



Effect of Cabin Pressure on the Piloted Ignition of Combustible Solids

**Sonia Fereres¹, Chris Lautenberger¹,
Carlos Fernandez-Pello¹, David Urban², Gary Ruff²**

¹University of California at Berkeley
Dept. of Mechanical Engineering
Berkeley, CA

²NASA Glenn Research Center
Cleveland, OH

***6th Triennial International Aircraft Fire and
Cabin Safety Research Conference,
October 28, 2010, Atlantic City, New Jersey***



Motivation

- Fires in pressurized vehicles (aircraft, spacecraft or submarines) are extremely hazardous
 - Small compartments
 - Difficulty to escape
- **Emphasis on fire prevention:**
 - Material flammability
 - Effect of environmental conditions (oxygen concentration, pressure, radiant heat flux, etc) on ignition



Today's Talk

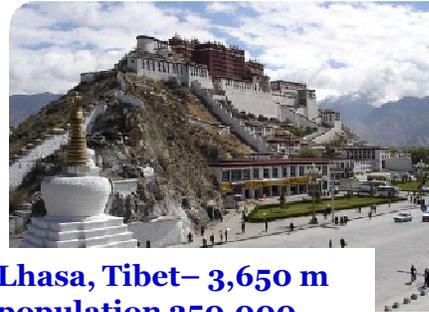
- Understand the physical mechanisms responsible for **ignition of solid combustibles** under **low pressure**
- Aircraft cabin pressure is typically pressurized to a "cabin altitude" of 8000 feet or less (~ 75 kPa)
- Are reduced pressure environments a higher fire risk?
 - Piloted ignition experiments with air at low P
 - *Forced Ignition and Spread Test (FIST)* apparatus at UC Berkeley to analyze material flammability
 - Analytical explanation of results





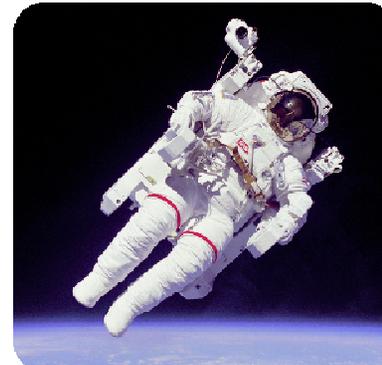
Lower ambient pressure can be found at...

- High Altitude
- Inside Aircraft
- Inside Spacecraft



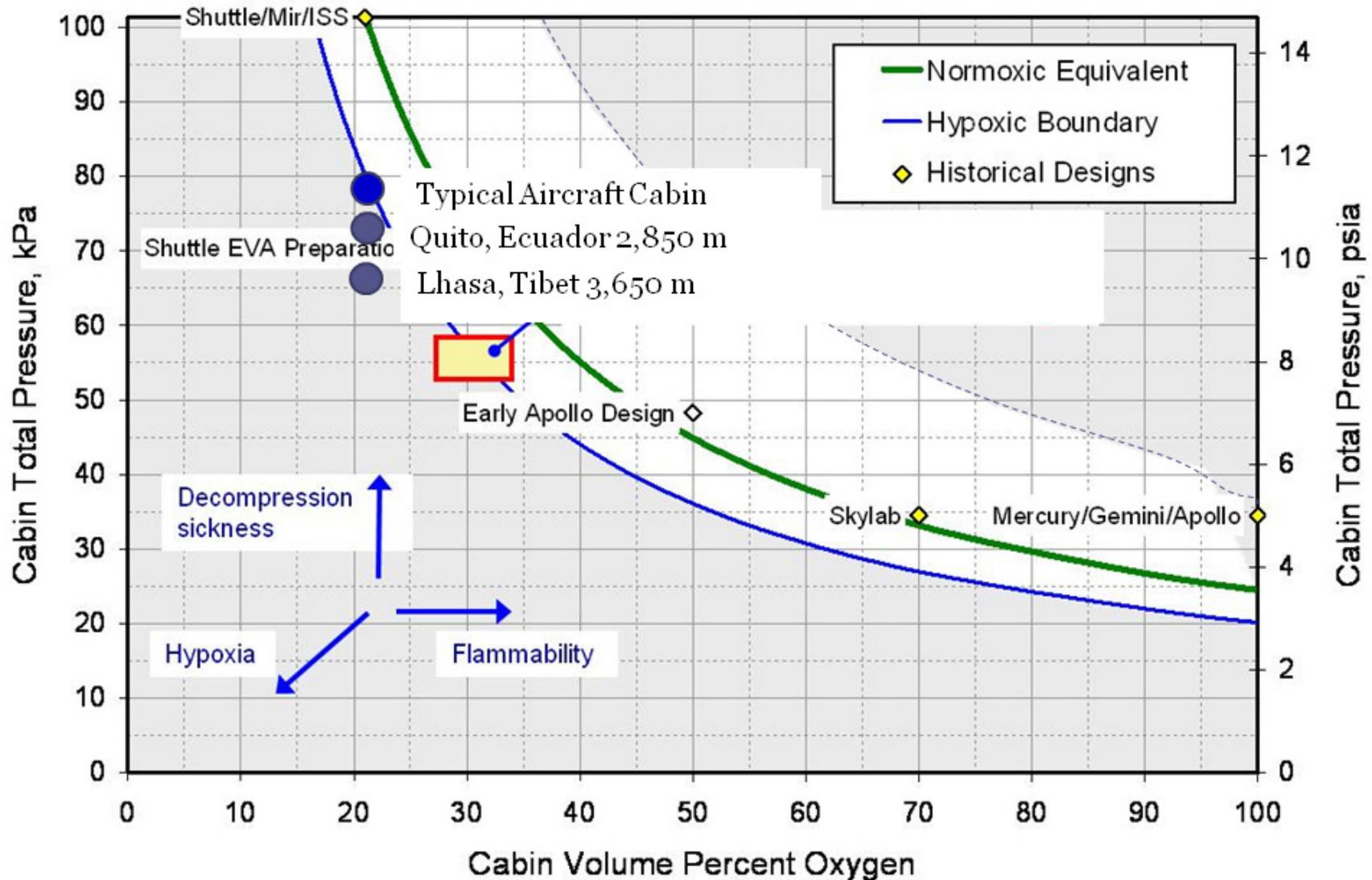
Lhasa, Tibet- 3,650 m
population 250,000

Quito, Ecuador- 2,850 m
population 2M





Cabin Environments



How does a solid fuel ignite?

Piloted ignition process:

1. Solid heating & pyrolysis
2. Mixing of gaseous fuel and air
3. Chemistry: fuel/air mixture reaches lean flammability limit at high temperature igniter
4. If sufficient pyrolysis gases are generated: a diffusion flame will anchor on solid (burning) → critical mass flux at ignition





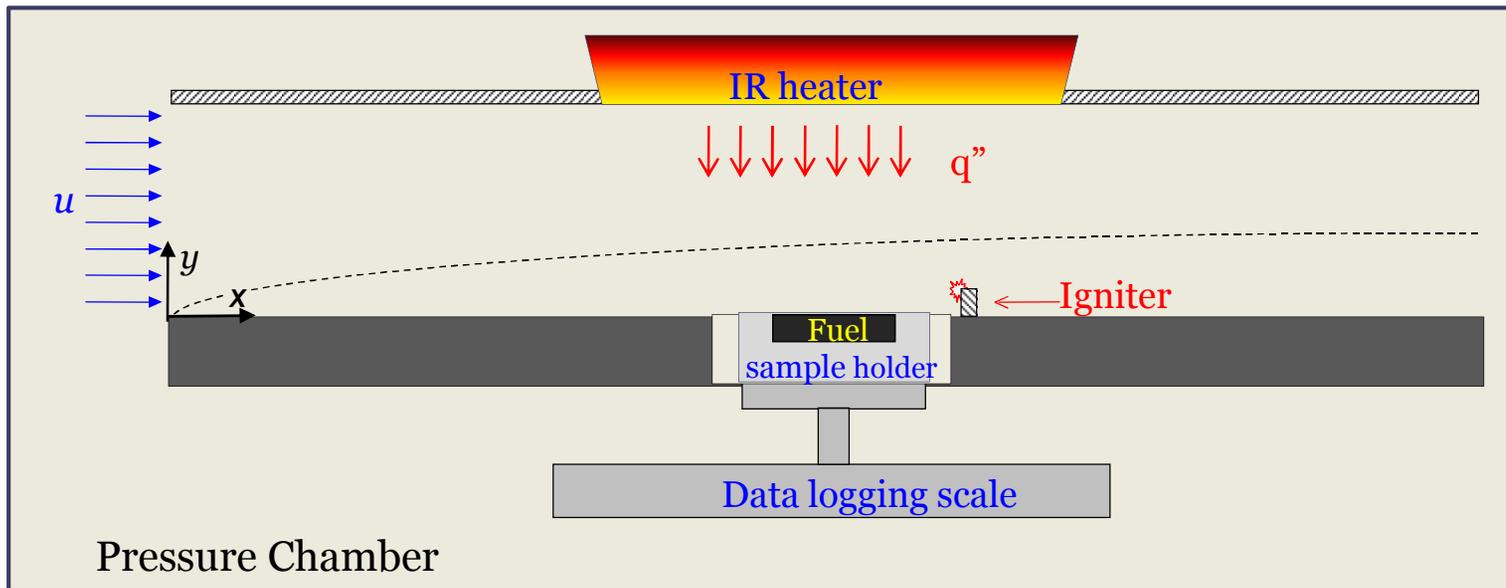
Possible Fire Scenario



- **Heat source: electronic component overheating**
- **Fuel: polymeric materials used in panels, blocks, covers**
- **Ignition source: spark from electrical arcing**



Forced Ignition and Spread Test (FIST)



Variables:

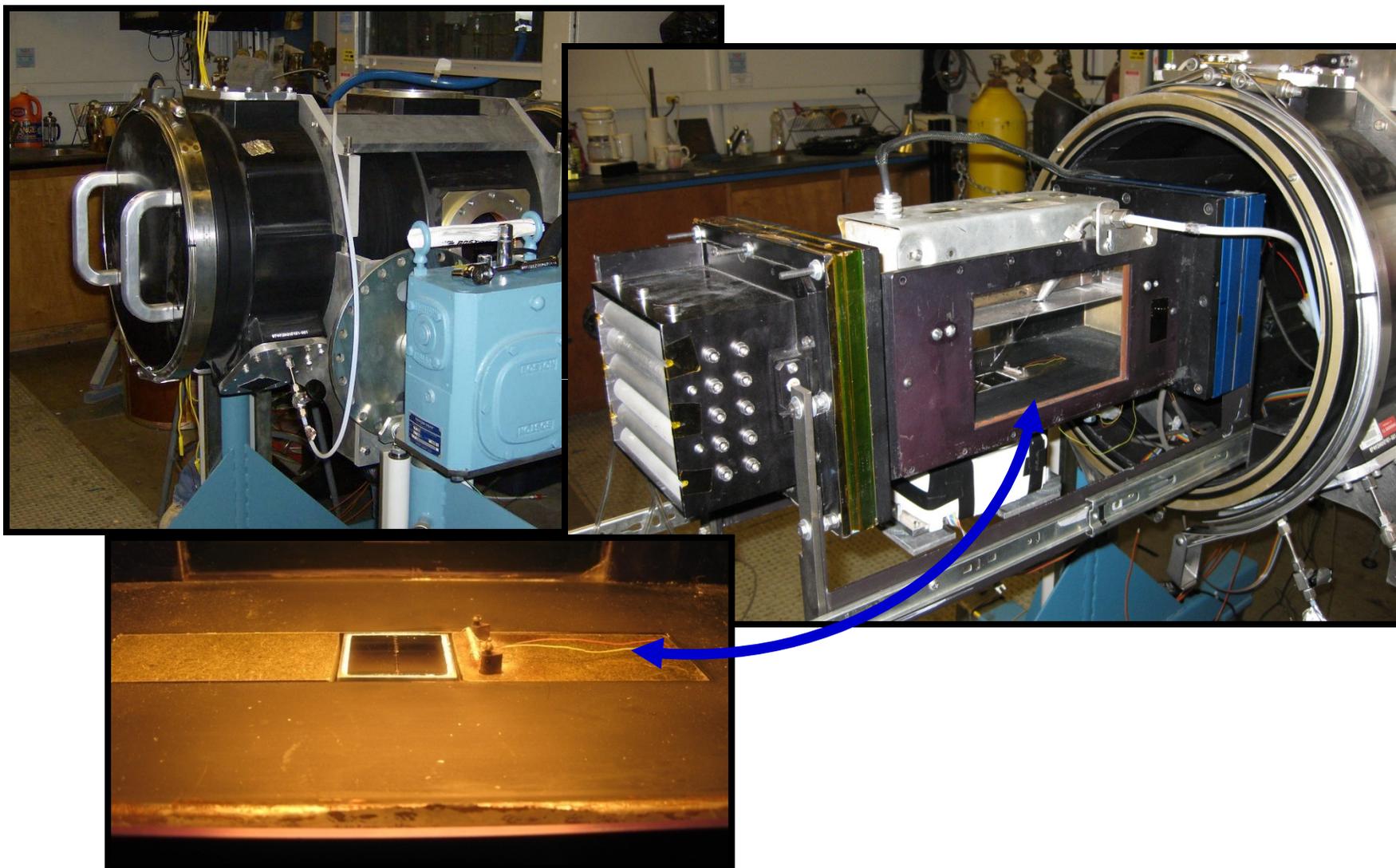
- air flow velocity
- incident heat flux
- ambient pressure

Material: PMMA

Measure :

- T_{surface} vs. time $\rightarrow t_{\text{ig}}$, **time to ignite**
- Mass vs. time $\rightarrow (dm/dt)|_{t_{\text{ig}}}$ **mass loss rate at ignition**

FIST Apparatus





Video of Test

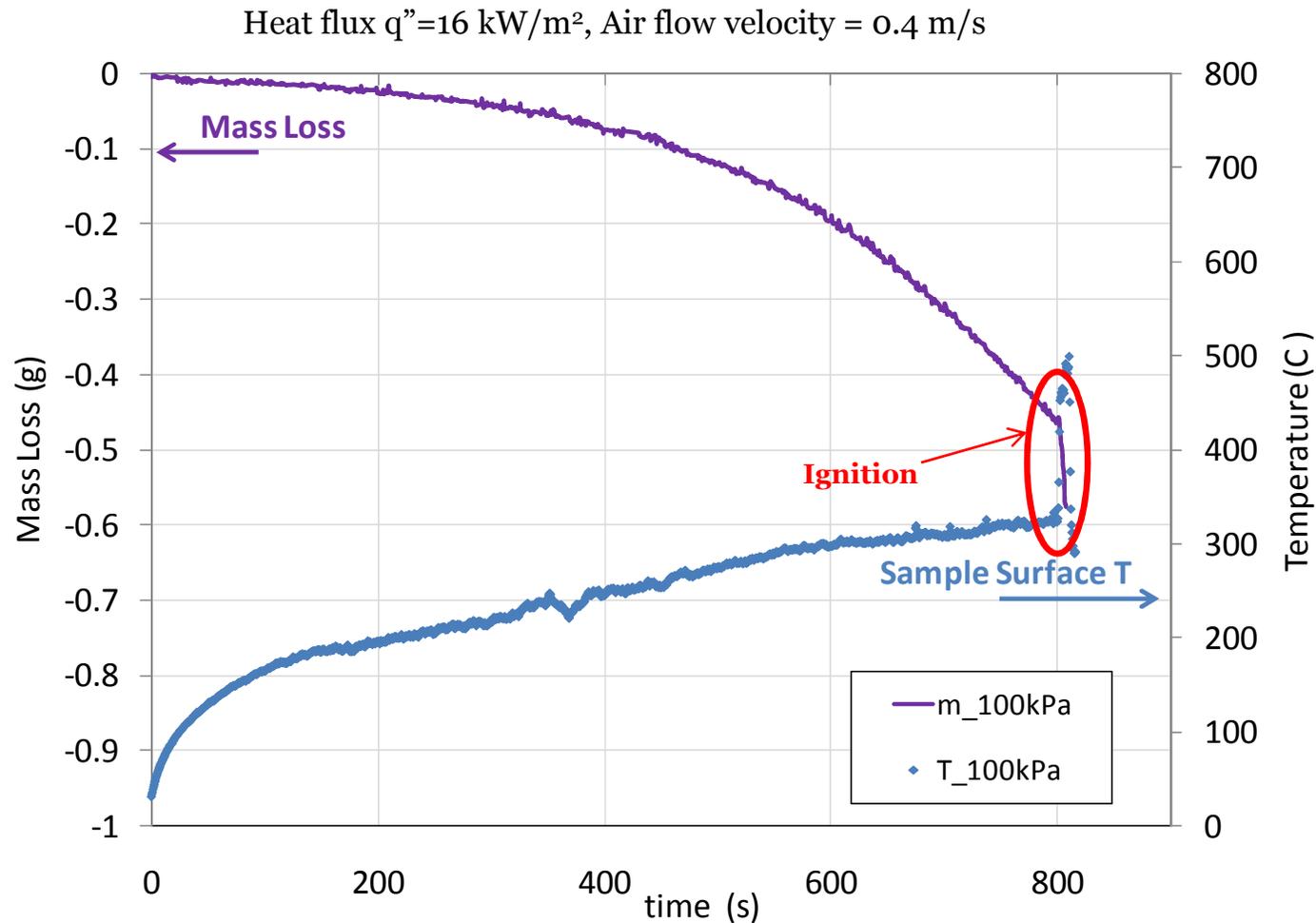


Air velocity = 0.4 m/s , $q'' = 16 \text{ kW/m}^2$, $P = 12 \text{ psi}$



Experiment Example

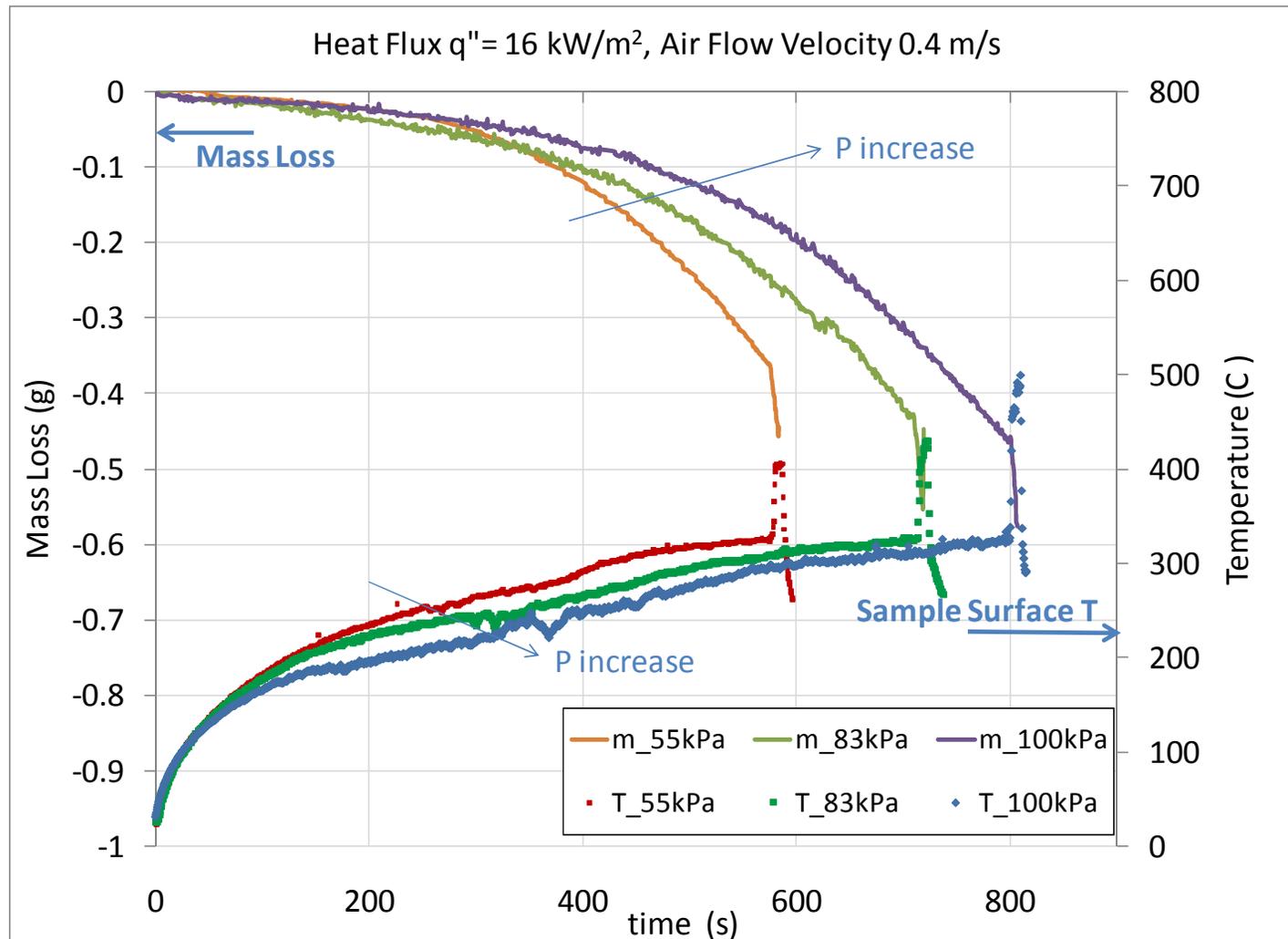
- 100 kPa & Air (21% O₂ by volume) -Raw Data





Experimental Results

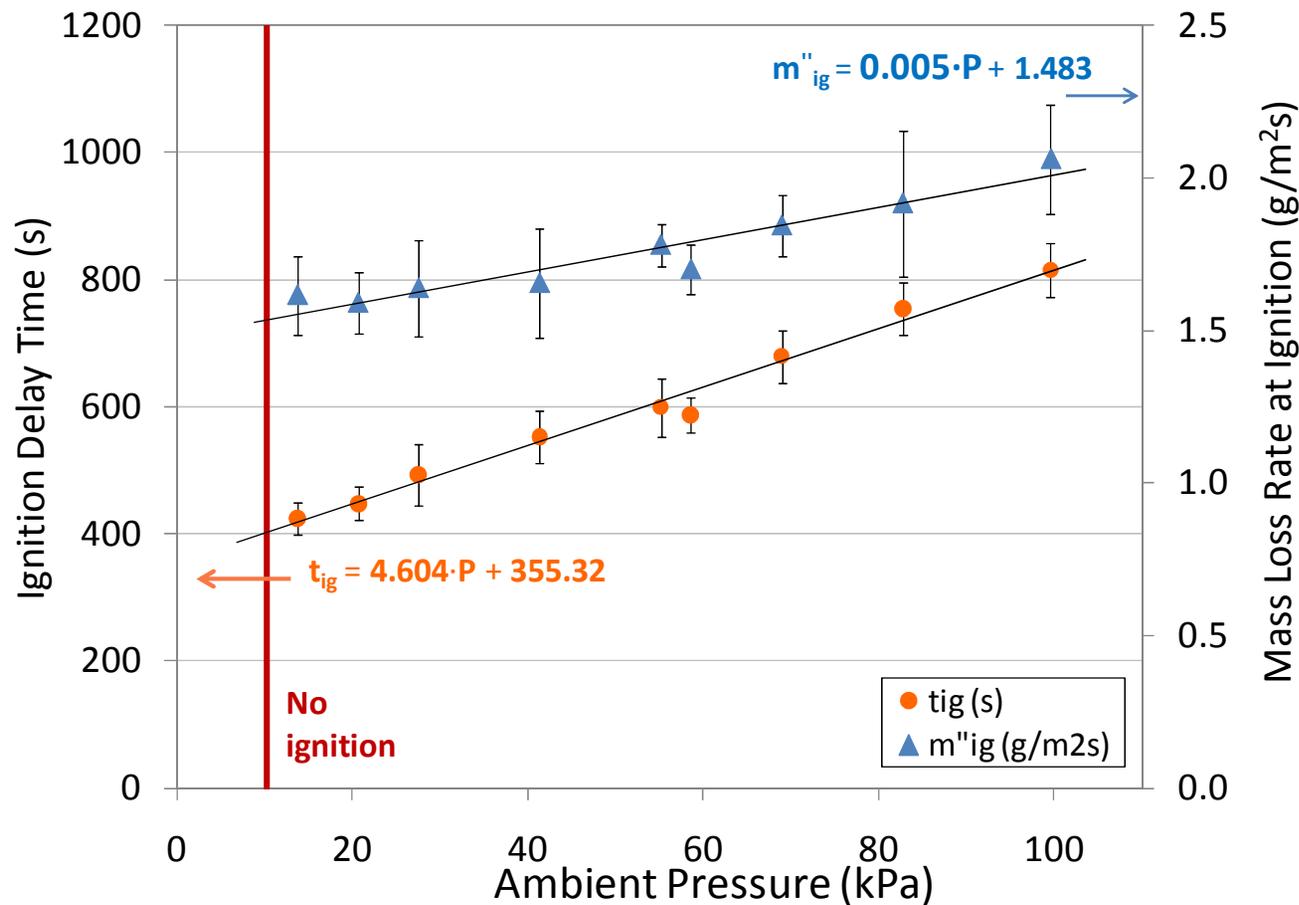
- Pressure Comparison : 55, 83 & 100 kPa (Raw Data)





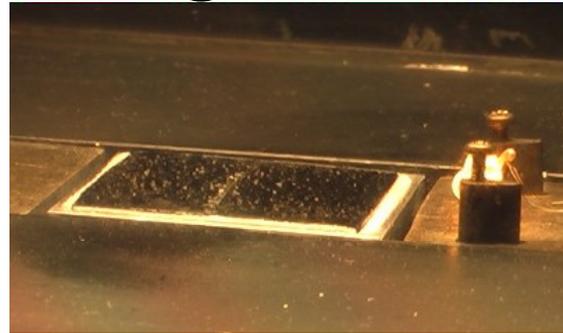
Experimental Results

- Ignition Delay & Mass Loss Rate at Ignition vs. Total Pressure
Air (21% O₂ by volume) at 0.4 m/s , $q''=16$ kW/m²

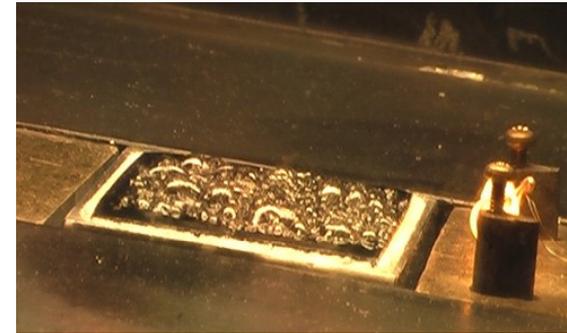


Visual Observations

- Different surface behavior: bubble formation, size and bursting characteristics



3 psi (21 kPa)



12 psi (83 kPa)

- Flame establishment over solid surface also different



3 psi (21 kPa)



12 psi (83 kPa)



Effect of Pressure (Part1)

- Ignition delay time, t_{ig} :
 - $t_{ig} = t_{heating} + t_{mixing/transport} + t_{induction}$
- Heating time: convective heat loss over flat plate

- Forced flow: $h \propto Re^{1/2} Pr^{1/3}$

- Natural convection: $h \propto Gr^{1/4} Pr^{1/4}$

- Mixed flow $h \propto Re^{1/2} Pr^{1/3} \sqrt[4]{1 + \frac{Gr}{Re^2} Pr^3}$

$$\left. \begin{array}{l} Re = \rho UL / \mu, Re \sim P \\ Pr \neq f(P) \\ \text{Ideal gas: } Gr \sim P^2 \end{array} \right\}$$



$$h \sim P^{1/2}$$

As pressure decreases, convective heat loss of material to surroundings is lower → heats more rapidly



Effect of Pressure (Part 2)



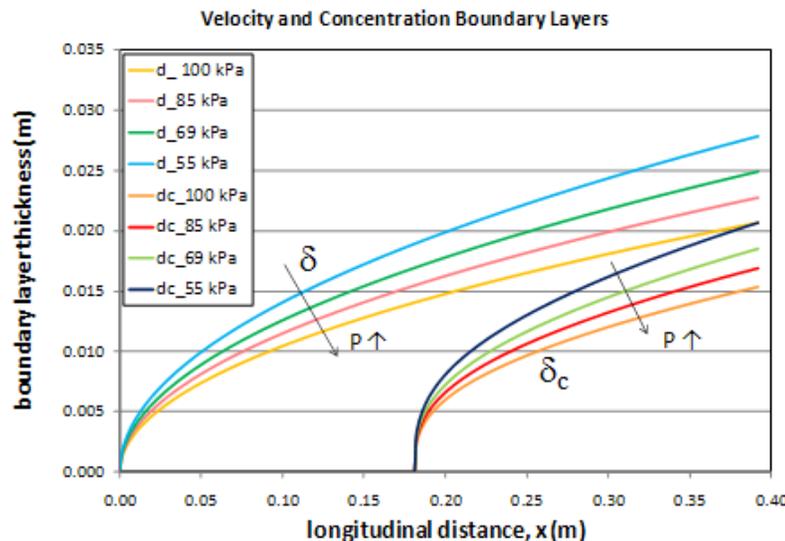
- Mixing/transport time: Mass loss rate at which a flammable concentration (LFL) is obtained at the pilot

Simplified Analysis : Boundary Layer – Integral Method

- 3rd order polynomials for velocity, temperature and species profiles:

$$\frac{u}{U_\infty} = \frac{3y}{2\delta} - \frac{1}{2}\left(\frac{y}{\delta}\right)^3 \quad \frac{T - T_0}{T_\infty - T_0} = \frac{3y}{2\delta_T} - \frac{1}{2}\left(\frac{y}{\delta_T}\right)^3 \quad \frac{Y_F - Y_{FO}}{Y_{F\infty} - Y_{FO}} = \frac{3y}{2\delta_c} - \frac{1}{2}\left(\frac{y}{\delta_c}\right)^3$$

- Integrate BL Eqns. → analytical expressions for hydrodynamic, thermal and concentration BL thicknesses:



$$\delta = \sqrt{\frac{280 \nu x}{13 U_\infty}} \approx 4.64 \sqrt{\frac{\nu x}{U_\infty}}$$

$$\delta_T = \left[\frac{10 \alpha x \delta}{U_\infty} \left(1 - \frac{x_T}{x} \right) \right]^{1/3}$$

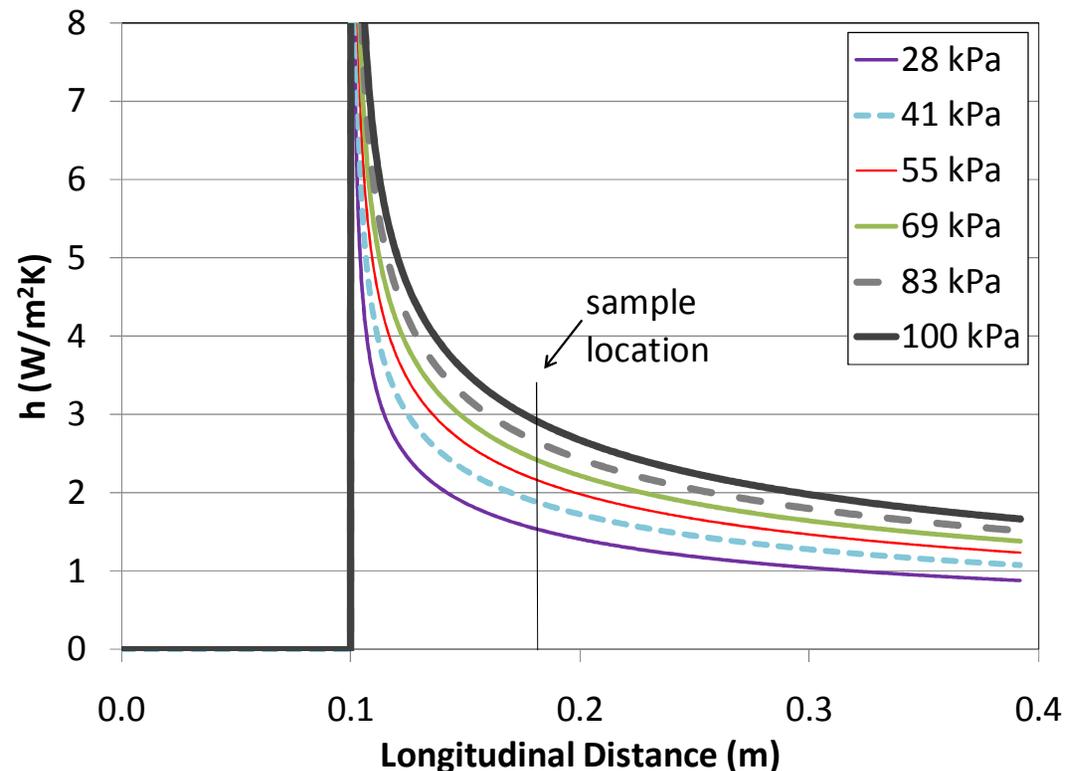
$$\delta_c = \delta \left\{ \frac{13 D}{14 \nu} \left[1 - \left(\frac{x_c}{x} \right)^{3/4} \right] \right\}^{1/3}$$



Simplified Analysis

Heat Transfer Coefficient

- $h \approx k/\delta_T$
- At the sample location, h decreases by 13% when the pressure is reduced from 100 kPa to 75 kPa



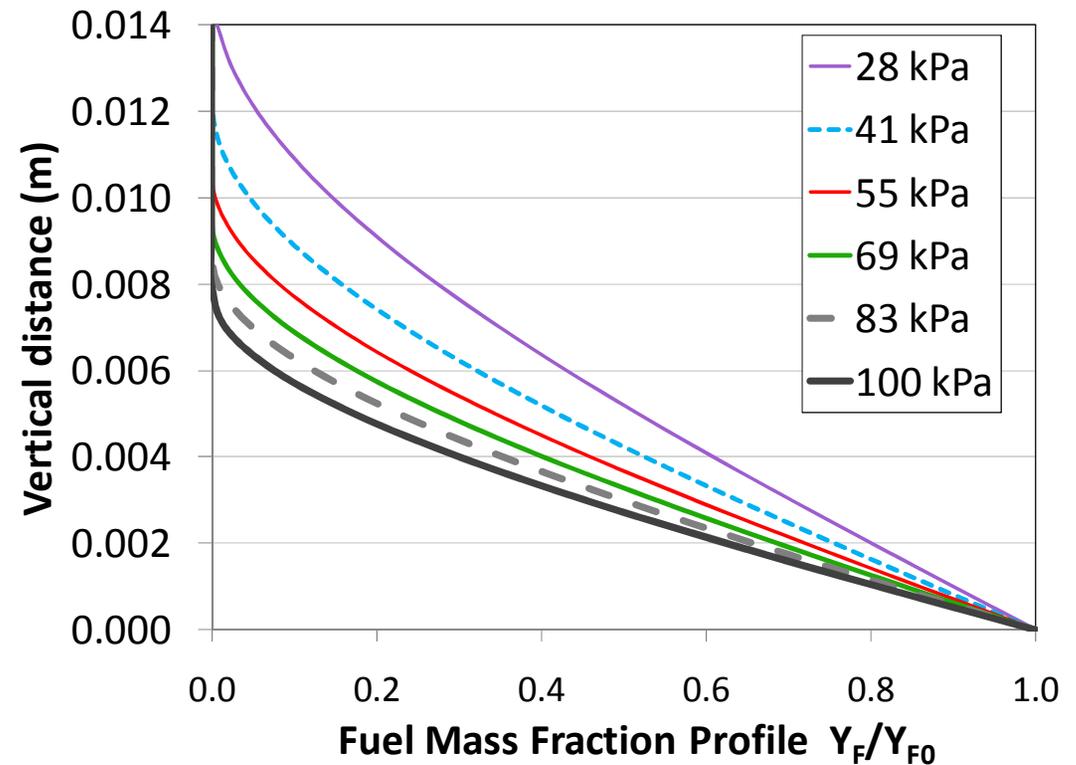


Simplified Analysis Species Concentration



- Reduced pressure leads to a thicker species boundary layer

$$\frac{Y_F}{Y_{F0}} = 1 - \frac{3y}{2\delta_c} + \frac{1}{2}\left(\frac{y}{\delta_c}\right)^3$$



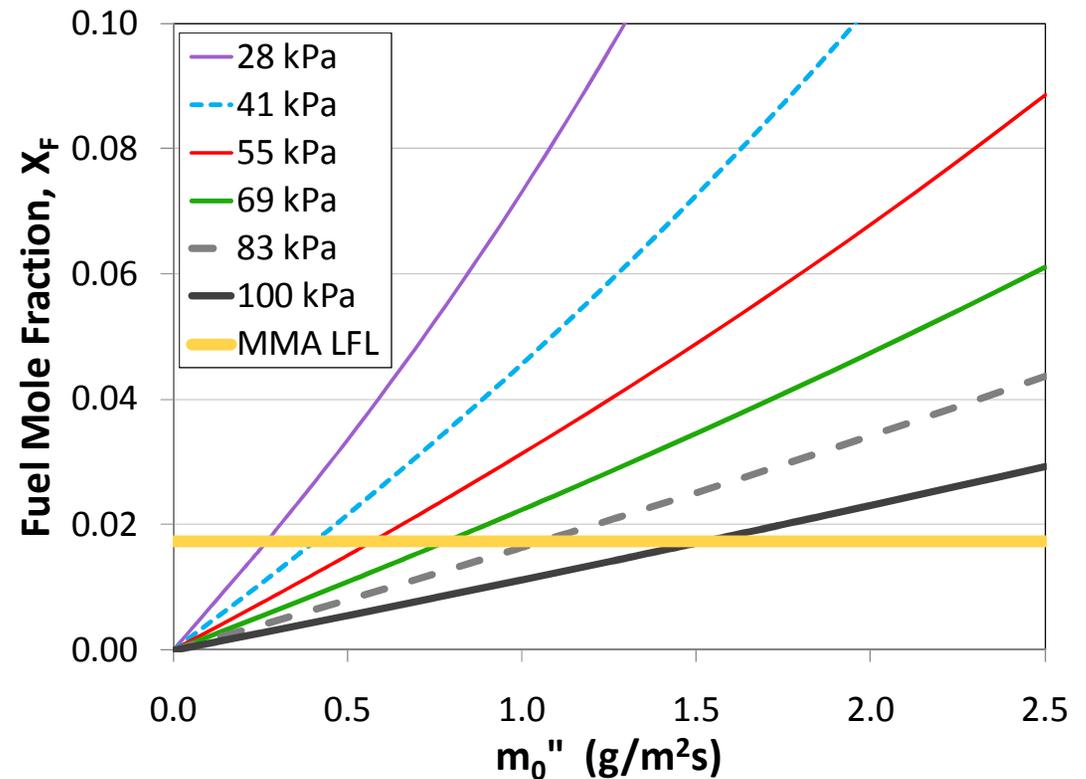


Simplified Analysis

- To determine mass loss rate:

$$\dot{m}'' \approx -\rho D \left(\frac{\partial Y_F}{\partial y} \right)_y = \frac{3}{2\delta_c} \rho D Y_{F0} \left(1 - \left(\frac{y}{\delta_c} \right)^2 \right)$$

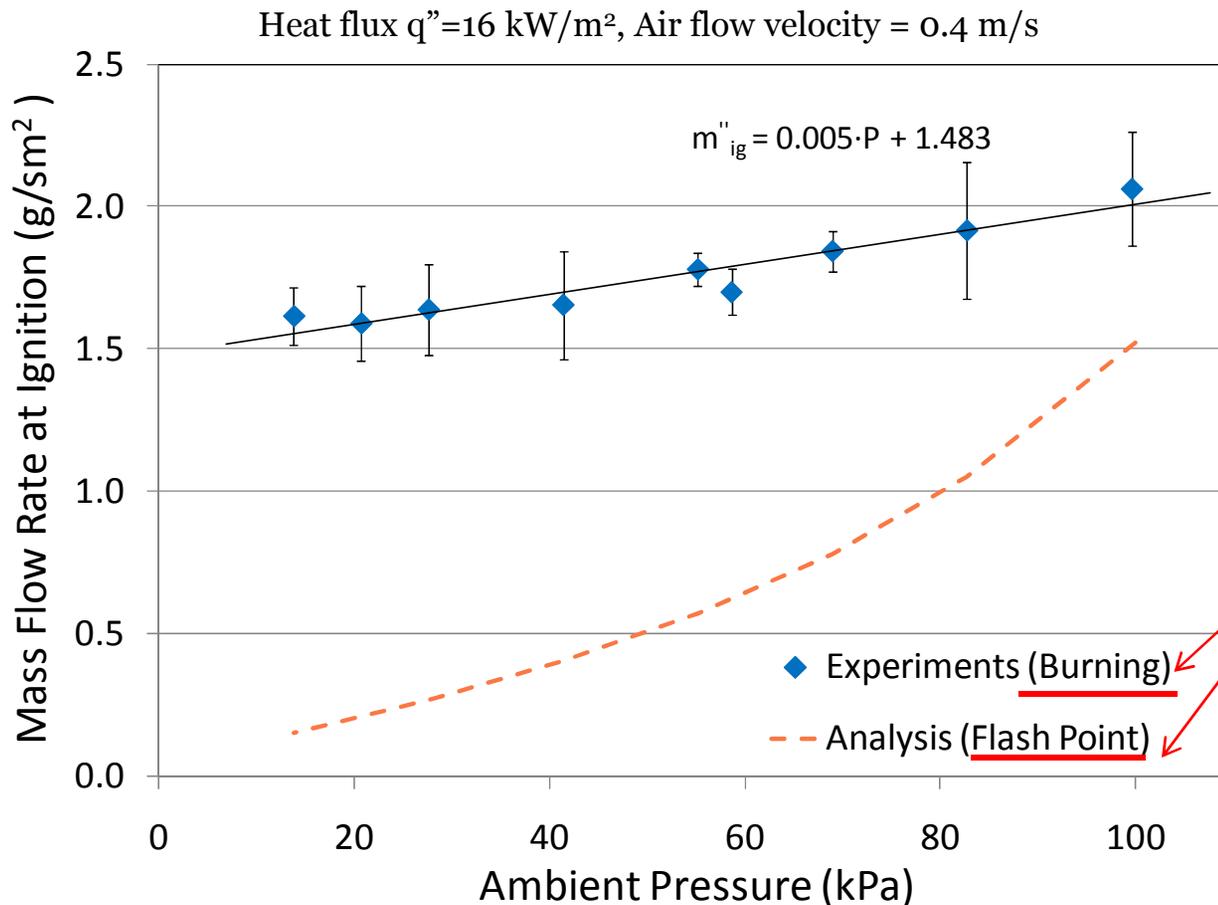
- At lower P, required mass flow rate of fuel to reach lean flammability limit at igniter location is reduced





Comparison of Trends

- Mass Loss Rate at Ignition vs. Pressure



Not the same:
reaching LFL is
necessary but not
sufficient
condition for
sustained burning!



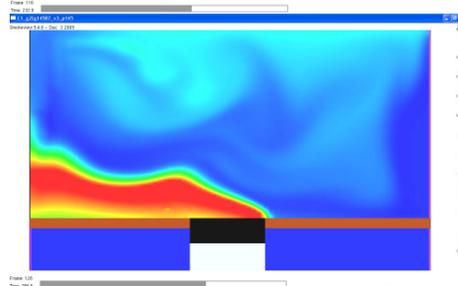
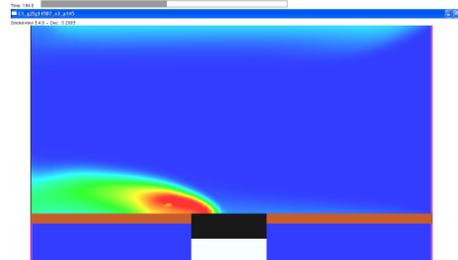
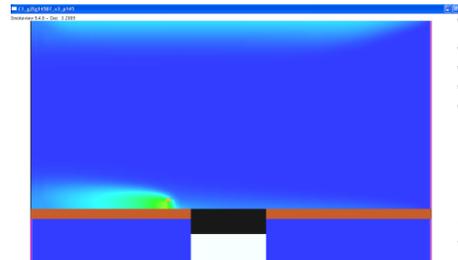
Current Work

Fire Dynamics Simulator (FDS) 2D model

- PMMA is irradiated under a prescribed heat flux → the solid decomposes and the products of the pyrolysis ignite in the gas phase.



HRR



Temperature

Premixed flame appears in the gas phase

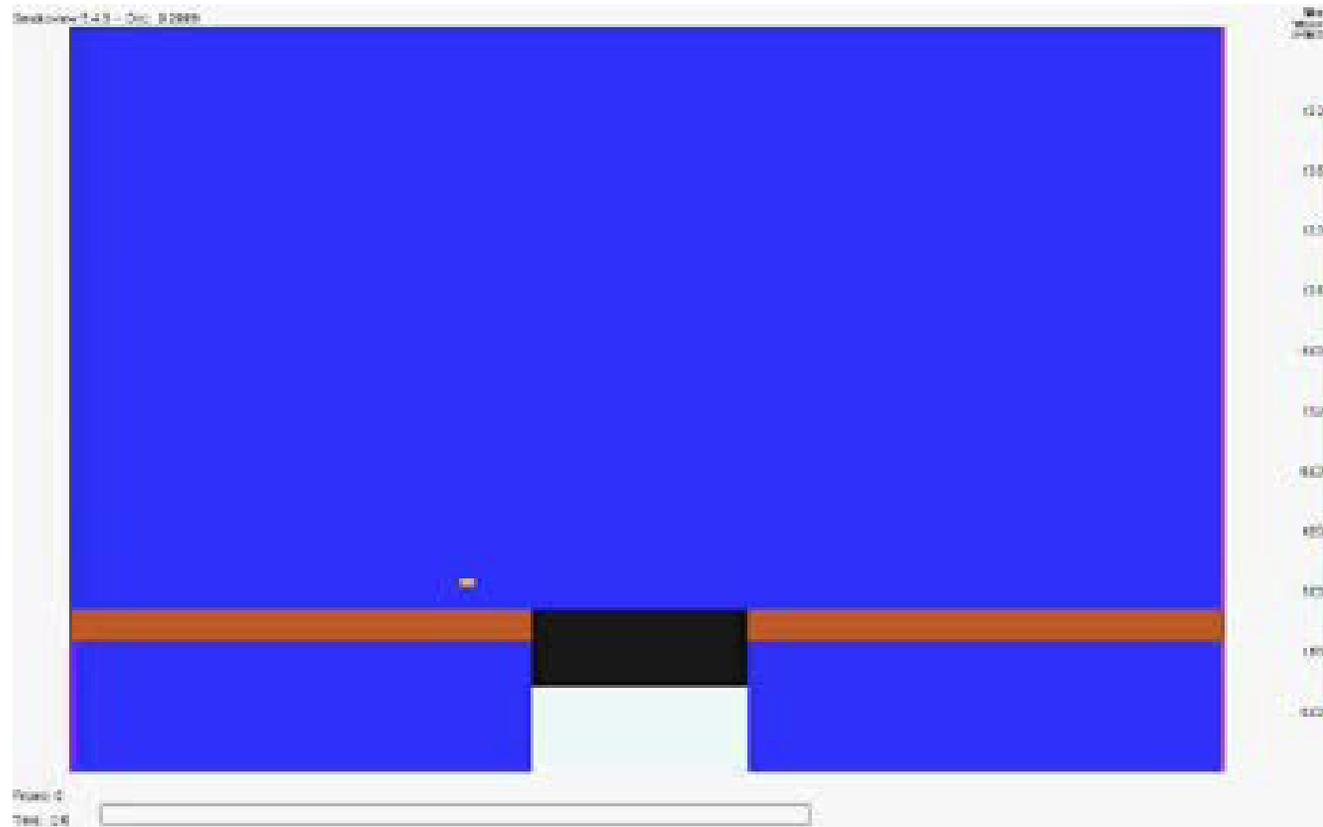
Flame 'jumps' on to solid fuel surface

Diffusion flame anchored on solid surface travels



Current Work

- Fire Dynamics Simulator (FDS) 2D model
 - Heat release rate/volume:





Summary & Conclusions

- Experimental results from piloted ignition show that t_{ig} & m''_{ig} decrease with pressure:
 - At 75 kPa , $\Delta t_{ig} = -15\%$
 $\Delta m''_{ig} = -7\%$
- A theoretical explanation provides insight on the effect of pressure on:
 - Heat transfer coefficient
 - Mass loss rate required to reach a flammable mixture
- Next steps include developing a numerical model using FDS to compare to experiments
- Overall, **a reduction in ambient pressure leads to an increased fire risk**



Acknowledgements

- *The research at University of California at Berkeley was supported by NASA on grant NNX08BA77A*