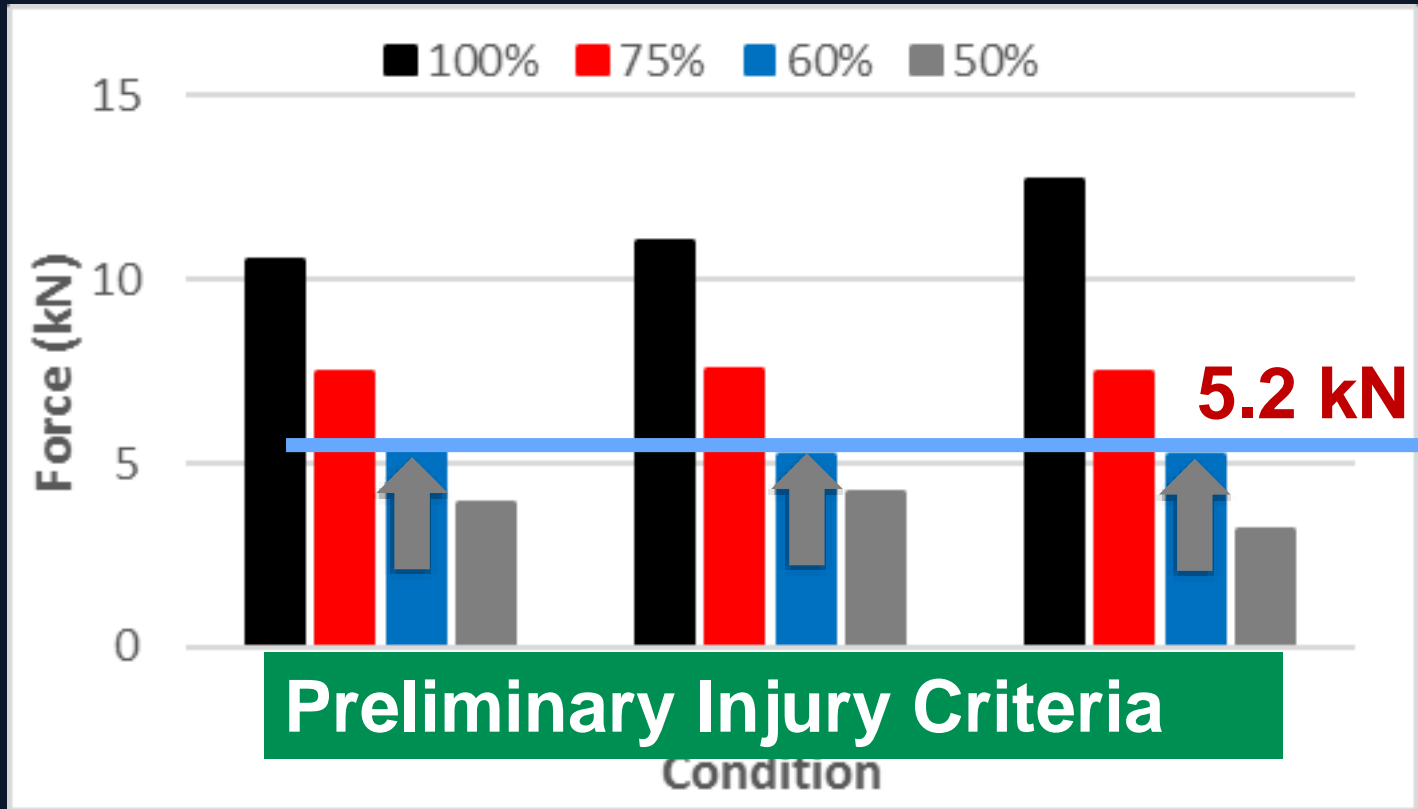
Conducted 60% Pulse PMHS for Conditions 1 and 2

No Injury

LUMBAR SPINE ATD TENSION FORCE



LUMBAR SPINE ATD TENSION FORCE



SUMMARY



FEDERAL REGISTER

The Daily Journal of the United States Government

0

Sign in Sign up

Rule

Special Conditions: Flight Structures, Inc., Boeing Model 777-200 Dynamic Test Requirements for Single-Occupant, Oblique (Side-Facing) Seats With Airbag Devices

A Rule by the Federal Aviation Administration on 09/30/2015



FAA res

4. Spine and Torso Injury Criteria

a. The shoulders must remain aligned with the hips throughout the impact sequence, or support for the upper torso must be provided to prevent forward or lateral flailing beyond 45 degrees from the vertical during significant spinal loading. Alternatively, the lumbar spine tension (F_2) cannot exceed **1200 lb** ≈ 5.2 kN

Seats



object:
ugh



Thanks

