

FuelShield™ – Alternate Fuel Tank Inerting That Also Protects Against Debris Impacts

By

N. Albert Moussa and Venkat Devarakonda

BlazeTech Corporation

24 Thorndike St. Cambridge, MA 02141

617-661-0700/ 617-661 9242(fax)

amoussa@blazetech.com

Presented at the

International Aircraft Fire and Cabin Safety Research Conf.

Atlantic City, NJ, 22-25 October 2001

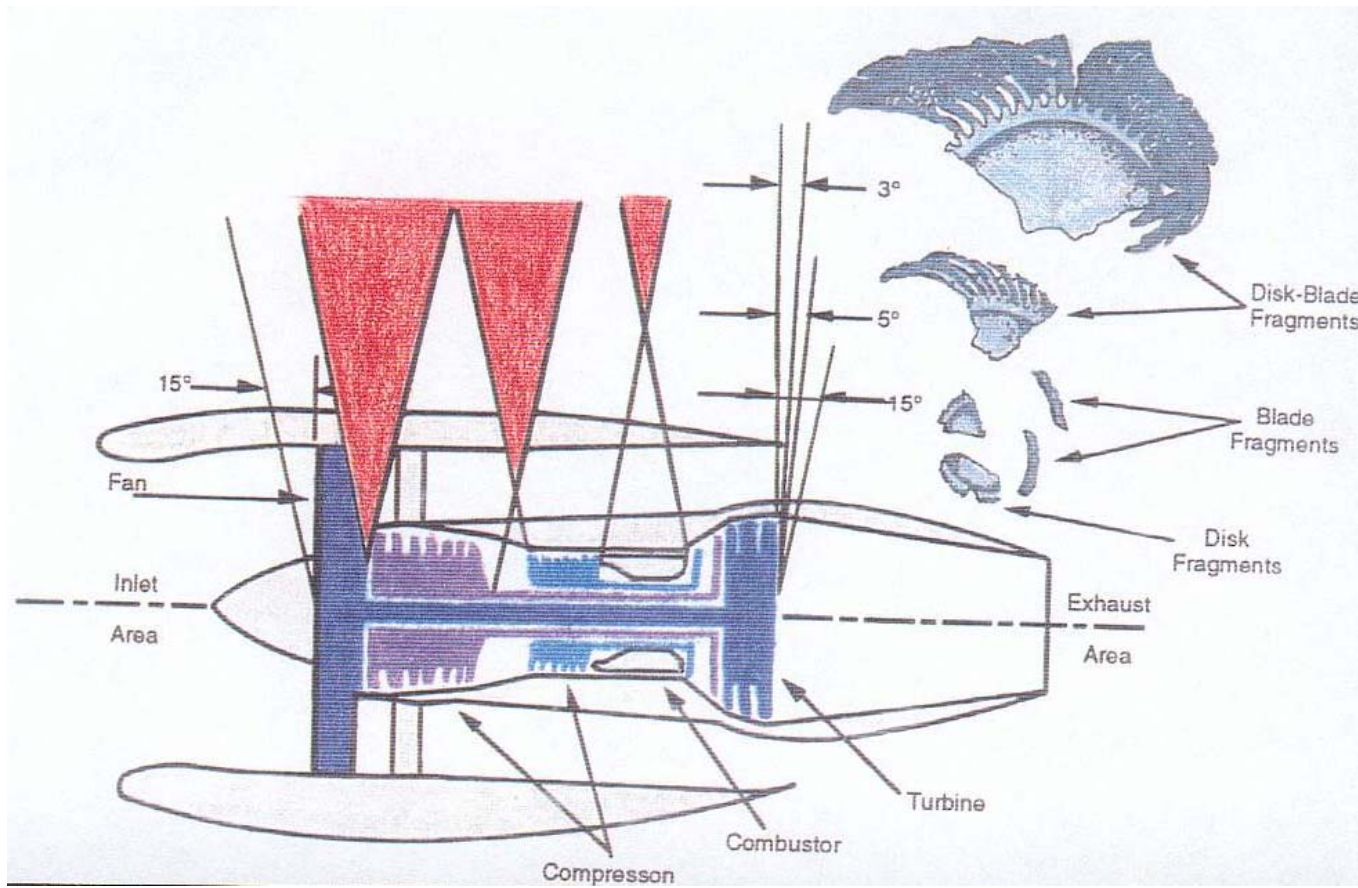
Introduction

- FAA concerned about uncontained engine failures
- Driven by accidents in Manchester, England, and in Sioux City, IA
- FAA Project at BlazeTech (1995): Effects of engine debris impacts on fuel tanks
- Two hazards of interest due to debris impacts
 - Fire/explosion in ullage
 - Hydrodynamic ram

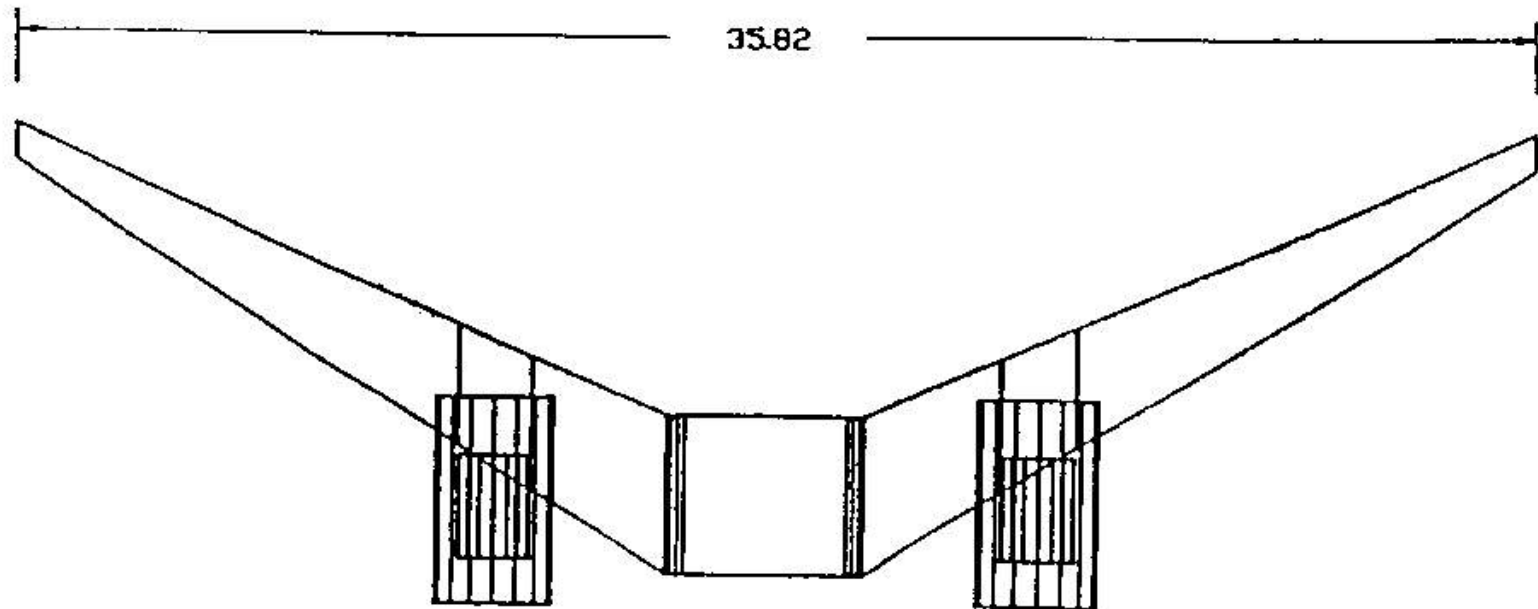
Outline

- Background on FAA study
- Summary of results on both hazards
- Related work funded by Air Force, Navy and BlazeTech
- FuelShield™:
 - Technology description
 - status of development

Debris from Uncontained Engine Failure

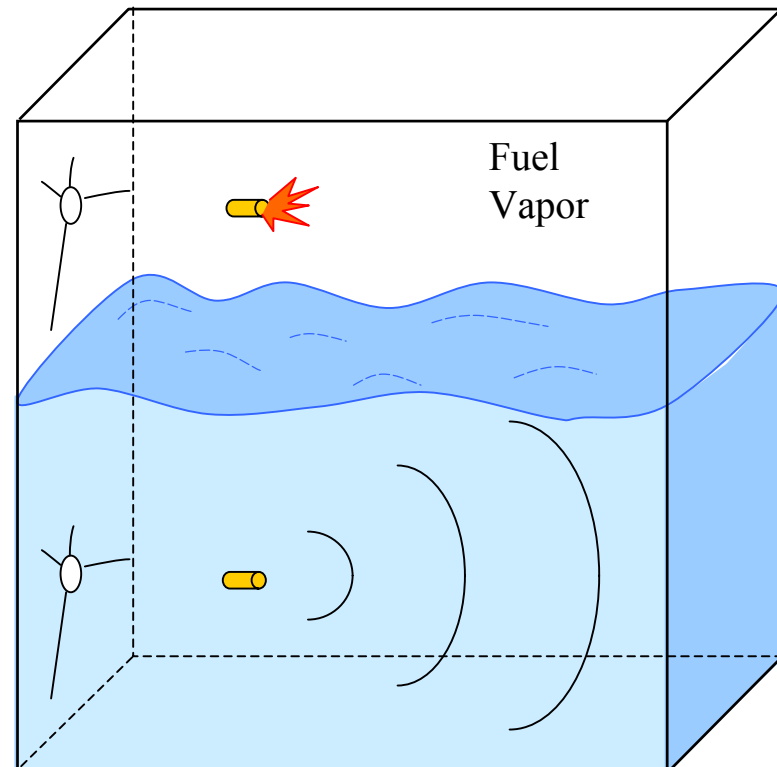


Wing Mounted Engines



Hazards in Fuel Tanks Impacted by Debris

1. Ullage Fire/Explosion
2. Hydrodynamic Ram



Key Events in Ram

1. Debris enters tank
2. Pressure rise in tank due to motion of debris
3. Structural response and tank failure
4. Enlargement of penetration hole (more fuel leakage)
5. Fuel leak can produce fire and loss of aircraft

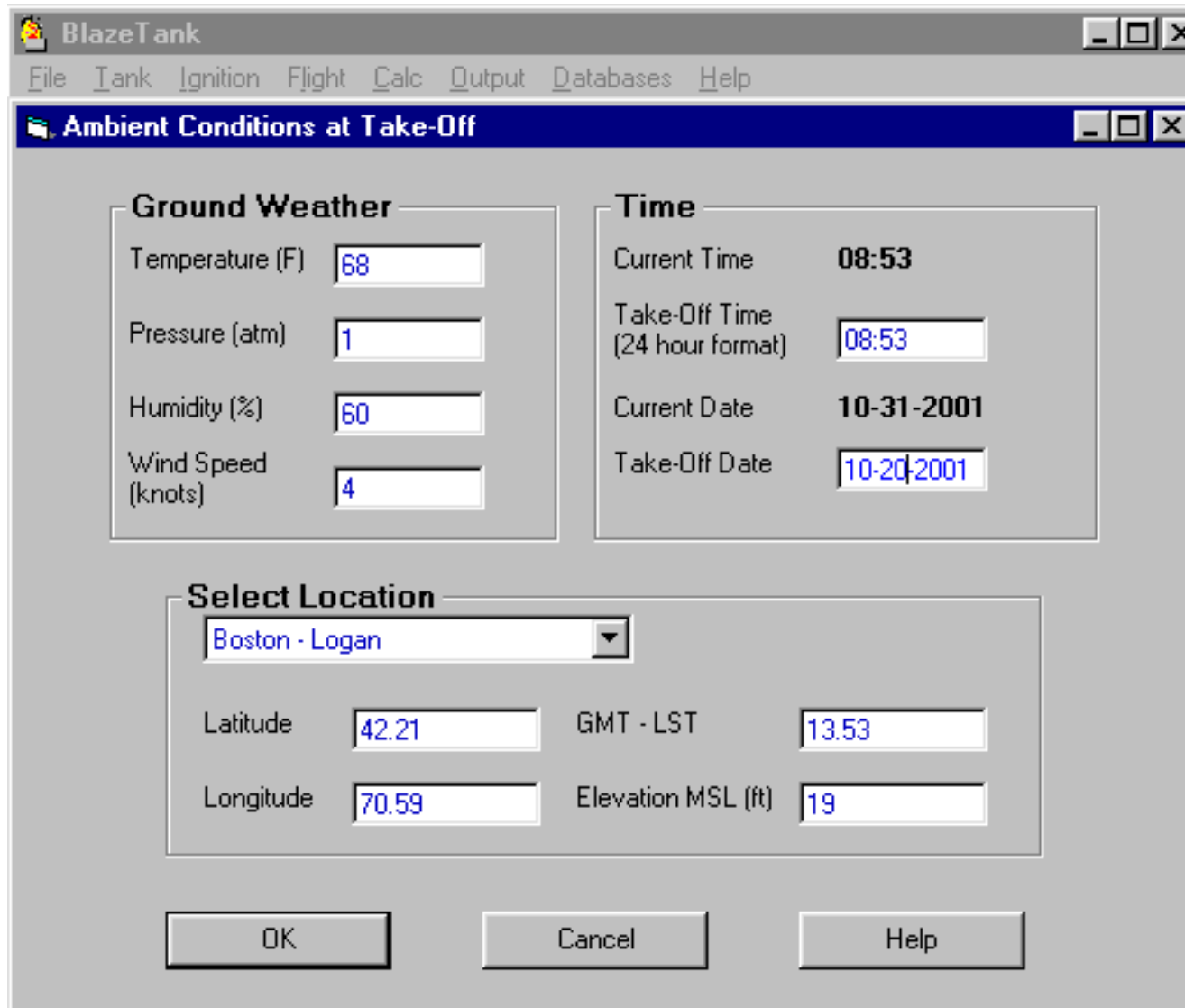
Ullage Fire/Explosion

- In FAA study, we characterized conditions leading to ignition
- Results were discounted until TWA 800 occurred
- Packaged the results into a model (BlazeTank) that predicts flammability, ignition and overpressure as functions of fuel properties, fuel tank design, flight profiles and impact conditions
- Presented it at last Fire and Cabin Safety Research Conference

Model for Ullage Fire/Explosion



BlazeTank Model (contd.)



The image shows a screenshot of the 'BlazeTank' software interface, specifically the 'Ambient Conditions at Take-Off' dialog box. The dialog box has a title bar with the text 'BlazeTank' and standard window controls. Below the title bar is a menu bar with the following options: File, Tank, Ignition, Flight, Calc, Output, Databases, and Help. The main area of the dialog box is divided into three sections: 'Ground Weather', 'Time', and 'Select Location'. The 'Ground Weather' section contains four input fields: Temperature (F) with the value 68, Pressure (atm) with the value 1, Humidity (%) with the value 60, and Wind Speed (knots) with the value 4. The 'Time' section contains four input fields: Current Time with the value 08:53, Take-Off Time (24 hour format) with the value 08:53, Current Date with the value 10-31-2001, and Take-Off Date with the value 10-20-2001. The 'Select Location' section contains a dropdown menu with the value 'Boston - Logan' and four input fields: Latitude with the value 42.21, Longitude with the value 70.59, GMT - LST with the value 13.53, and Elevation MSL (ft) with the value 19. At the bottom of the dialog box are three buttons: OK, Cancel, and Help.

BlazeTank

File Tank Ignition Flight Calc Output Databases Help

Ambient Conditions at Take-Off

Ground Weather

Temperature (F) 68

Pressure (atm) 1

Humidity (%) 60

Wind Speed (knots) 4

Time

Current Time 08:53

Take-Off Time (24 hour format) 08:53

Current Date 10-31-2001

Take-Off Date 10-20-2001

Select Location

Boston - Logan

Latitude 42.21 GMT - LST 13.53

Longitude 70.59 Elevation MSL (ft) 19

OK Cancel Help

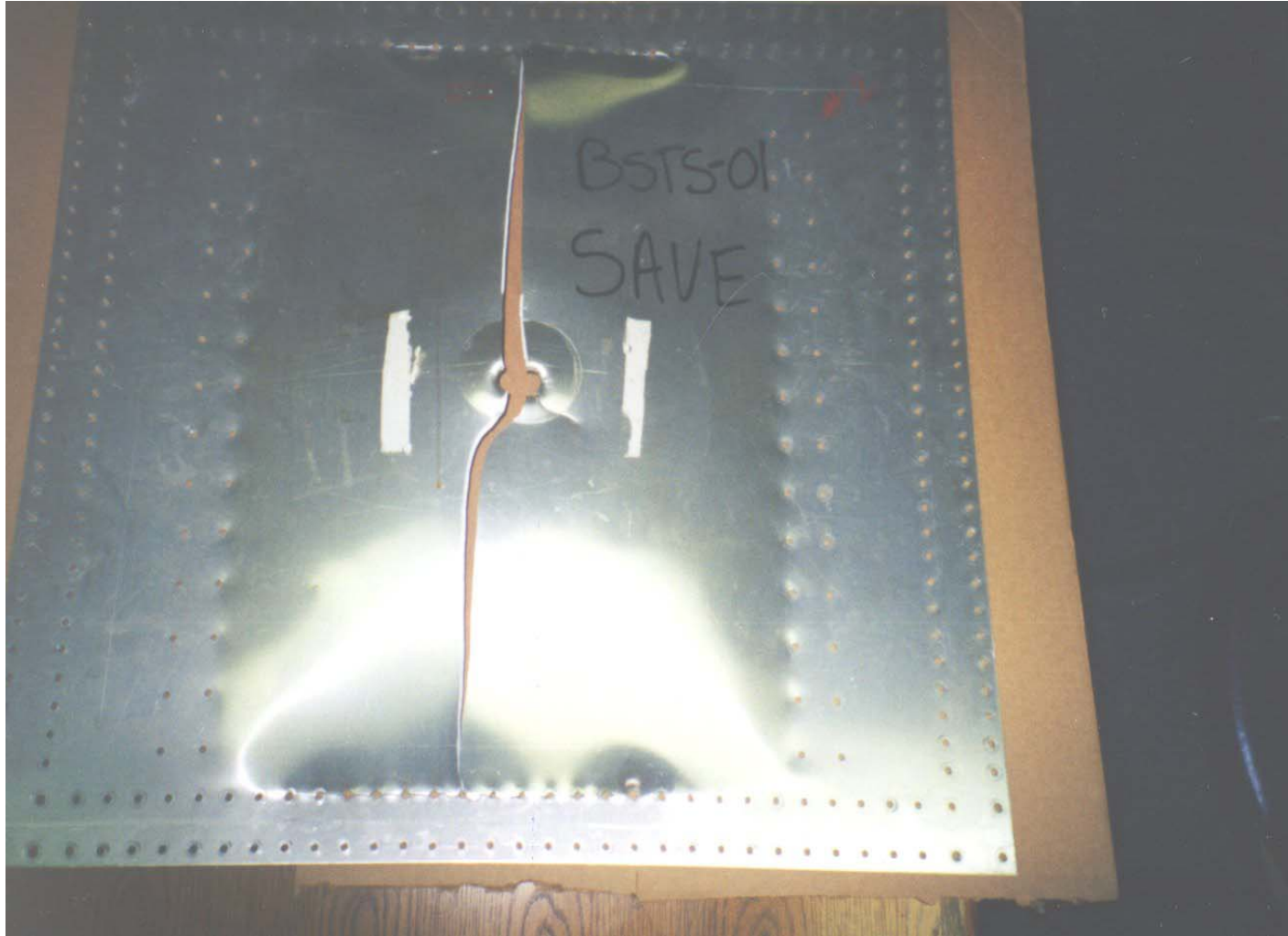
Hydrodynamic Ram

- In FAA study, we characterized conditions leading to ram
- Results completely ignored until Concorde accident
- Ram effect is well recognized in military aircraft
- How can you tell a ram effect?
 - Examine damage to tank wall
 - Tank tears out even though it is punched in
 - Tear in wall is much larger than punched hole

Bullet Hole and Circumferential Cracks



Overview of Front Panel Deformation & Rupture



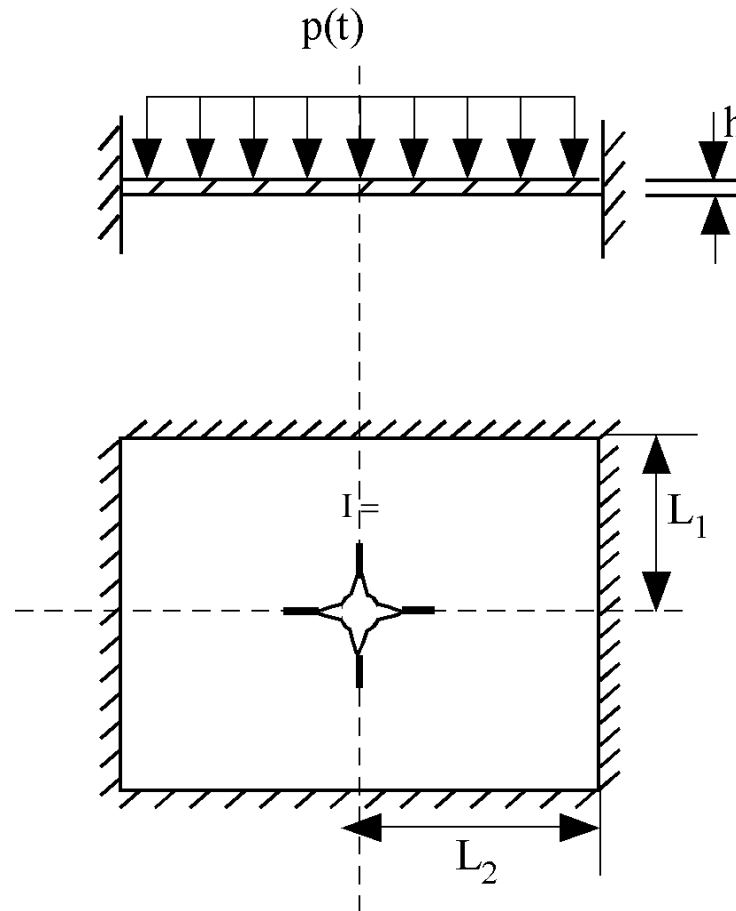
Tank Wall Damage Due to Hydrodynamic Ram: Test Data

Source	Impulse (Psi-s)	Final Deflection (in)	Crack Length (in)	Crack Area (in ²)	$\frac{A_{\text{crack}}}{A_{\text{projectile}}}$	$\frac{A_{\text{crack}}}{A_{\text{plate}}}$
Navy ¹	0.097	3.15	>4.9	12.6	64	0.133
Air Force ²	0.071	3.74	>15.8	18.6	29	0.027
	0.128	5.31	>15.8	37.2	54	0.054
	0.017	Minimal	Perforation only	N/A	1	0

¹Panel: Al 7075-T6, Curved, 9.84x9.65x0.0787 (in)

²Panel: Al 2024-T3, Flat, 31.5x21.7x0.157 (in)

TankCrack: Model of Structural Response and Failure



Assumptions:

Thin, clamped, rectangular plate

Small diameter hole at the center with a distribution of starter cracks

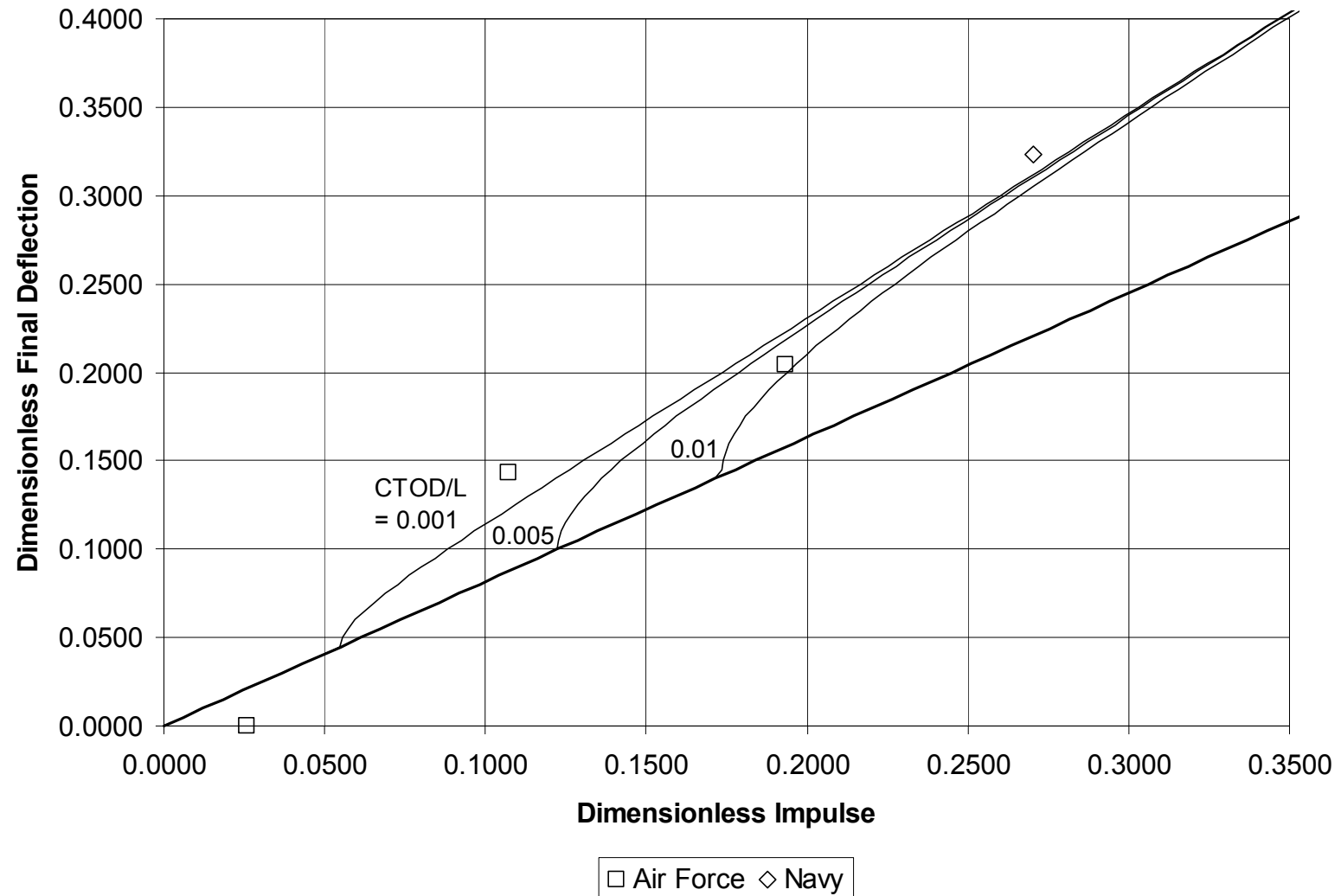
Impulsive loading, $I = \int p(t)dt$

Plastic deformation and failure

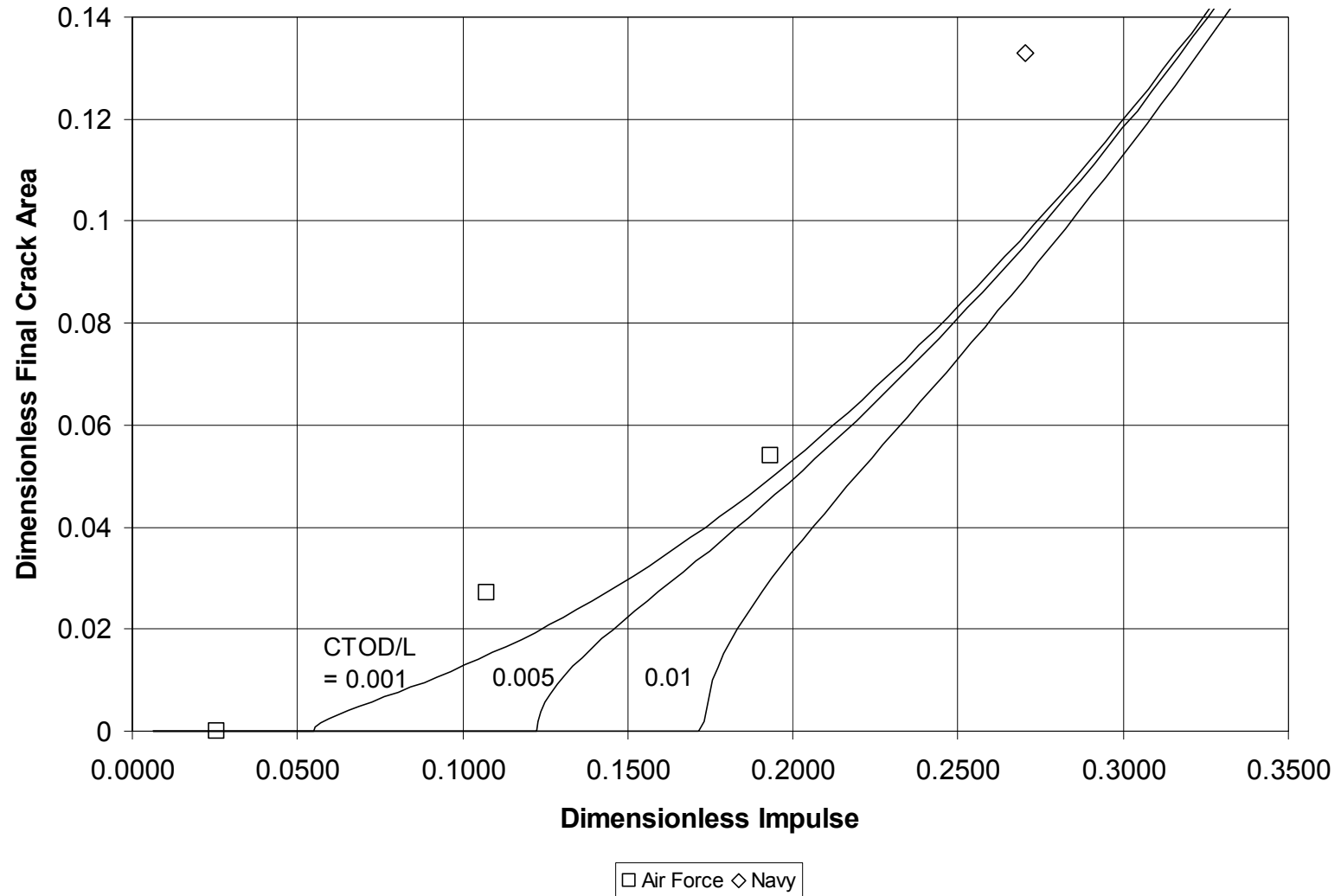
Plate Failure Criterion

- Plate deflection grows due to applied impulse.
- Crack will grow when the Crack Tip Opening Displacement (CTOD) reaches a critical value.
- The critical value of the CTOD is a material property; a value of ~ 10 mils matched the data well.

Final Deflection at Plate Center: Predictions vs. Tests



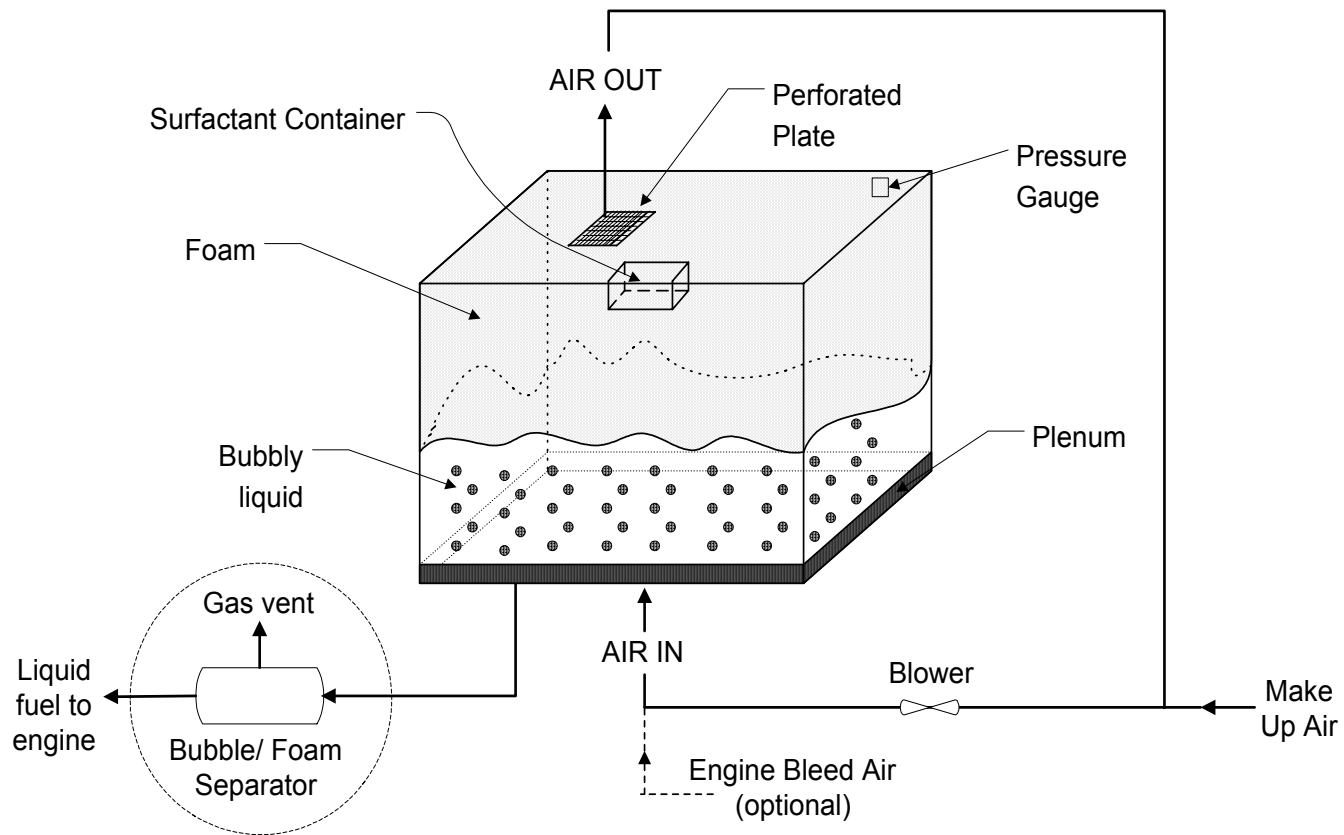
Crack Area: Model Predictions vs. Test Data



FuelShield™

- Mechanical/chemical treatment of fuel to protect fuel tanks against both ullage fire/explosion and hydrodynamic ram
- Technology under development for military aircraft
- Is it useful to civilian aircraft as an alternate to inerting, particularly in view of attacks on Sep. 11?

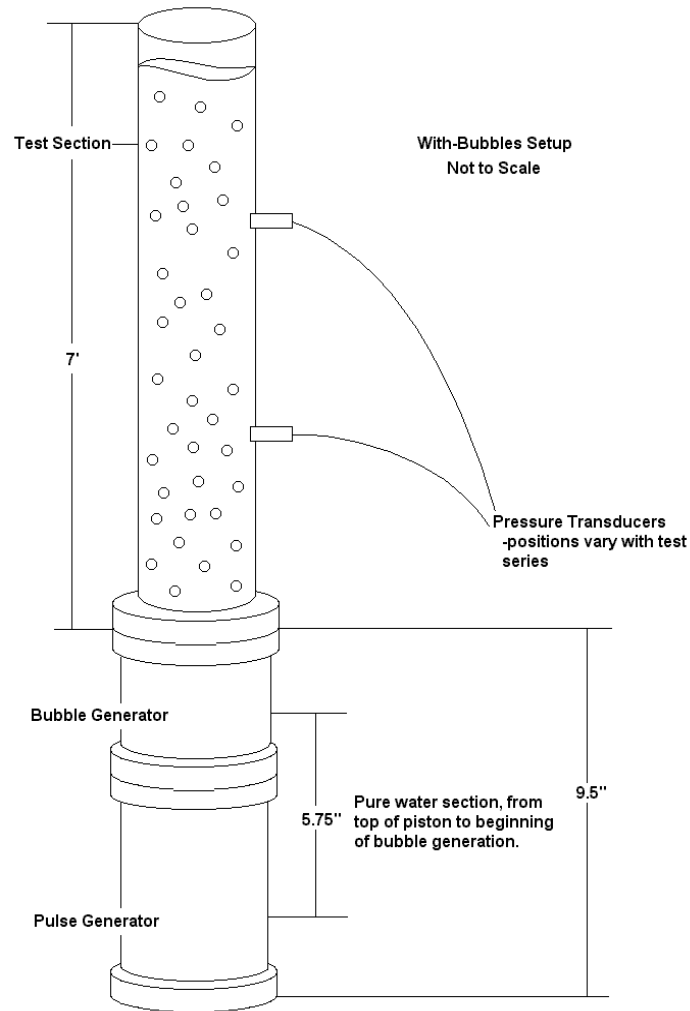
FuelShield™: Schematic of Overall Design



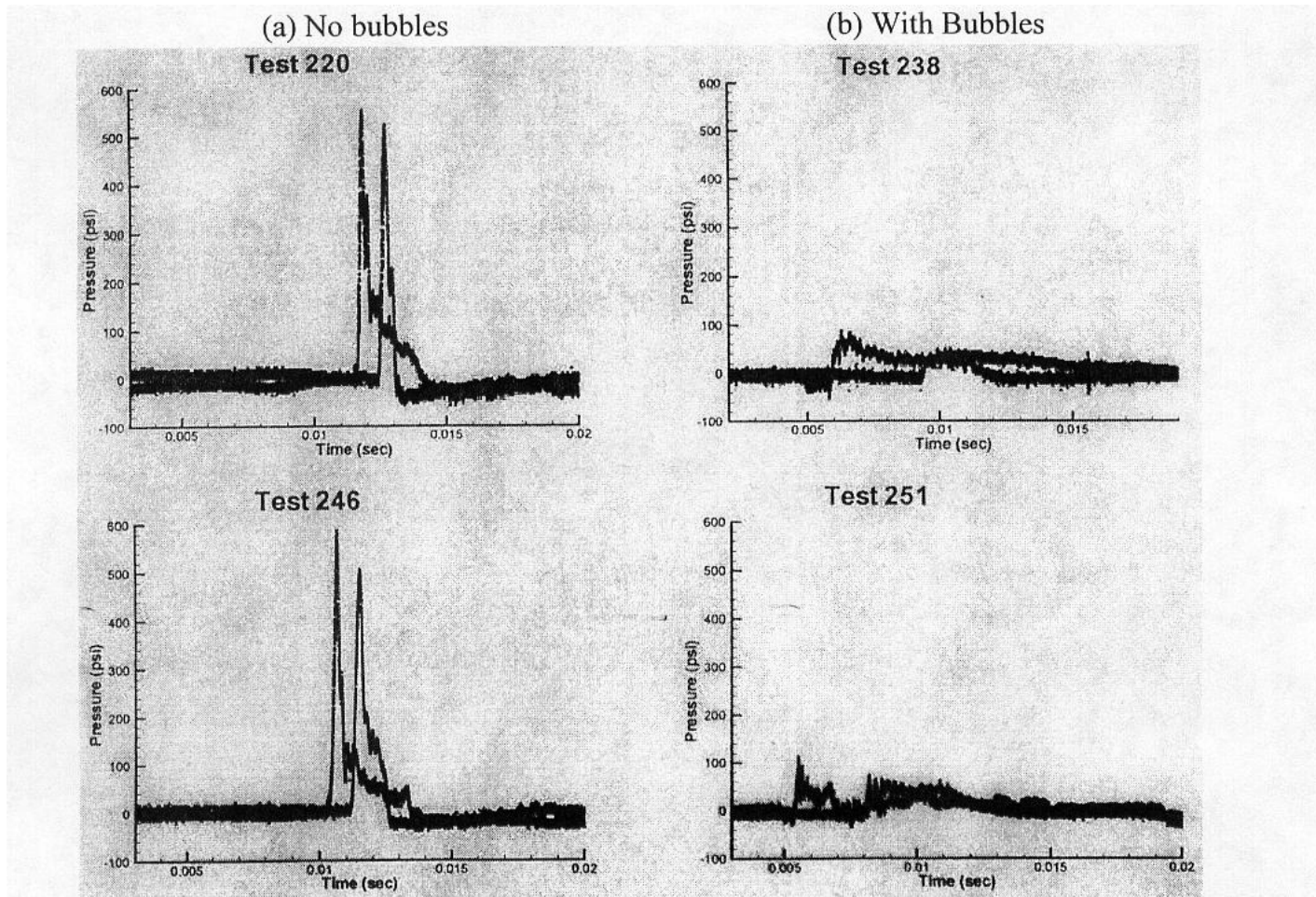
Technology Description

- Bubbles in liquid mitigate pressure wave from ram
- Foam in ullage mitigates ignition and flame spread
- Status of Development
 - Successful lab scale tests at BlazeTech
 - Successful preliminary gunfire tests at Navy's China Lake facility
 - Two patents pending

Bubbly Liquid Shock Tube



Representative Attenuation of Pressure Pulses



Fuel Foam Results

- Tested around 30 surfactants: two showed good foaming potential in a range of hydrocarbon fuels
- Organic based surfactants and $< 1\%$ needed – burn with minimum impact on combustion or emission
- Foam cell characteristics:
 - Expansion ratios > 20
 - Small foam cell size (1-5 mm)
 - Extremely stable and reproducible foam
- Bubbling action sufficient to initiate foaming
- No corrosion (iron)

Future Work

- Full scale gunfire tests, planned in 2002 at the Air Force
- Examine practical considerations:
 - Effect on engine parts
 - Operational and environmental effects of surfactant
 - Foam activation and flight profile
 - Mode and time of surfactant addition
 - Applicability to various airplanes and fuel tanks
- Seek partners for additional development and commercialization

Summary

- Hydrodynamic ram can be a hazard even on commercial airplanes
- Presented a model (TankCrack) of hole enlargement via plate cracking under hydrodynamic ram that agrees well with the available data
- Presented a protection method (FuelShield™) against both ram and ullage fire/explosion.